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**SINGLE STAGE EXPERIMENTAL EVALUATION
OF
VARIABLE GEOMETRY GUIDE VANES
AND STATOR BLADING**

**PART IV - DATA AND PERFORMANCE FOR VARIABLE
CAMBER GUIDE VANE AND STATOR B**

BY

B.A. JONES AND R.P. OSCARSON

**Prepared For
National Aeronautics And Space Administration**

Contract NAS3-7604

Pratt & Whitney Aircraft
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ABSTRACT

A single-stage experimental investigation was conducted to determine the extent that a variable camber inlet guide vane and a two-part stator having adjustable leading and trailing edge metal angles can increase the stable operating range of a compressor front stage. The stage was designed with zero rotor prewhirl at design equivalent rotor speed (SLTO conditions) and 35-degree hub - 20-degree tip prewhirl at 70% of design rotor speed (high Mach number cruise conditions). The cruise configuration permitted stable operation at a lower flow for a given speed than those attainable with the SLTO configuration. At cruise rotor speed and flow conditions, the adiabatic efficiency of the cruise configuration was approximately 7% higher than that of the SLTO configuration.

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SUMMARY

An experimental investigation was conducted with a 0.5 hub/tip ratio single-stage compressor to determine the extent that a variable camber guide vane and a two-part stator having adjustable leading and trailing edge metal angles can extend the range of stable compressor operation. The stage, which resembled the front stage of a typical high cruise flight Mach number compressor, was designed with zero rotor prewhirl at design equivalent rotor speed (sea level takeoff, SLTO conditions), and 35-degree hub - 20-degree tip prewhirl at 70% of design rotor speed (high Mach number cruise conditions). The design rotor tip relative Mach number for the SLTO configuration was 1.15, and the design stage pressure ratio and efficiency were respectively 1.32 and 85.1%. The SLTO configuration was tested over a range of speeds from 50 to 110% of design equivalent rotor speed. The cruise configuration and an intermediate configuration (approximately halfway between the SLTO and cruise IGV and stator settings) were tested from 50 to 100% of design equivalent rotor speed. Design performance goals were met with the SLTO and cruise configurations. The cruise configuration permitted stable operation at a lower flow for a given speed than that attainable with the SLTO configuration. At cruise rotor speed and flow conditions, adiabatic efficiency was approximately 7% higher for the cruise configuration. Moreover, the cruise configuration exhibited a substantial improvement in stall margin over that obtained with the SLTO configuration in the range of rotor speeds between 50 and 80% of design equivalent rotor speed. The intermediate setting of the guide vanes and stator (halfway between the SLTO and cruise configurations) resulted in flow reductions at SLTO and cruise design pressure ratio equal to approximately 40% of the reduction in flow between the SLTO and cruise configuration. The stall limit flow for the intermediate configuration at cruise rotor speed

was approximately 2% less than that for the SLTO configuration and 10% greater than that for the cruise configuration.

INTRODUCTION

The supersonic cruise operating point of fixed-geometry front stages of turbojet engine compressors tends to approach the stall line as cruise Mach number increases. This results from the flow limitation in the compressor rear stages at cruise rotor speed and pressure ratio conditions. This imposes a reduction in compressor inlet flow and axial velocity below that for which the front stages are designed to operate efficiently. When the reduction in front-stage axial velocity becomes proportionally greater than the reduction in blade tip velocity, rotor incidence angle increases above the sea level takeoff design value. If the increase in incidence angle is sufficiently large, the front stage(s) may be subject to stall, which can lead to reduced compressor performance and possibly to compressor surge. One approach in overcoming this problem is to provide mechanical adjustment of the guide vane and stator leading and/or trailing edge metal angles to permit selection of the proper stage velocity triangles over the required operating range.

A single-stage compressor experimental investigation was conducted to evaluate the extent to which variable geometry concepts can be applied to increase the stable operating range of a compressor front stage. A variable camber inlet guide vane, rotor, and two variable geometry stator configurations were designed, fabricated, and tested over a range of equivalent rotor speeds from 50 to 110% of the design rotor speed. The stage was designed with zero rotor prewhirl at design equivalent rotor speed and 35-degree hub - 20-degree tip prewhirl at cruise conditions. Design point (sea level takeoff, SLTO) equivalent rotor speed and flow were respectively 6050 rpm and 265 lb/sec; off-design (simulated high Mach number cruise) equivalent rotor speed and flow were respectively 4235 rpm and 143 lb/sec.

The variable-camber guide vanes comprised modified 63-series airfoil sections and were configured with a fixed leading-edge segment and two articulated flap segments. The rotor, designed with circular arc airfoils, had a tip inlet relative Mach number of 1.15. The stator configurations, comprised of 65-series airfoil sections, were designed to turn the flow back to the near axial direction at the SLTO operating point and to provide a (hypothetical) 2nd-stage rotor prewhirl distribution of

approximately 30-degree hub - 25-degree tip at the cruise operating point. One stator design had a fixed-geometry leading edge and an adjustable trailing edge (Stator A), and the other stator design had adjustable leading and trailing edge geometry (Stator B).

Details of the experimental stage aerodynamic and mechanical design are presented in Reference 1. The results of an annular cascade test program for the preliminary evaluation of candidate variable geometry concepts are presented in Reference 2. The data and performance obtained with the variable camber guide vane, rotor, and Stator A are presented in Reference 3. This report presents the data and performance obtained with the variable camber guide vane, rotor, and Stator B design that had variable leading and trailing edge geometry. The variable-geometry stage was tested with the guide vanes and stator in their SLTO position, cruise position, and an intermediate position halfway between the SLTO and cruise positions.

TEST EQUIPMENT

Facility

The compressor test facility is shown in figure 1. The compressor rotor is driven by a single-stage free turbine that is powered by exhaust gases from a J75 slave engine. Drive turbine speed is controlled by means of the engine throttle. Air entered the compressor test section through a 103-ft combined inlet duct, plenum, and bellmouth inlet and exhausted through an exit diffuser to the atmosphere. A 7-degree diffuser at the plenum inlet and 10:1 bellmouth contraction ratio ensured uniform flow conditions at the compressor inlet.

Compressor Test Rig

The compressor rig, shown in figure 2, comprises bellmouth inlet, test section, and exhaust section. The rotor assembly and shaft are supported on two bearings that transmit loads to the outer case through the stationary leading-edge segments of the variable-geometry inlet guide vanes and the struts in the exhaust case. A set of motor-driven throttle vanes is located in the exhaust case to vary flowrate. A section view of the flowpath is shown in figure 3. The guide vane inlet station is preceded by a flow-straightening section approximately 10-in. long. The inner flowpath wall converges from a diameter of 18.70 in. at the guide vane inlet to a diameter of 25.60 in. at the stator exit. The outer wall diameter converges slightly from 43.20 to 42.10 in. The hub/tip ratio at

the rotor leading edge is 0.492. In general, the flowpath is designed to resemble the front-stage environment of a state-of-the-art high cruise Mach number compressor. The blade rows were spaced sufficiently apart to permit the installation of instrumentation.

A machined aluminum shroud was used for the rotor to prevent damage to the blade tips in the event a rub occurred during operation near or in stall. The nominal blade tip clearance was 0.050 in.

Blading Design

Inlet Guide Vane

The variable camber inlet guide vane was designed to provide axial inlet flow (zero swirl) in the SLTO configuration and 35-degree hub - 20-degree tip swirl distribution in the cruise configuration. Slightly-modified 63-series airfoil sections were used. The basic camber was selected to provide approximately two-thirds of the anticipated air turning requirement at the cruise operating point.

The SLTO and cruise positions were achieved by actuating two articulated flap segments in the appropriate direction to reduce or increase camber angle from the basic design camber angle. A photograph of the guide vane is shown in figure 4. Guide vane geometry details are summarized in table 1a; symbols are defined in Appendix A. Table 1a includes the leading and trailing edge metal angles for an intermediate guide vane setting that was tested following the tests with the SLTO and cruise geometries. Additional design information for the inlet guide vane is presented in Reference 1.

Rotor

The rotor was comprised of 34 blades, designed with circular-arc airfoil sections. The design point inlet relative Mach number at the tip was 1.15 and the design diffusion factor was nominally 0.40. Blade chord length and solidity were representative state-of-the-art values for a high cruise Mach number front stage. A part span shroud was added at 40% span from the tip to reduce blade resonance stress. Rotor geometry details are summarized in table 1b, and additional rotor design information is contained in Reference 1.

Table 1. Design Data

a. Inlet Guide Vane Geometry

Airfoil Series: 63
 No. of Blades: 20
 Aspect Ratio: 2.642
 Thickness Ratio: 0.09

SLTO Configuration

Percent Span (from tip)	κ_1	κ_2	ϕ	γ°	c	σ	δ°
90	-20.2	5.4	25.6	1.2	4.55	1.412	Approximately 1.0 Degree Estimated
70	-20.2	3.7	23.9	0.5	4.55	1.245	
50	-20.2	1.9	22.1	-0.2	4.55	1.080	
30	-20.2	0.3	20.5	-0.9	4.55	0.913	
10	-20.2	-1.6	18.6	-1.6	4.55	0.748	

5 Cruise Configuration

Percent Span (from tip)	κ_1	κ_2	ϕ	γ°	c	σ	δ°
90	-20.2	36.0	56.2	21.1	4.47	1.412	2.2
70	-20.2	34.3	54.5	20.5	4.47	1.245	3.2
50	-20.2	32.5	52.7	19.9	4.47	1.080	4.3
30	-20.2	30.9	51.1	19.2	4.47	0.913	5.9
10	-20.2	29.0	49.2	18.5	4.47	0.748	7.1

Intermediate Configuration

Percent Span (from tip)	κ_1	κ_2
90	-20.2	20.7
70	-20.2	19.0
50	-20.2	17.2
30	-20.2	15.6
10	-20.2	13.7

Table 1. Design Data (Continued)

b. Rotor Geometry

Airfoil Series: Circular Arc
 No. of Blades: 34
 Aspect Ratio: 2.829

SLTO Configuration

Percent Span (from tip)	κ_1	κ_2	ϕ	γ°	i_m	c	σ	t/c	δ	$\bar{\omega}'$
90	48.6	18.8	29.8	31.2	1.4	3.24	1.42	0.0744	6.1	0.024
70	52.5	32.5	20.0	41.2	1.4	3.43	1.30	0.0641	4.3	0.024
50	56.1	42.1	14.0	48.2	1.2	3.63	1.20	0.0542	3.8	0.033
30	59.6	48.6	11.0	53.6	0.4	3.82	1.12	0.0445	3.3	0.058
10	63.2	52.9	10.3	57.9	0.0	4.01	1.06	0.0355	3.8	0.097

Cruise Configuration

Percent Span (from tip)	κ_1	κ_2	ϕ	γ°	i_m	c	σ	t/c	δ	$\bar{\omega}'$
90	Same as SLTO	Same as SLTO	Configuration	Configuration	-1.7	Same as SLTO	Configuration	Configuration	6.2	0.030
70	Same as SLTO	Same as SLTO	Configuration	Configuration	2.3	Same as SLTO	Configuration	Configuration	4.3	0.018
50	Same as SLTO	Same as SLTO	Configuration	Configuration	4.3	Same as SLTO	Configuration	Configuration	2.9	0.030
30	Same as SLTO	Same as SLTO	Configuration	Configuration	4.9	Same as SLTO	Configuration	Configuration	2.3	0.064
10	Same as SLTO	Same as SLTO	Configuration	Configuration	4.5	Same as SLTO	Configuration	Configuration	2.5	0.125

Table 1. Design Data (Continued)

c. Stator B geometry

Airfoil Series: 65
 No. of Blades: 40
 Aspect Ratio: 2.939
 Thickness Ratio: 0.08

SLTO Configuration

Percent Span (from tip)	κ_1	κ_2	ϕ	γ°	i_m	c	σ	δ°	$\bar{\omega}$
90	40.9	-8.7	49.6	16.0	-4.2	2.75	1.28	12.8	0.026
70	37.1	-7.5	44.6	14.5	-3.3	2.88	1.19	10.1	0.025
50	34.7	-7.5	42.2	13.5	-3.2	3.00	1.12	9.3	0.026
30	34.0	-8.0	42.0	13.2	-3.2	3.13	1.07	9.8	0.029
10	36.0	-8.9	44.9	13.5	-4.8	3.27	1.02	11.0	0.032

Cruise Configuration

Percent Span (from tip)	κ_1	κ_2	ϕ	γ°	i_m	c	σ	δ°	$\bar{\omega}$
90	49.9	17.3	----	----	-2.2	----	1.28	11.9	0.040
70	46.1	18.5	----	----	1.7	----	1.19	9.0	0.038
50	43.7	18.5	----	----	3.2	----	1.12	8.2	0.030
30	43.0	18.0	----	----	3.2	----	1.07	8.4	0.024
10	45.0	17.1	----	----	1.2	----	1.02	9.3	0.027

Intermediate Configuration

Percent Span (from tip)	κ_1	κ_2
90	45.4	4.3
70	41.6	5.5
50	39.2	5.5
30	38.5	5.0
10	40.5	4.1

Stator B

A photograph and section view of Stator B are shown in figure 4. The design point geometry was developed from 65-series airfoil sections; the vanes were designed to preserve as much as possible of the basic 65-series airfoil shape. Pursuant to this objective, the two flap segments were selected to have approximately the same chord length and an overlap of 15% of the basic 65-series chord length. The leading edge of the forward flap is identical to the basic 65-series airfoil up to 30% of the forward flap chord. The geometry of the remainder of the forward flap was generated by combining the thickness distribution for a 65-series airfoil (having a chord length equal to the flap chord and thickness at 30% chord as defined by the basic airfoil) with the basic airfoil suction surface contour. For the rear flap, the uncovered section (55 to 100% of the basic airfoil chord) is identical to the basic 65-series airfoil geometry. A NACA-4-digit series airfoil shape was used for the covered leading edge section (40 to 55% basic chord) to permit low-loss operation over a wide range of incidence angles. A small slot was provided between the two segments in this position to reduce recirculation in the pressure surface cavity that is formed by the two segments.

Minimum loss incidence angle, loss, and deviation angle for the SLTO geometry airfoil were predicted from conventional 65-series airfoil cascade data. Midspan values of stator inlet Mach number and diffusion factor were 0.56 and 0.33, respectively. Additional design information is presented in Reference 1.

To obtain the desired incidence and exit angle distributions for operation at the cruise conditions, the forward segment was rotated to increase the leading edge metal angle 9 degrees and the rear segment was rotated to increase the trailing edge metal angle 26 degrees. The net effect was a decrease in camber angle of 17 degrees. For the intermediate geometry setting, the forward segment was rotated 4.5 degrees and the rear segment was rotated 13 degrees. Stator design details are presented in table 1c.

Instrumentation

Instrumentation was provided to obtain overall and blade element performance data for each blade row. Axial locations of instrumentation stations are indicated in figure 2 and the circumferential locations of instrumentation at each axial station is shown in figures 5 through 7.

Airflow was measured with an ASME standard thin-plate orifice located in the inlet duct, as indicated in figure 1. Inlet total pressure and temperature were measured in the plenum by means of three Kiel-type total pressure probes and six half-shielded total temperature probes. Four equally spaced static pressure taps were located on both the inner and outer wall at Station 0. The static pressure measurements at this station were used to check the orifice weight flow.

Air angles were measured by means of 20-degree wedge traverse probes. Behind the inlet guide vane and stator these probes were located circumferentially as required to keep them out of vane wakes. Each 20-degree wedge probe includes a total pressure tube for total pressure measurement as shown in figure 8.

Total pressure measurement behind the guide vane and stator was accomplished by means of two stationary wake probes at each station. One wake probe had sets of 14-tube rakes at 10, 50, and 90% span locations; the other wake probe had sets of 14-tube rakes at 30 and 70% span locations. The angle settings of both the rake heads and the probe stem were selected to permit alignment with the airstream within approximately +5 degrees for both SLTO and cruise geometry. The acceptance yaw angle (pressure accuracy within +1% of $P - p$) of the rake tubes is +10 degrees. The wake probe was translated circumferentially to keep the wake centered on the rakes when the guide vane and stator flaps were in the cruise position. A typical rake head is shown in figure 9. The 20-degree wedge traverse probes used for air angle measurement behind the guide vane and stator also measured total pressure, which provided a check on the midchannel total pressures obtained with wake probes. Total pressures behind the rotor were measured with the two 20-degree wedge traverse probes and two radial rakes with Kiel heads at 10, 30, 50, 70, and 90% span. The traverse probes and radial rakes were spaced so as to approximately average the circumferential total pressure profile generated by the inlet guide vanes. Total temperatures were measured behind the stator using Kiel-type probes mounted on four radial rakes. The temperature rakes were spaced as shown in figure 7; they were located with respect to each other at 0, 1/4, 1/2, and 3/4 of a guide vane gap. Total pressure and temperature rakes are shown in figure 10.

Radial distribution of static pressures behind the guide vanes, rotor, and stator were measured by means of 8-degree wedge traverse probes, which are shown in figure 11. Four approximately equally spaced static pressure taps were provided on the inner and outer walls behind

the guide vane and stator. Four taps were located on the inner wall and twelve taps were located on the outer wall behind the rotor. The eight additional outer wall taps behind the rotor were spaced across a guide vane gap to determine the static pressure variation associated with guide vanes wakes and/or stator flow fields.

Three Kistler high response pressure transducers were installed approximately 30 and 60 degrees apart at 10% span behind the rotor to measure the number, size, and rotational speed of rotating stall zones.

Rotor blades and stator vanes were instrumented with strain gages to measure torsional and bending stresses.

Rotor speed was measured with an electromagnetic sensor mounted adjacent to a 60-tooth gear on the rotor shaft. Gear-tooth-passing frequency was displayed as rpm on an Anadex digital counter.

Steady-state pressure measurements were recorded on a scannivalve multichannel pressure transducer system. Traverse probe data (total and static pressures, air angle, and radial position) were recorded on magnetic tape at the rate of 60 samples (2.0 in. probe travel) per minute. Temperatures were indicated on precision potentiometers and manually recorded.

Orifice static pressure, bellmouth static pressure, plenum pressure, and representative stage exit pressure instrumentation were close-coupled to transducers for high response transient recording of flow and pressure ratio during operation into and out of stall. Similar instrumentation was connected to manometers in the control room to permit setting the desired steady-state pressure ratio and flow conditions.

PROCEDURES

Test Procedure

Overall and blade element performance data were obtained at three settings of the variable camber inlet guide vane and variable geometry Stator B: sea level takeoff, cruise, and an intermediate setting approximately halfway between the SLTO and cruise settings. Data were obtained at 50, 70, 80, 100, and 110 percent of design equivalent rotor speed with the SLTO configuration, and 50, 70, 80, and 100 percent of design rotor speed with the cruise and intermediate configuration. Data points were distributed between maximum flow and near stall flow at each rotor speed.

For the SLTO configuration, points near stall at 100 and 110% of design speed were obtained with a reduced inlet pressure (approximately 11 psia) because of the test stand horsepower limit. For each data point the fixed pressure and temperature instrumentation data were recorded with the traverse probes withdrawn. The traverse probes were subsequently immersed in the airstream and recorded as they were traversed from the inner to the outer wall.

The compressor was operated into and out of stall at each of the test rotor speeds; orifice and bellmouth static pressures, exit total pressure, and rotor speed were recorded on magnetic tape at the rate of 600 samples per minute. Kistler transducer data were also recorded during the excursions into stall to define the rotating stall patterns. Time correlation between the Kistler and stall transient pressure recordings permitted correlation of the Kistler and stall transient data.

Data Reduction Procedure

Data reduction was accomplished in four steps. Initially, the raw data compiled on magnetic tape and IBM cards were converted to engineering units that were subsequently recompiled on a storage tape and tabulated for preliminary inspection. The second data reduction step accomplished the following:

1. Mach number corrections, as required, for traverse probe data
2. Correction of pressure and temperature data to NASA Standard Day conditions
3. Mass averaging of guide vane and stator wake pressures (these were subsequently performed in a separate computer program)
4. Calculation of corrected flow
5. Calculation of radial distribution of flow parameter (for radial mass flow averaging of total pressures in subsequent procedure)
6. Circumferential arithmetic averaging of fixed and traverse instrumentation data.

The results of this data reduction step were printed to permit a more detailed inspection of the data to eliminate apparent inconsistencies prior to the calculation of performance and vector diagram variables.

Pressure ratios were calculated for the rotor, guide vane-rotor, and the entire stage. Stage inlet total pressure was corrected to the NASA Standard Day conditions, 14.69 psia. The guide vane, rotor, and stator exit total pressures were weighed according to local mass flow to obtain the average values. The guide vane and stator wake total pressures at each radial measuring station were mass-averaged using the local total pressures in the wake and the 8-degree wedge probe static pressure to define the local Mach numbers. Mass flux was then obtained from the relationship

$$\bar{m} = \frac{W\sqrt{T}}{PA} = \sqrt{\gamma g/R} \quad M \left[1 + \frac{\gamma-1}{2} M^2 \right]^{1/2} p/P$$

where T is measured total temperature and A is the flow area associated with each total pressure tube. With the radial distribution of total pressure and mass flux thus calculated, the total pressures were mass-averaged in the radial direction. Behind the rotor, the total pressures obtained with the two 20-degree wedge probes were arithmetically averaged, and the resulting radial distribution was mass-flow-averaged using the 8-degree wedge probe static pressure and stator exit radial temperature distribution to define weight flow.

Performance and vector diagram calculations were performed two ways: one method used the measured static pressure distributions to define the velocity distributions at each axial station; the second method, a streamline analysis, calculated a static pressure distribution that satisfied radial equilibrium and continuity. For the streamline analysis calculation, adjustments could be made to an input boundary layer displacement factor to achieve agreement between the calculated and measured static pressure. The primary purpose for using the streamline analysis computer program was to permit translation of the velocity diagram data from instrumentation axial stations to the respective adjacent blade row leading and trailing edge stations for comparison with the design velocity diagram results (Reference 1). The translation was performed with the assumption of isentropic flow and constant angular momentum along design stream annuli whose centers intersected each blade row leading edge at 10, 30, 50, 70, and 90% span positions from the tip. The velocity diagram data presented herein were calculated by means of the streamline analysis computer program and correspond to the leading and trailing edges of the rotor and stator blade rows as defined by the dashed lines in figure 3. Loss coefficient calculations for the guide vane and stator were performed in a separate computer program to permit use of the wake freestream total pressure

as the upstream pressure and to permit the elimination of certain rake tube pressures that were apparently faulty and affected the credibility of the calculated loss coefficient. Rotor overall performance and loss coefficients were based on inlet total pressures that resulted from integration of the as-measured guide vane wake pressures. The discrepancies in guide vane wake pressures that influenced guide vane loss coefficients had a negligible effect on the calculated rotor loss. Because of the sensitivity of the calculated rotor blade relative total pressure to static pressure, the rotor loss coefficients were based on the static pressure values measured at the instrumentation stations rather than those calculated for the leading and trailing edge stations. Consequently, the loss coefficient data for the three blade rows correspond to the instrumentation stations upstream and downstream of each blade row (along the aforementioned design streamline) rather than to the blade row leading and trailing edges.

PRESENTATION OF DATA

Overall Performance

Overall performance data are presented in terms of pressure ratio and adiabatic efficiency as functions of corrected weight flow, $W\sqrt{\theta}/\delta$; specific weight flow, $W\sqrt{\theta}/\delta A_A$; and equivalent rotor speed, $N/\sqrt{\theta}$. Overall performance is also summarized in terms of pressure rise coefficient, Ψ , as a function of flow coefficient, ϕ . Definitions of calculated performance variables and symbols are presented in Appendix A.

Pressure ratio and efficiency for the rotor, guide vane-rotor, and guide vane-rotor-stator B are presented respectively in figures 12, 13, and 14 for the SLTO configuration; in figures 15, 16, and 17 for the cruise configuration; and in figures 18, 19, and 20 for the intermediate configuration. The dashed-line portion of the curves in figures 14, 17, and 20 refer to transient data, and the data point on the stall line is the incipient stall point determined from the transient data. The corrected flows and pressure ratios for steady-state data points are tabulated in table B-1. Efficiency contours are presented for the three configurations in figure 21, and the pressure rise coefficient, Ψ , is presented as a function of the flow coefficient, ϕ , in figure 22.

Two qualifying statements must be made concerning the efficiency data presented in figures 12, 13, 15, 16, 18, and

19. First, the mass-average temperature used in the efficiency calculations was based on the temperature and flow distributions measured at Station 2A downstream of the stator. Redistribution of the flow through the stator may produce a slight difference between the mass-average temperature behind the rotor and the value based on stator exit conditions. This temperature difference would be reflected in the rotor and guide vane-rotor efficiencies. The second qualification concerns the use of measured guide vane wake total pressure (as discussed under Data Reduction Procedure) in the calculation of rotor pressure ratio and efficiency. Discrepancies in rotor efficiency, resulting from observed discrepancies in the guide vane wake profiles, are within ± 1.0 percent.

Although the performance levels in figures 12, 13, 15, 16, 18 and 19 are generally correct for the rotor and guide vane-rotor combination, a more valid assessment of stage and blade-row performance is presented in the overall performance data in figures 14, 17, and 20, and the detailed blade element data in the following section.

Figure 14 indicates that the SLTO configuration slightly exceeded its predicted performance. The predicted stage pressure ratio and efficiency were 1.32 and 85.1% compared with measured values of 1.347 and 87.8% at the design flowrate. The higher measured performance is partially attributable to slightly high rotor tip loading due to slightly negative swirl produced by the guide vanes in their SLTO position, as seen in figure 23.

(Note in figures 12 through 14 and in table B-1 that only three data points were processed for the 110% of design rotor speed condition. Discrepancies in the plenum pressure measurement were noted for the other three data points, which resulted in questionable performance values. The source of error was not discovered and the three questionable data points were, therefore, omitted from the data presentation).

Figure 17 shows that the predicted stage pressure ratio of 1.129 and efficiency of 88.3% for the cruise configuration geometry were exceeded with a measured pressure ratio of 1.15 and corresponding efficiency of 90% at the cruise design flow.

The intermediate configuration efficiency for cruise design flow in figure 20 is about halfway between the SLTO and cruise configuration efficiencies for the same flow in figures 14 and 17. The corresponding pressure ratio for the intermediate configuration was close to that obtained with the SLTO geometry. The improvement in

efficiency at cruise design conditions, as the stage geometry was changed from the SLTO configuration to the cruise configuration, is indicated by the efficiency contours in figure 21.

The pressure coefficient-flow coefficient characteristics in figure 22 indicate that the SLTO and cruise configurations are fairly well matched with their respective design conditions.

Blade Element Performance
for
Sea Level Takeoff Configuration

Inlet Guide Vane; SLTO

Guide vane exit air angle distributions for the sea level takeoff configuration are presented in figure 23. The design distribution is shown for comparison with the data. The distribution in figure 23 indicates only a slight variation from the zero swirl design distribution. The swirl distribution corresponding to design flow imposed slightly higher incidence on the rotor tip relative to design incidence conditions. The wide variation of swirl angle with flow in the tip region is attributable to (1) influence of the rotor on the static pressure gradient in the tip region ahead of the rotor and (2) possible influence of the relief cut in the tip of the guide vane flap to permit wall clearance in the cruise configuration. A similar trend of tip region swirl angle variation with flow was noted in the Stator A test program reported in Reference 3.

Guide vane loss coefficient for the hub, mean, and tip sections are presented in figure 24. Because the guide vane loss coefficients were calculated in a separate computer program and involved the detailed inspection of each guide vane wake, only the data for the 70 and 100% design rotor speeds at the hub, mean, and tip sections were processed. These results provide assessment of the guide vane losses with respect to spanwise variation and Mach number. The loss coefficient values obtained with the Stator A SLTO configuration of Reference 3, also presented in figure 24, indicate good agreement with the Stator B SLTO configuration results.

Guide vane velocity diagram data for all five spanwise measuring stations are presented in table B-2.

Rotor; SLTO

The rotor inlet relative Mach number and air angle distributions achieved with the variable camber guide vane and Stator B in the sea level takeoff position at design equivalent rotor speed are shown in figure 25. The distributions of both Mach number and air angle that correspond to near design flow are in good agreement with the indicated design distribution.

Rotor diffusion factor, deviation angle, and loss coefficient for design equivalent rotor speed are shown in figures 26 through 30. The measured diffusion factors and deviation angles are in generally good agreement with the design values.* The rotor losses are generally greater than the predicted values. Minimum loss incidence angles for the 10 and 30% span locations (figures 26 and 27) are approximately 4 degrees greater than the predicted incidence angles for minimum loss.

Static pressure measurements obtained behind the rotor at the outer wall locations shown in figure 6 are presented as a function of equivalent guide vane and stator blade gap in figure 31. The pressure distribution indicates two regions of high and low pressure relative to a guide vane gap and one region of high and low pressure relative to a stator gap. The circumferential variation in static pressure is considered to be associated with the stator flow field because (1) the measured guide vane wake at 10% span from the tip was relatively small (figure 32) and (2) the static pressure taps at Station 2 were in close proximity to the stator leading edge (figure 3).

The outer wall value of static pressure obtained from extrapolation of the 8-degree wedge traverse probe static pressure data is shown in figure 31. The extrapolated value is in good agreement with the average of the wall static pressures.

Additional rotor velocity diagram and blade element performance data are presented in table B-2.

* The rotor diffusion factor design points presented in figures 23 through 27 are defined by the corresponding equation in Appendix A. This definition accounts for the effect of the streamline diameter change across the rotor. The effect of this change was significant in the rotor hub region. The predicted rotor diffusion factors presented in Reference 1 do not account for the streamline diameter change and should not be used for comparison with the data in these figures.

Stator B; SLTO

Stator B inlet conditions for the sea level takeoff configuration are shown in figure 33. The air angle distribution for design flow is in fair agreement with the indicated design distribution. The Mach numbers for design flow are low relative to the design values between hub and midspan.

Stator B diffusion factor, deviation angle, and loss coefficient are presented as functions of incidence angle at the five spanwise measuring stations in figures 34 through 38. In general, the measured diffusion factor, deviation angle, and loss coefficient are in favorable agreement with the respective predicted values with the exception of the hub section loss coefficient, which is substantially greater than the predicted value.

Exit air angle distributions for the Stator B SLTO configuration are compared in figure 39 with the design exit air angle distribution. The measured air angles are slightly less (approximately 2 degrees) than the design values across the span.

Additional Stator B blade element data are presented in table B-2.

Blade Element Performance for Cruise Configuration

Inlet Guide Vane; Cruise

The swirl distribution achieved with the variable camber inlet guide vane in the cruise configuration at cruise (70% of design equivalent) rotor speed is compared with the design distribution in figure 40. The measured swirl distribution is relatively flat in the hub and tip regions with the exception of the unusually high angle at 10% span for the near stall flow conditions. The swirl distributions in figure 40 are characteristically different from the distributions obtained with the same guide vane configuration in the Stator A tests of Reference 3. Possible reasons for this difference will be discussed in the final report under the contract program. Guide vane loss coefficients for the hub, mean, and tip sections are presented as a function of Mach number in figure 41. Additional guide vane data are contained in table B-3.

Rotor; Cruise

Rotor inlet relative Mach number and air angle achieved with the cruise guide vane configuration at cruise rotor speed are shown in figure 42. The measured Mach number distribution is close to the design Mach number distribution. The Mach numbers in the hub region are slightly high compared to the design values. The experimental values of inlet relative air angle for design flow agree with the design values between 10 and 50% span and increase to about 4 degrees higher than the design value at 90% span.

As a matter of interest, the relative air angle distribution obtained with the SLT0 guide vane configuration at cruise rotor speed and flow conditions is included in figure 42 for comparative purposes. This comparison illustrates the high incidence angles that a fixed sea level takeoff geometry stage would encounter at the specified cruise conditions.

Rotor diffusion factor, deviation angle, and loss coefficient for cruise rotor speed conditions are presented in figures 43 through 47. In general, the experimental values compare well with the predicted values at all span locations. The rotor losses are slightly higher than predicted between the midspan and hub region. The minimum loss incidence angle in the tip region appears to occur at a lower incidence than the predicted minimum loss incidence.

Stator B; Cruise

Stator B inlet conditions for the cruise configuration at cruise rotor speed are shown in figure 48. The experimental Mach number distribution for design flow is in good agreement with the design values. The experimental air angles for design flow agree with the design values in the hub and tip regions and are slightly higher at midspan.

Stator B diffusion factor, deviation angle, and loss coefficient for cruise configuration at cruise rotor speed are shown in figures 49 through 53. The measured diffusion factors are slightly lower than the values predicted for the design air turning at cruise conditions. Deviation angles are in good agreement with the predicted values. The stator losses are appreciably higher than the losses predicted for a conventional stator designed for the same air turning and operation at minimum loss incidence. The relatively high losses are attributed to the relatively

broad wake generated by the two segments of the Stator B cruise configuration. A midspan wake total pressure profile for near cruise design flow conditions is shown in figure 54.

Exit air angles for the Stator B cruise configuration are compared with the design exit air angles in figure 55. Good agreement is seen between the measured and design air angle values, particularly for the cruise design flow point.

Additional Stator B velocity diagram and performance data for the cruise geometry configuration are presented in table B-3.

Blade Element Performance for Intermediate Configuration

Blade element performance was obtained with the variable guide vanes and Stator B positioned midway between the SLTO and cruise configuration positions. The blade element performance results obtained with this configuration at the 70, 80, and 100% equivalent rotor speed operating conditions are presented in figures 56 through 69. In each figure the predicted design information for the SLTO and cruise operating points are presented for comparison with the data. Inlet guide vane exit air angles and losses are presented only for the 70 and 100% rotor speeds. Additional velocity diagram and blade element performance data for the intermediate configuration are presented in table B-4.

Inlet Guide Vane; Intermediate

Inlet guide vane swirl distributions for 70 and 100% design equivalent rotor speed conditions are shown in figure 56. The swirl distribution is similar to that obtained with the cruise geometry (figure 40).

Guide vane loss coefficient at the root, midspan, and tip sections for the 70 and 100% rotor speed conditions are presented in figure 57. The loss coefficient levels are similar to those obtained with the SLTO guide vane configuration.

Rotor; Intermediate

Rotor inlet relative Mach number and air angle distributions for the 70, 80, and 100% rotor speed conditions are shown in figure 58. The maximum and near

stall flow data points are presented to show the overall range at each rotor speed condition.

Rotor diffusion factor, deviation angle, and loss coefficient are presented as functions of incidence angle in figures 59 through 63. The general agreement between experimental results and those predicted for the respective SLTO and cruise configurations indicates that the intermediate guide vane configuration permits the rotor to operate within its low-loss range at both SLTO and cruise rotor speeds. It should be recalled, however, that the intermediate configuration overall pressure ratio and flow did not match either the SLTO or cruise design conditions, and would not be a suitable fixed-geometry stage for the specified operating range.

Stator B; Intermediate

Stator inlet Mach number and air angle distributions for maximum and near stall flow conditions are compared in figure 64 with the SLTO and cruise design distributions.

Blade element performance for the Stator B intermediate configuration is presented in figures 65 through 69. The values of diffusion factor, deviation angle, and loss coefficient obtained with the intermediate position are consistent with SLTO and cruise configuration values.

Stall Transient Performance

Stall transient data are presented in figures 70, 71, and 72 in terms of pressure ratio versus corrected weight flow. Each figure includes (1) the steady-state overall performance curve for reference, (2) the transient overall performance curve based on the arithmetic average of stage exit total pressures measured with a single radial rake, and (3) a tabulation of rotating stall pattern results derived from Kistler transducer data. The transient data pressure ratios may be less than or slightly greater than the steady-state (mass flow average) pressure ratios depending upon the radial rake position with respect to stator wakes.

The rotating stall patterns determined from the three Kistler transducers located at 10% span are characterized by a single stall zone that covers approximately one-half of the annulus (circumferentially) and rotates at about one-half of the rotor speed. The stall zone appearance is coincident with the stall point determined from the stall transient data. This point is indicated as point number (1) in the figures. When the throttle vanes are opened and the stage comes out of stall, the size of the

stall zone diminishes, as indicated by the values of percent area tabulated for the last point number in each figure. For the SLTO configuration operating at 70% rotor speed (figure 71) the rotating stall zone disappeared after the stage's initial stall, and operation in stall was fairly steady. The stall zone reappeared as the stage was operated out of stall, and finally diminished in size as indicated in the figure. A similar result was obtained with the Stator A configuration of Reference 3.

A sample of the stall transient data and Kistler transducer traces are presented for the SLTO configuration operating at 100% rotor speed in figure 73. The method used for determining the rotating stall patterns from Kistler transducer data is discussed in Appendix C.

CONCLUDING DISCUSSION

Tests were conducted with a 0.5 hub/tip ratio compressor stage comprising a variable camber inlet guide vane, rotor, and a variable geometry stator having adjustable leading and trailing edge metal angles. Two design configurations that corresponded to SLTO and supersonic Mach number cruise operating conditions, and a third configuration geometrically midway between the two design configurations, were evaluated. Each of the three configurations were tested over a range of rotor speed and flow that encompassed both the SLTO and cruise design point conditions.

At the SLTO design conditions the SLTO configuration achieved a pressure ratio and adiabatic efficiency of 1.347 and 87.8%, respectively, compared to predicted values of 1.32 and 85.1%. At cruise design conditions the cruise configuration achieved a pressure ratio and efficiency of 1.15 and 90%, compared to predicted values of 1.129 and 88.3%. The efficiency of the SLTO configuration at cruise rotor speed and flow was 83%. In general, the cruise configuration permitted operation at lower flows and with correspondingly higher efficiencies than the SLTO configuration at the cruise rotor speed. Thus, the variable geometry features of the stage permitted efficient, stall-free operation over a wider range of corrected flow than that possible with the SLTO (fixed-geometry) configuration.

The Stator B configuration permitted adjustment of both the leading and trailing edge metal angles for operation at other than SLTO design conditions. This adjustment resulted in the desired match between rotor exit and stator inlet, and stator exit and hypothetical 2nd-stage rotor inlet conditions at the cruise design point. However, for the cruise setting the stator minimum

losses at all span locations were significantly higher than those predicted for a conventional stator designed with the same air turning requirement. The high losses resulted from the generation of a broad wake that was considered to be the product of mixed wakes from the two airfoil segments of Stator B. Although the Stator B rear airfoil segment leading edge was designed to operate with low losses over a wide range of incidence angle (Reference 1), it is apparent that additional design criteria, pertinent to tandem airfoil design, are necessary to define a low-loss Stator B type configuration.

REFERENCES

1. "Single Stage Experimental Evaluation of Variable Geometry Guide Vanes and Stators, Part I - Analysis and Design," NASA CR-54554, PWA FR-2112, August, 1968.
2. "Single Stage Experimental Evaluation of Variable Geometry Guide Vanes and Stators, Part II - Annular Cascade Investigation of Candidate Variable Geometry Designs," NASA CR-54555, PWA FR-2297, July, 1967.
3. "Single Stage Experimental Evaluation of Variable Geometry Guide Vanes and Stators, Part III - Data and Performance for Variable Camber Guide Vanes and Stator A," NASA CR-54556, PWA FR-2638, December 1968.

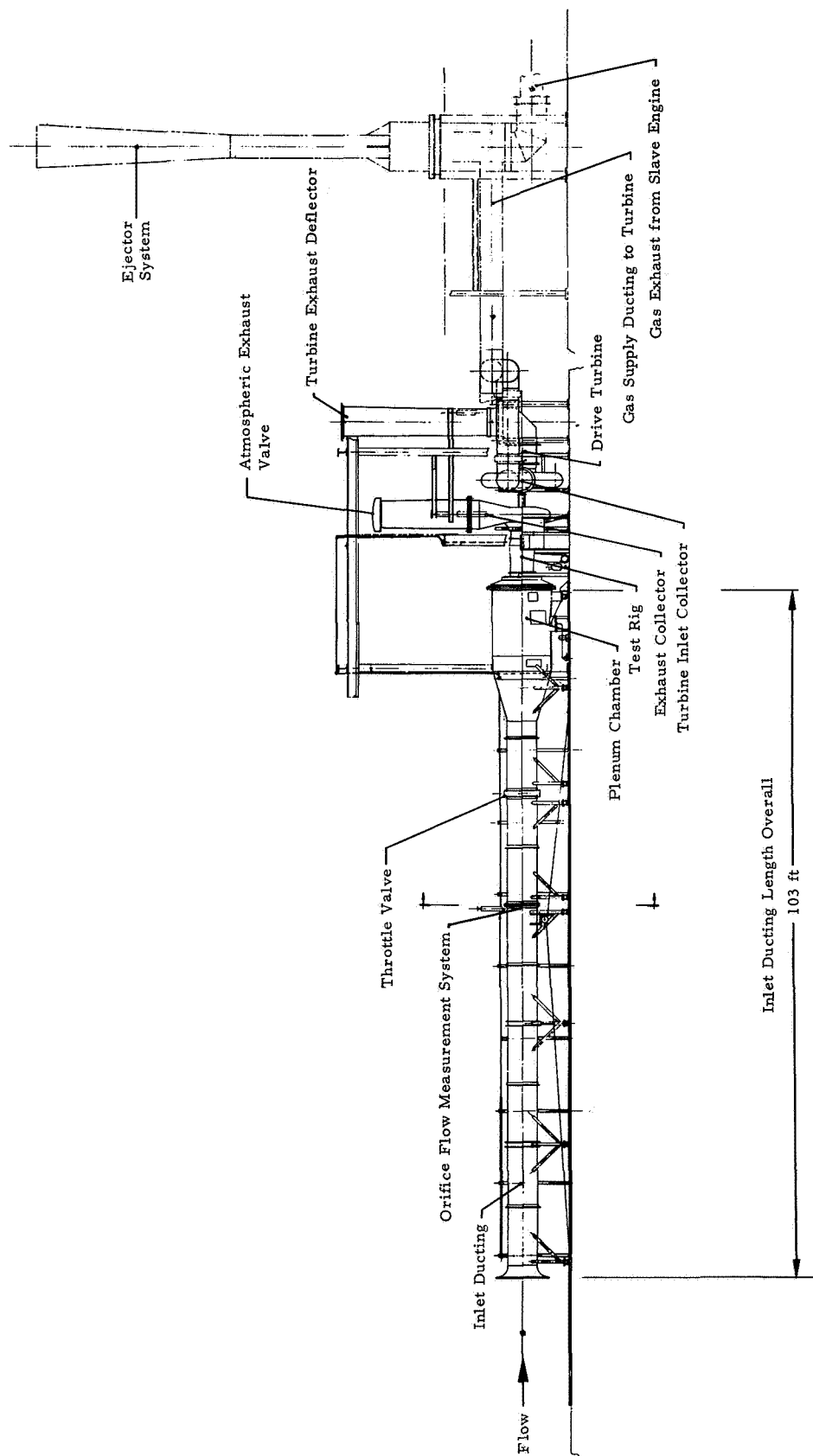
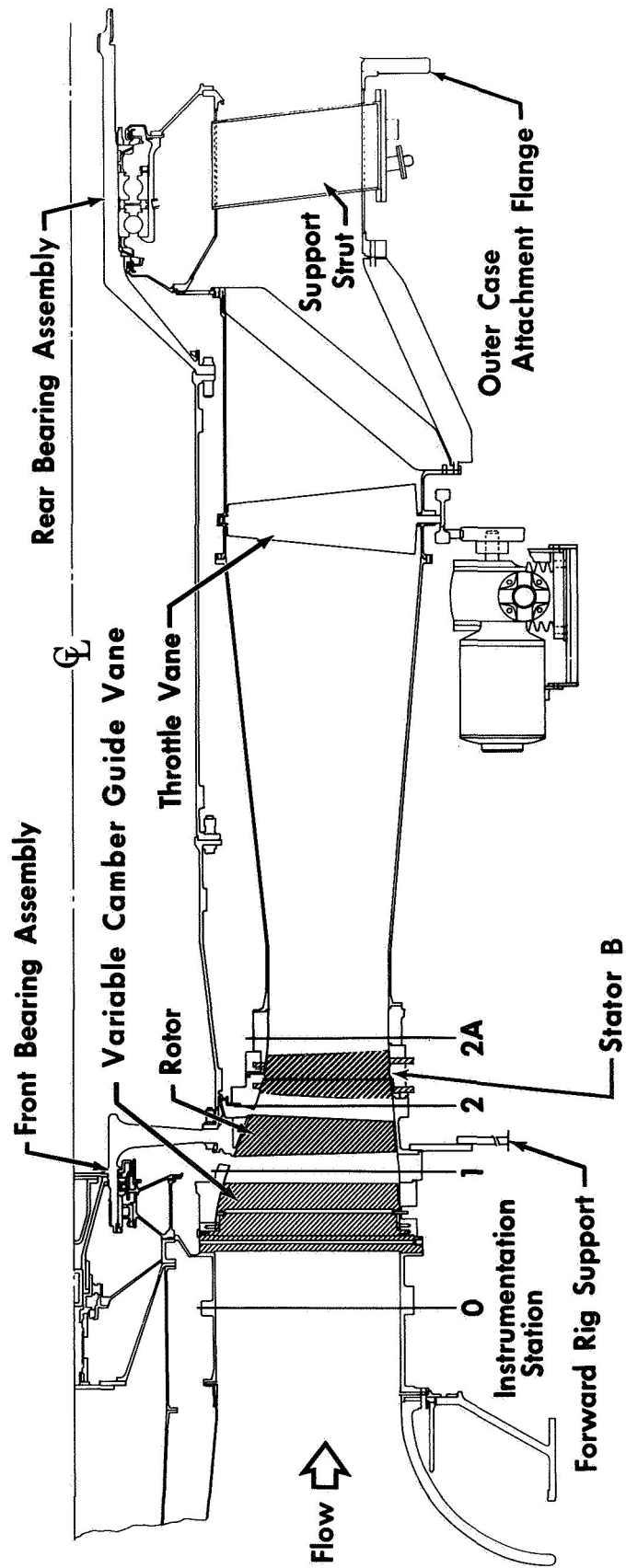


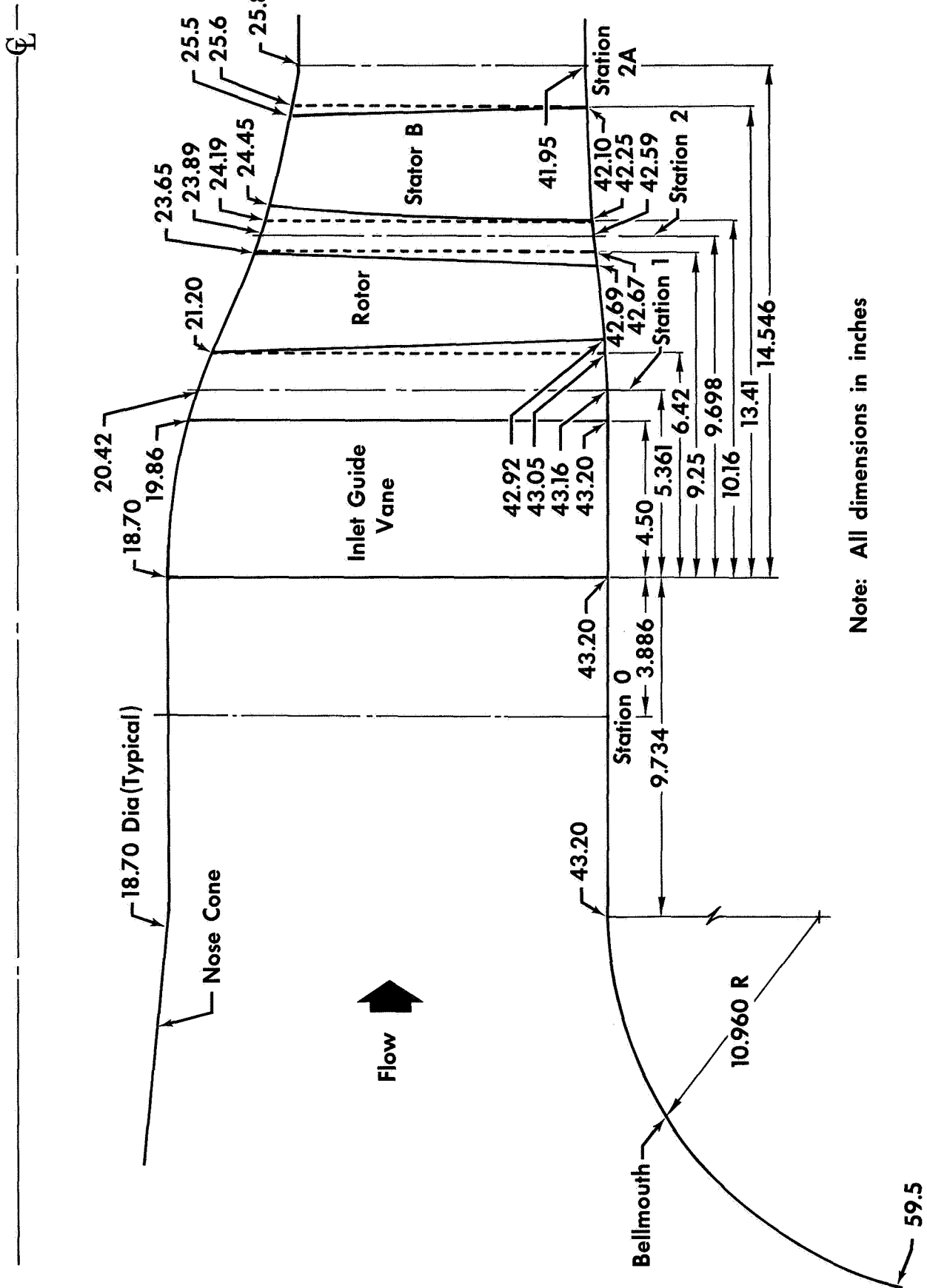
Figure 1. Compressor Research Facility

FD 10891D



GS 2120D

Figure 2. Compressor Rig

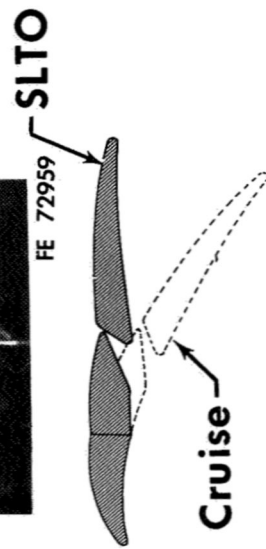
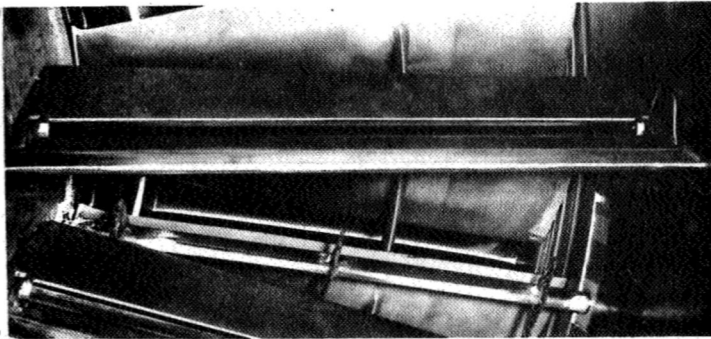


Note: All dimensions in inches

Figure 3. Section View of Flowpath

GS 6614A

**Variable-Camber
Inlet Guide Vane
(Cruise Position)**



**Stator B
(Cruise Position)**

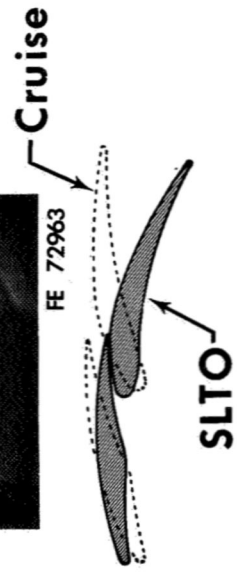
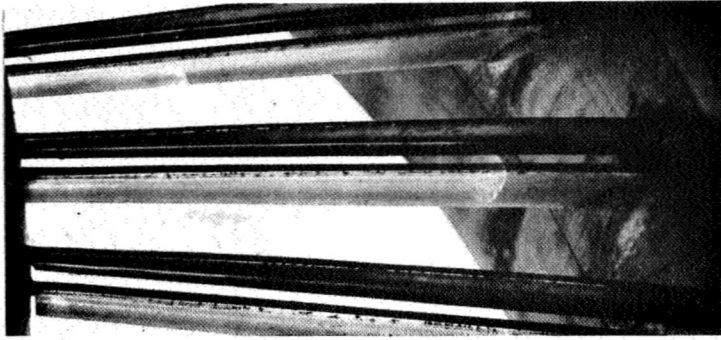
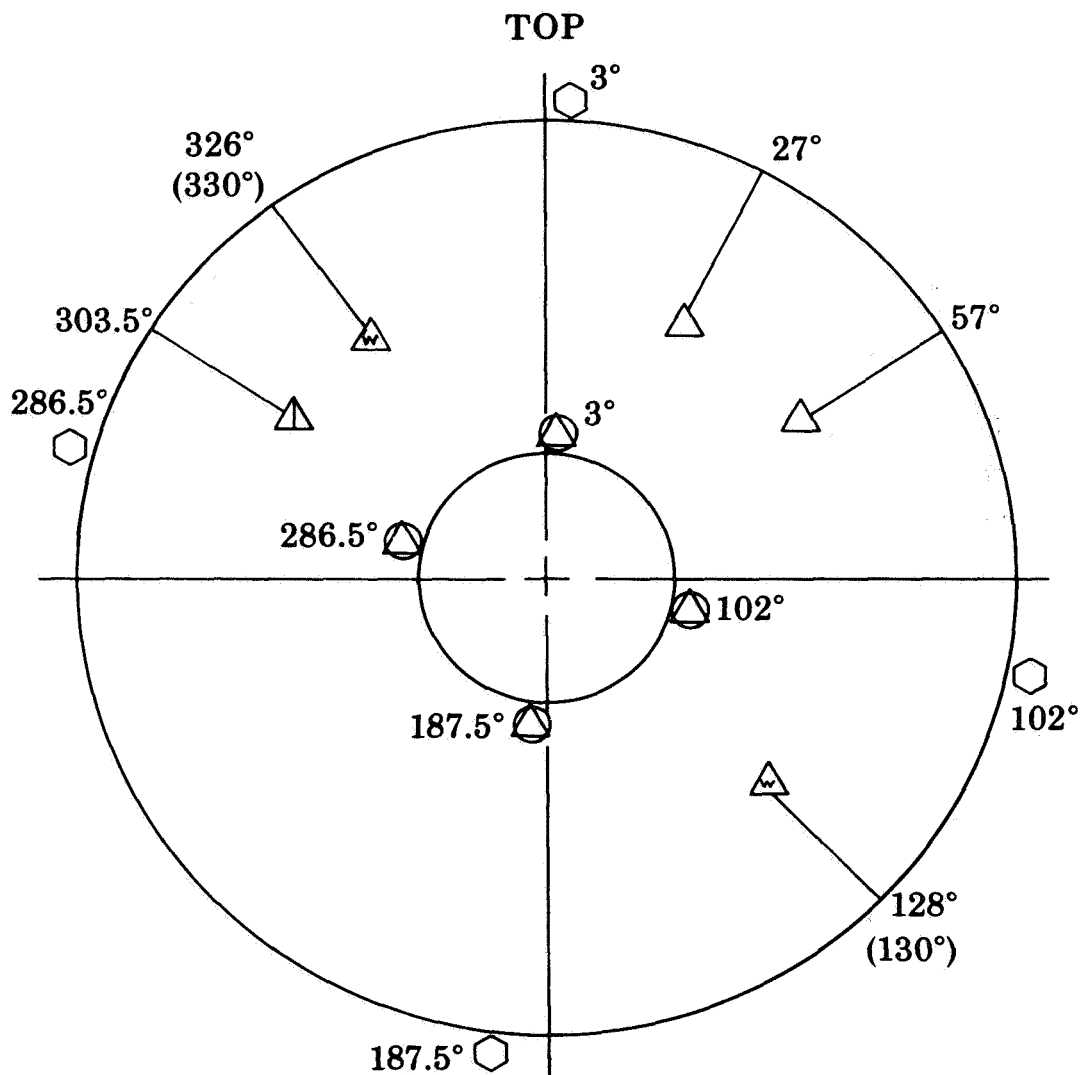


Figure 4. Variable Geometry Inlet Guide Vane and Stator B

GS 6613C



- | | |
|--------------------------------------|---|
| ⬡ OD Wall Static Pressure (4) | ⚙ 8-Degree Wedge Traverse Probe (1) |
| ⊗ ID Wall Static Pressure (4) | ⚙ Wake Total Pressure Rake (2) |
| △ 20-Degree Wedge Traverse Probe (2) | (°) Indicates Wake Rake Location For Cruise Configuration |

Figure 5. Instrumentation Station 1,
View Looking Forward

FD 25194

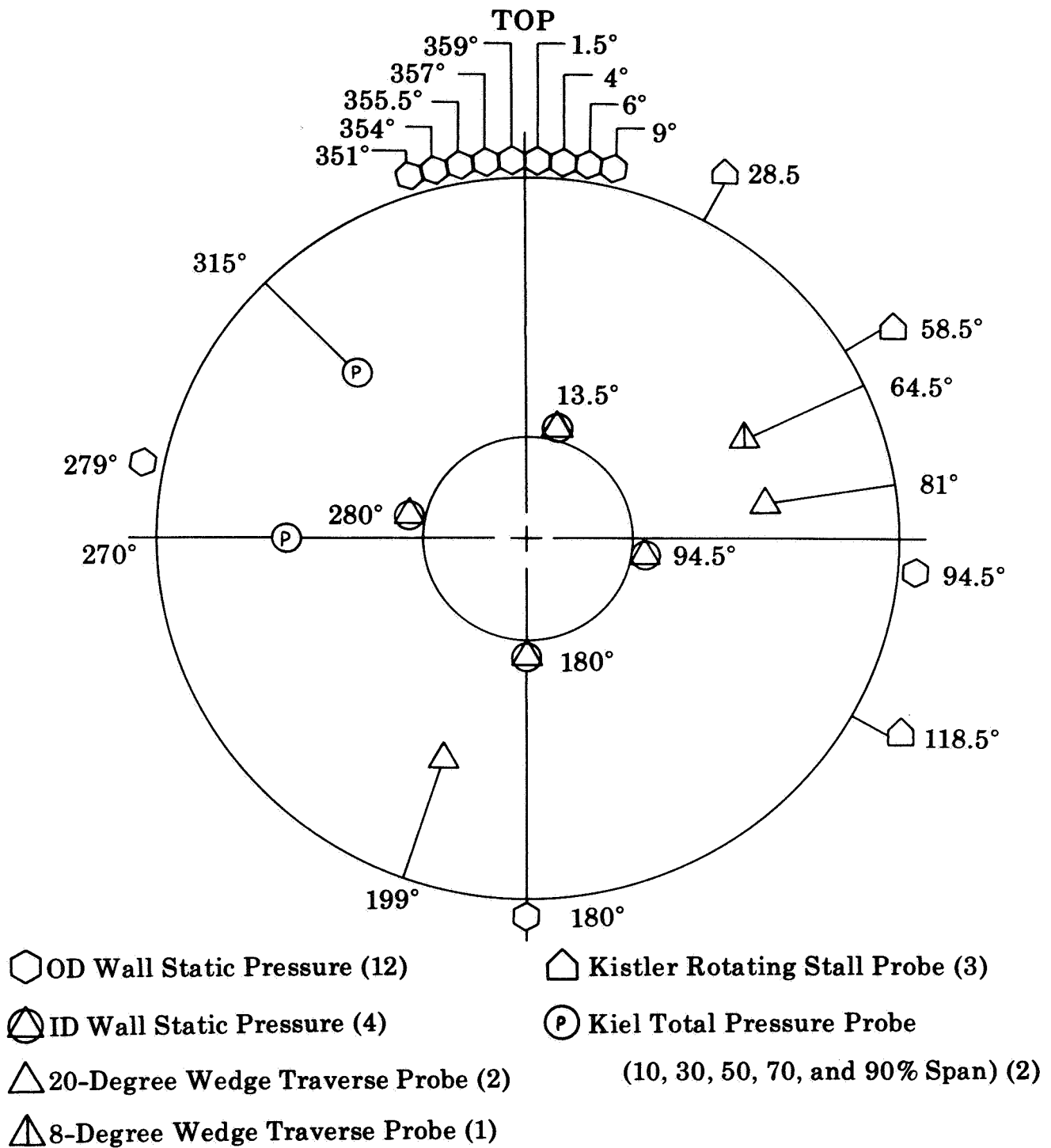
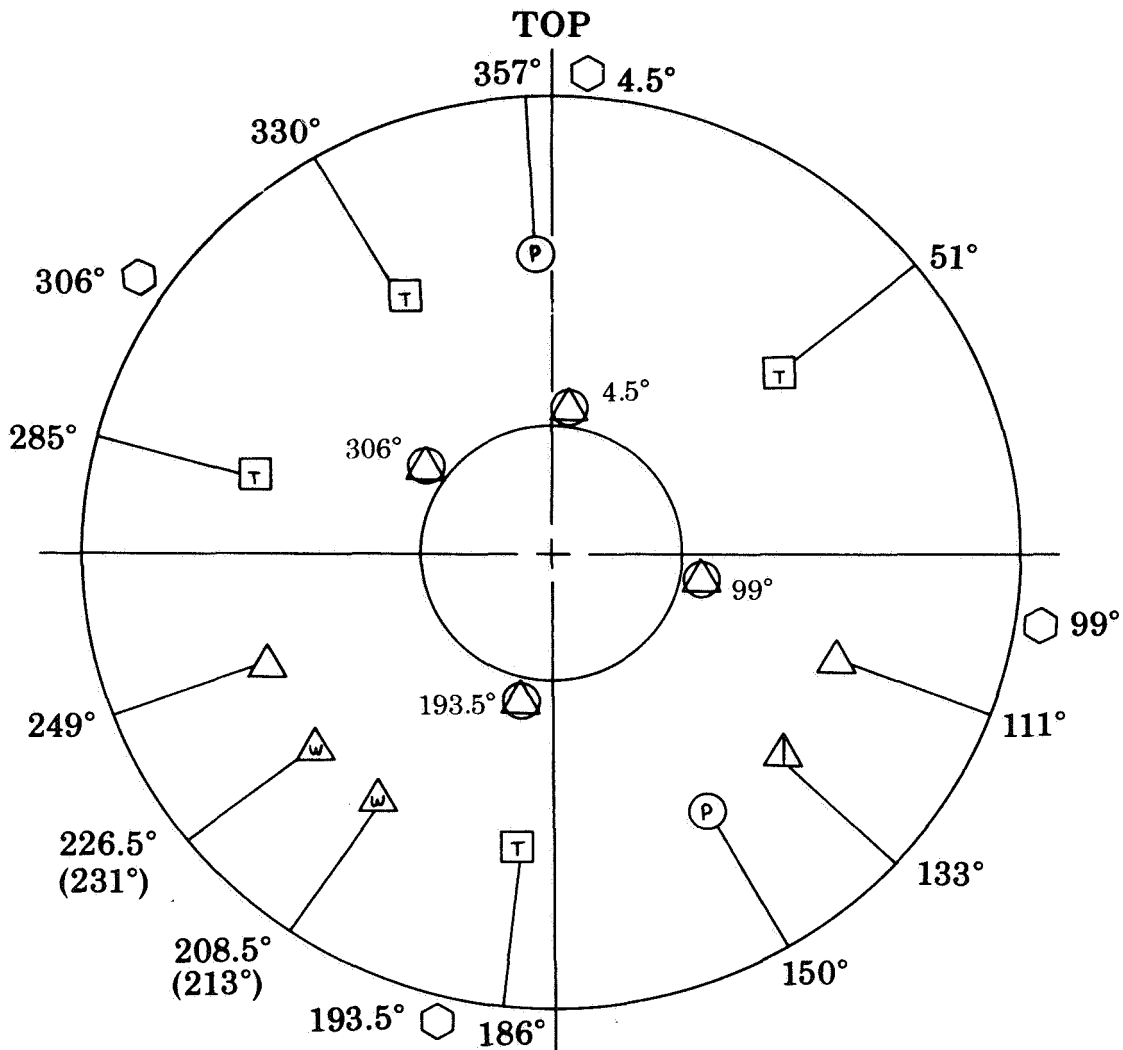


Figure 6. Instrumentation Station 2,
View Looking Forward

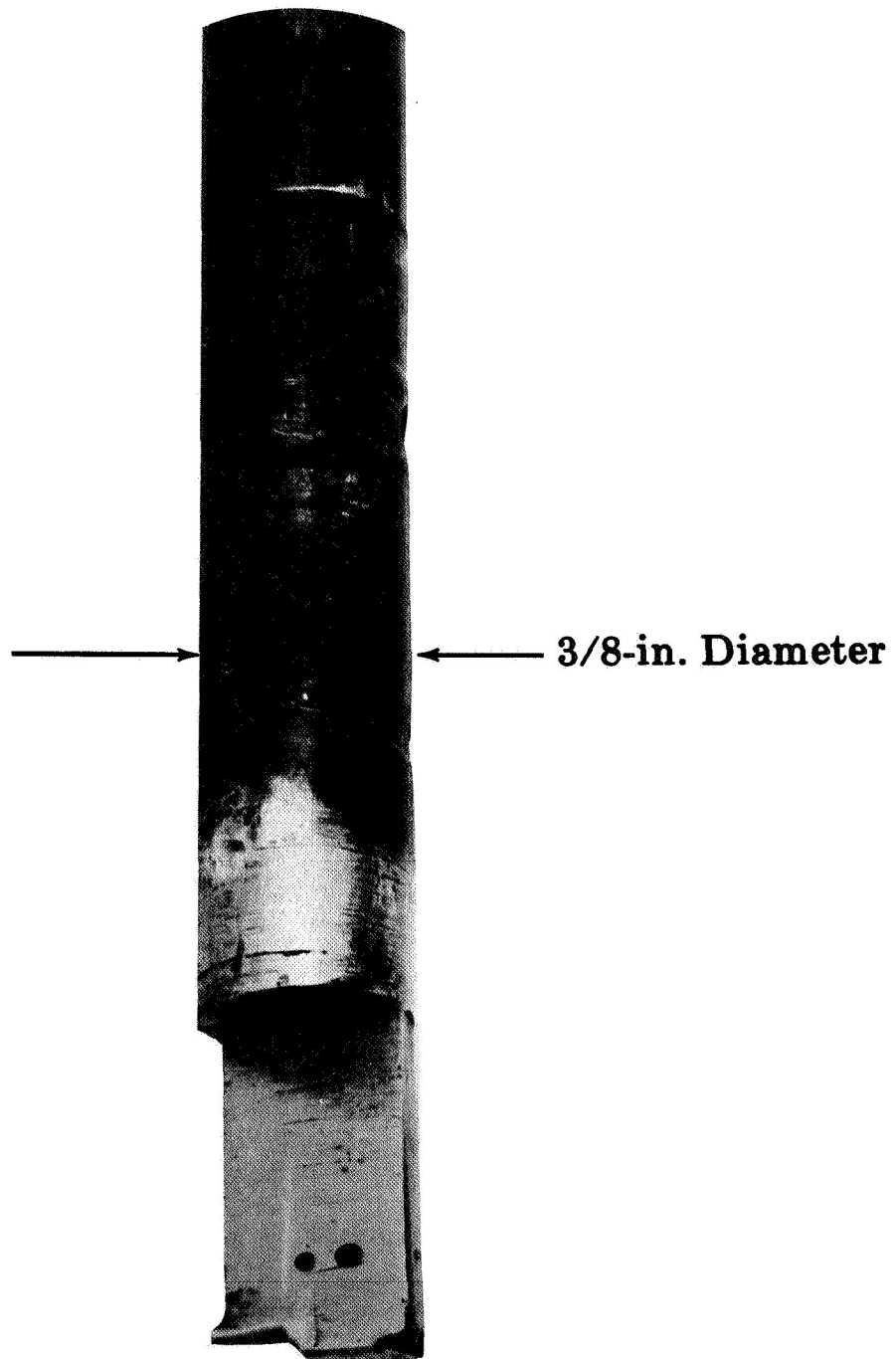
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- | | |
|--|---|
| <p> OD Wall Static Pressure (4)
 ID Wall Static Pressure (4)
 20-Degree Wedge Probe (2)
 8-Degree Wedge Traverse Probe (1)
 Wake Total Pressure Rake (2)
 (°) Indicates Wake Rake Location
 For Cruise Configuration </p> | <p> Temperature Rake
 (10, 30, 50, 70 and 90% Span) (4)
 Kiel Total Pressure Probe
 (10, 30, 50, 70 and 90% Span) (2) </p> |
|--|---|

Figure 7. Instrumentation Station 2A,
View Looking Forward

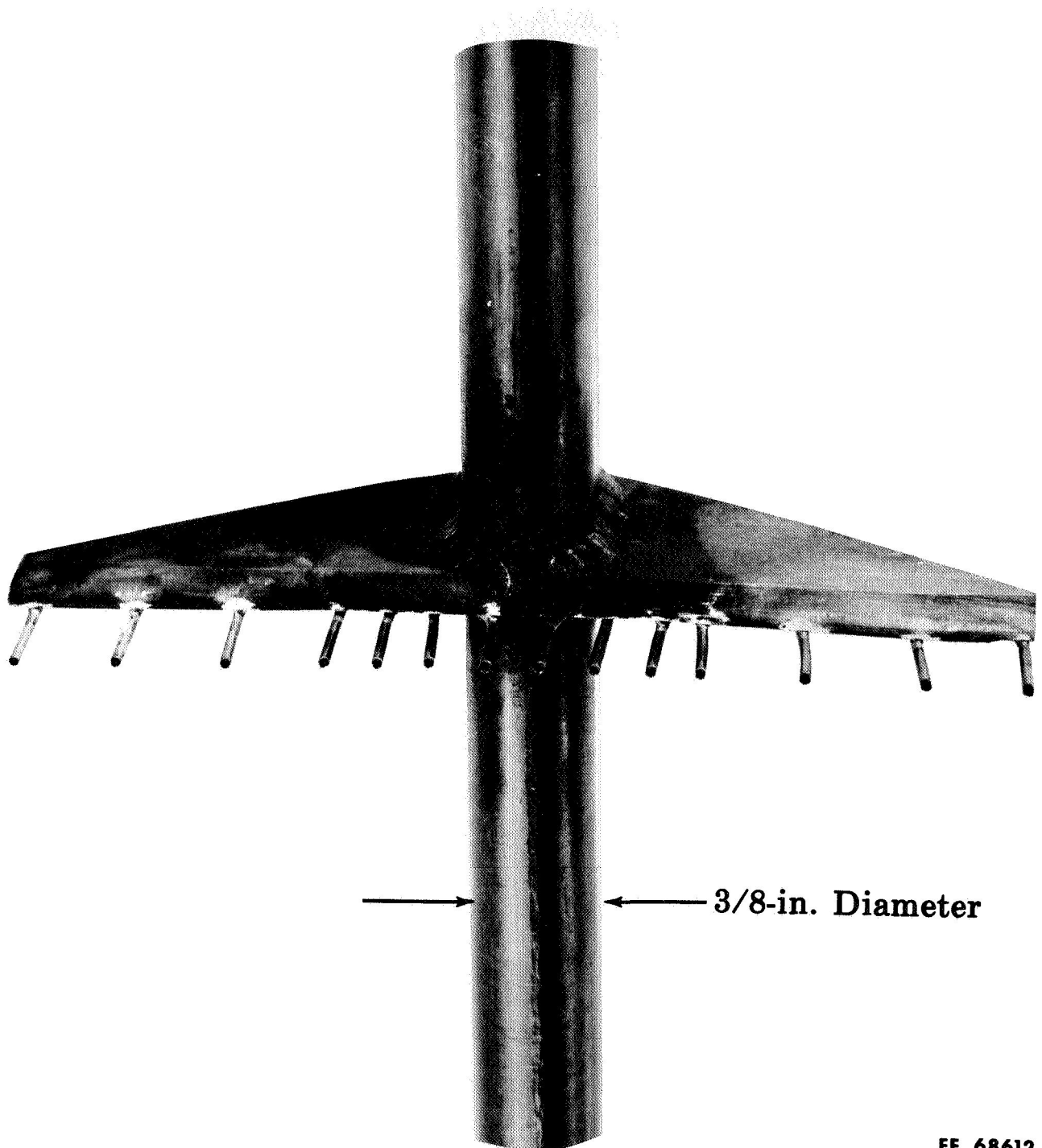
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FE 73903

Figure 8. 20-Degree Wedge Probe

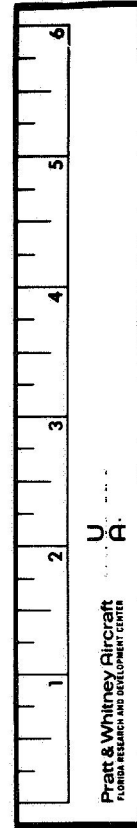
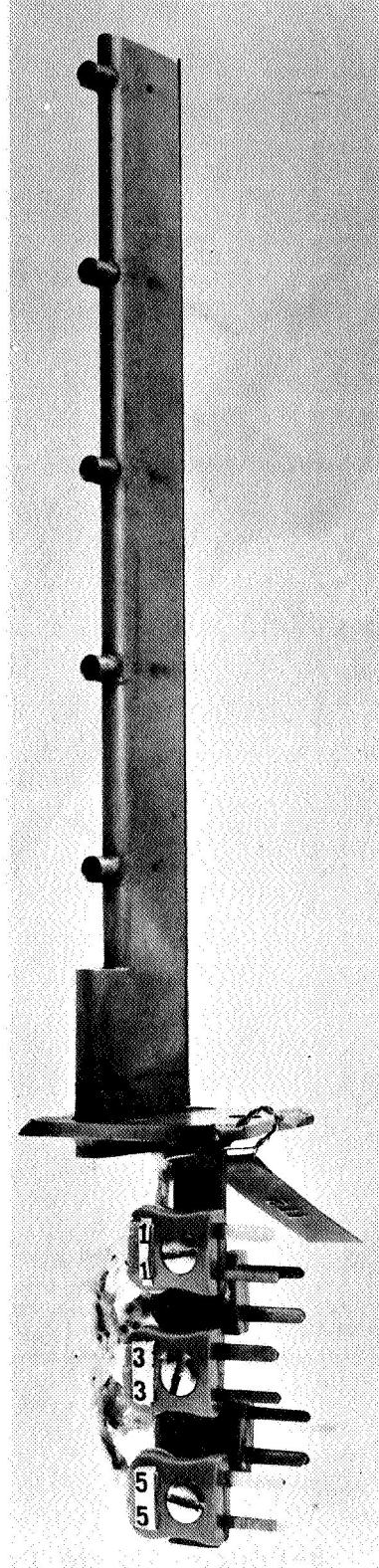
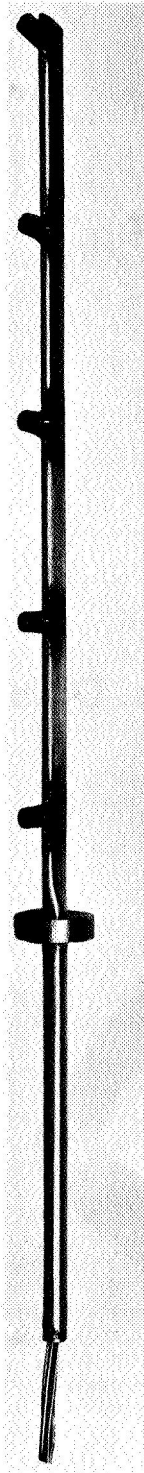
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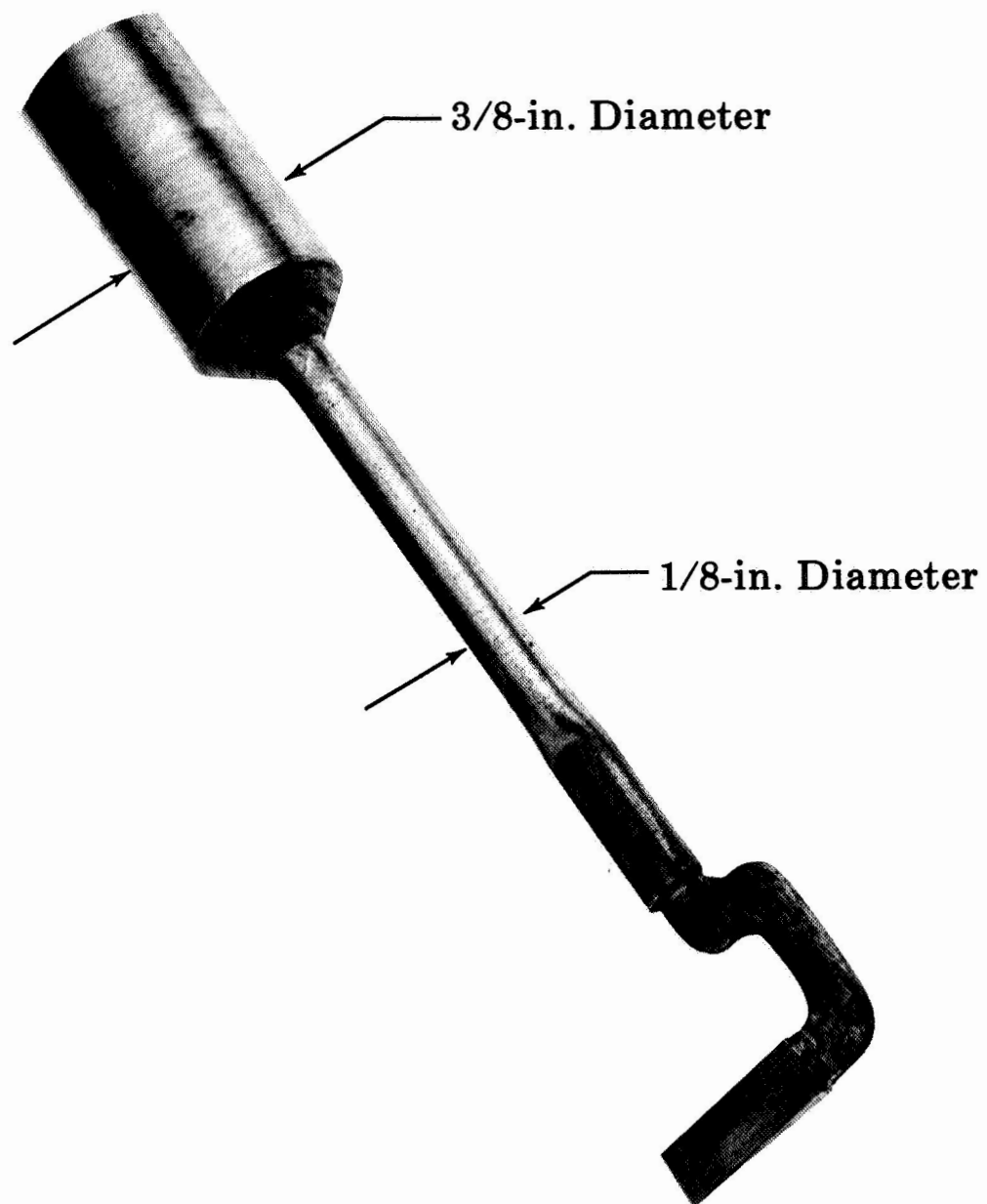
Figure 9. Typical Wake Probe Rake

FD 25198



FE 80200

Figure 10. Total Temperature and Total Pressure Rakes



FE 73904

Figure 11. 8-Degree Wedge Probe

FD 25200

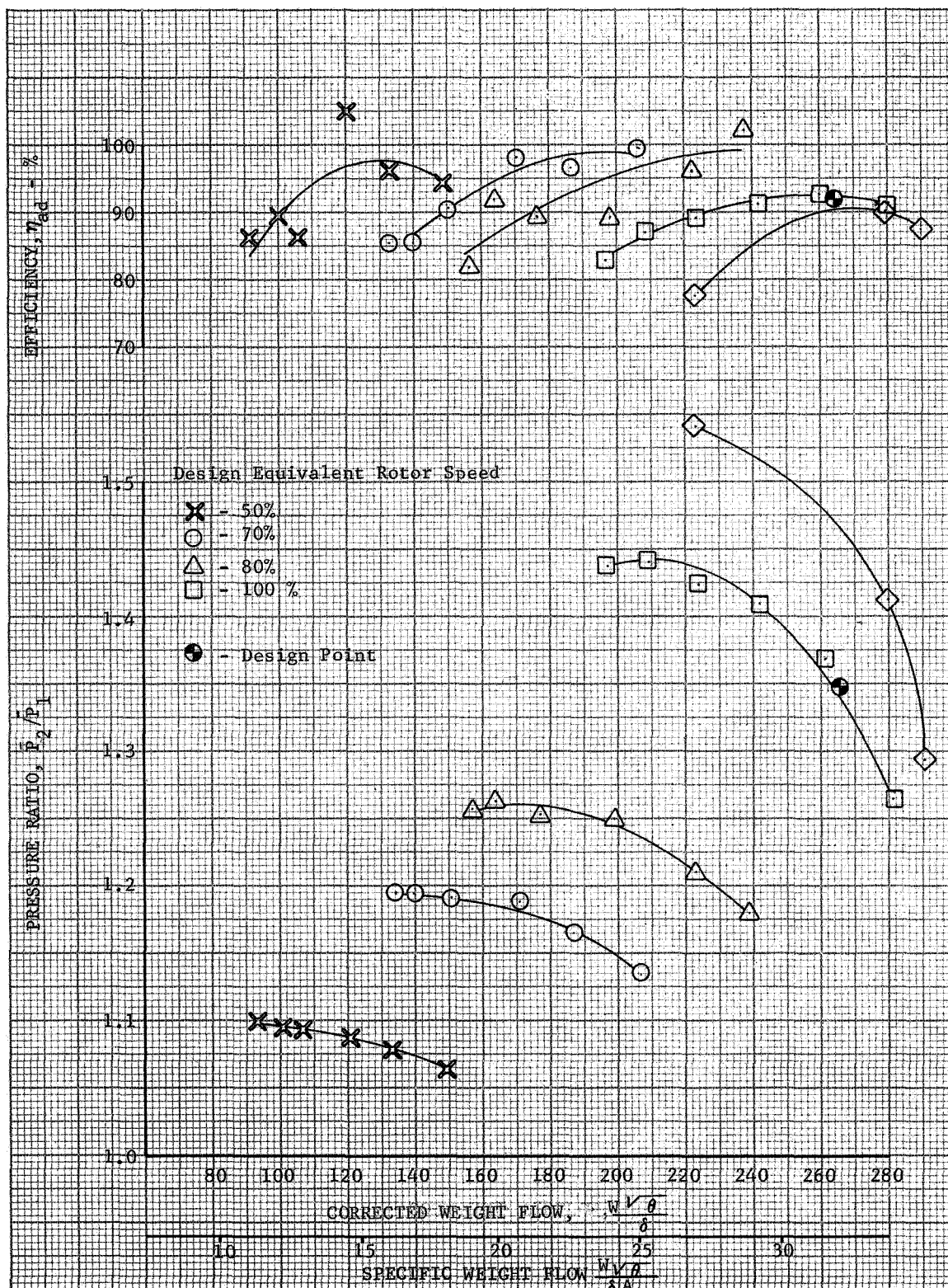


Figure 12. Overall Performance:
Rotor Only, SLTO Configuration

DF 68753

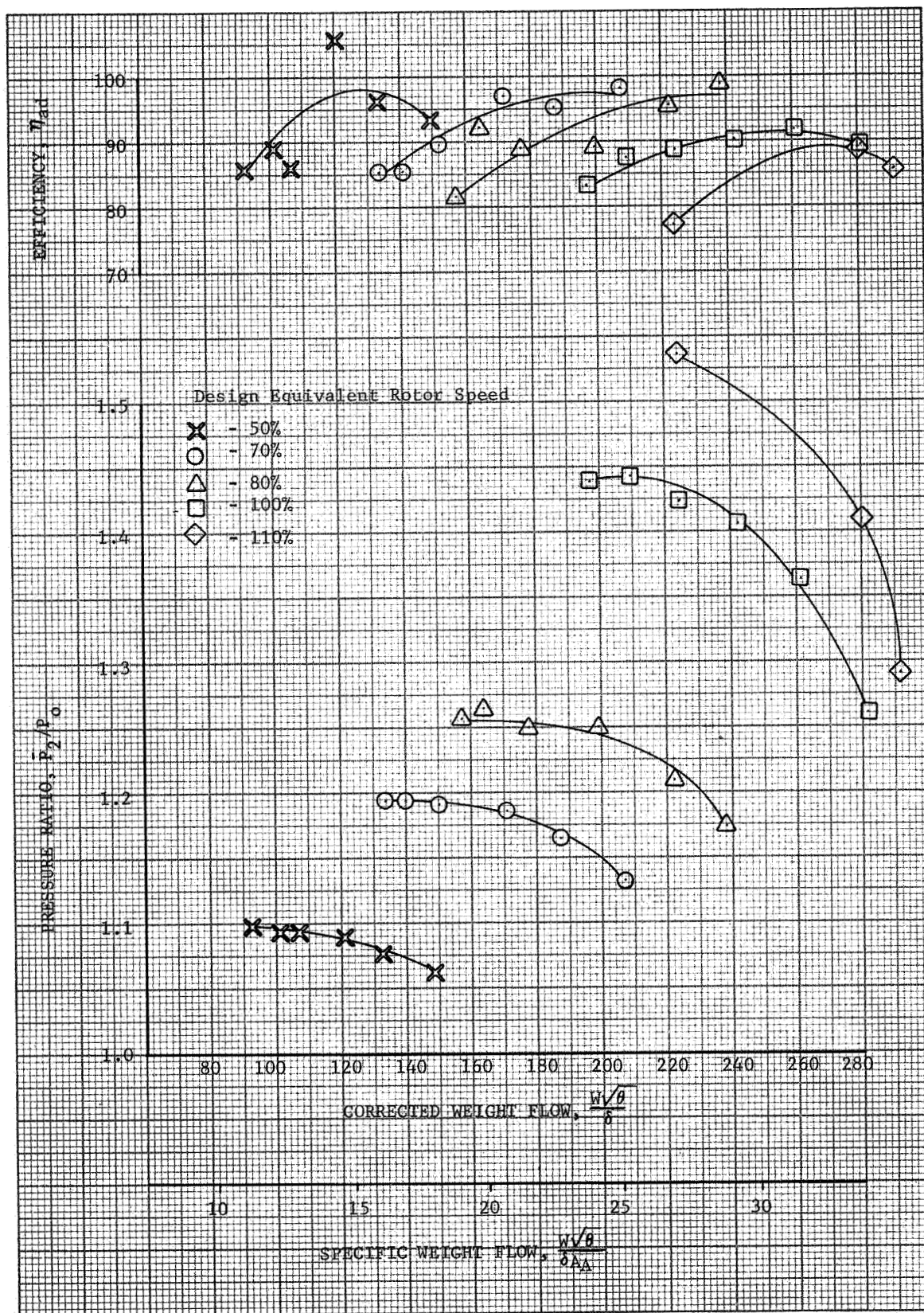


Figure 13. Overall Performance: Guide Vane Rotor, SLTO Configuration

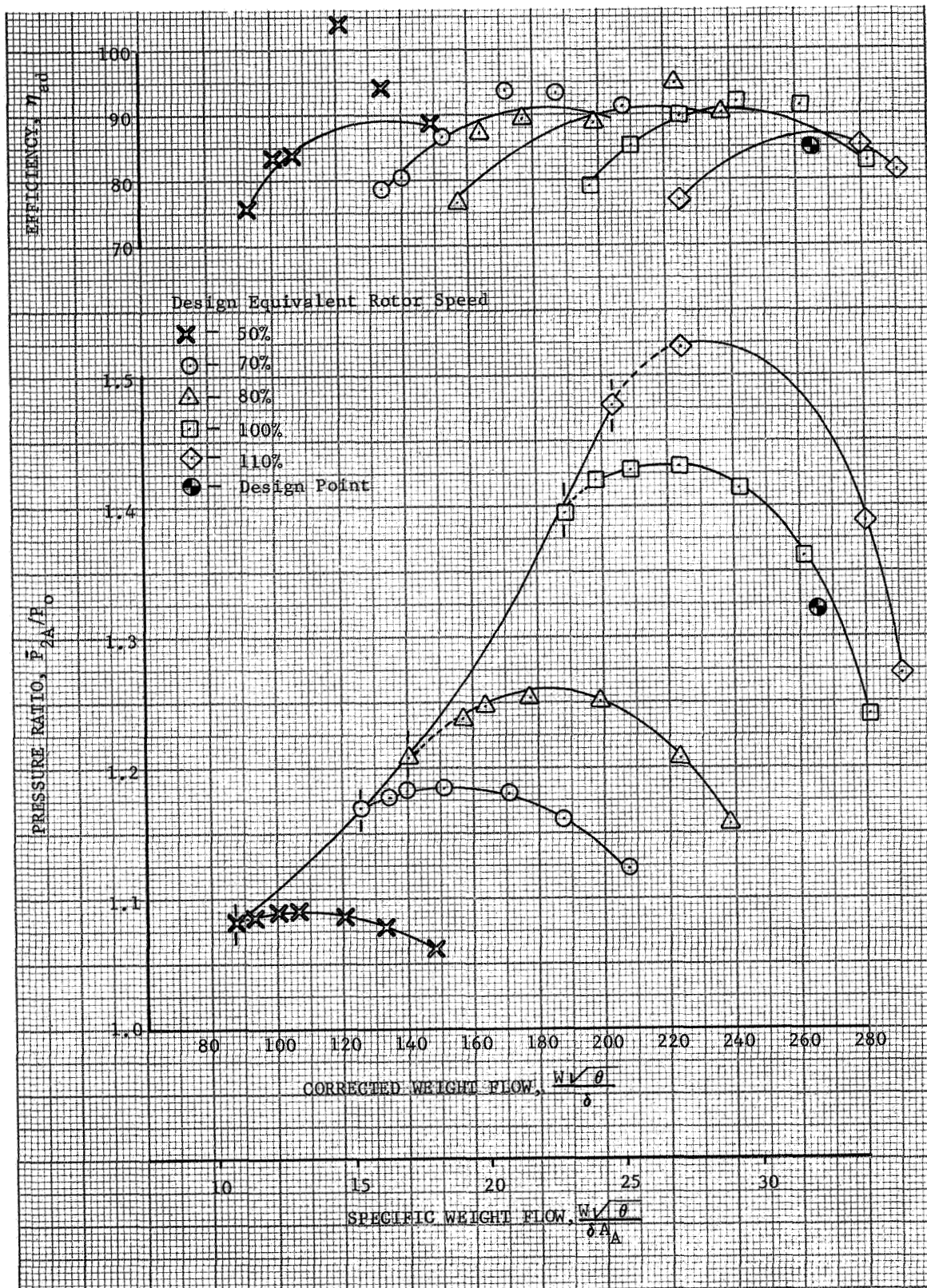


Figure 14. Overall Performance: Guide Vane, Rotor, Stator B, SLTO Configuration

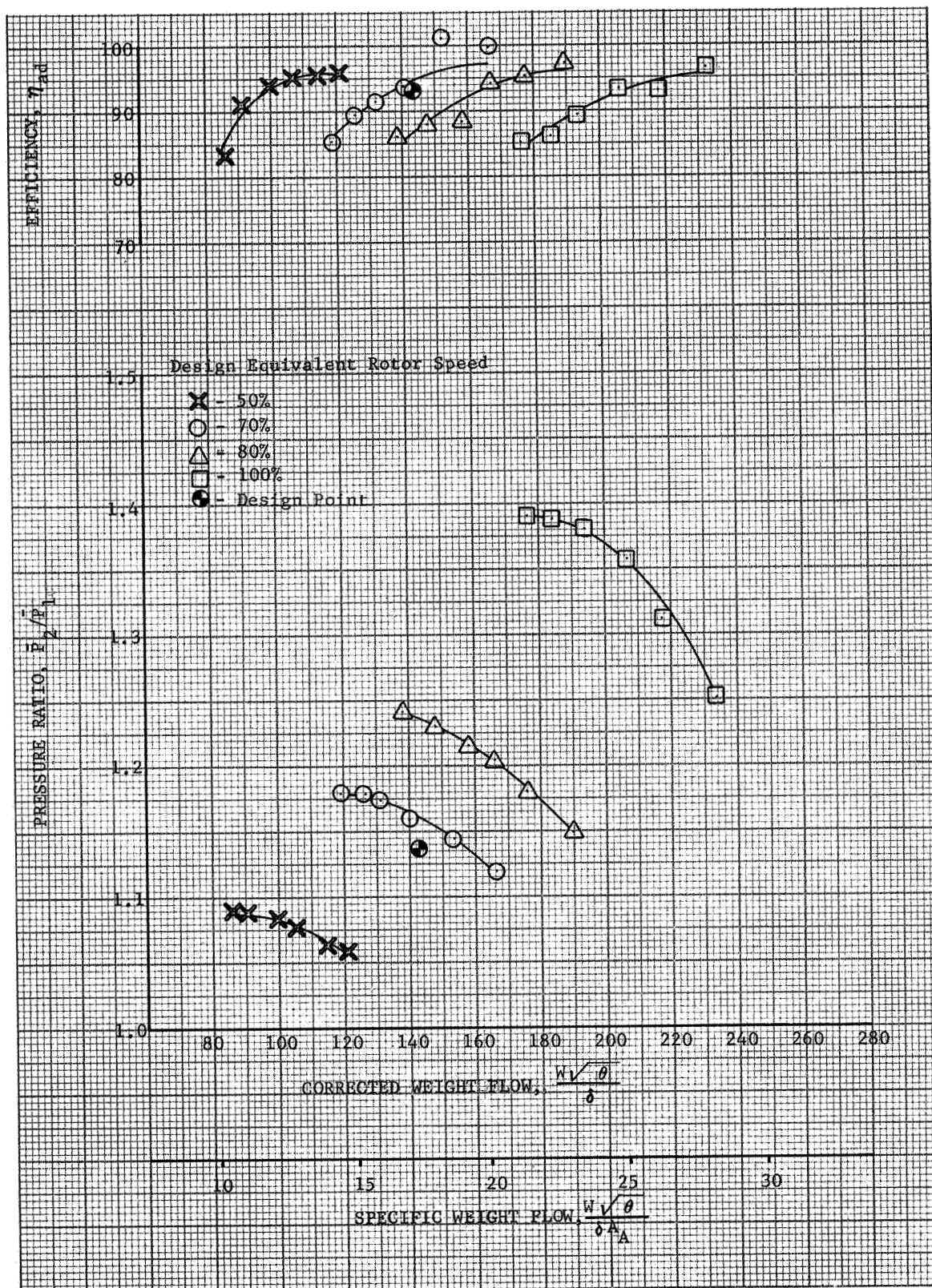


Figure 15. Overall Performance:
Rotor Only, Cruise Configuration

DF 68756

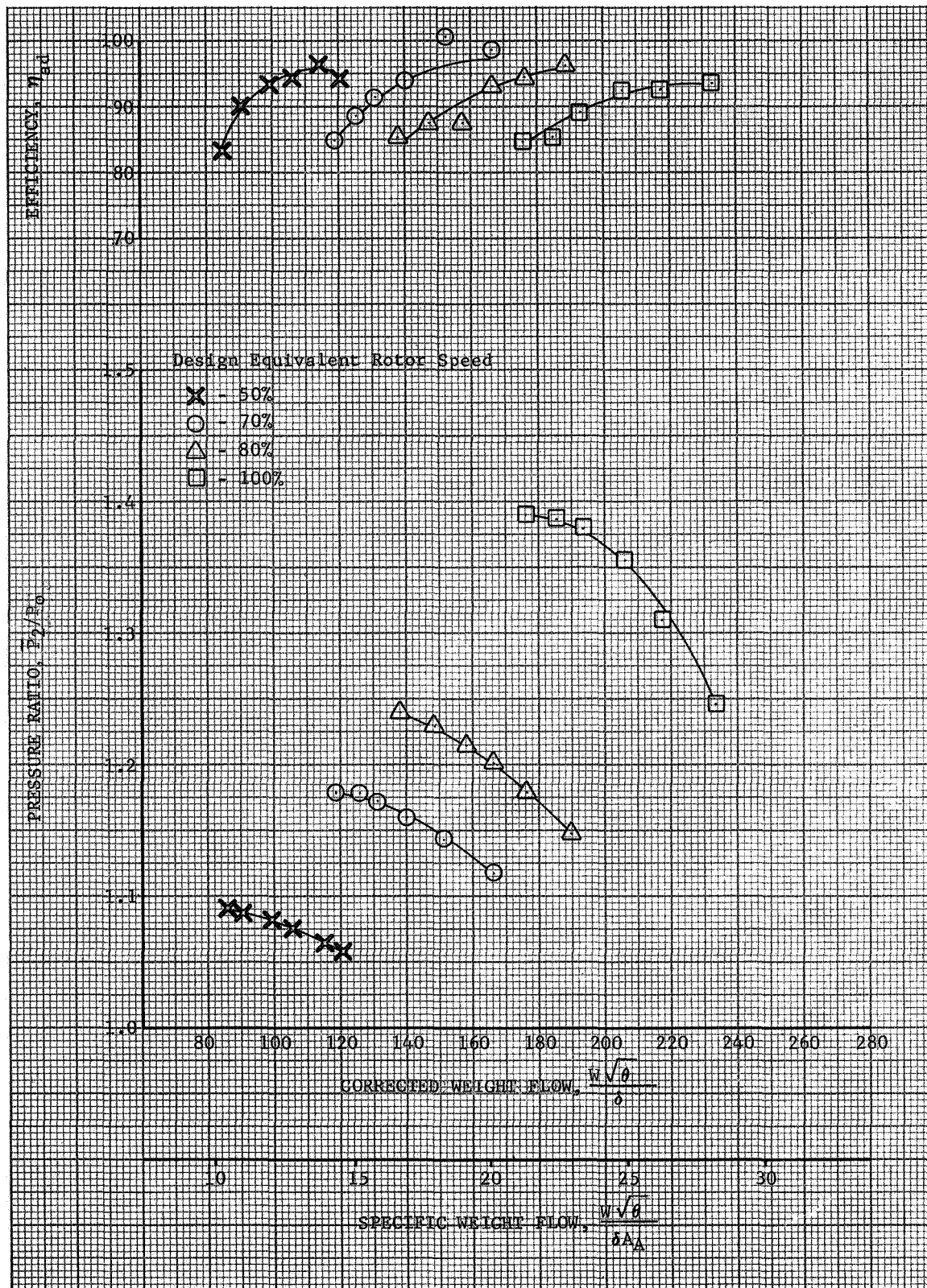


Figure 16. Overall Performance: Guide Vane, Rotor, Cruise Configuration

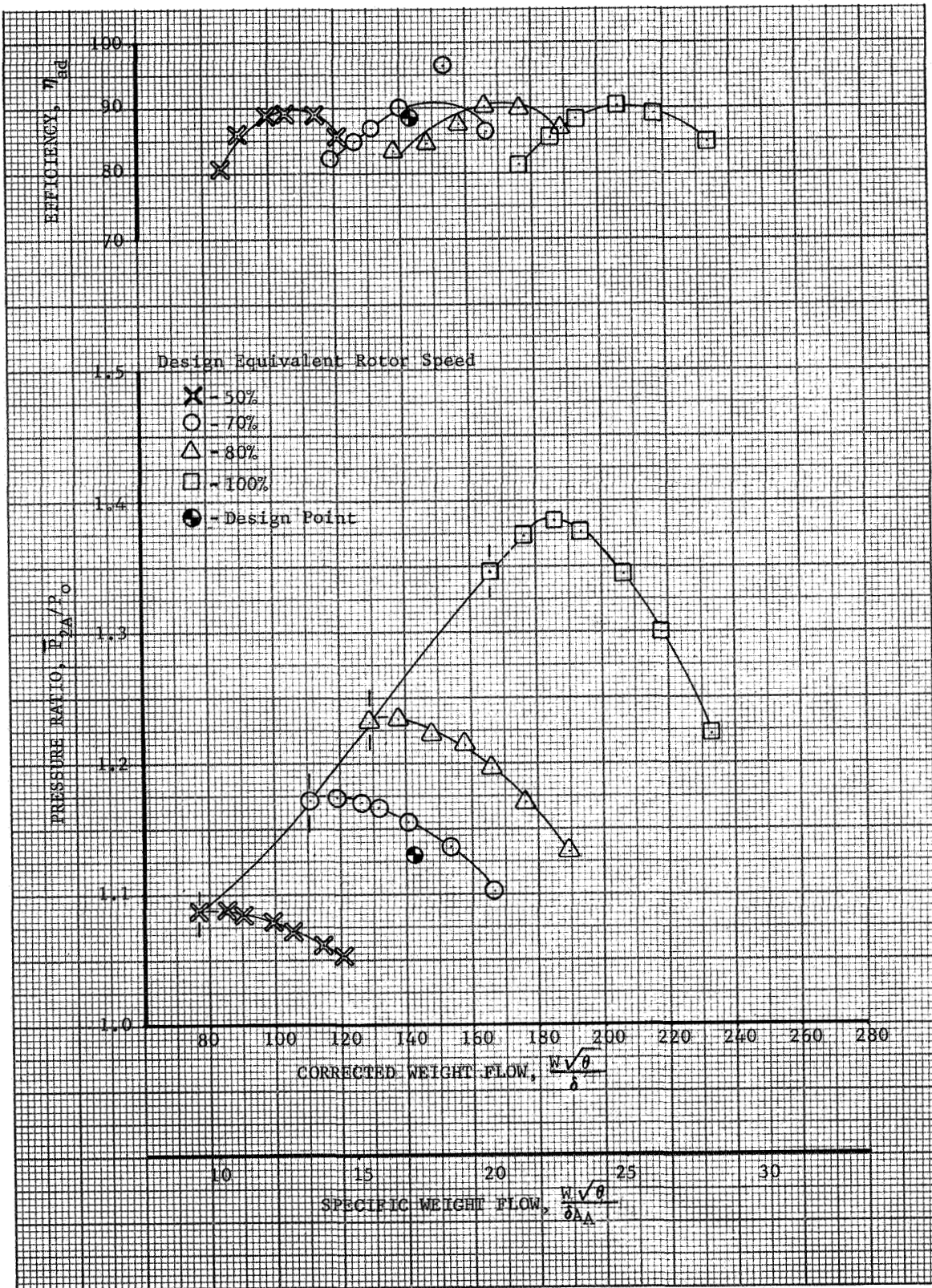
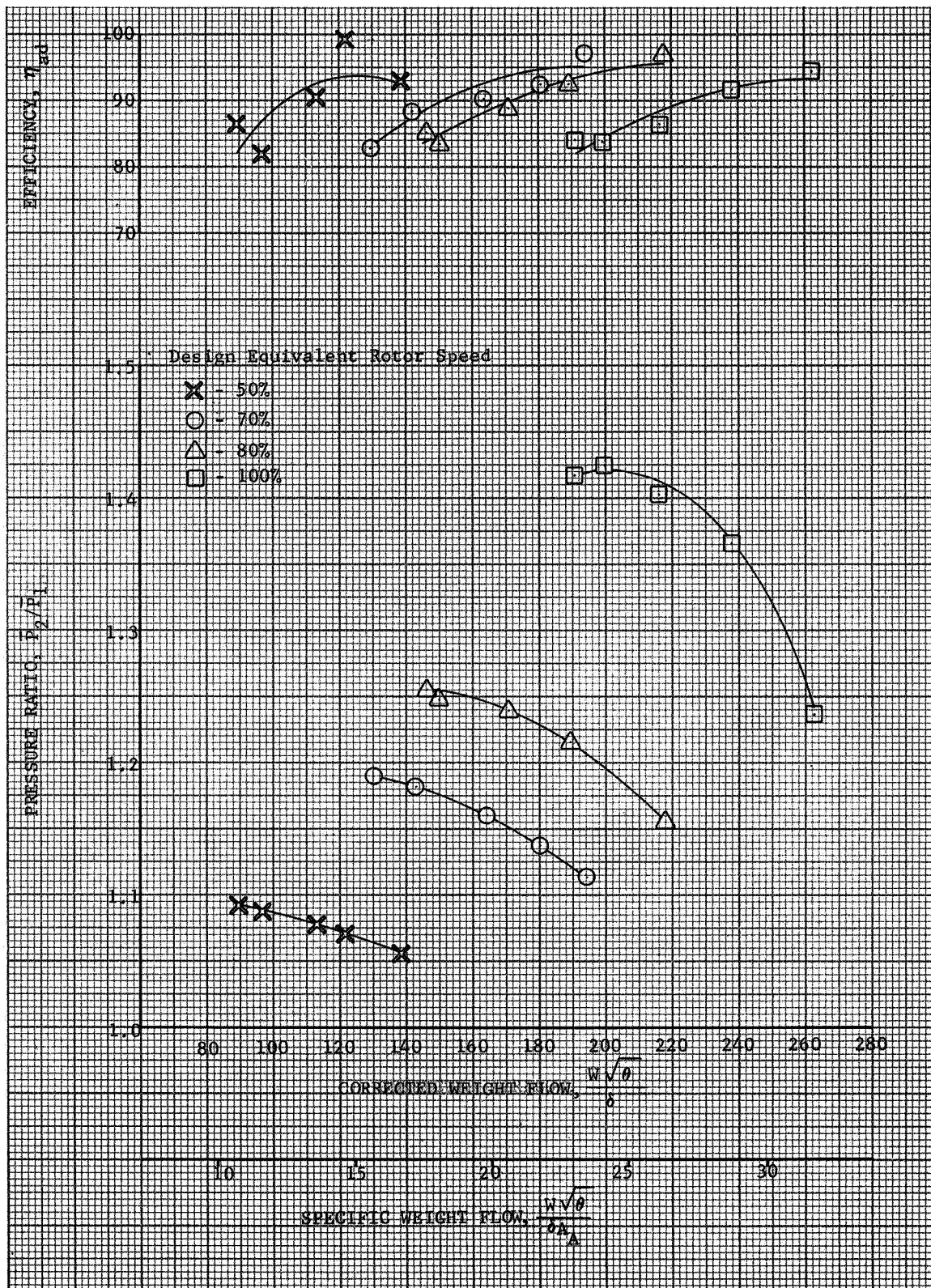


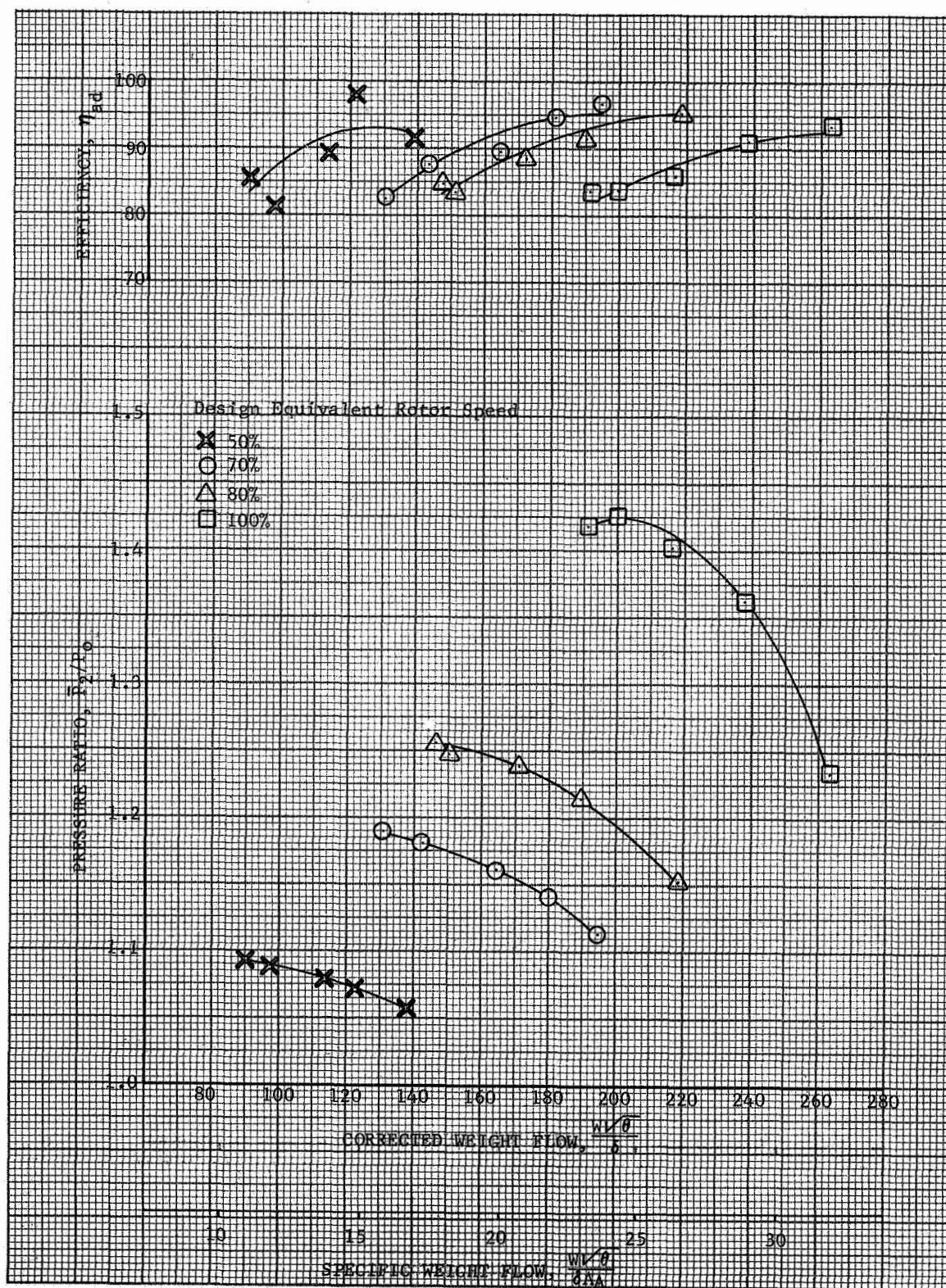
Figure 17. Overall Performance: Guide Vane, Rotor, Stator B, Cruise Configuration

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Figure 18. Overall Performance:
Rotor Only, Intermediate Configuration



DF 68760

Figure 19. Overall Performance: Guide Vane, Rotor, Intermediate Configuration

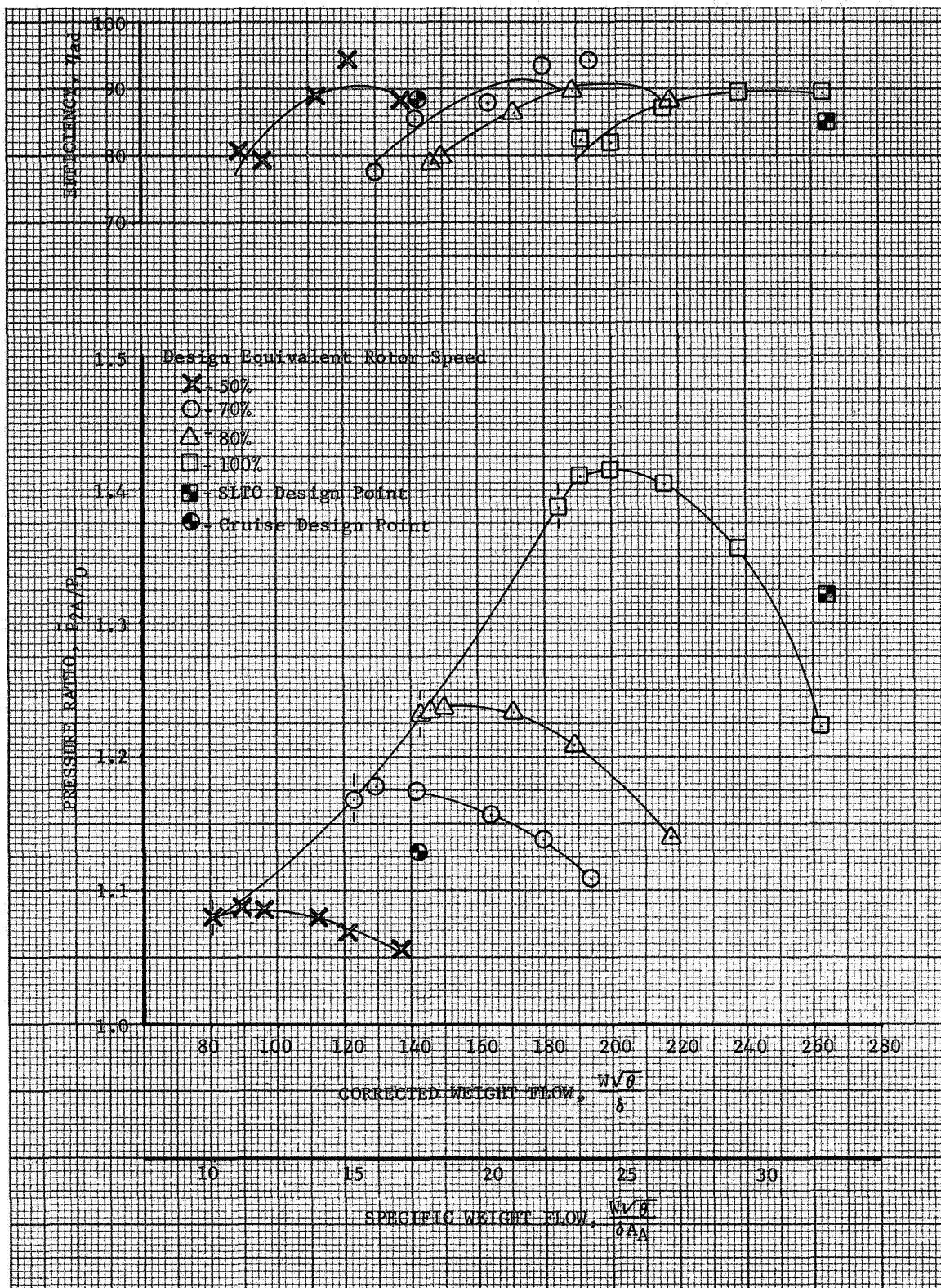
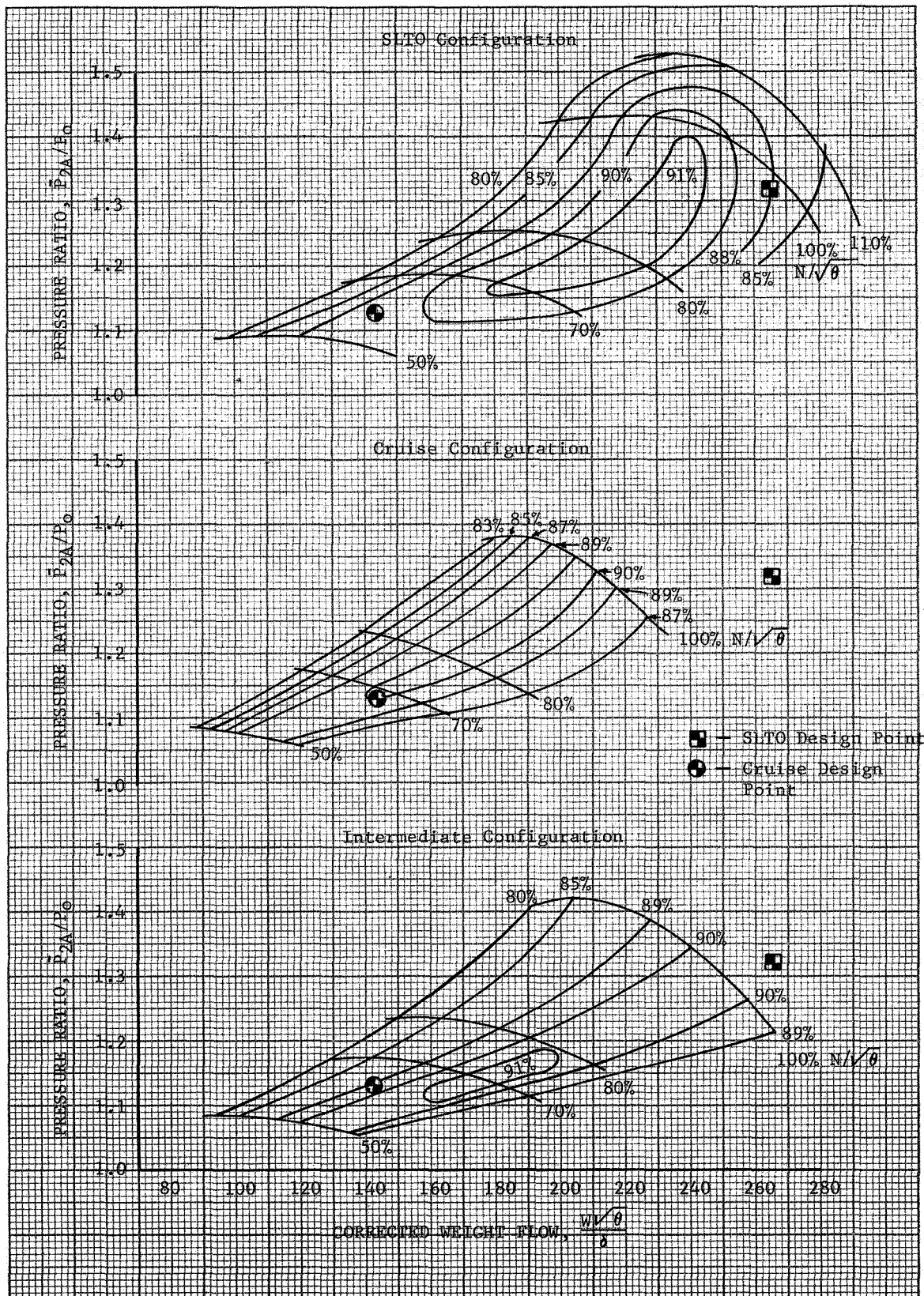


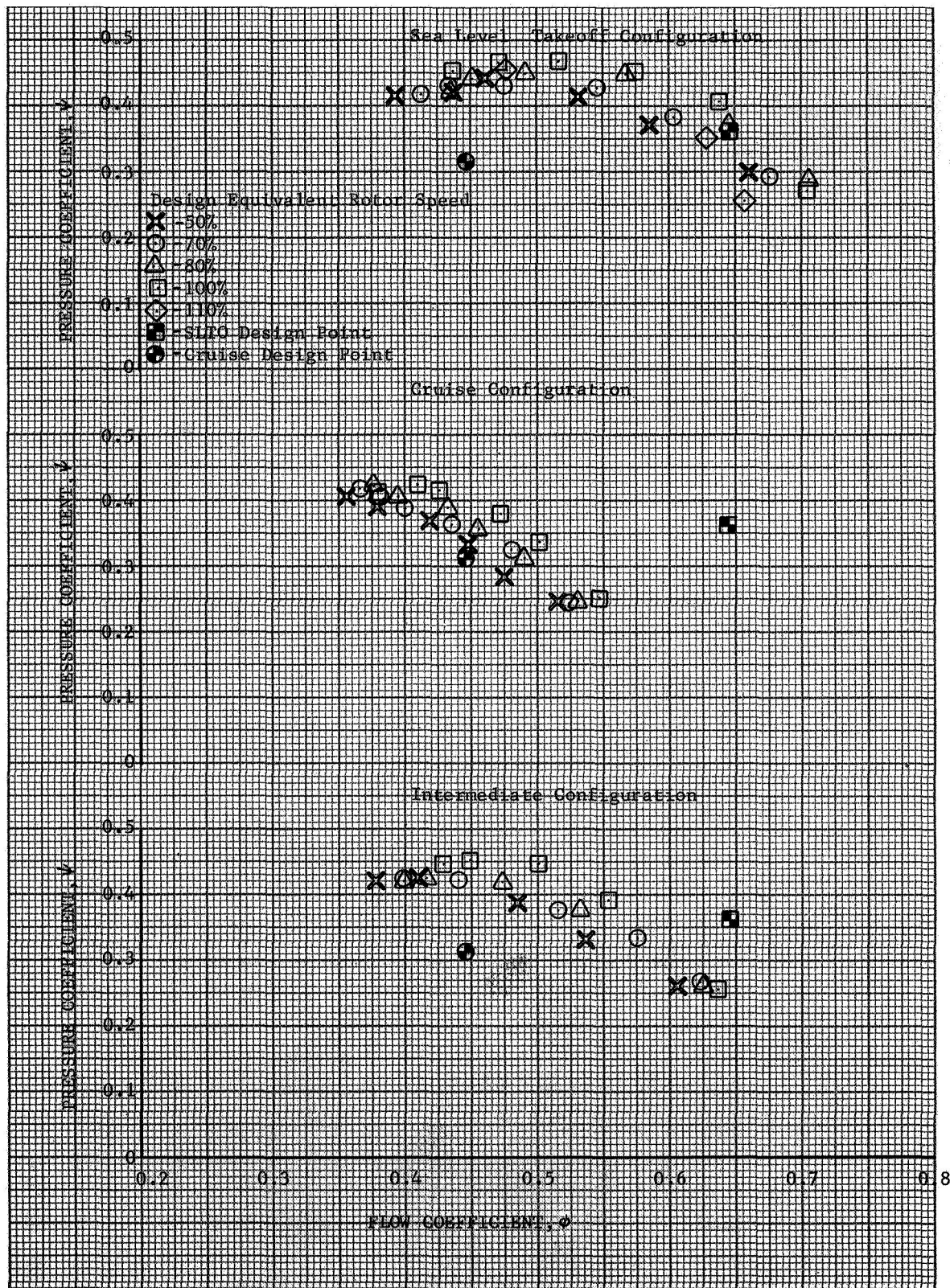
Figure 20. Overall Performance: Guide Vane, Rotor, Stator B, Intermediate Configuration

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DF 68762

Figure 21. Stage Efficiency Contours



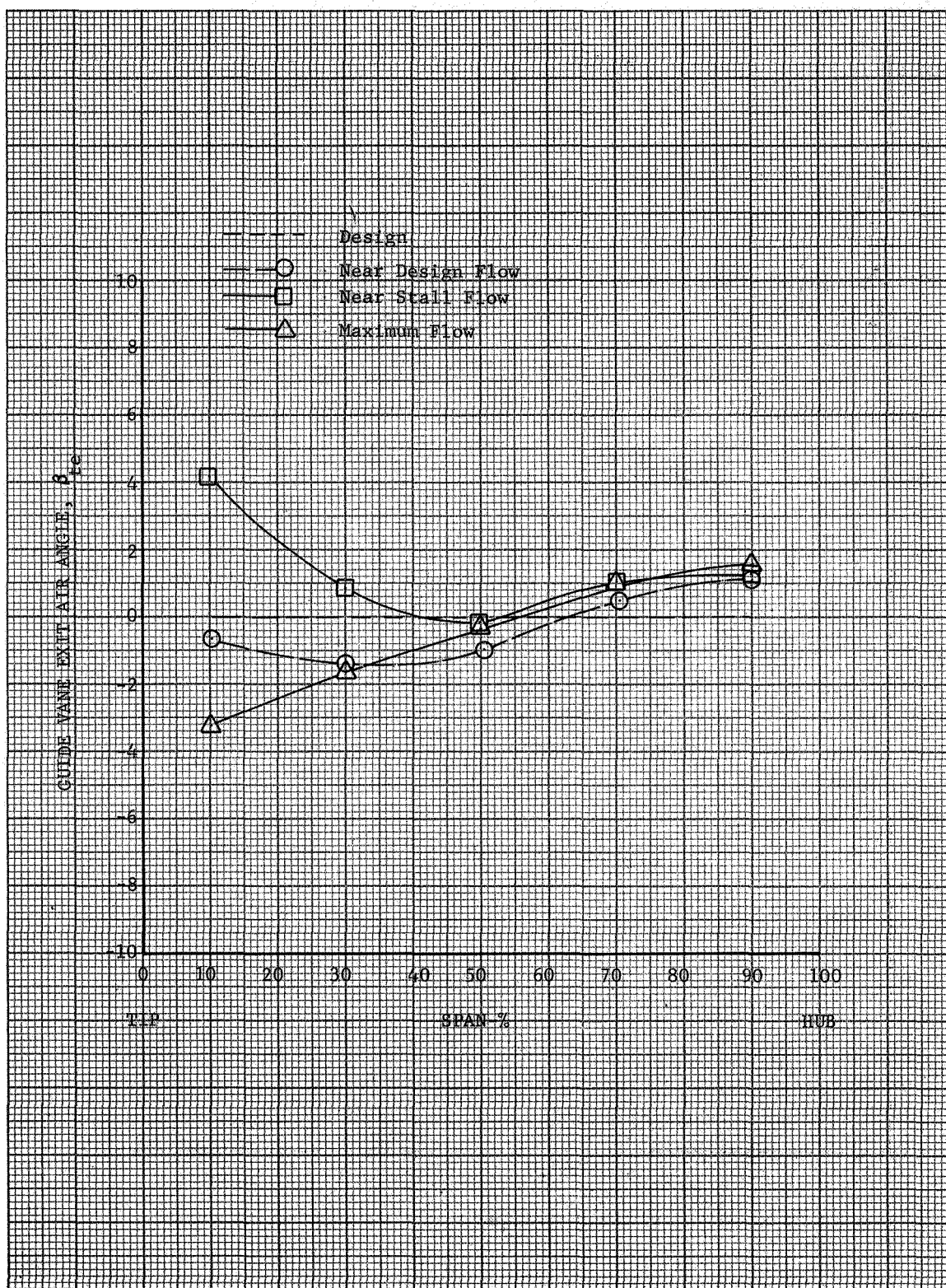


Figure 23. Guide Vane Exit Air Angle: SLTO Configuration, 100% Equivalent Rotor Speed

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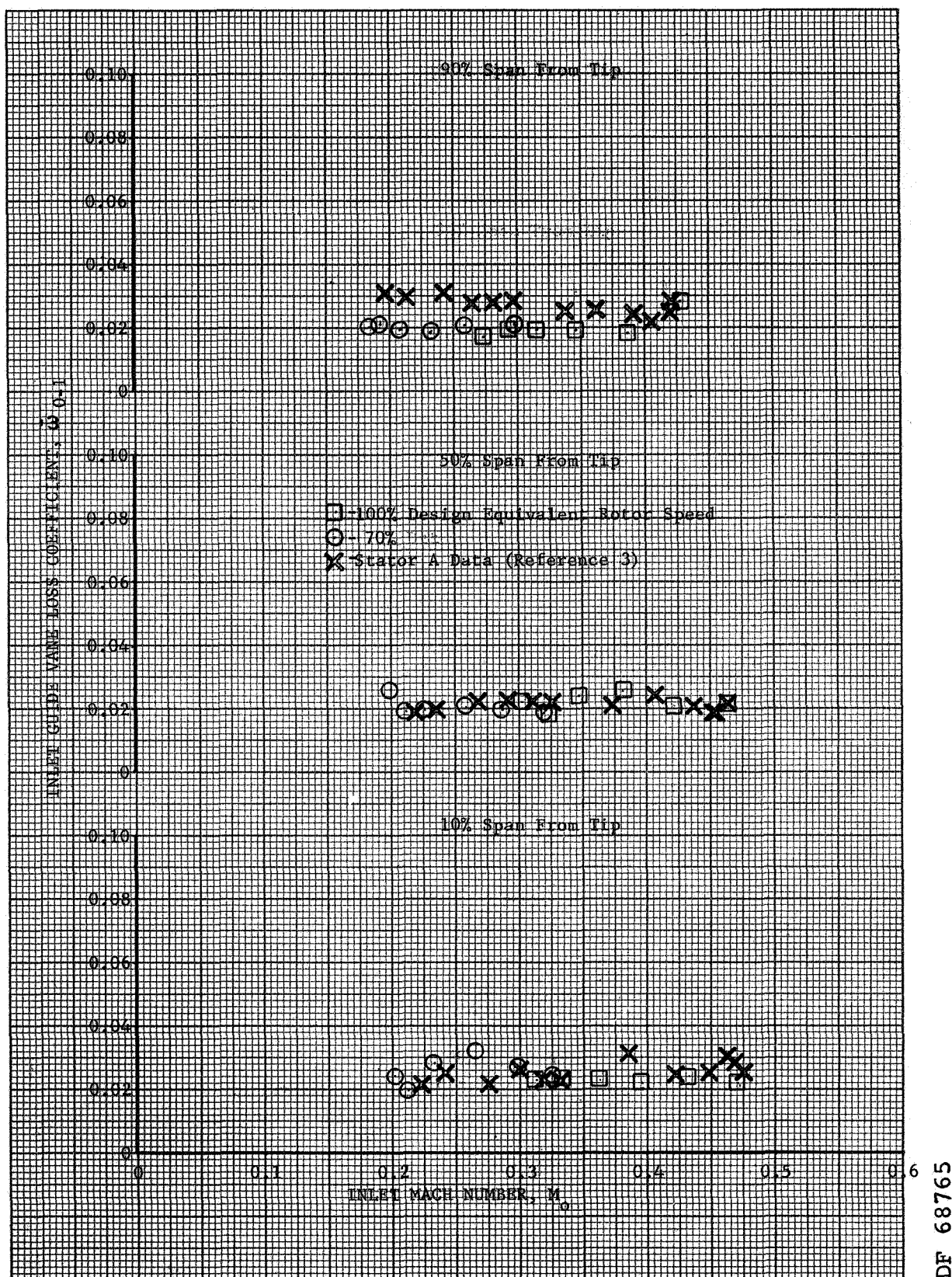


Figure 24. Inlet Guide Vane Loss Coefficient vs Inlet Mach Number: SLTO Configuration

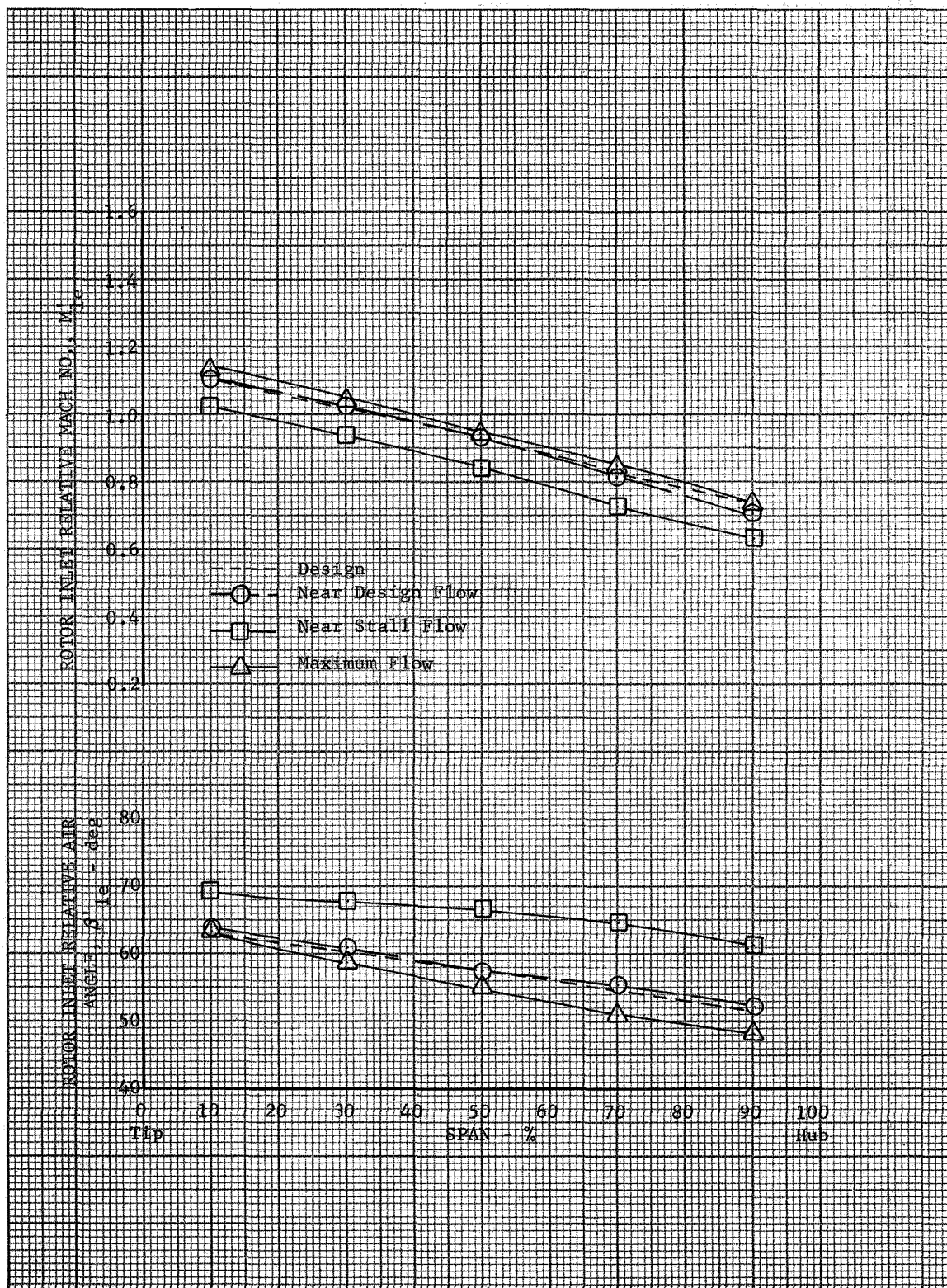
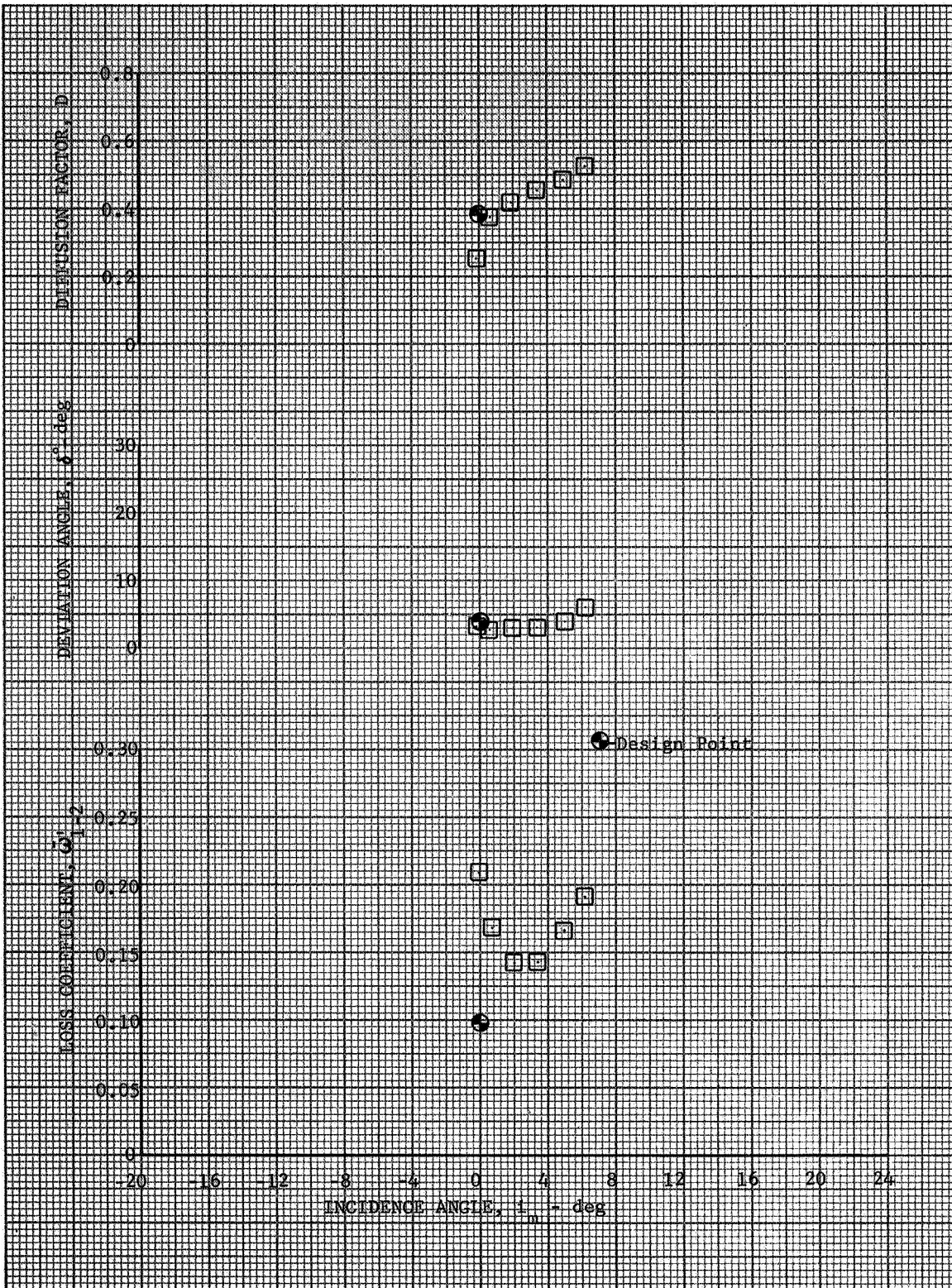


Figure 25. Rotor Inlet Air Angle and Mach Number Distribution: SLTO Configuration, 100% Equivalent Rotor Speed

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Figure 26. Rotor Blade Element Performance: SLTO Configuration, 100% Equivalent Rotor Speed, 10% Span From Tip

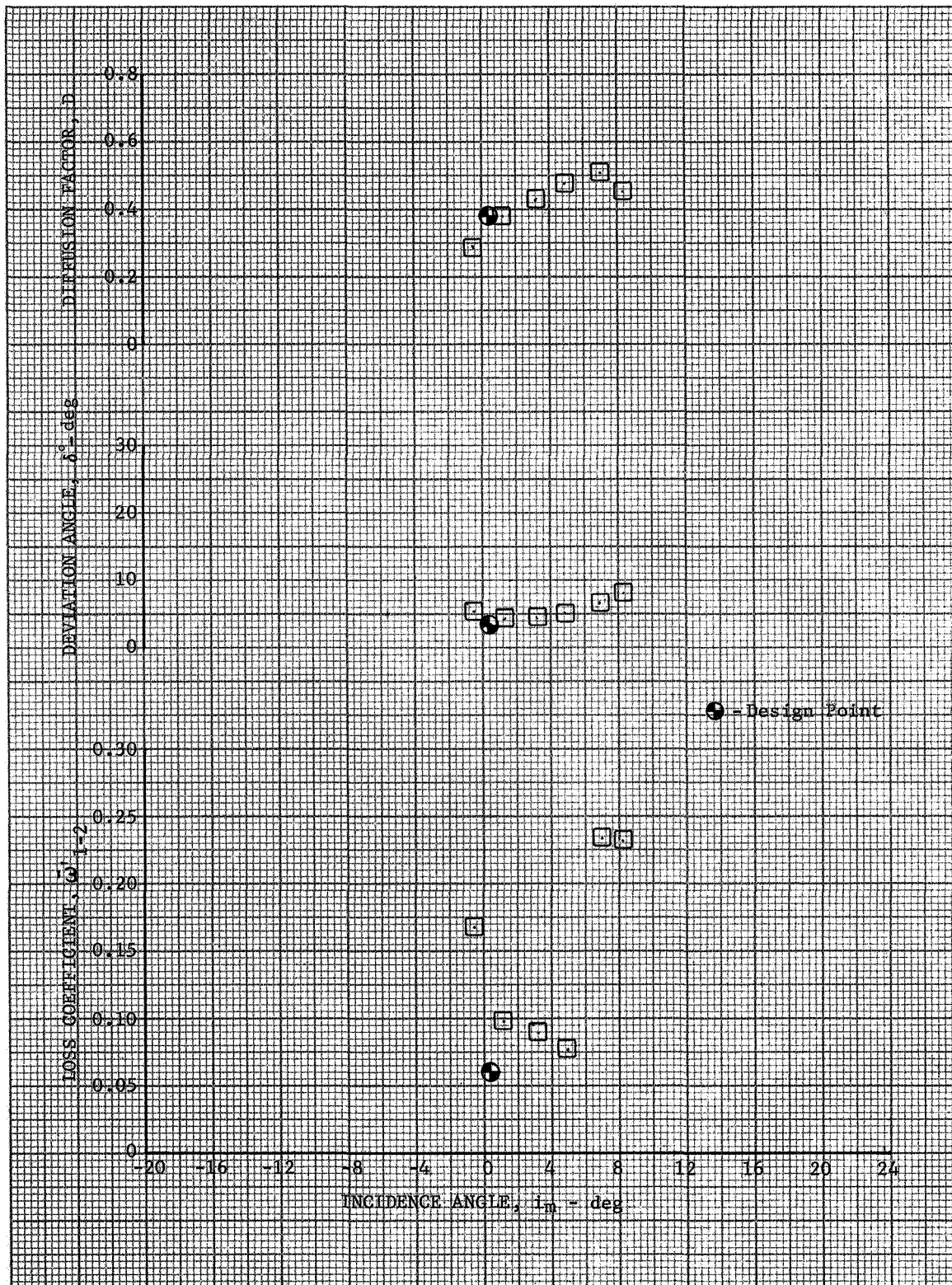


Figure 27. Rotor Blade Element Performance: SLTO Configuration, 100% Equivalent Rotor Speed, 30% Span From Tip

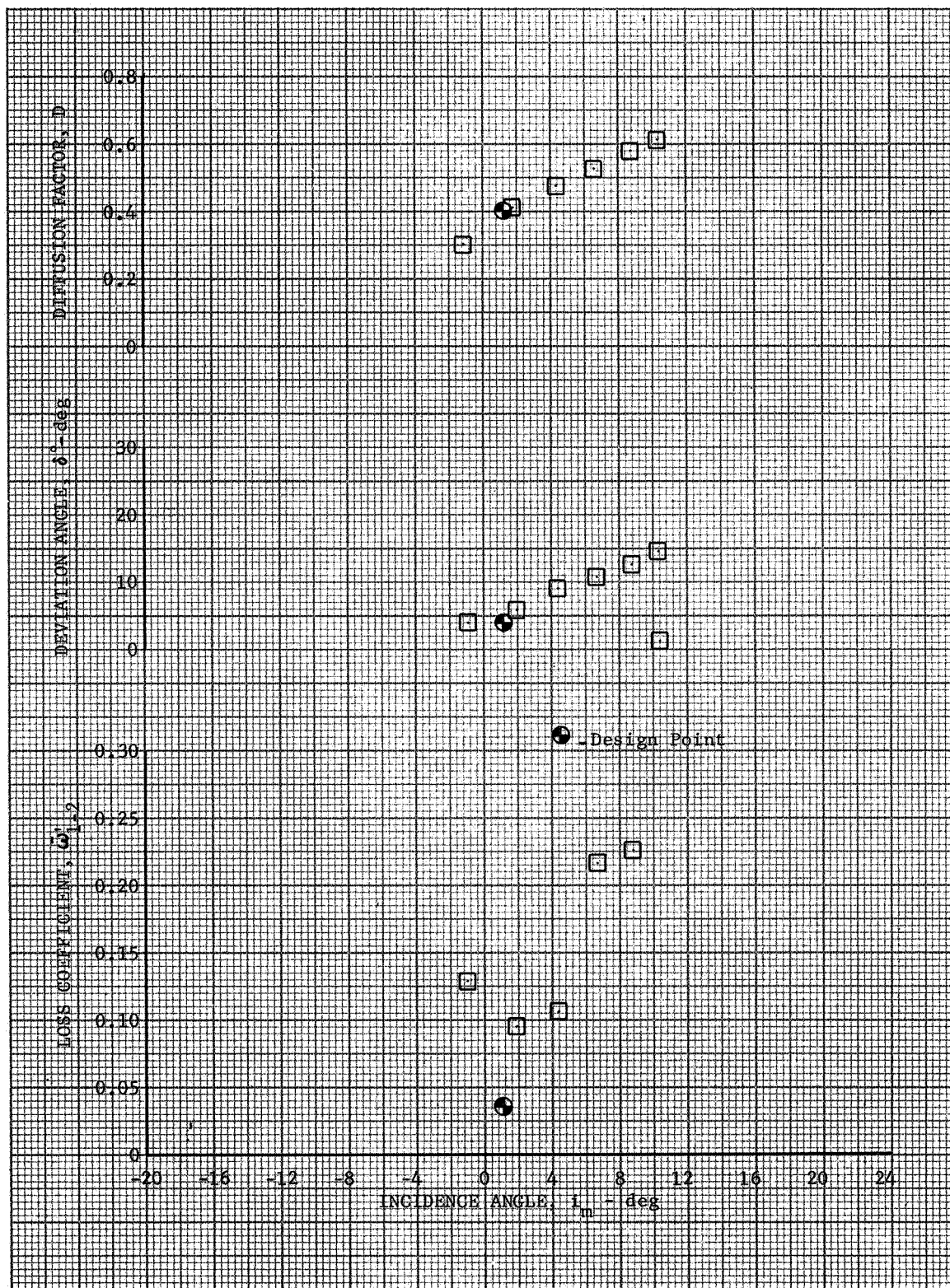
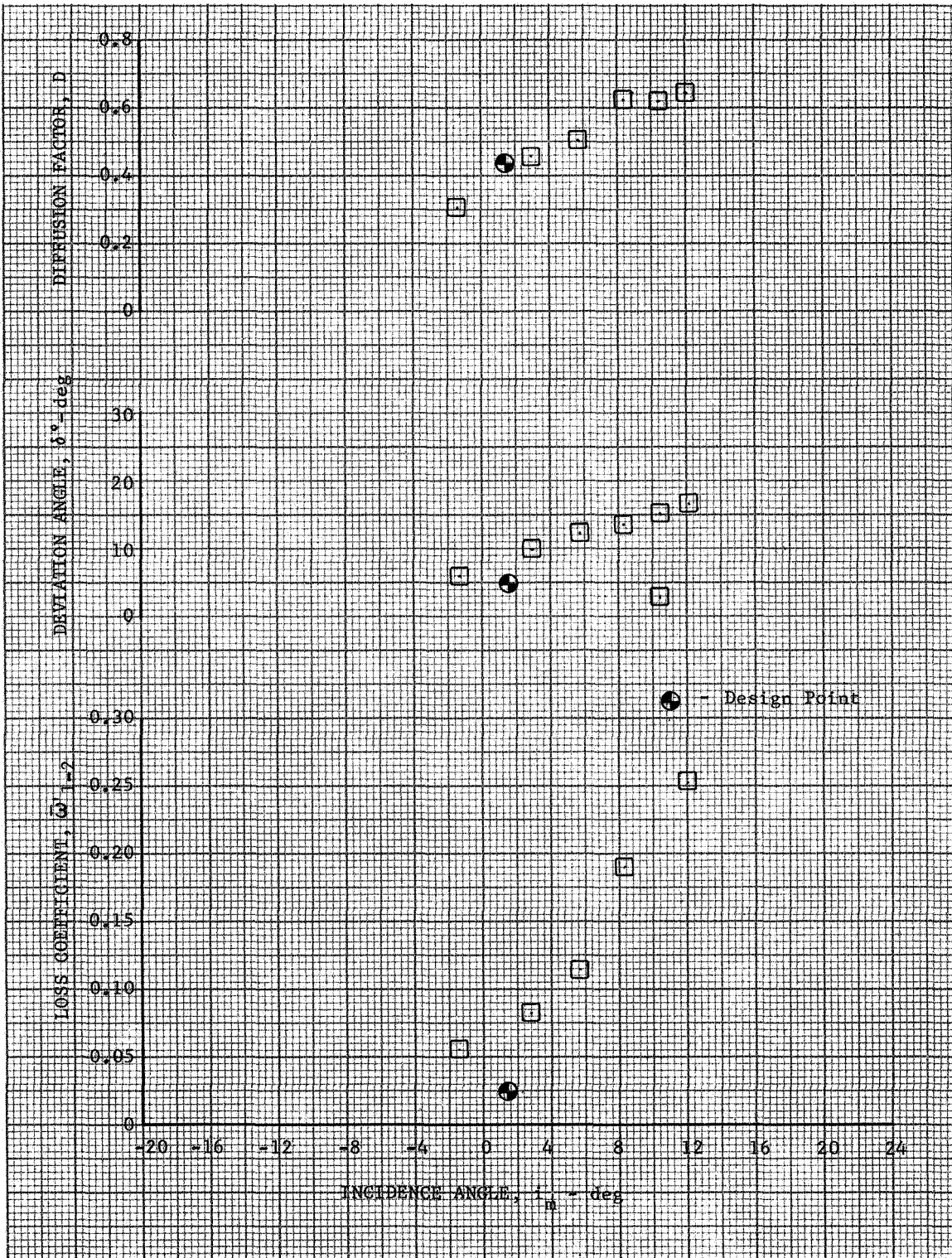


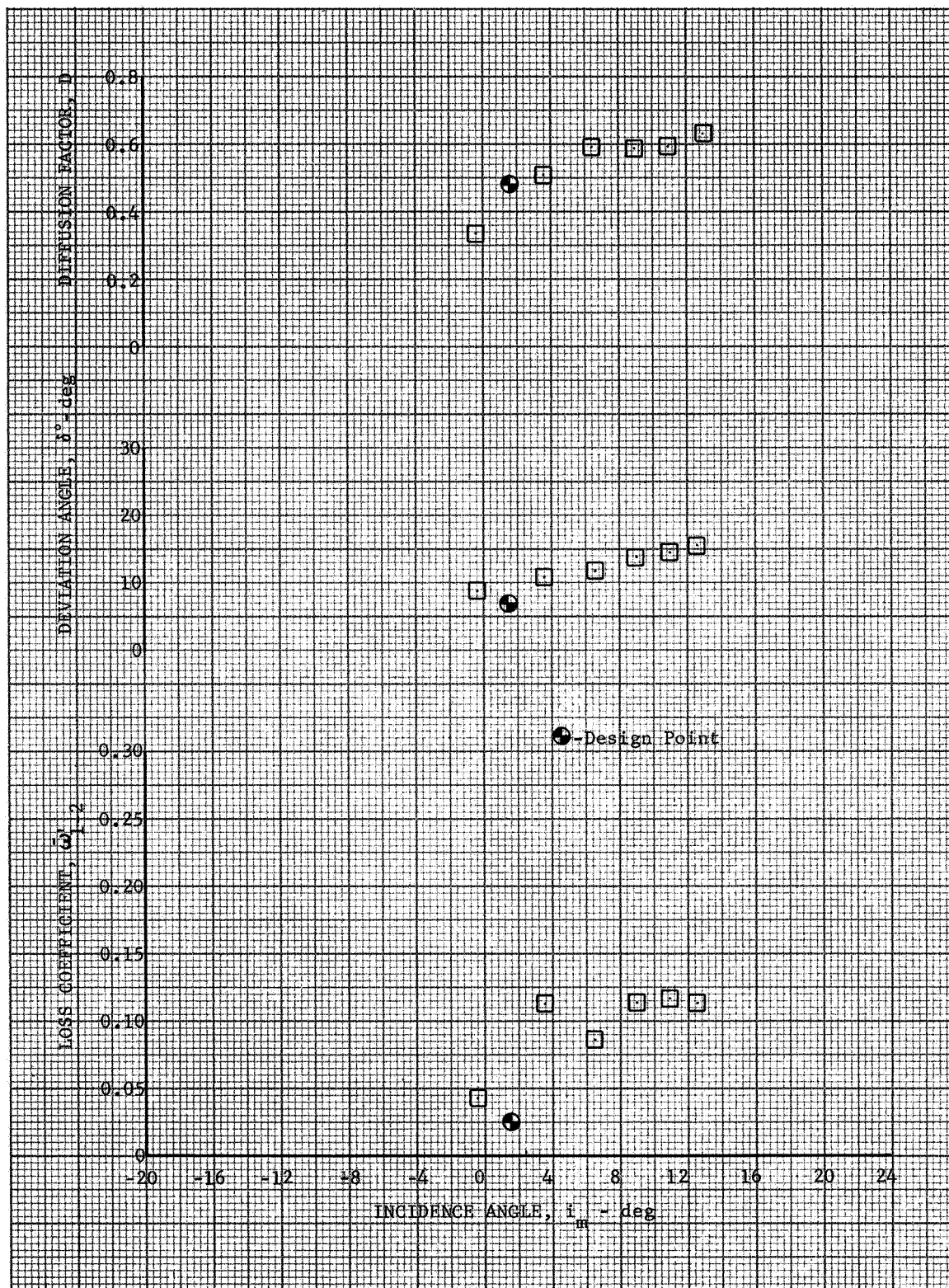
Figure 28. Rotor Blade Element Performance: SLTO Configuration, 100% Equivalent Rotor Speed, 50% Span From Tip

DF 68769



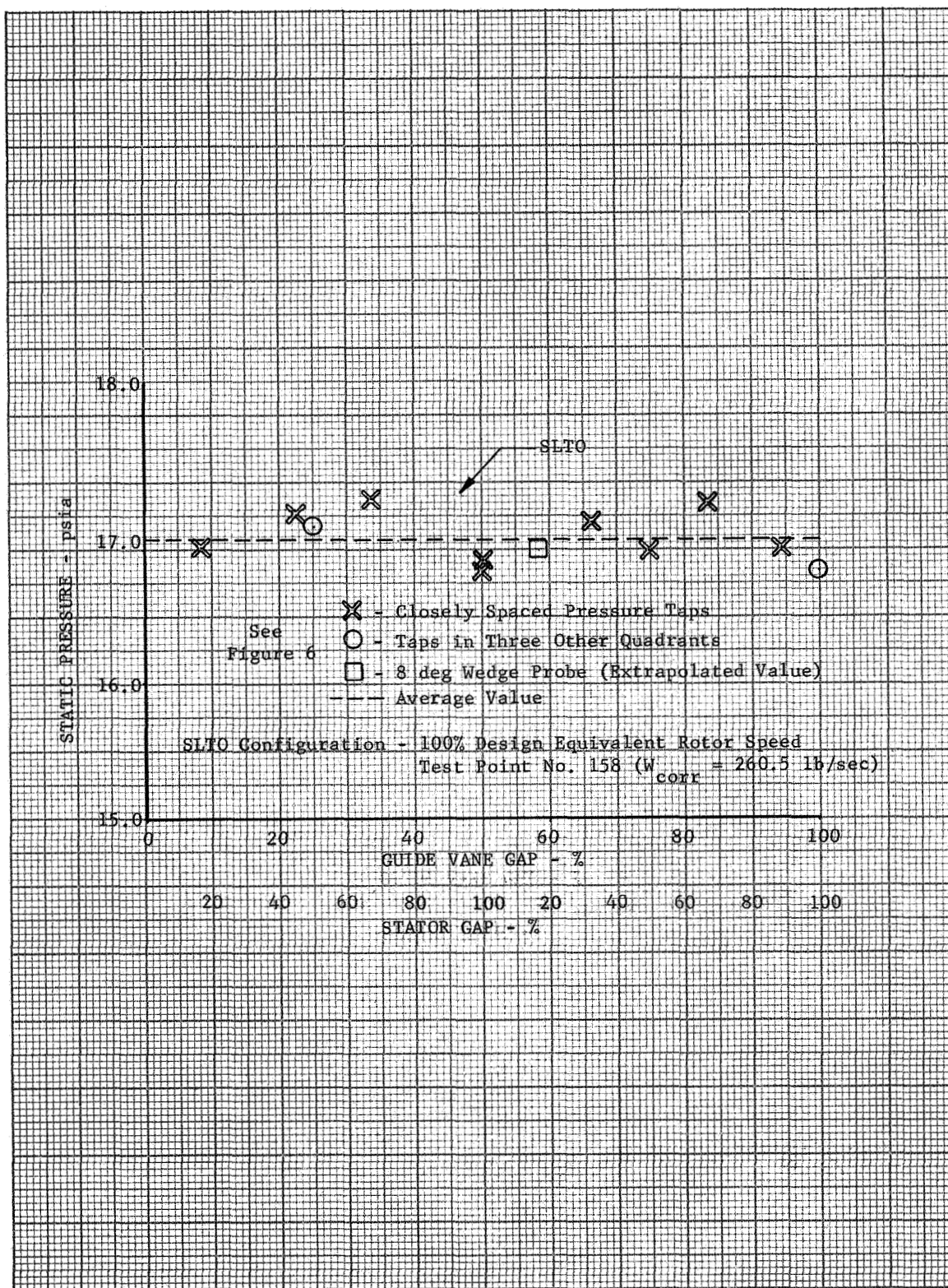
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Figure 29. Rotor Blade Element Performance: SLTO Configuration, 100% Equivalent Rotor Speed, 70% Span From Tip



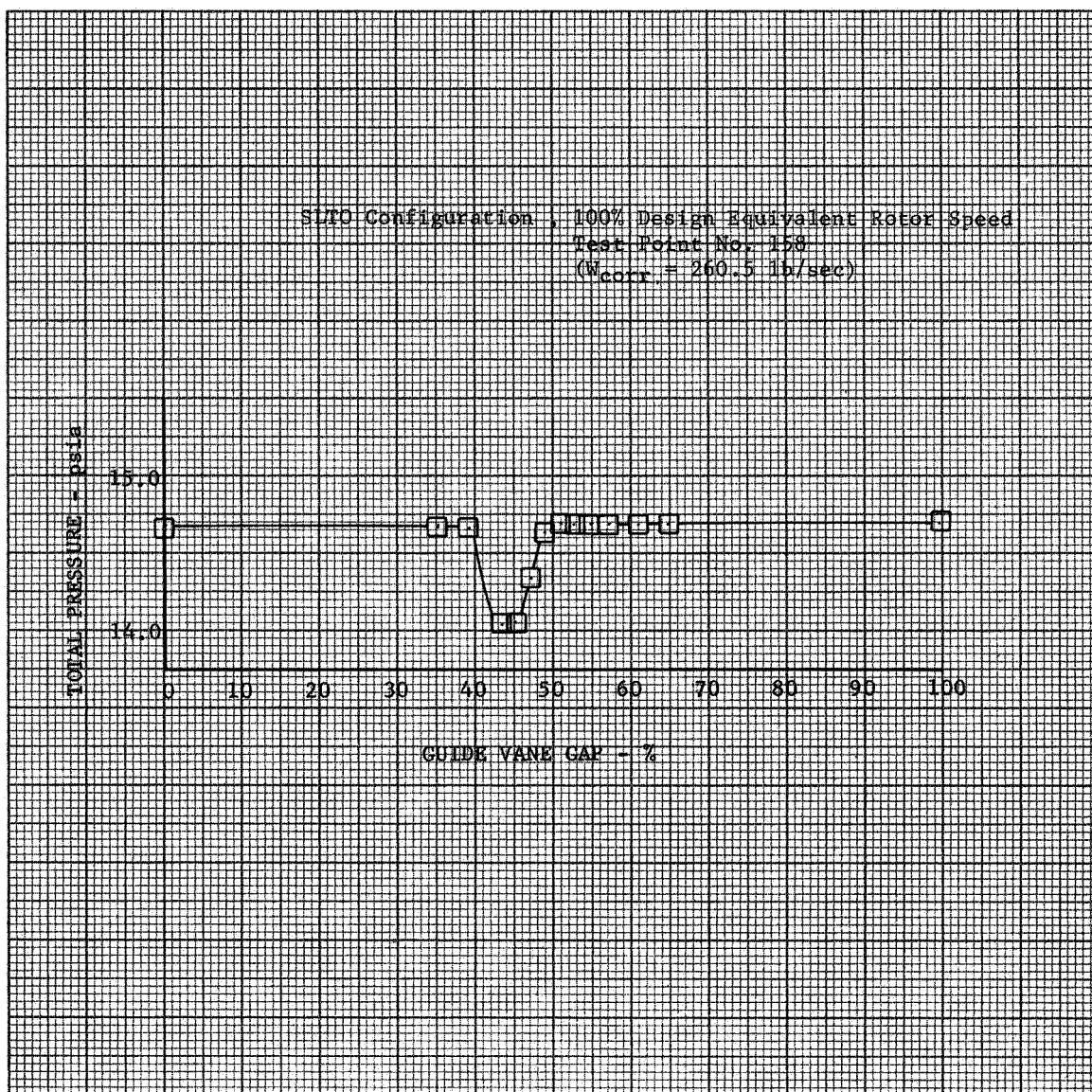
DF 68771

Figure 30. Rotor Blade Element Performance: SLTO Configuration, 100% Equivalent Rotor Speed, 90% Span From Tip



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Figure 31. Influence of Guide Vane Wake and Stator on Station 2 Outer Wall Static Pressure



DF 68992

Figure 32. Guide Vane Wake 10% From Tip

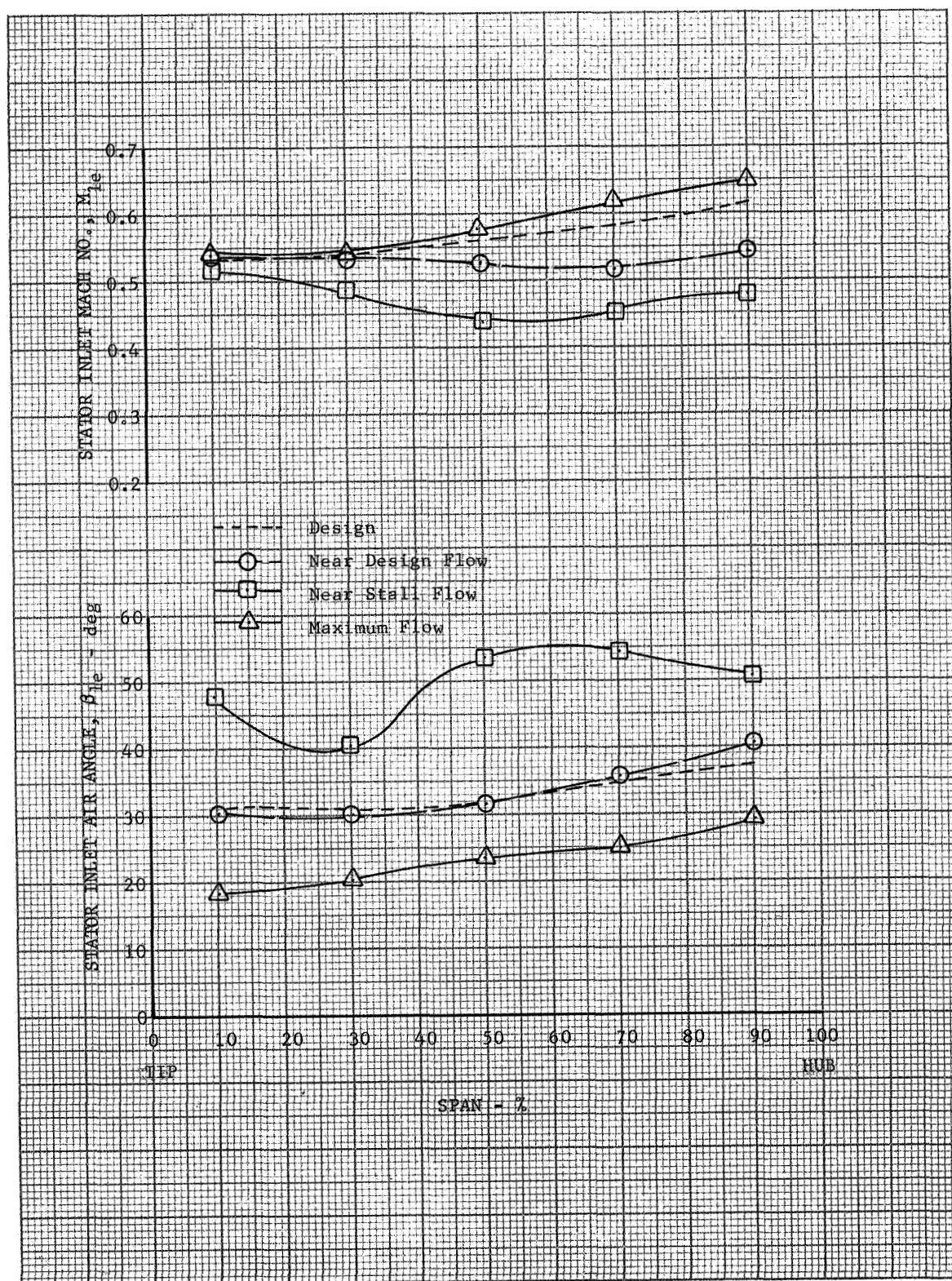


Figure 33. Stator Inlet Air Angle and Mach Number Distribution: SLTO Configuration, 100% Equivalent Rotor Speed

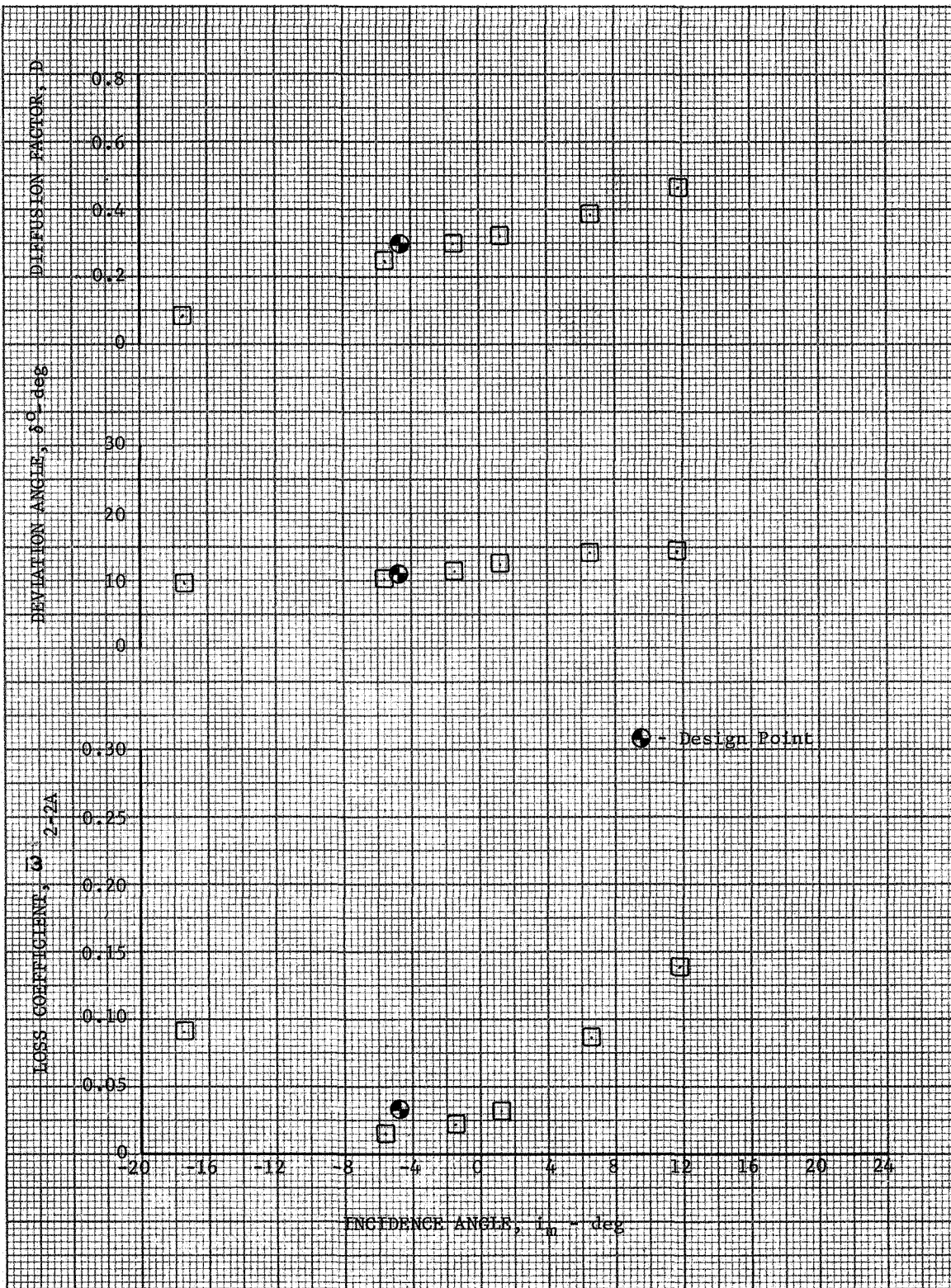


Figure 34. Stator Blade Element Performance: SLTO Configuration, 100% Equivalent Rotor Speed, 10% Span From Tip

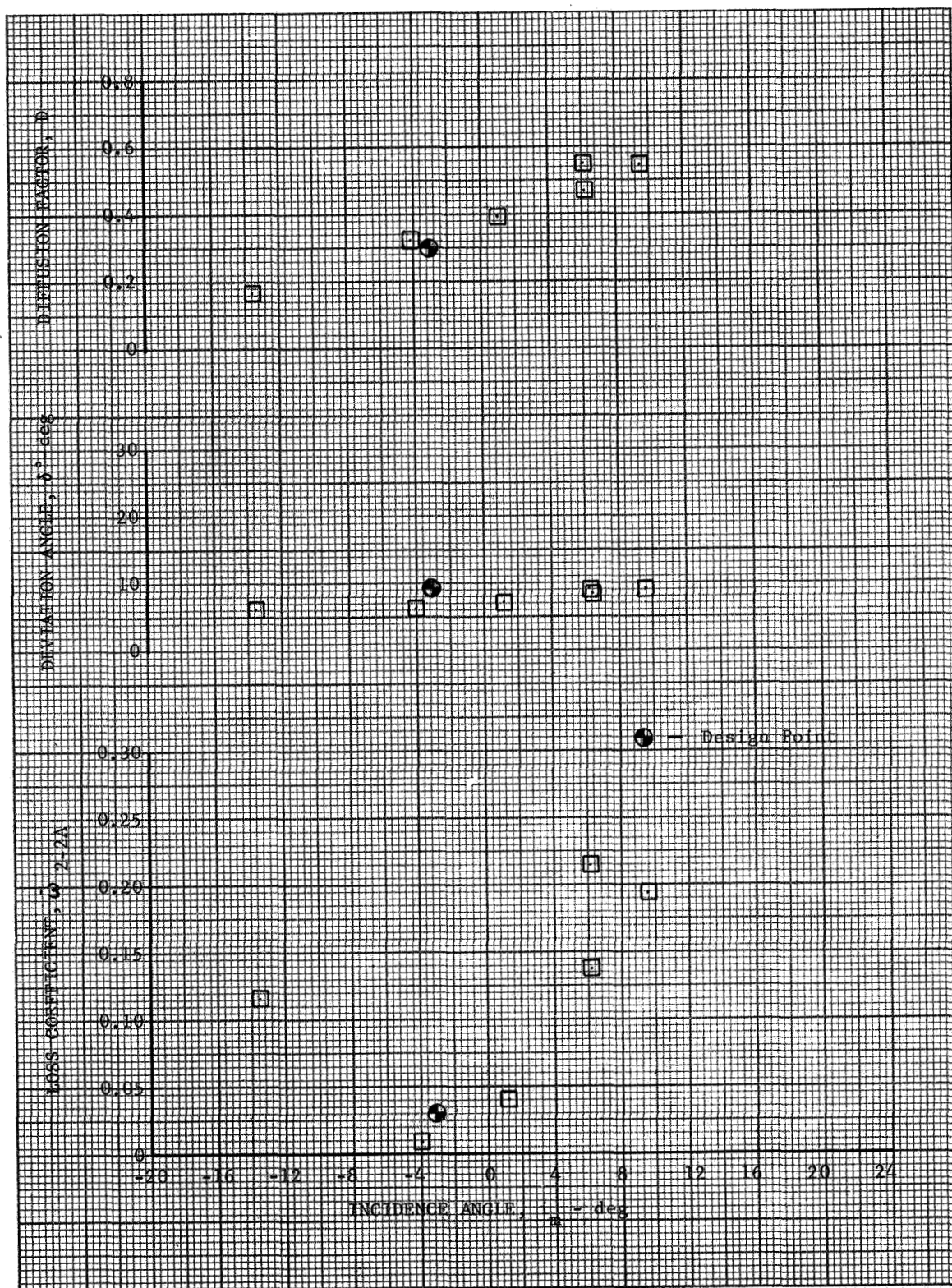


Figure 35. Stator Blade Element Performance: SLTO Configuration, 100% Rotor Speed, 30% Span From Tip

DF 68775

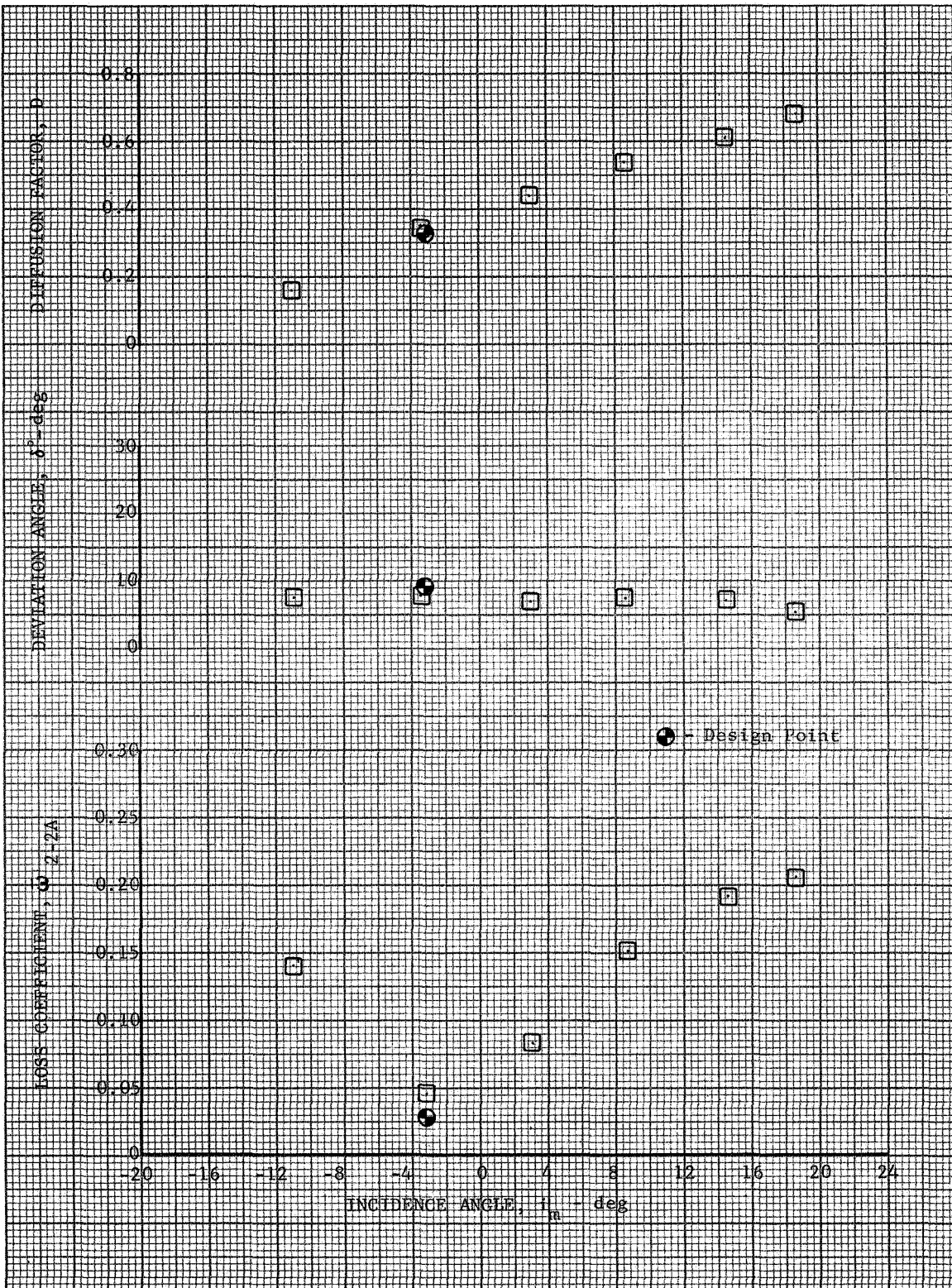


Figure 36. Stator Blade Element Performance: SLT0 Configuration, 100% Equivalent Rotor Speed, 50% Span From Tip

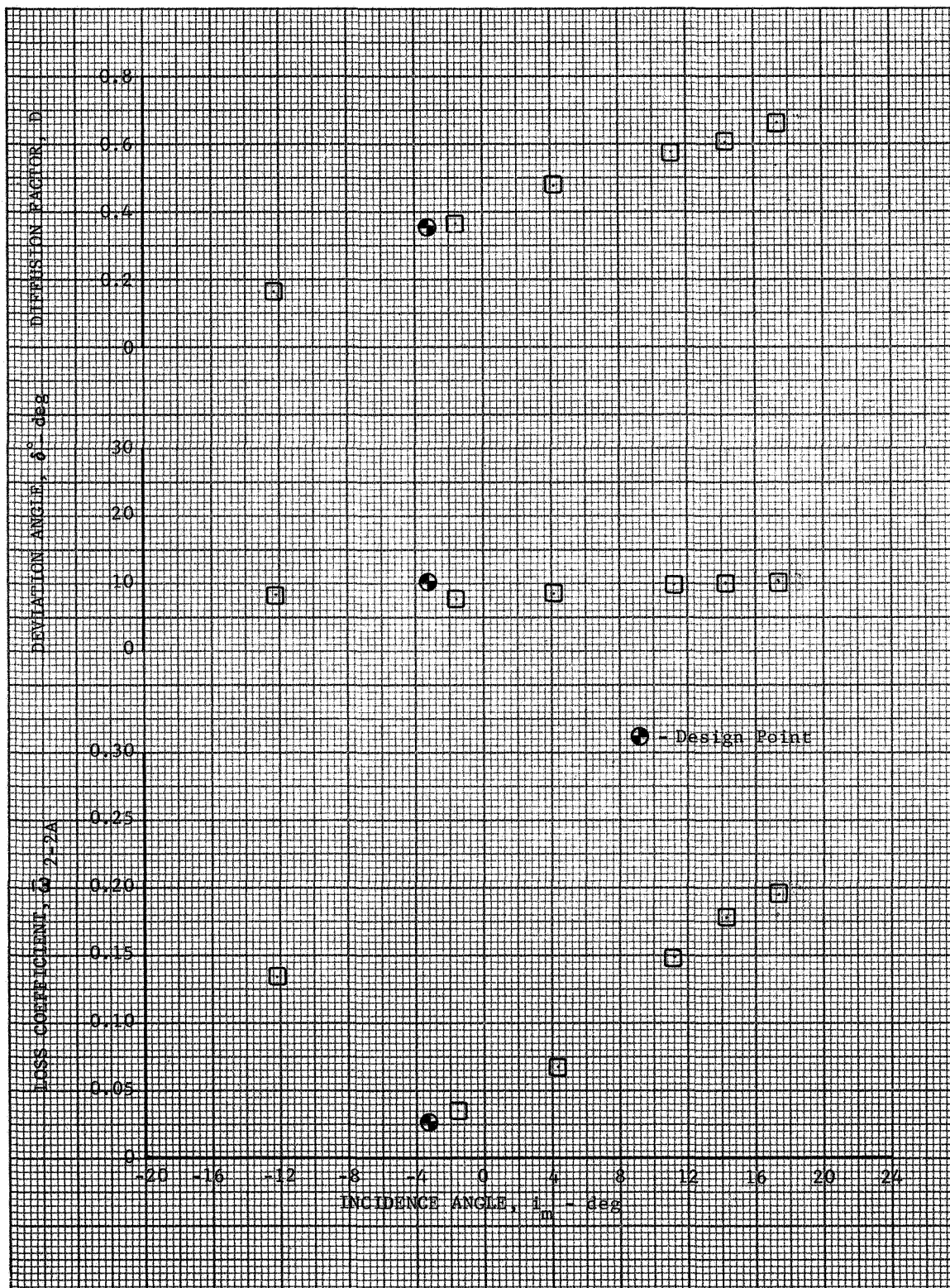


Figure 37. Stator Blade Element Performance: SLT0 Configuration, 100% Equivalent Rotor Speed, 70% Span From Tip

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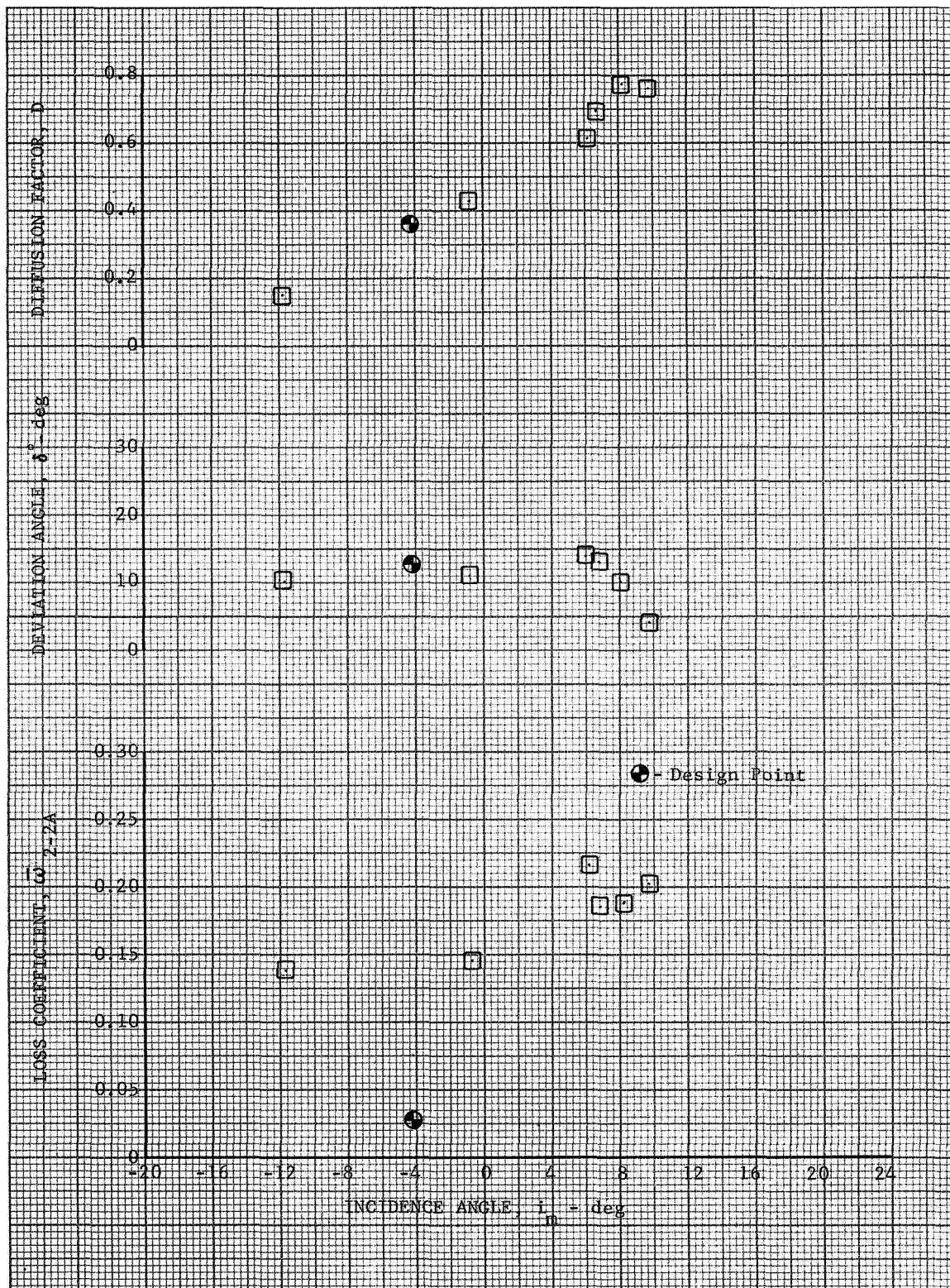


Figure 38. Stator Blade Element Performance: SLTO Configuration, 100% Equivalent Rotor Speed, 90% Span From Tip

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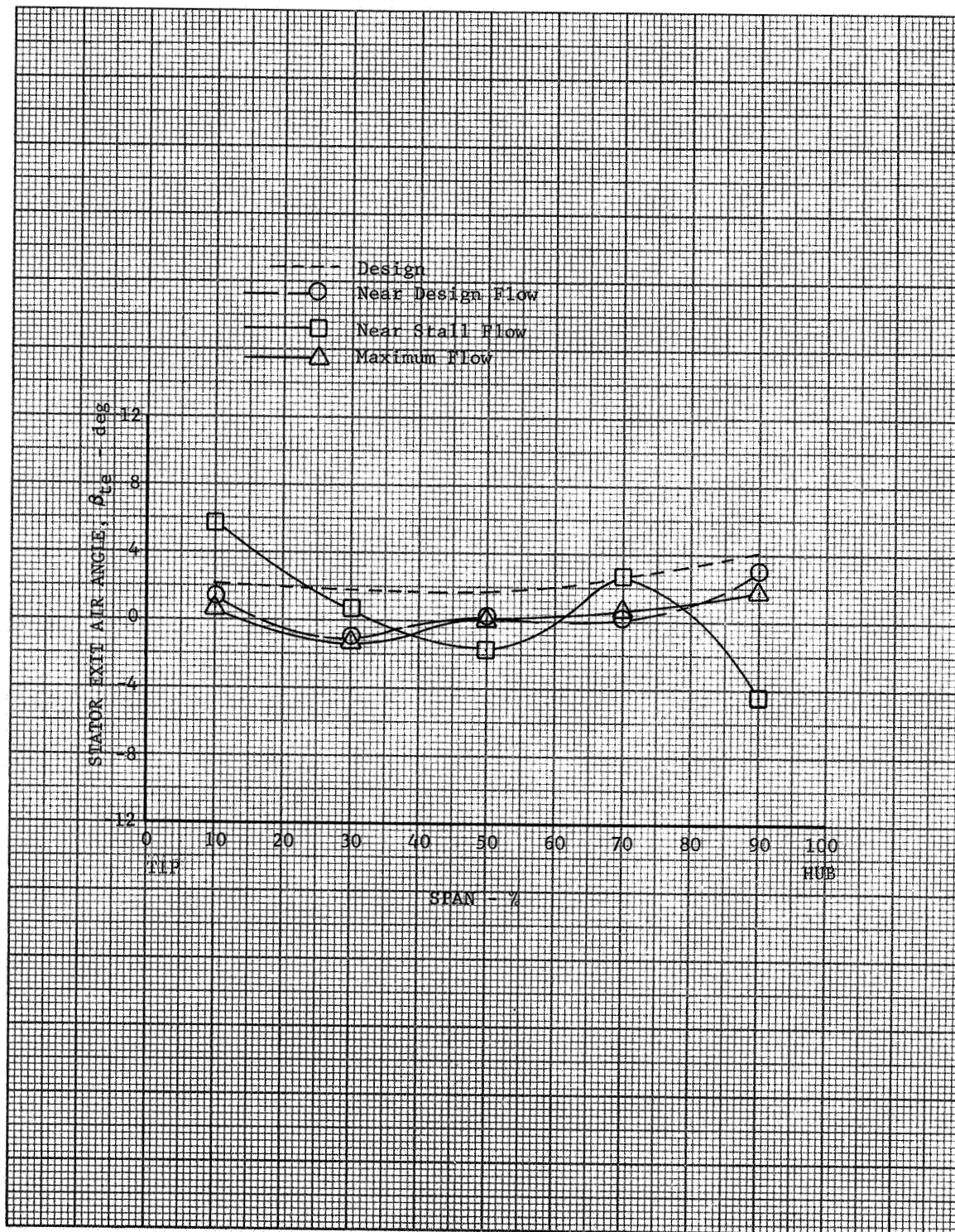


Figure 39. Stator Exit Air Angle vs Percent Span:
SLTO Configuration, 100% Equivalent
Rotor Speed

DF 68779

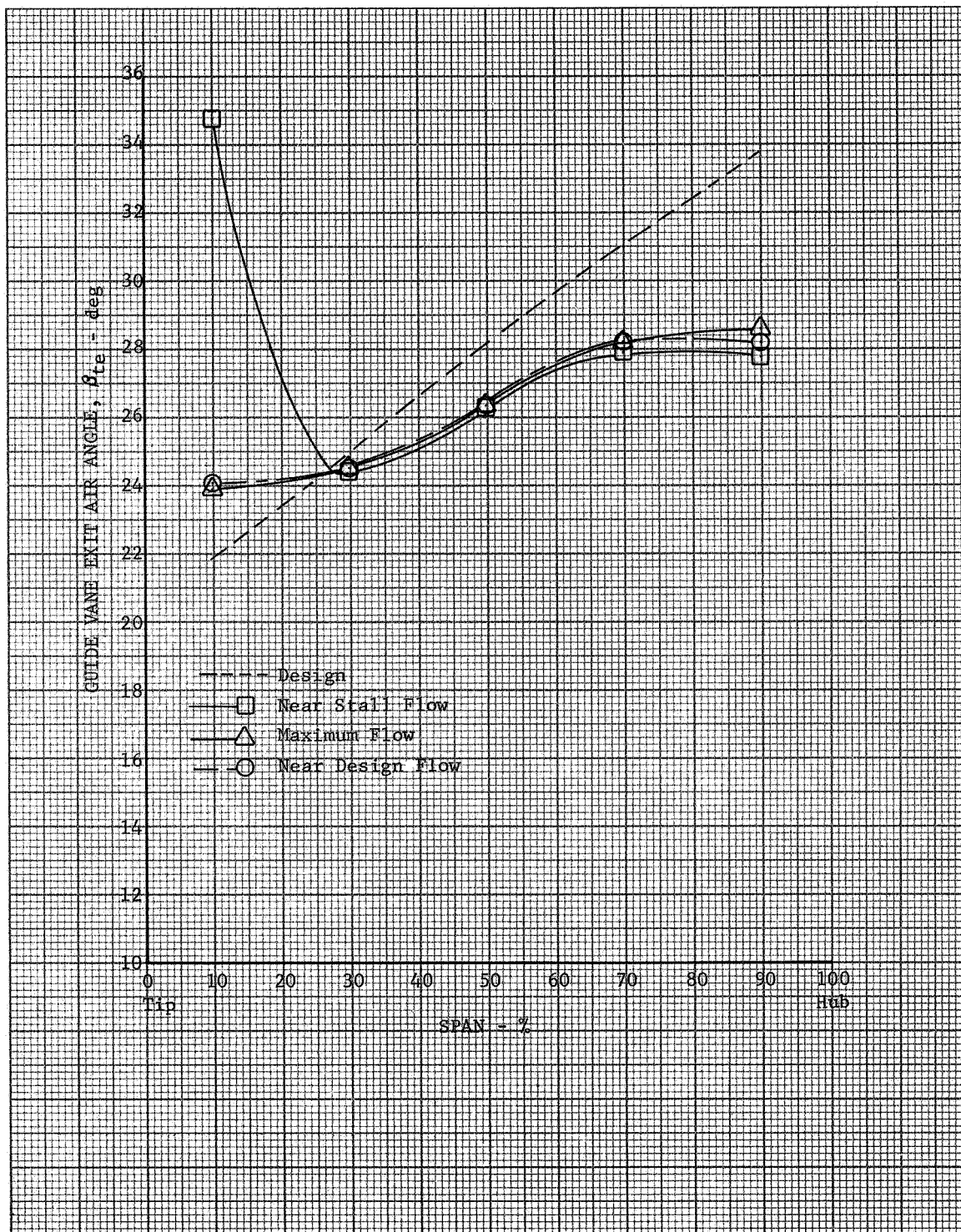


Figure 40. Guide Vane Exit Air Angle: Cruise Configuration, 70% Equivalent Rotor Speed

DF 68780

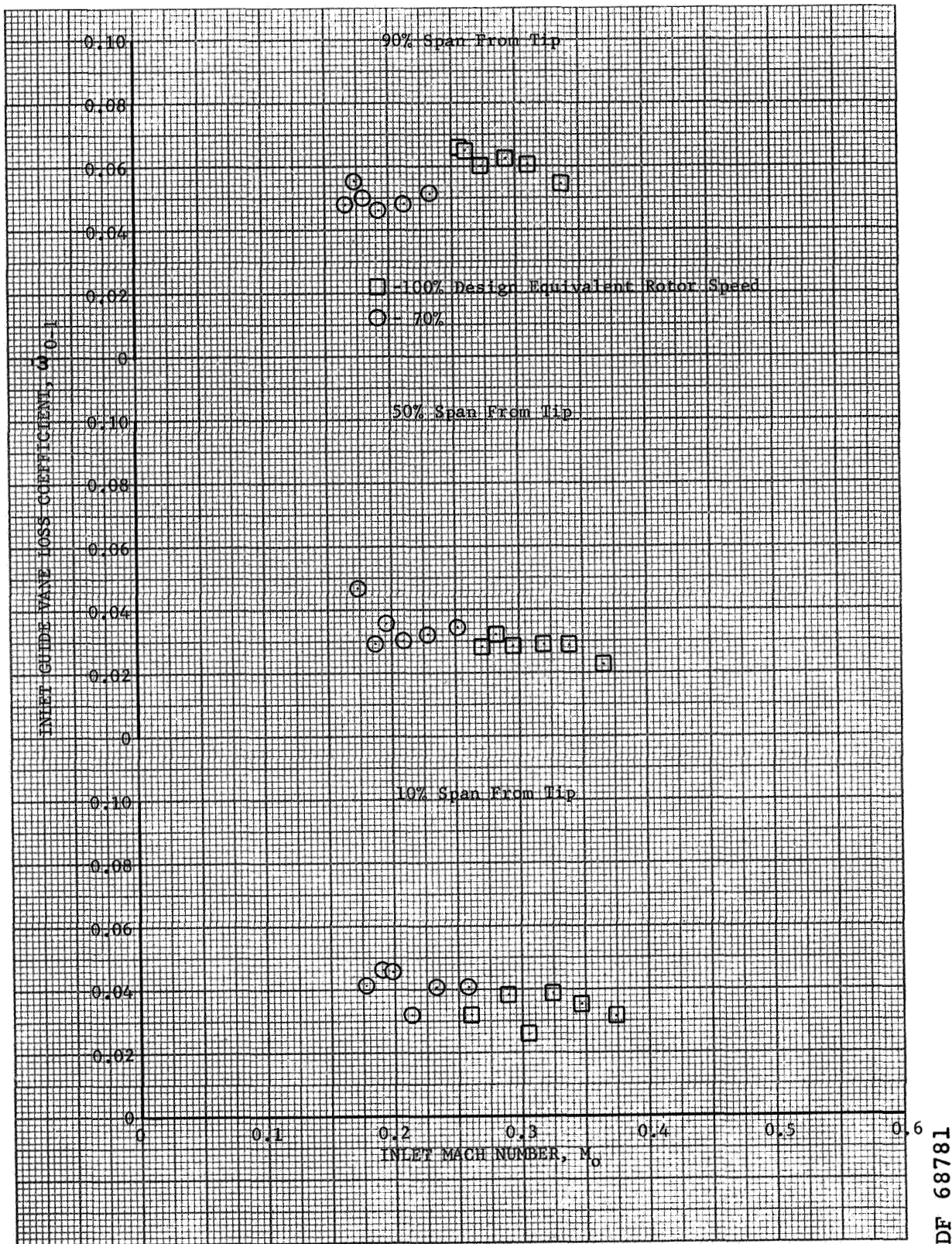


Figure 41. Inlet Guide Vane Loss Coefficient vs Inlet Mach Number: Cruise Configuration

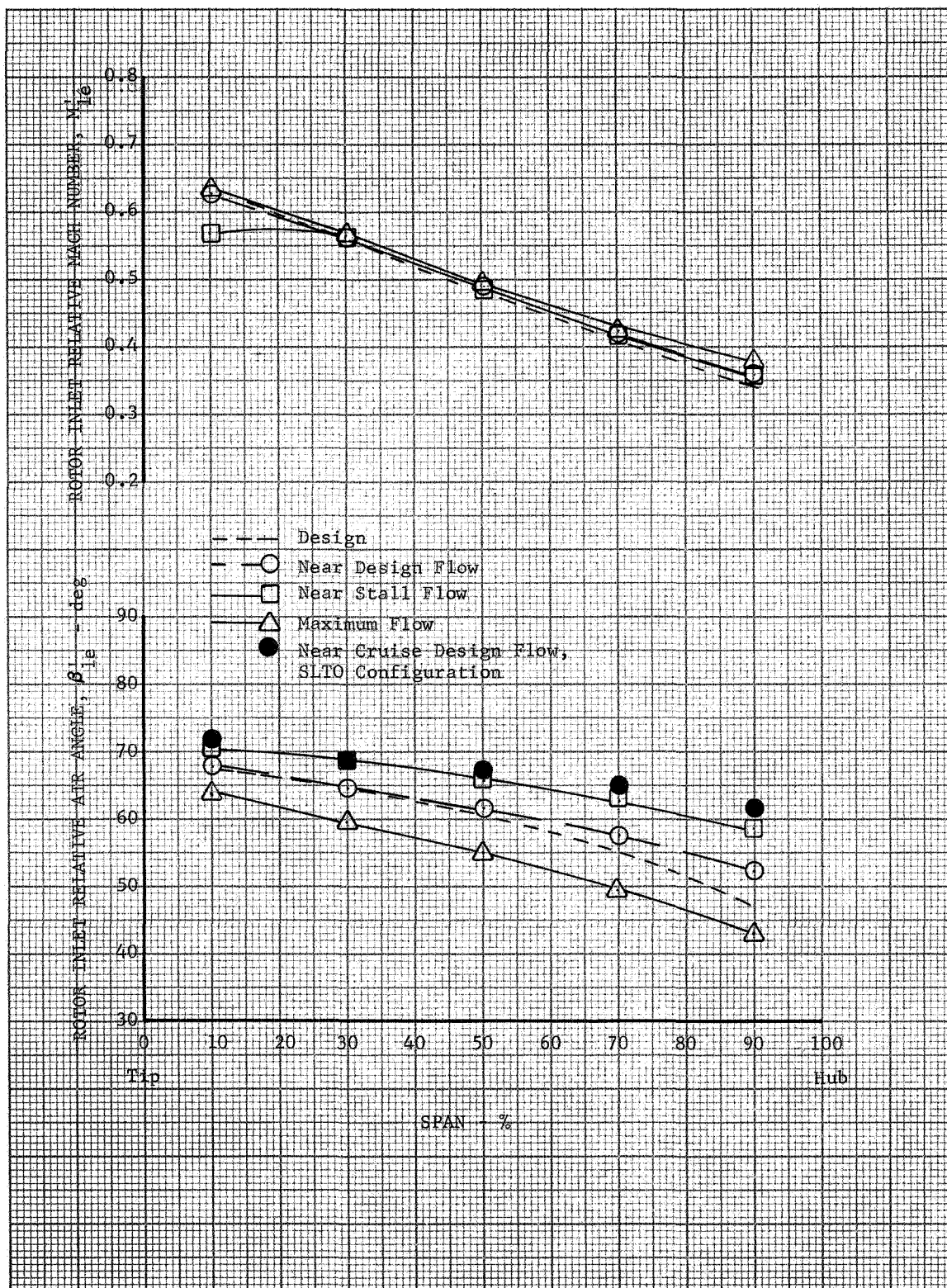


Figure 42. Rotor Inlet Air Angle and Mach Number Distribution: Cruise Configuration, 70% Equivalent Rotor Speed

DF 68782

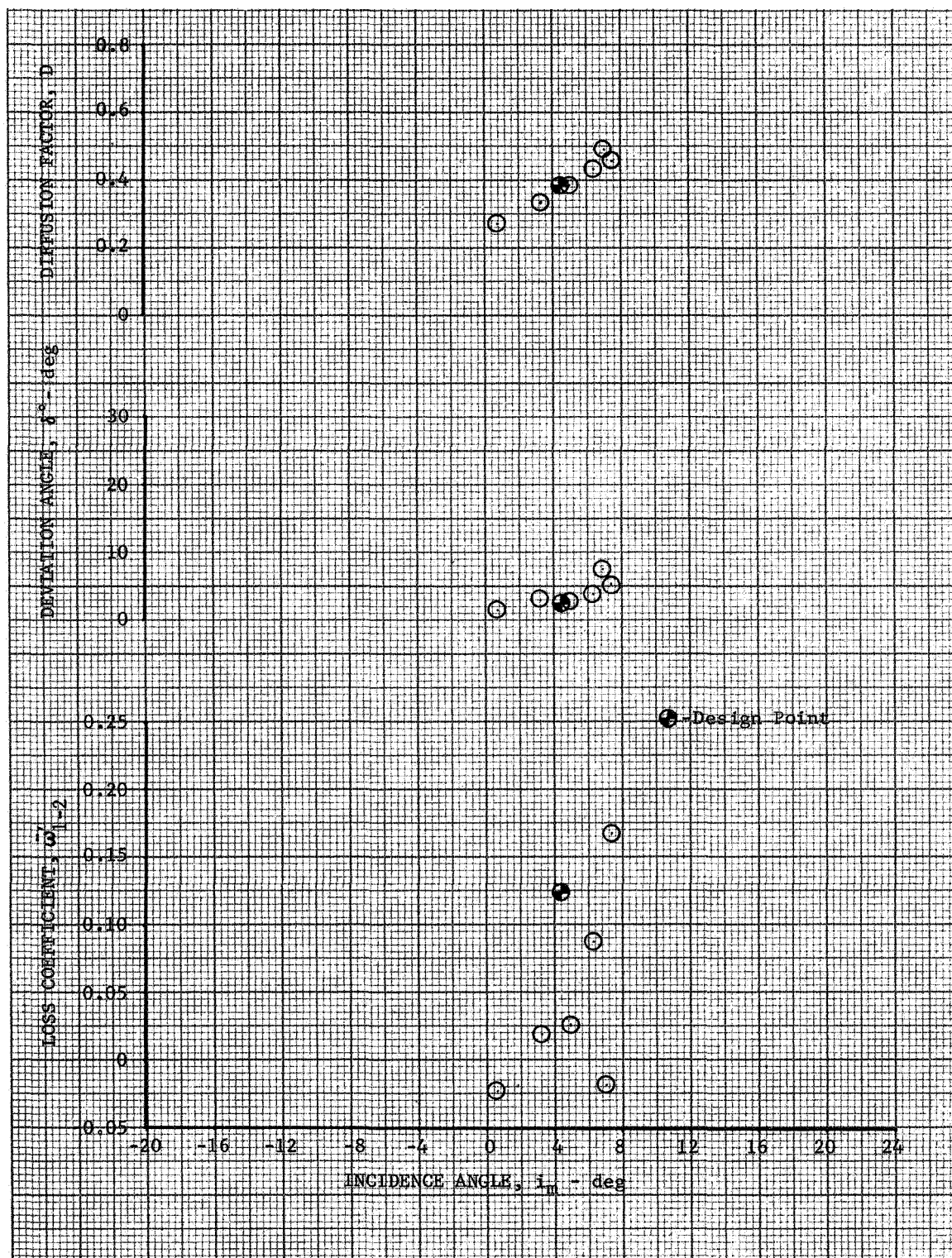
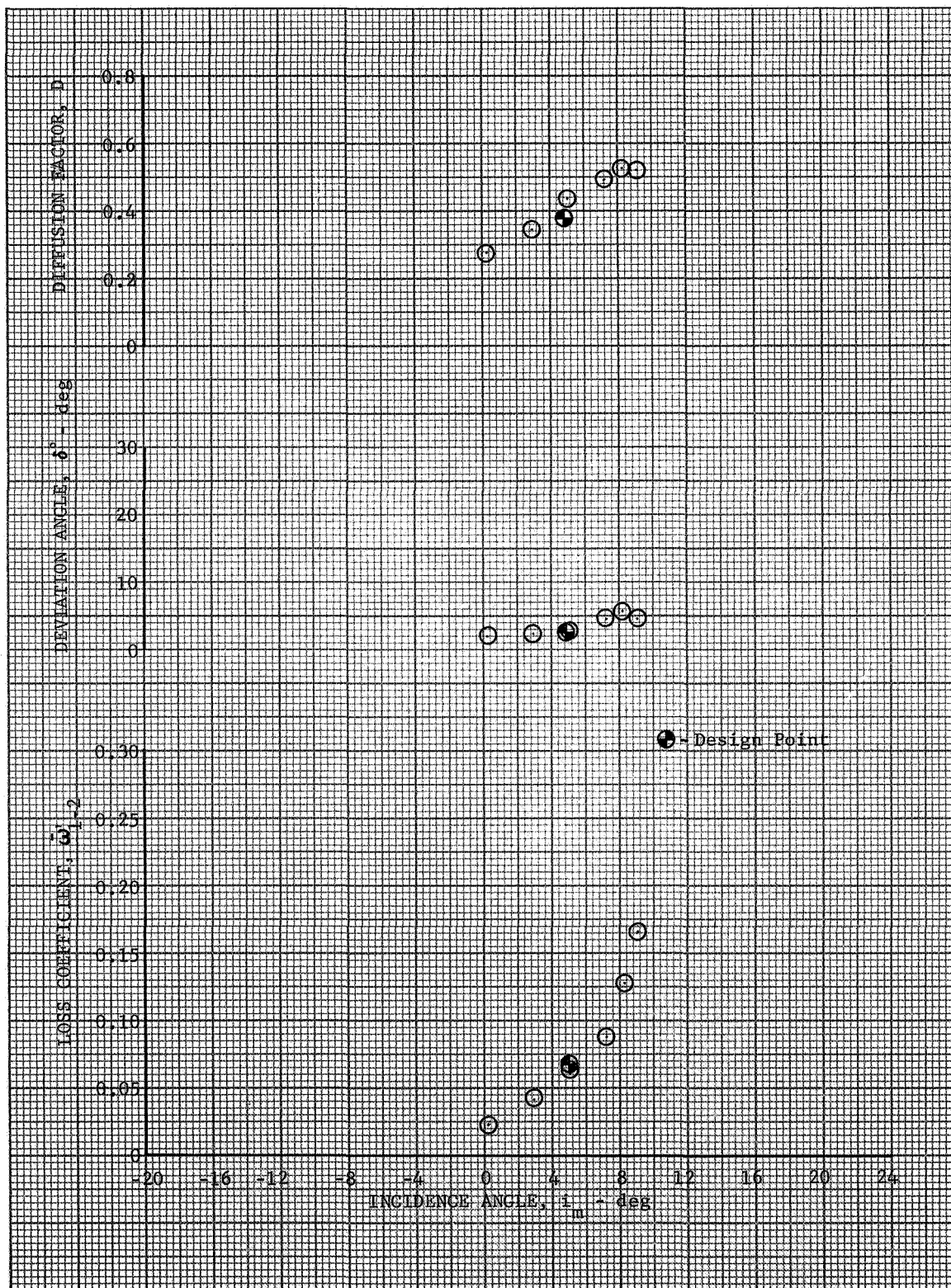


Figure 43. Rotor Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 10% Span From Tip

DF 68783



DF 68784

Figure 44. Rotor Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 30% Span From Tip

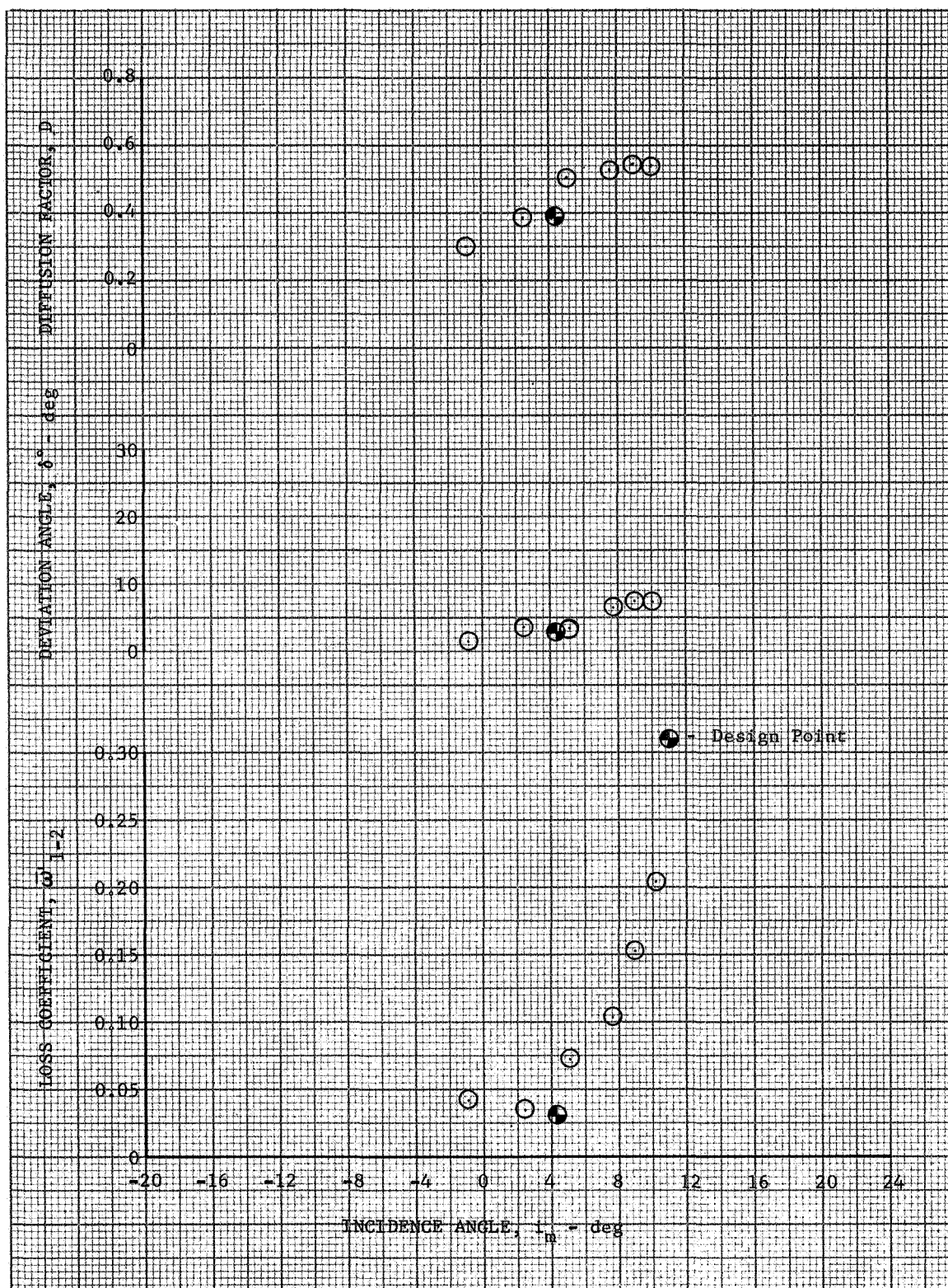


Figure 45. Rotor Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 50% Span From Tip

DF 68785

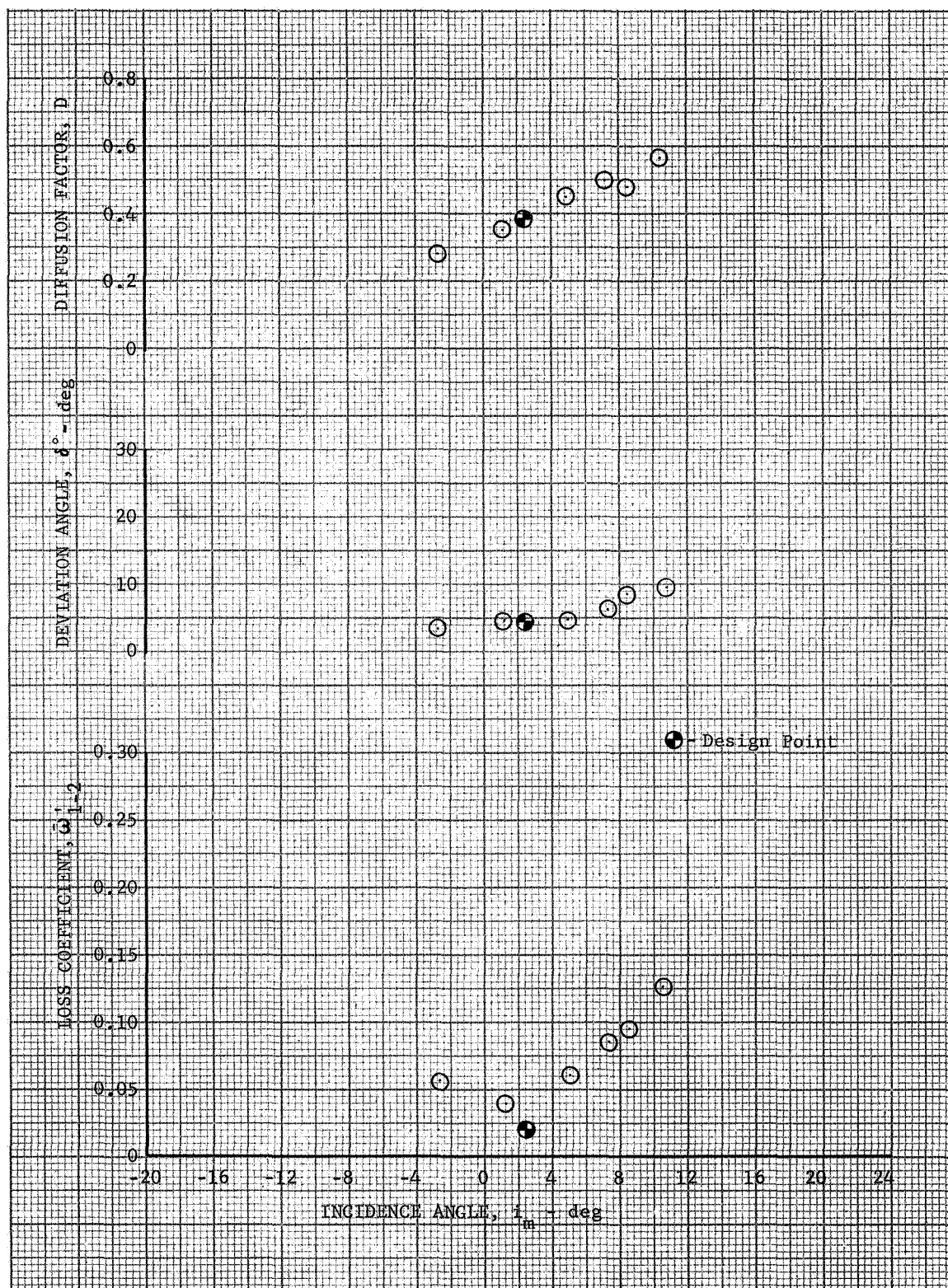


Figure 46. Rotor Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 70% Span From Tip

DF 68786

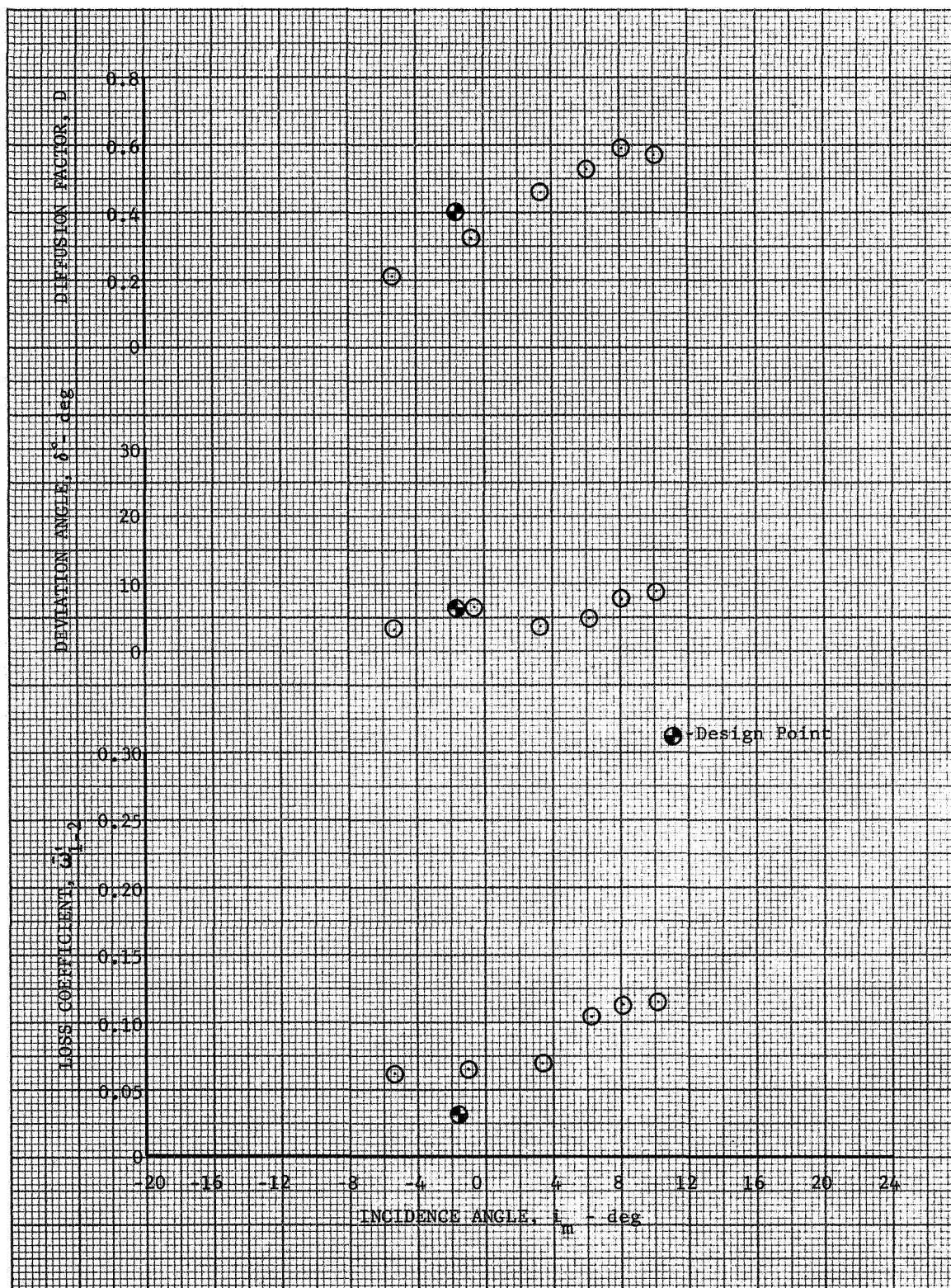


Figure 47. Rotor Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 90% Span From Tip

DF 68787

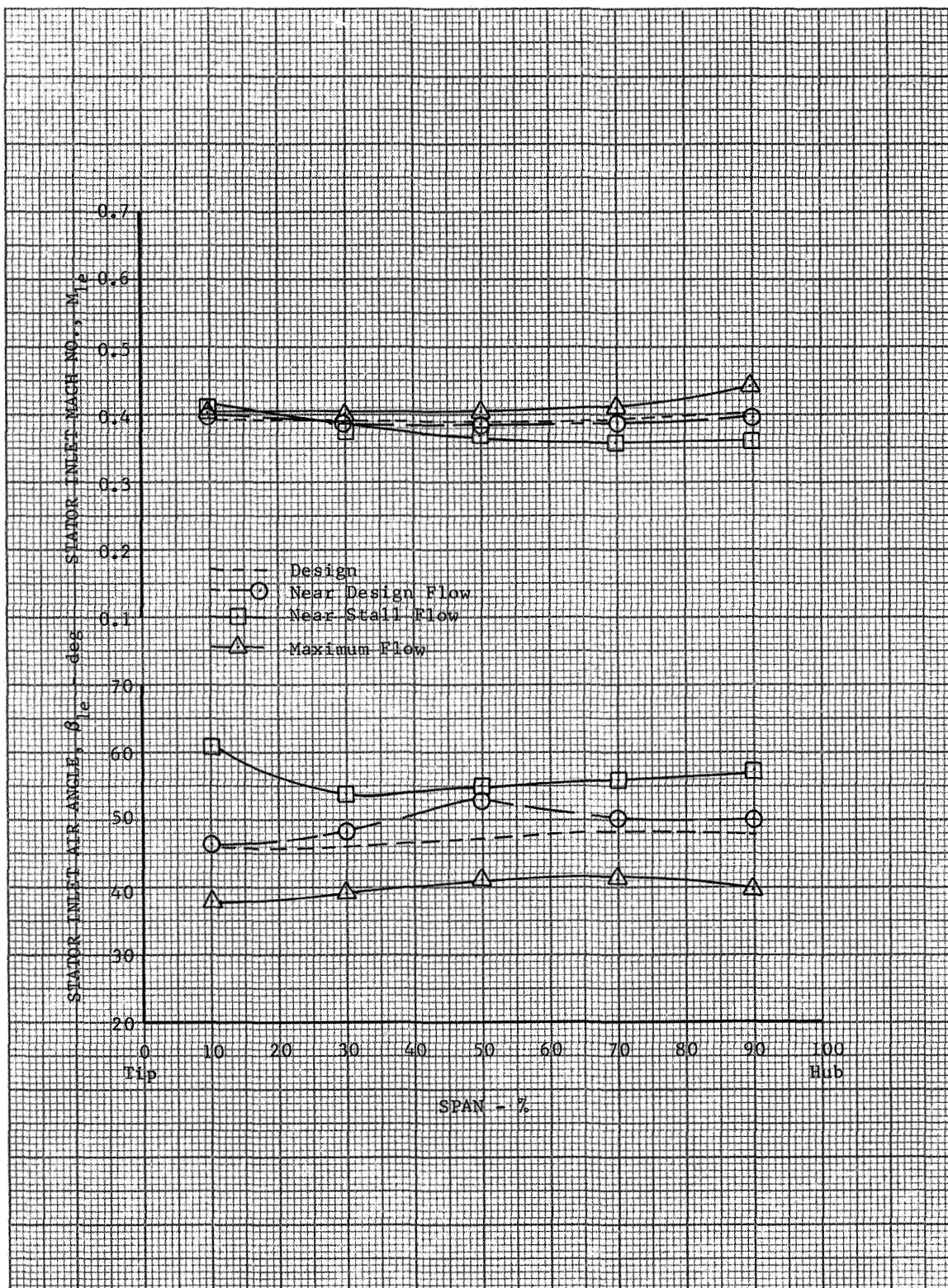
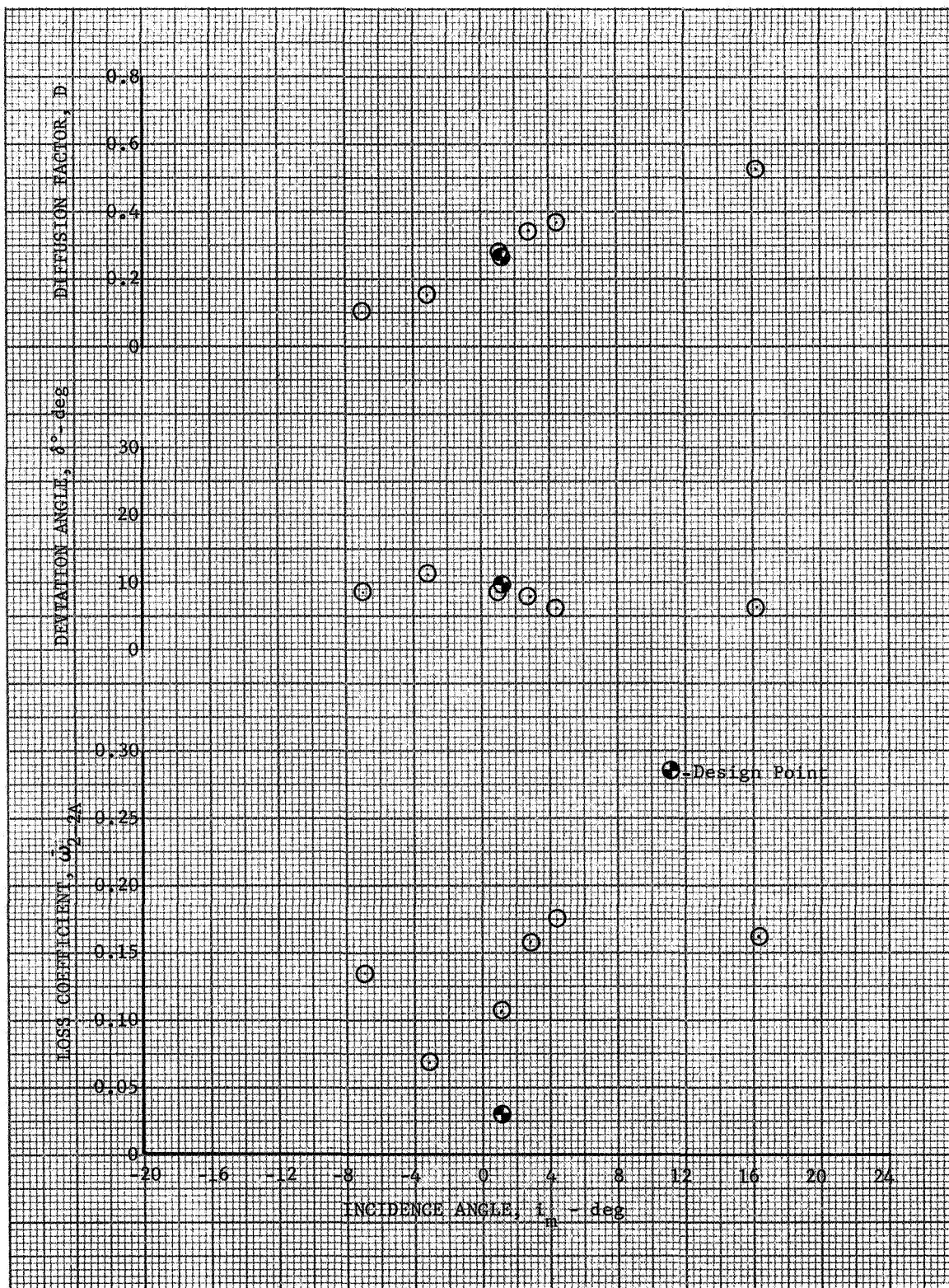
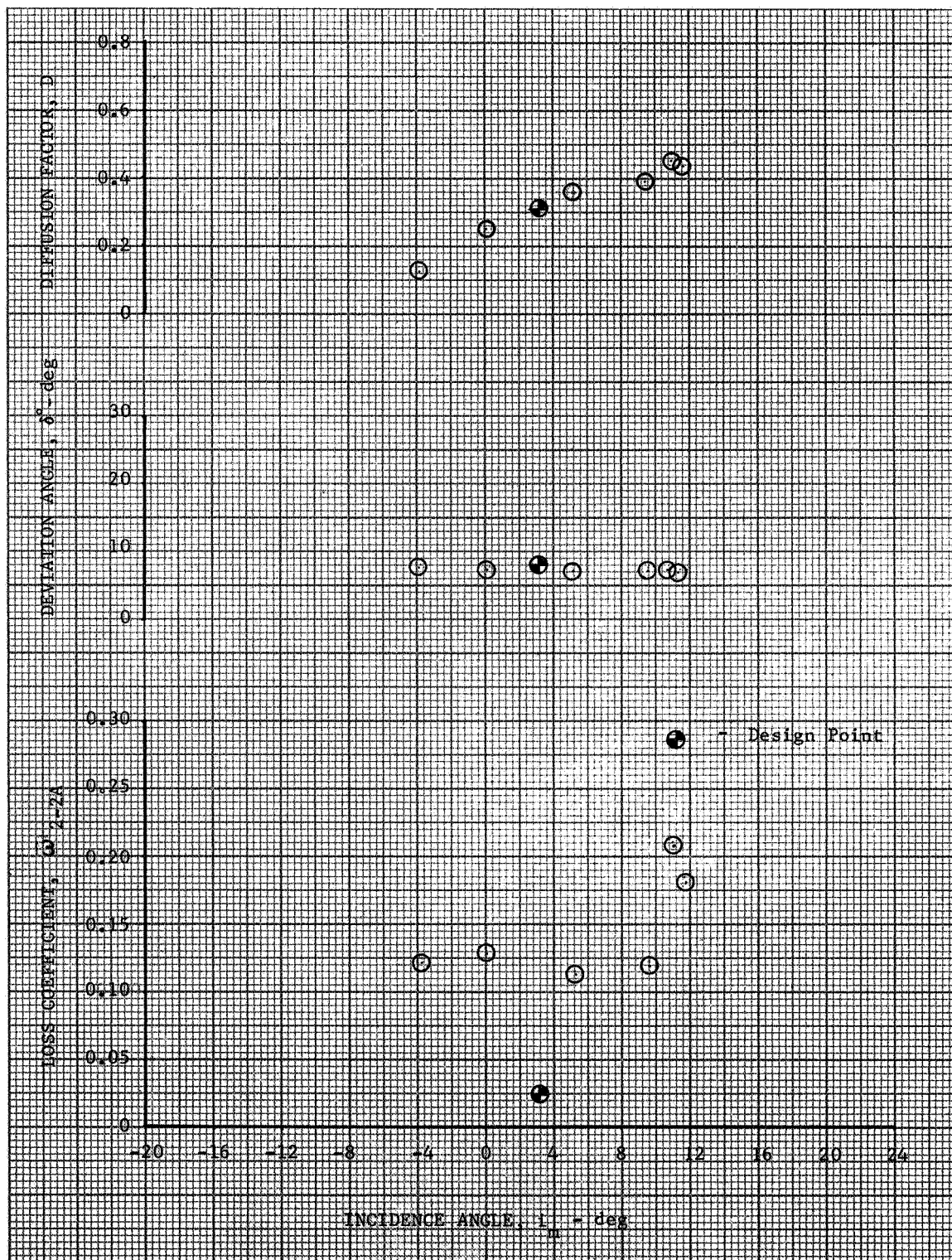


Figure 48. Stator Inlet Air Angle and Mach Number Distribution: Cruise Configuration, 70% Equivalent Rotor Speed



DF 68789

Figure 49. Stator Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 10% Span From Tip



DF 68790

Figure 50. Stator Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 30% Span From Tip

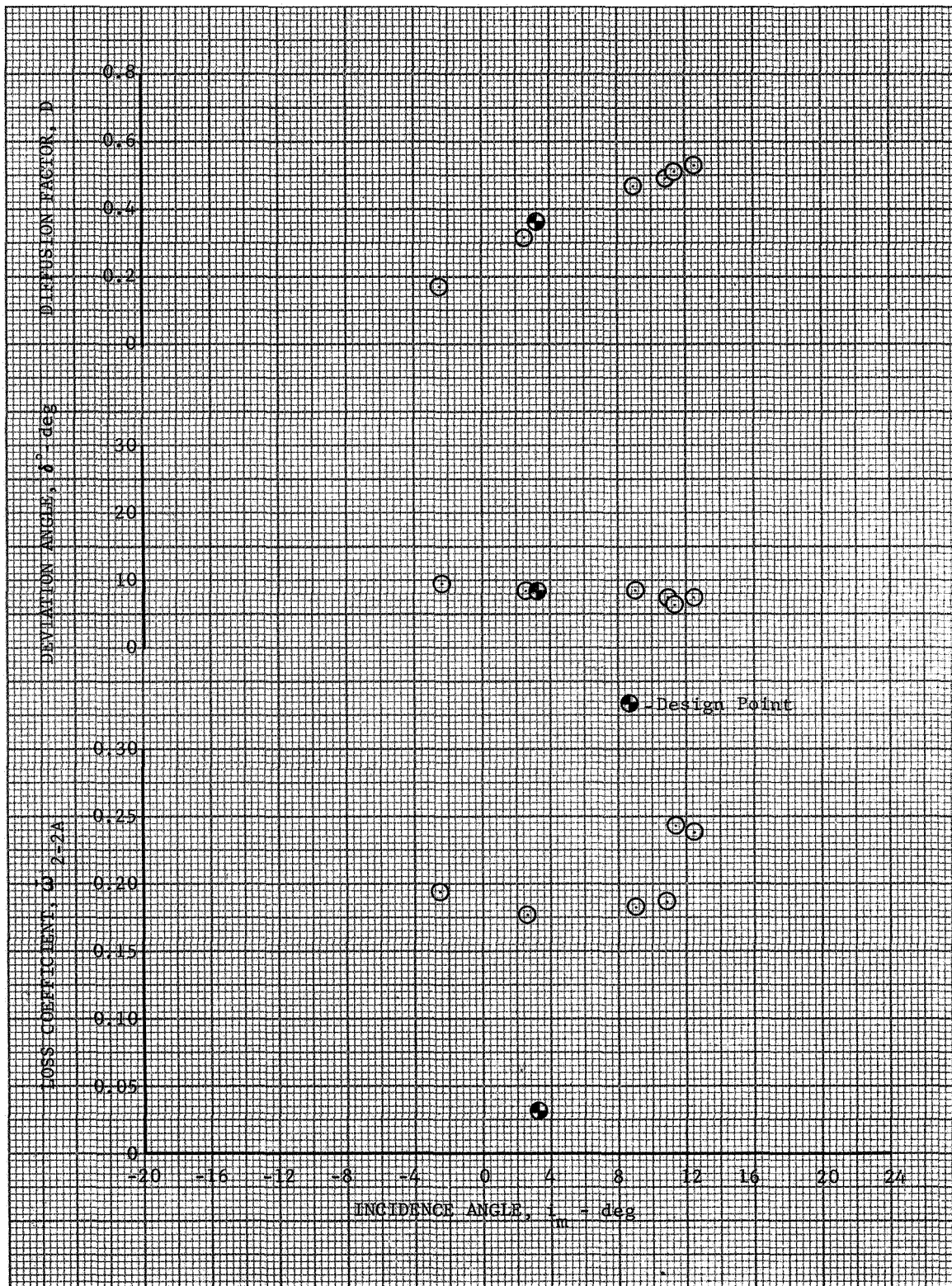


Figure 51. Stator Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 50% Span From Tip

DF 68791

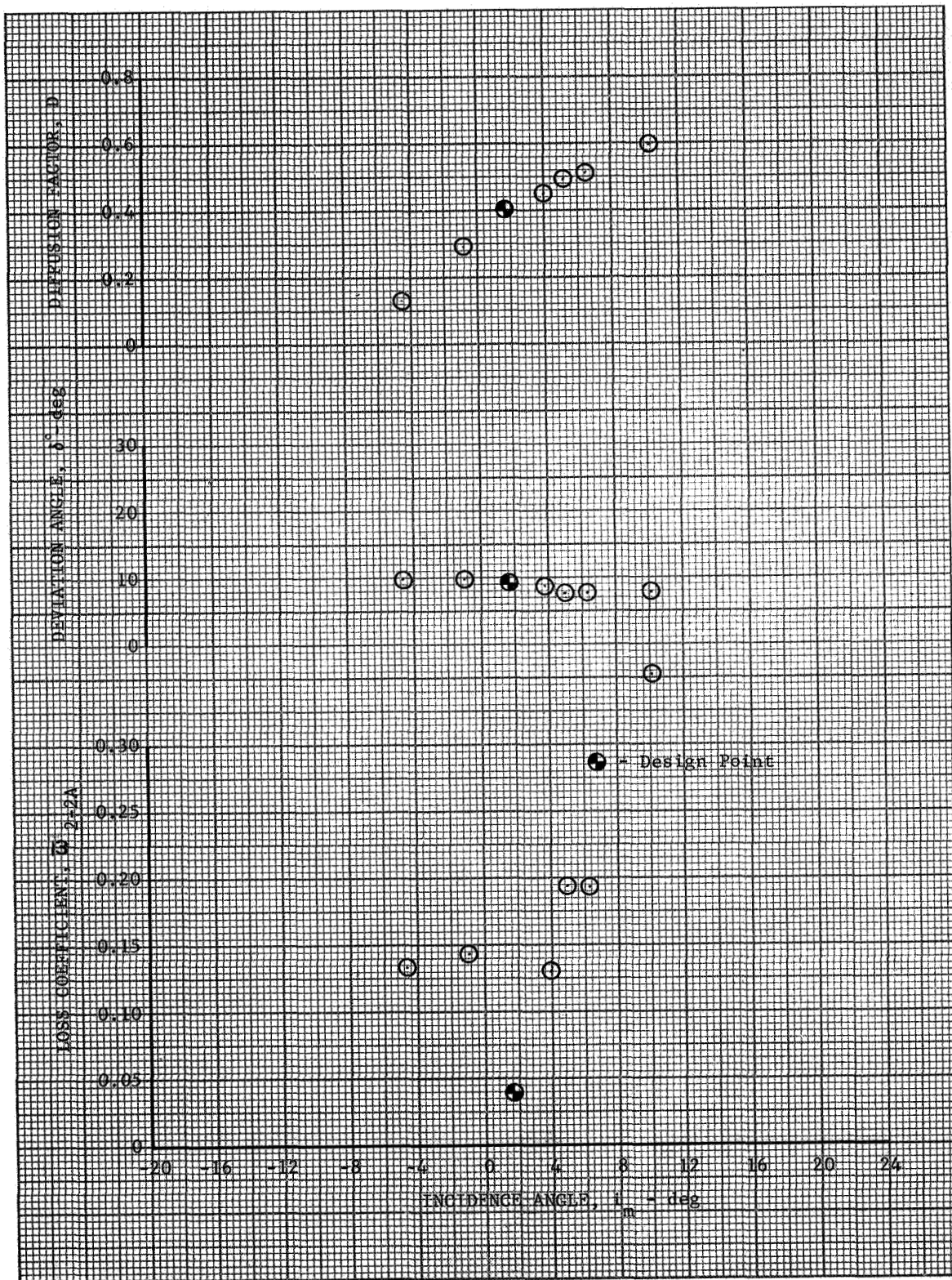


Figure 52. Stator Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 70% Span From Tip

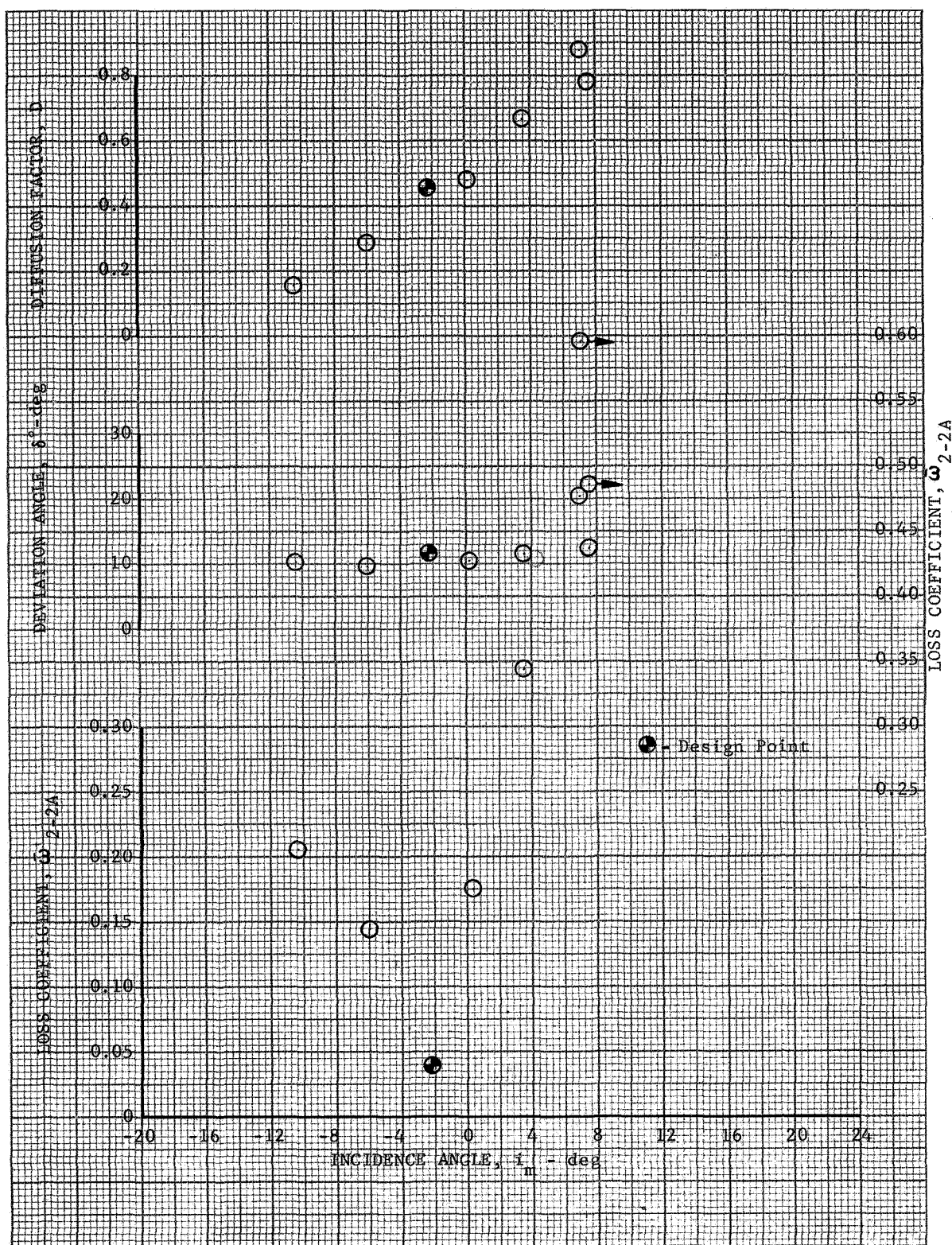
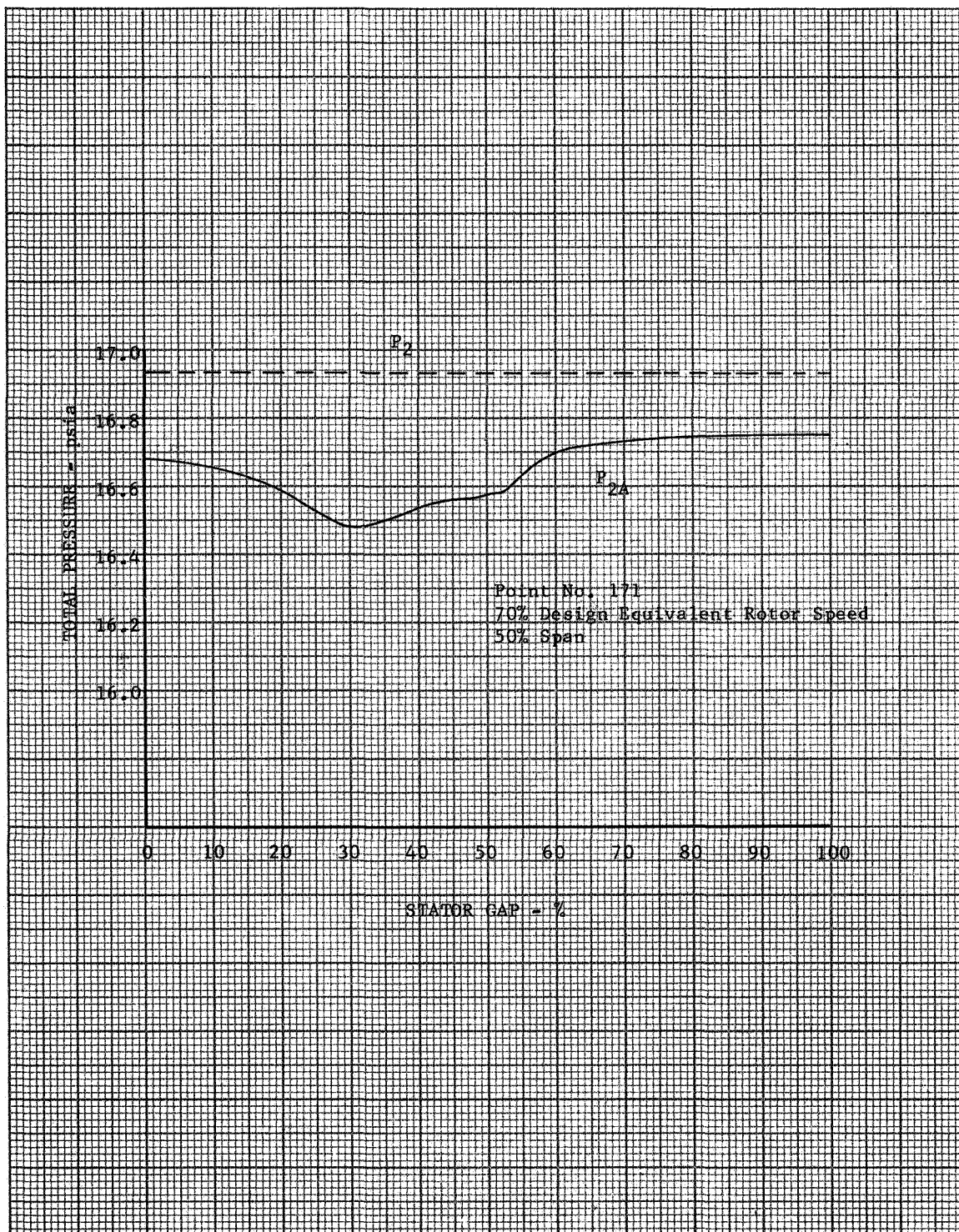


Figure 53. Stator Blade Element Performance: Cruise Configuration, 70% Equivalent Rotor Speed, 90% Span From Tip



DF 68795

Figure 54. Stator B Cruise Configuration Wake Total Pressure Profile

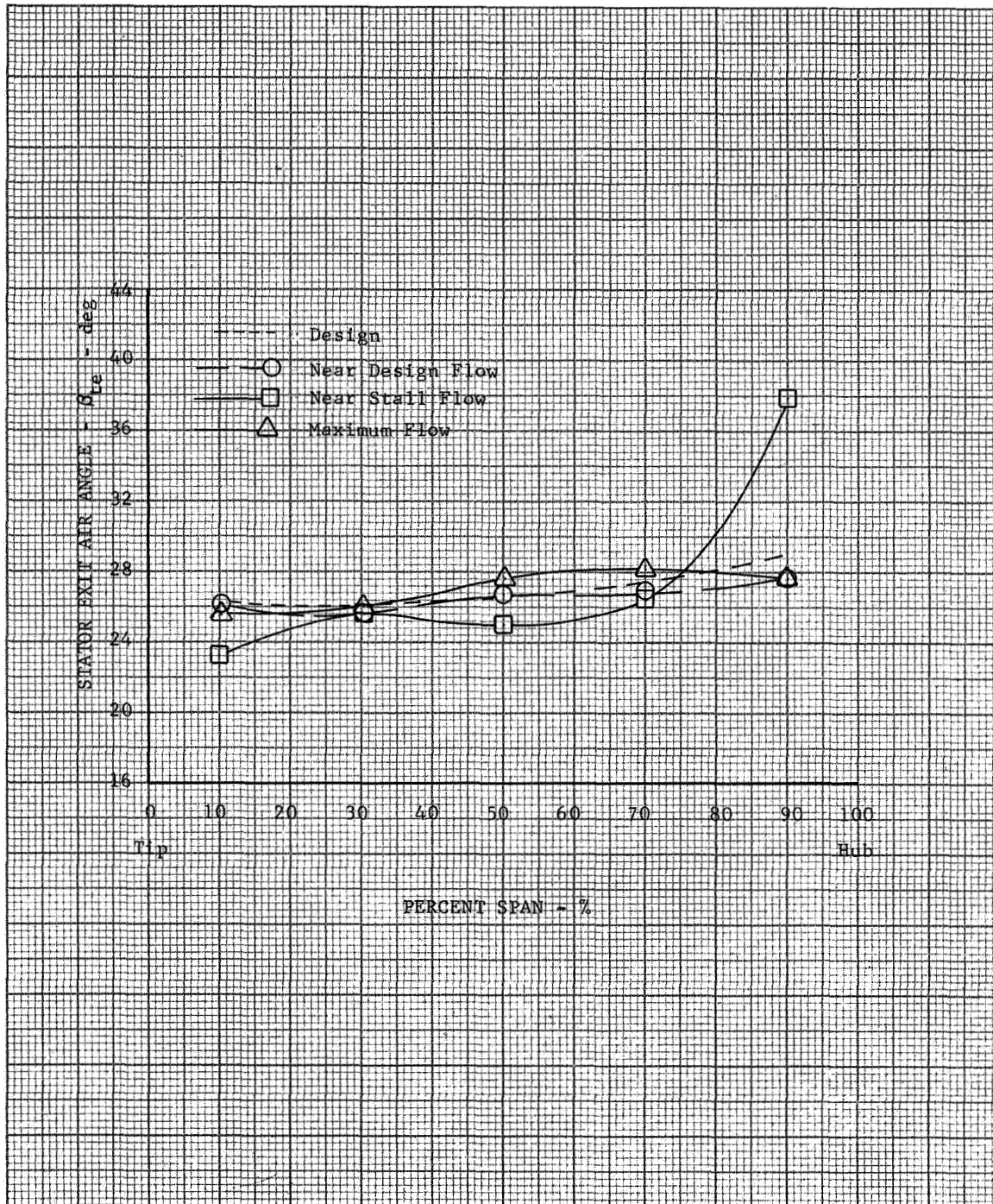
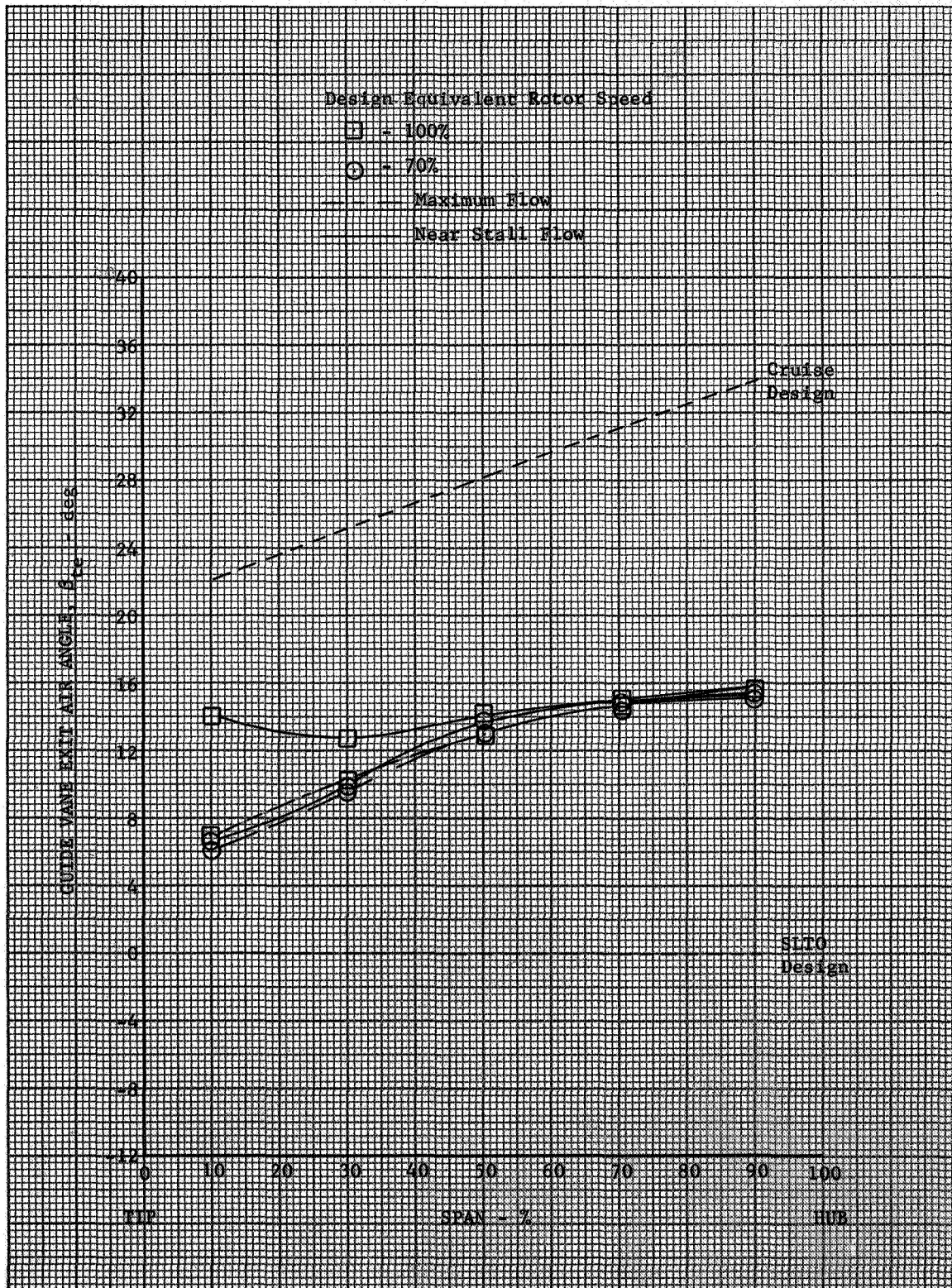


Figure 55. Stator Exit Air Angle vs Percent Span:
Cruise Configuration, 70% Equivalent
Rotor Speed

DF 68794



DF 68796

Figure 56. Guide Vane Exit Air Angle: Intermediate Configuration

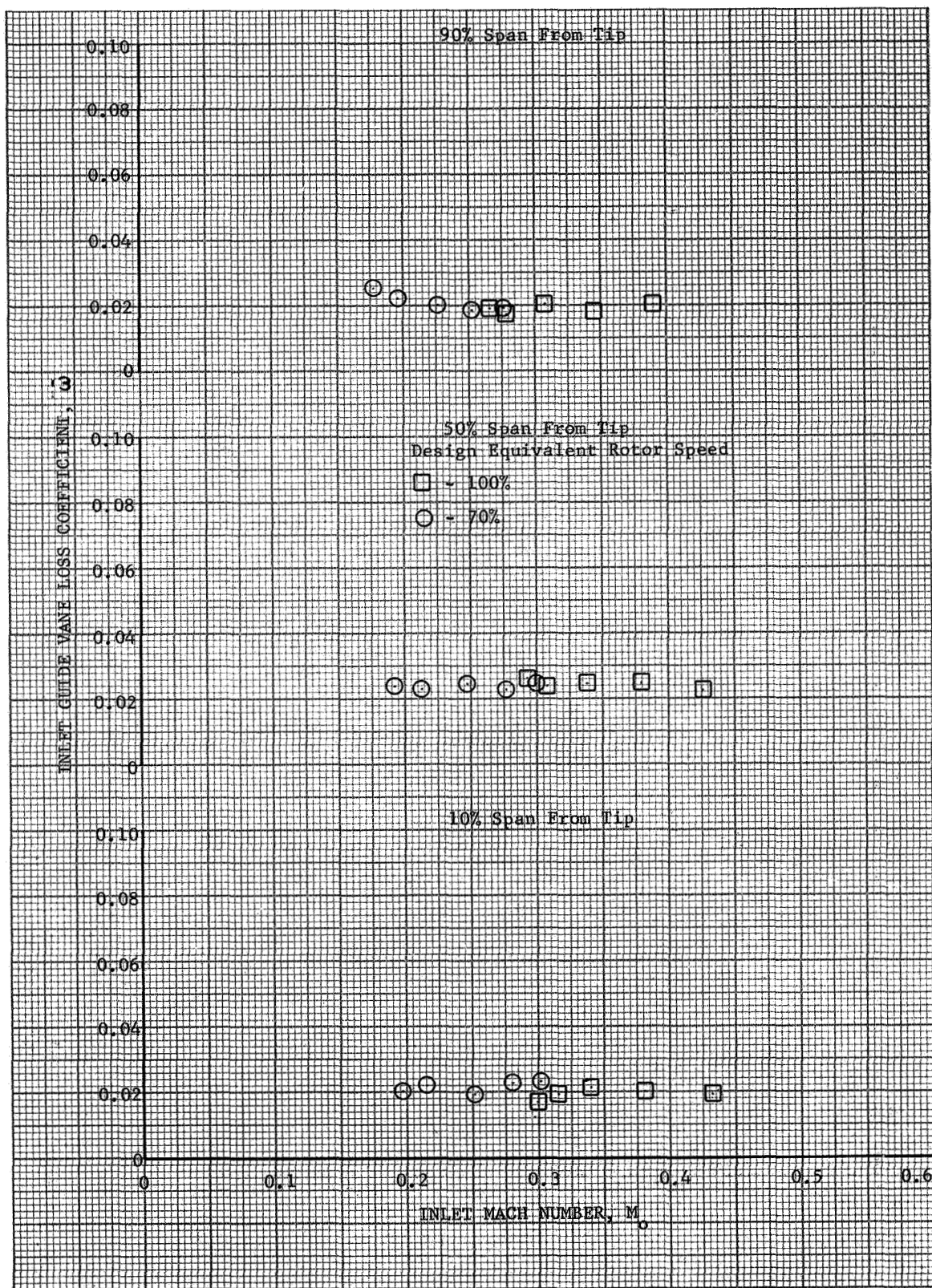


Figure 57. Inlet Guide Vane Loss Coefficient vs Inlet Mach Number: Intermediate Configuration

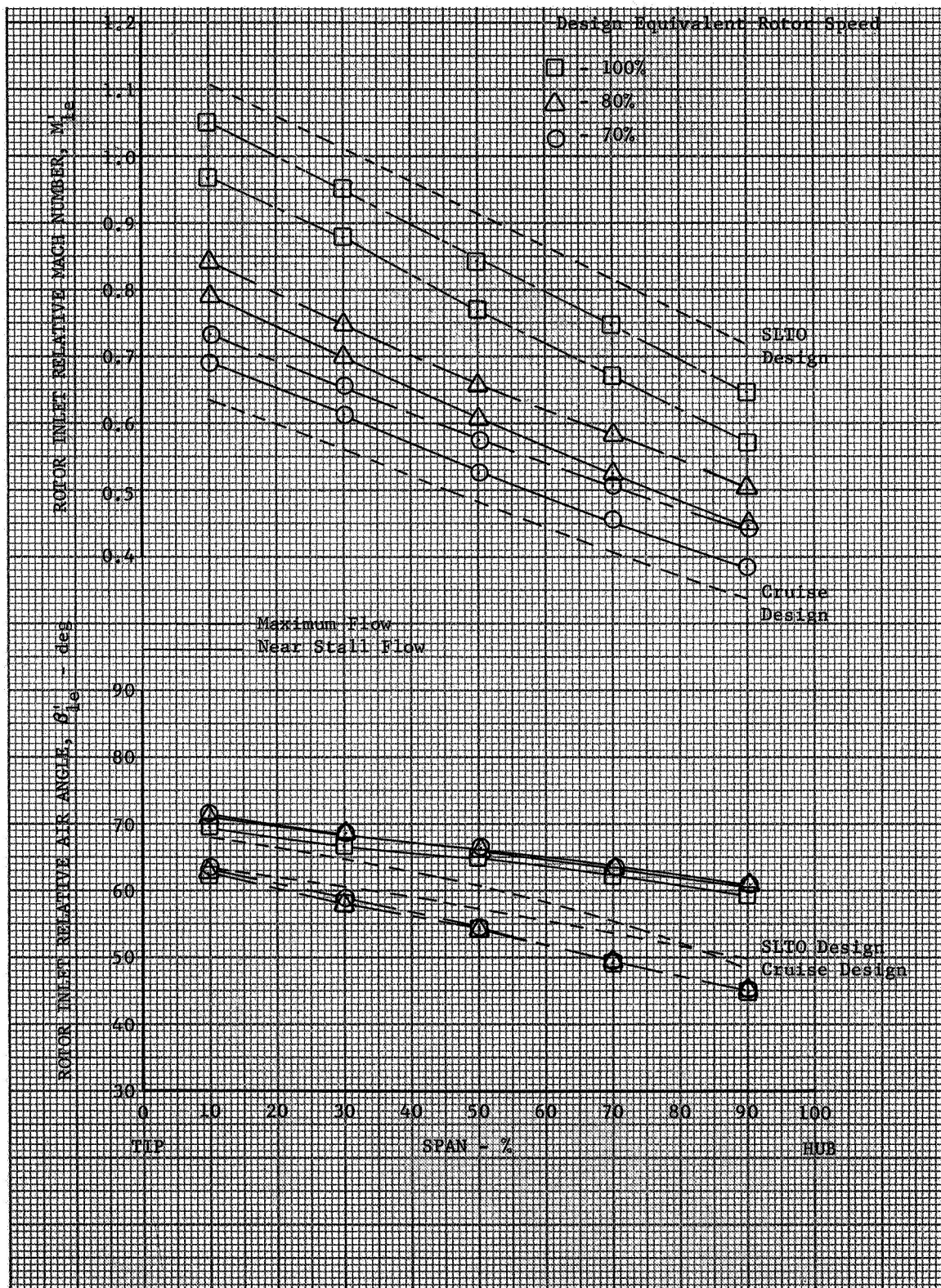
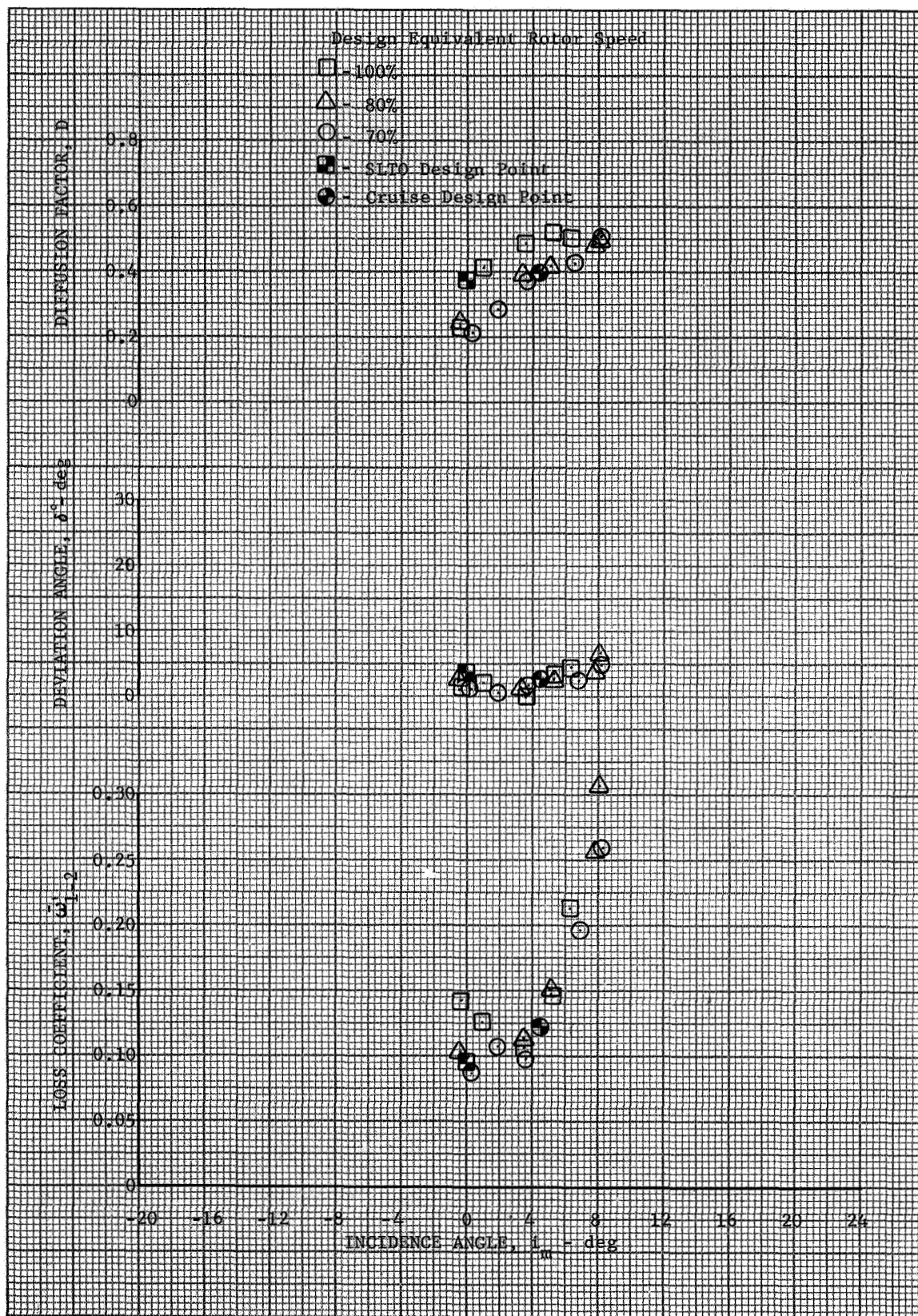


Figure 58. Rotor Inlet Relative Air Angle and Mach Number Distribution: Intermediate Configuration



DF 68799

Figure 59. Rotor Blade Element Performance: Intermediate Configuration, 10% Span From Tip

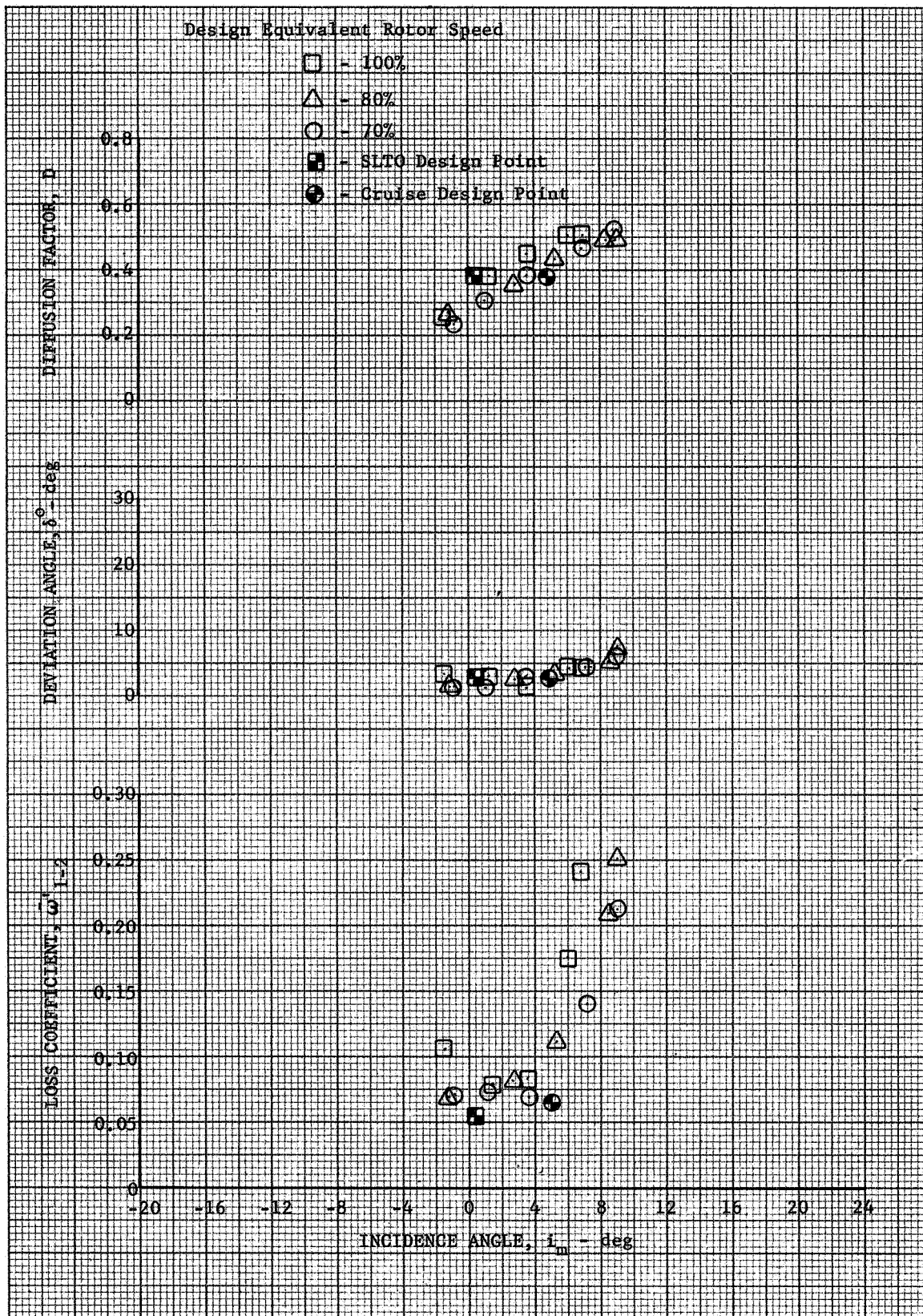
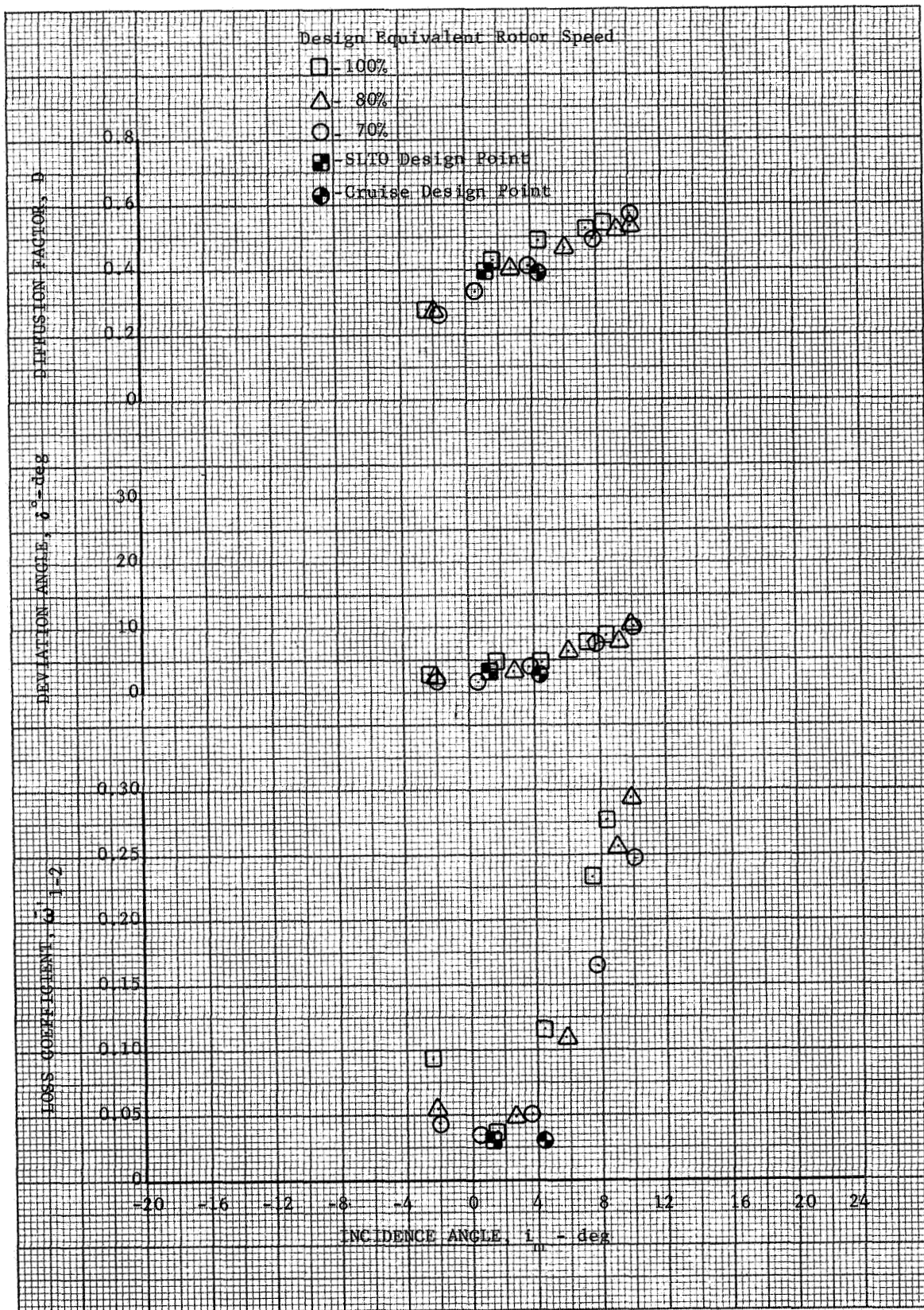


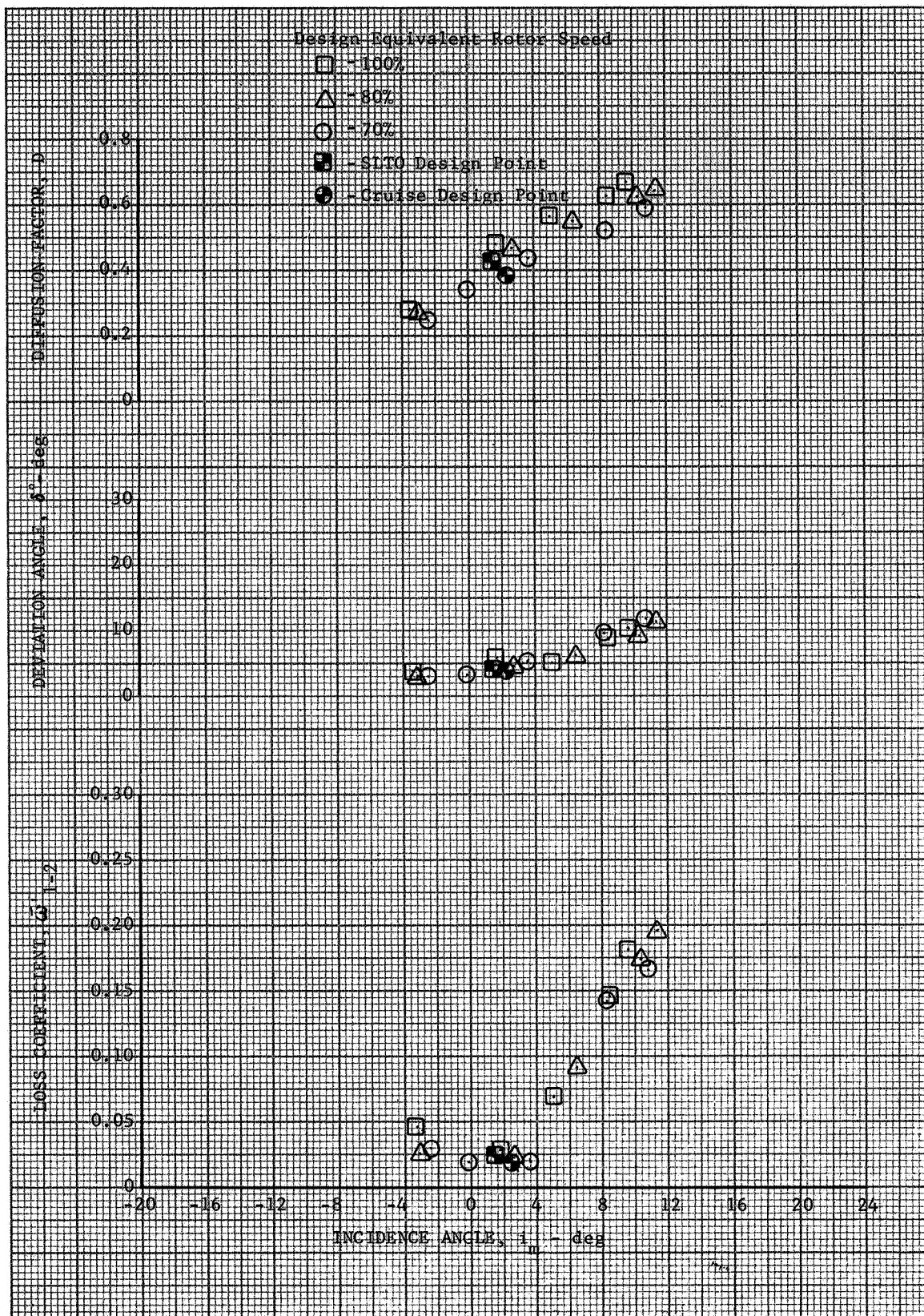
Figure 60. Rotor Blade Element Performance: Intermediate Configuration, 30% Span From Tip

DF 68941



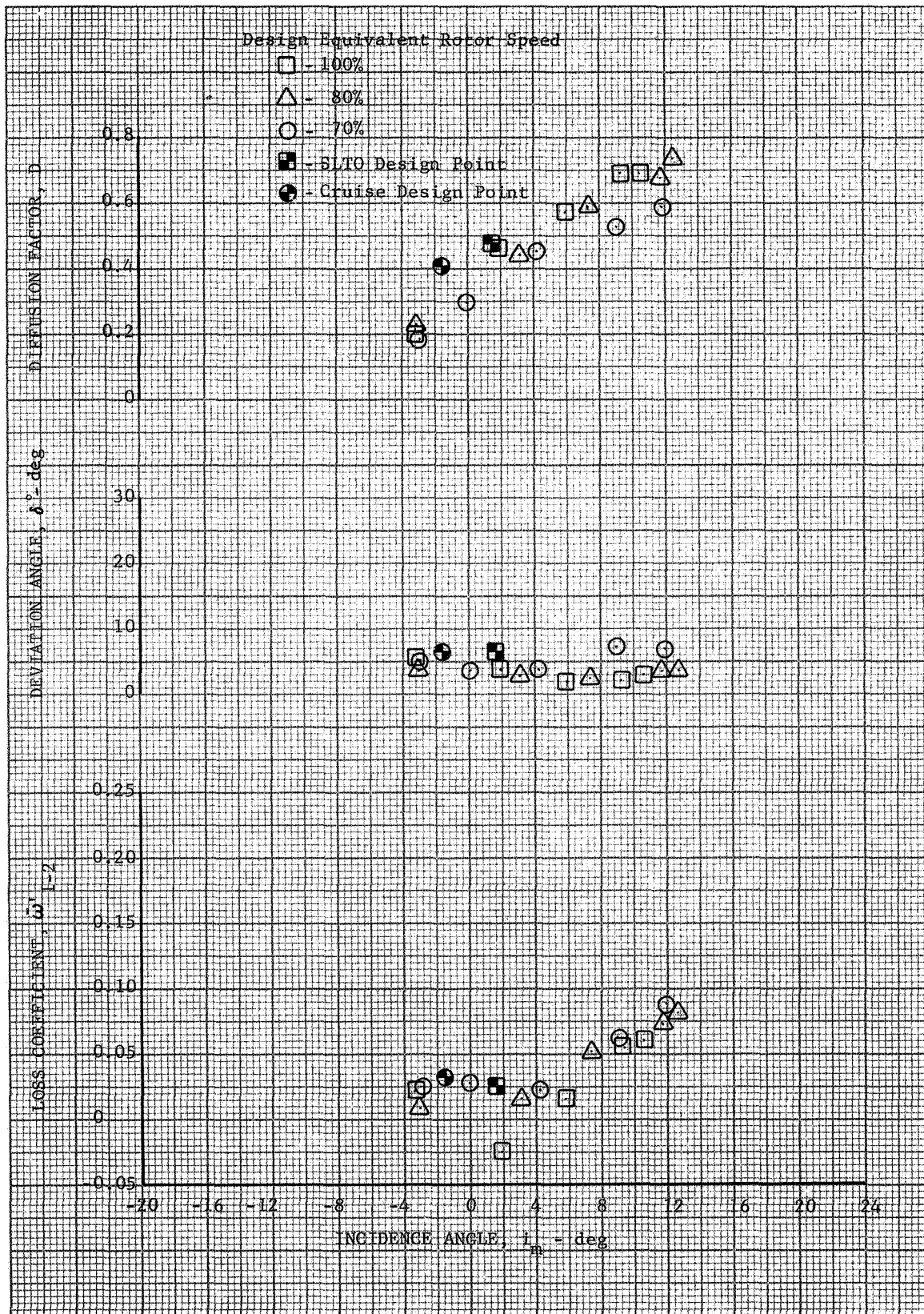
DF 68940

Figure 61. Rotor Blade Element Performance: Intermediate Configuration, 50% Span From Tip



DF 68939

Figure 62. Rotor Blade Element Performance: Intermediate Configuration, 70% Span From Tip



DF 68938

Figure 63. Rotor Blade Element Performance: Intermediate Configuration, 90% Span From Tip

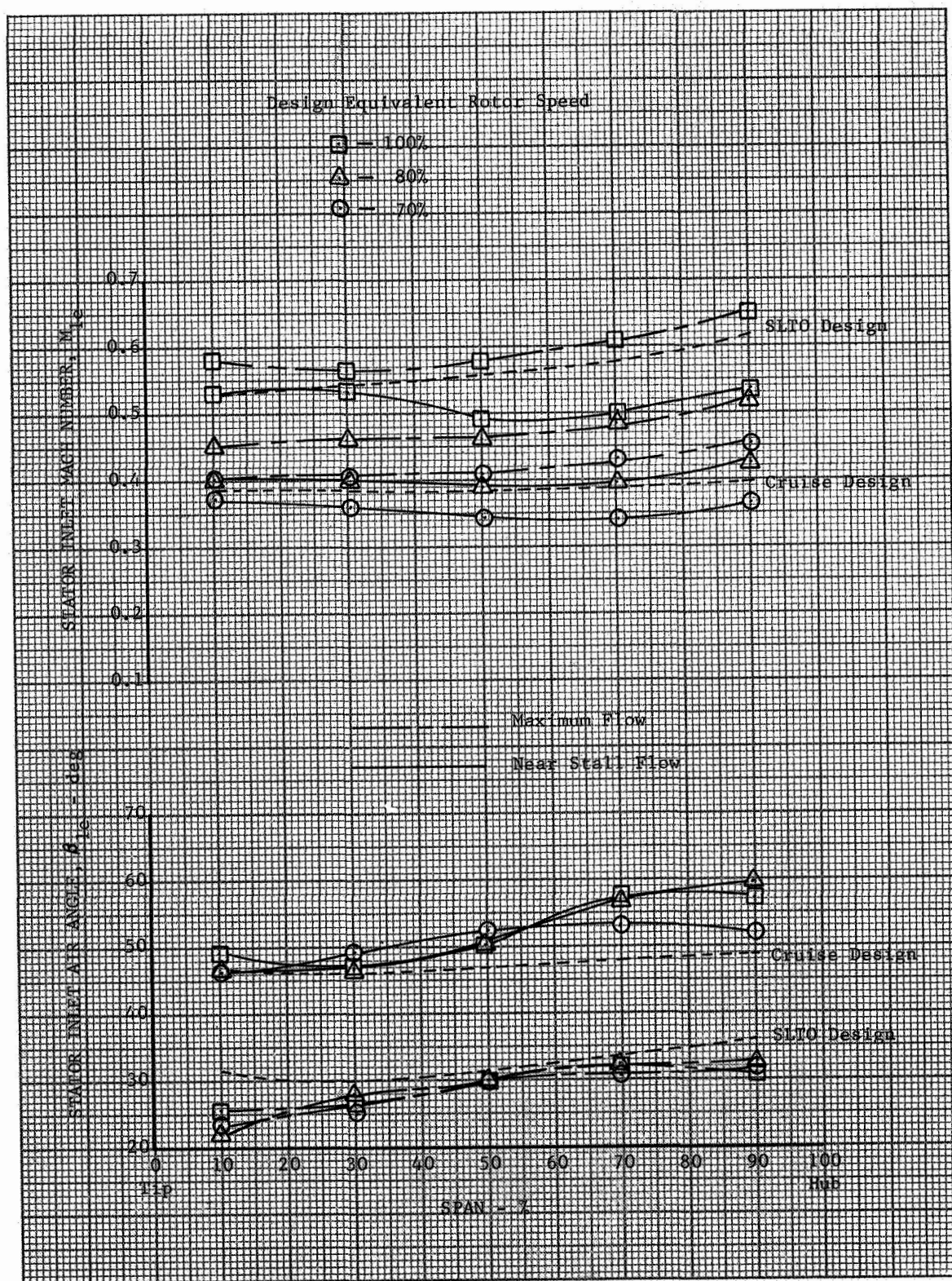
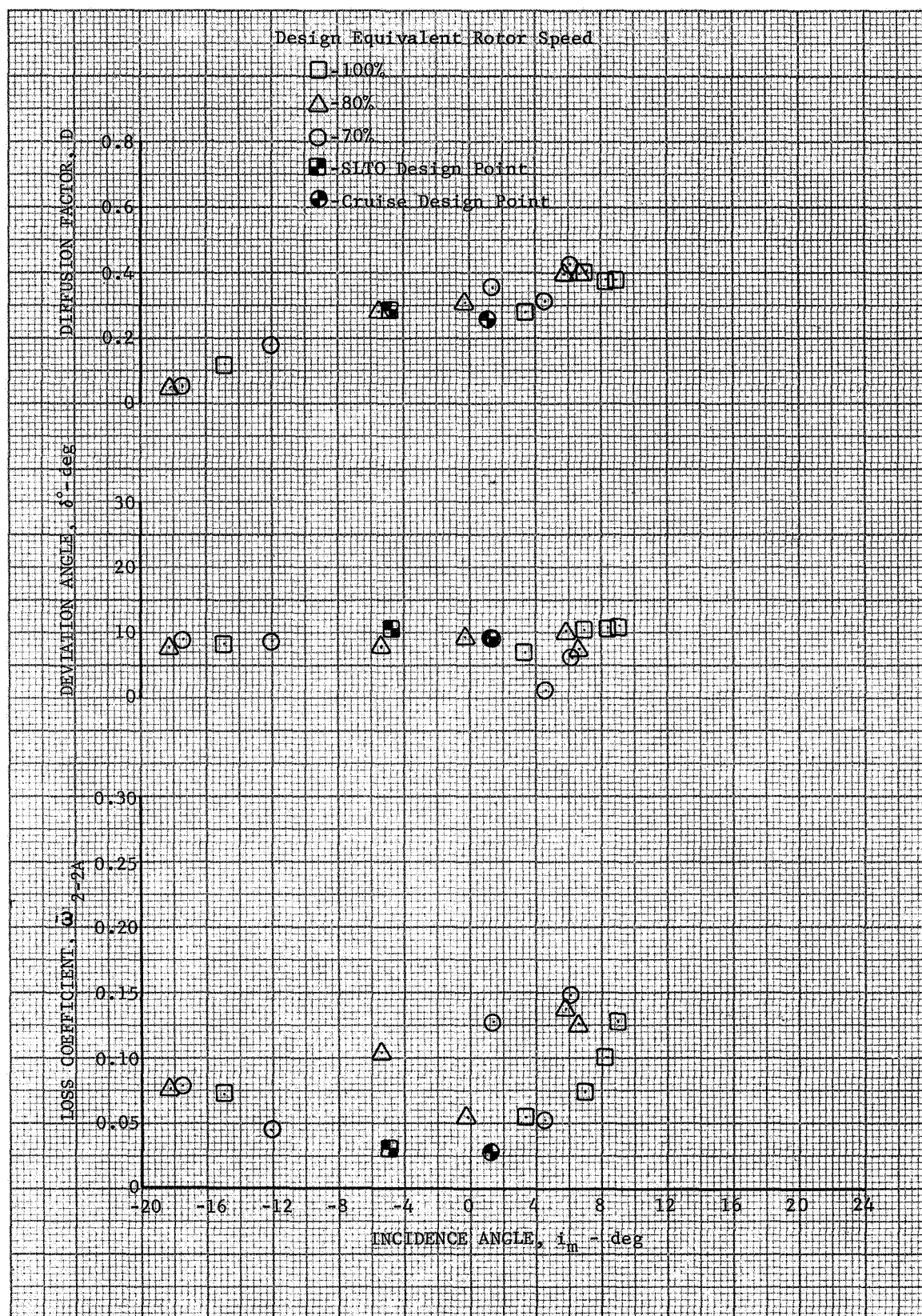


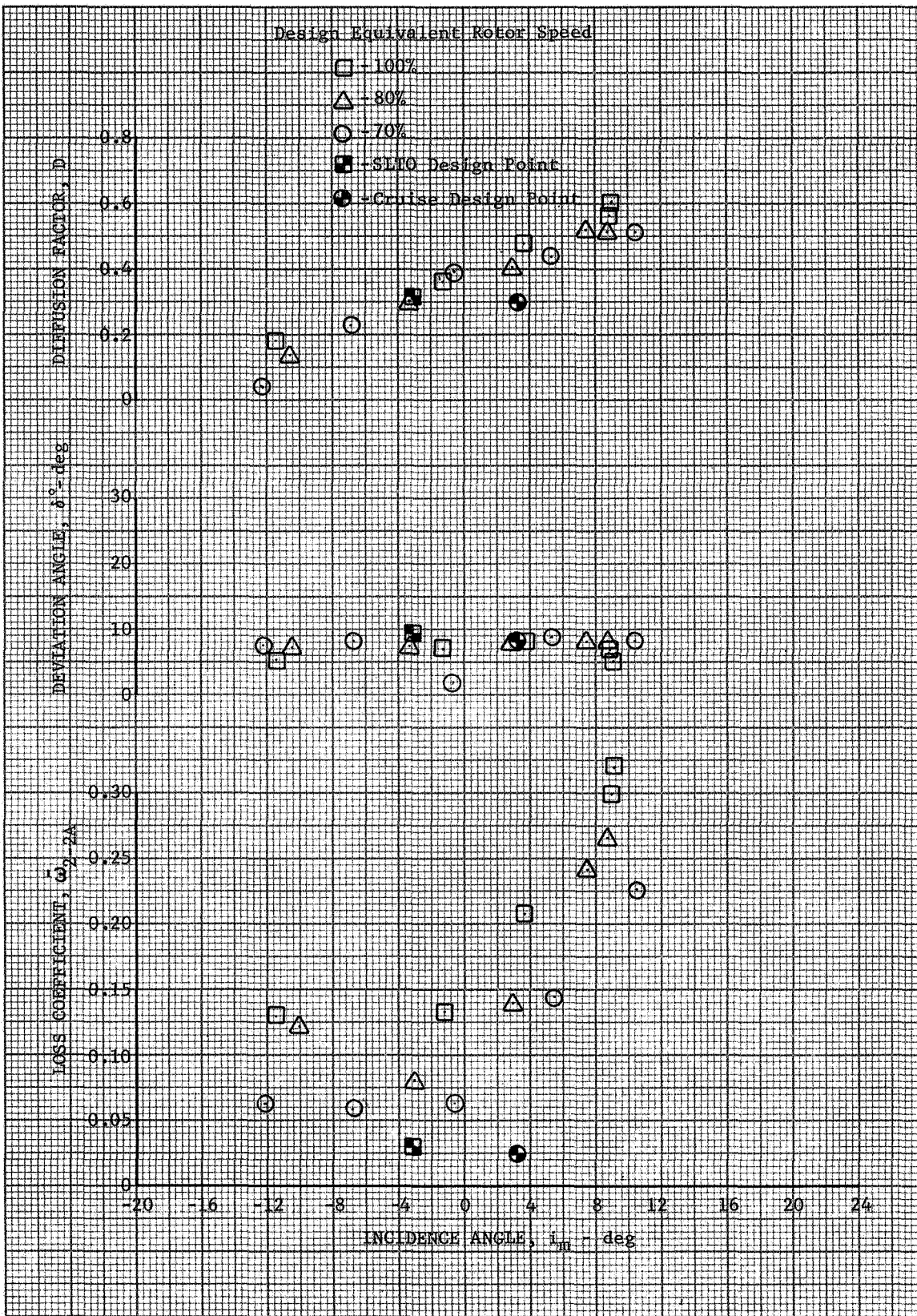
Figure 64. Stator Inlet Air Angle and Mach Number Distribution: Intermediate Configuration

DF 68937



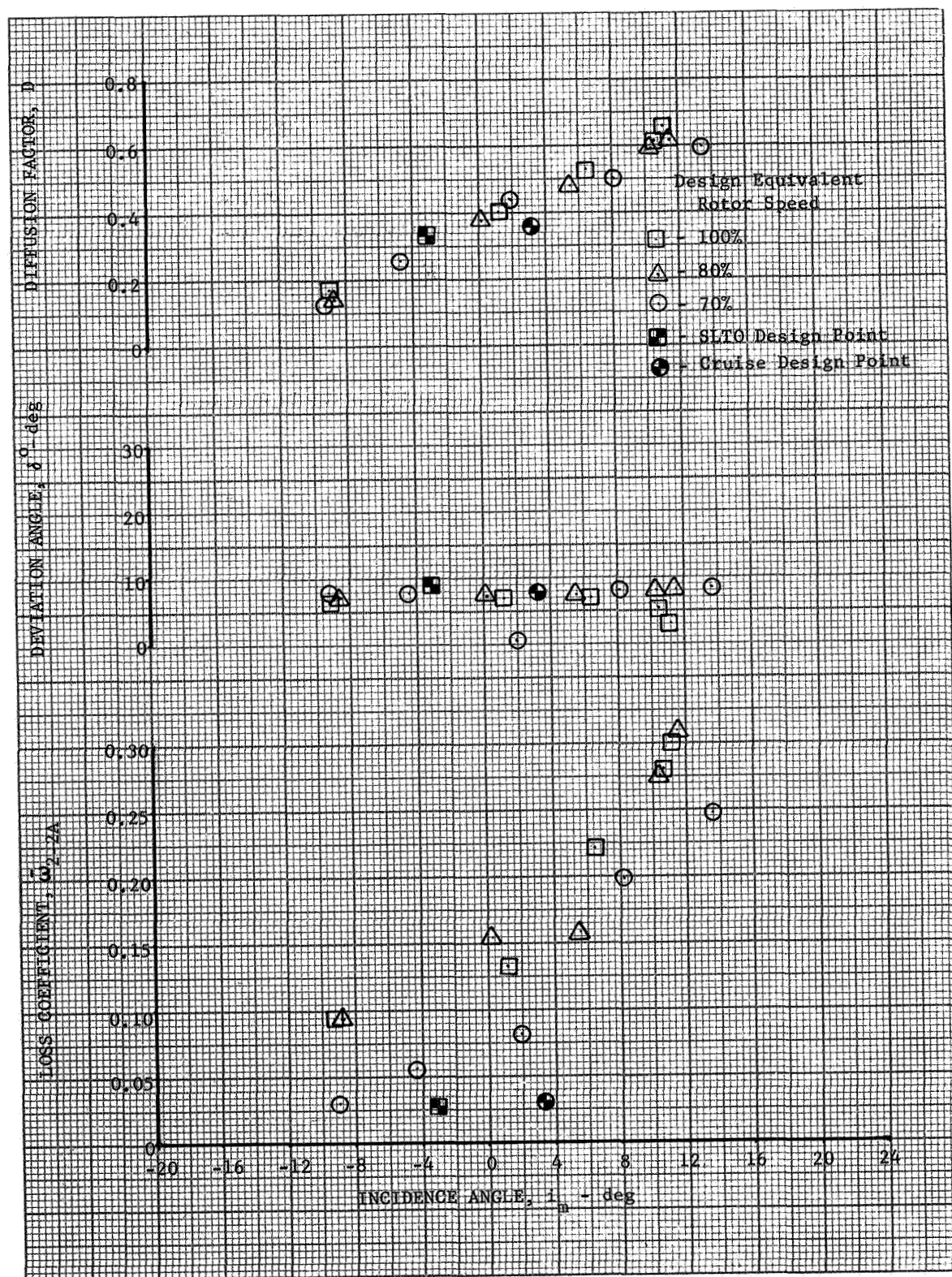
DF 68936

Figure 65. Stator Blade Element Performance: Intermediate Configuration, 10% Span From Tip



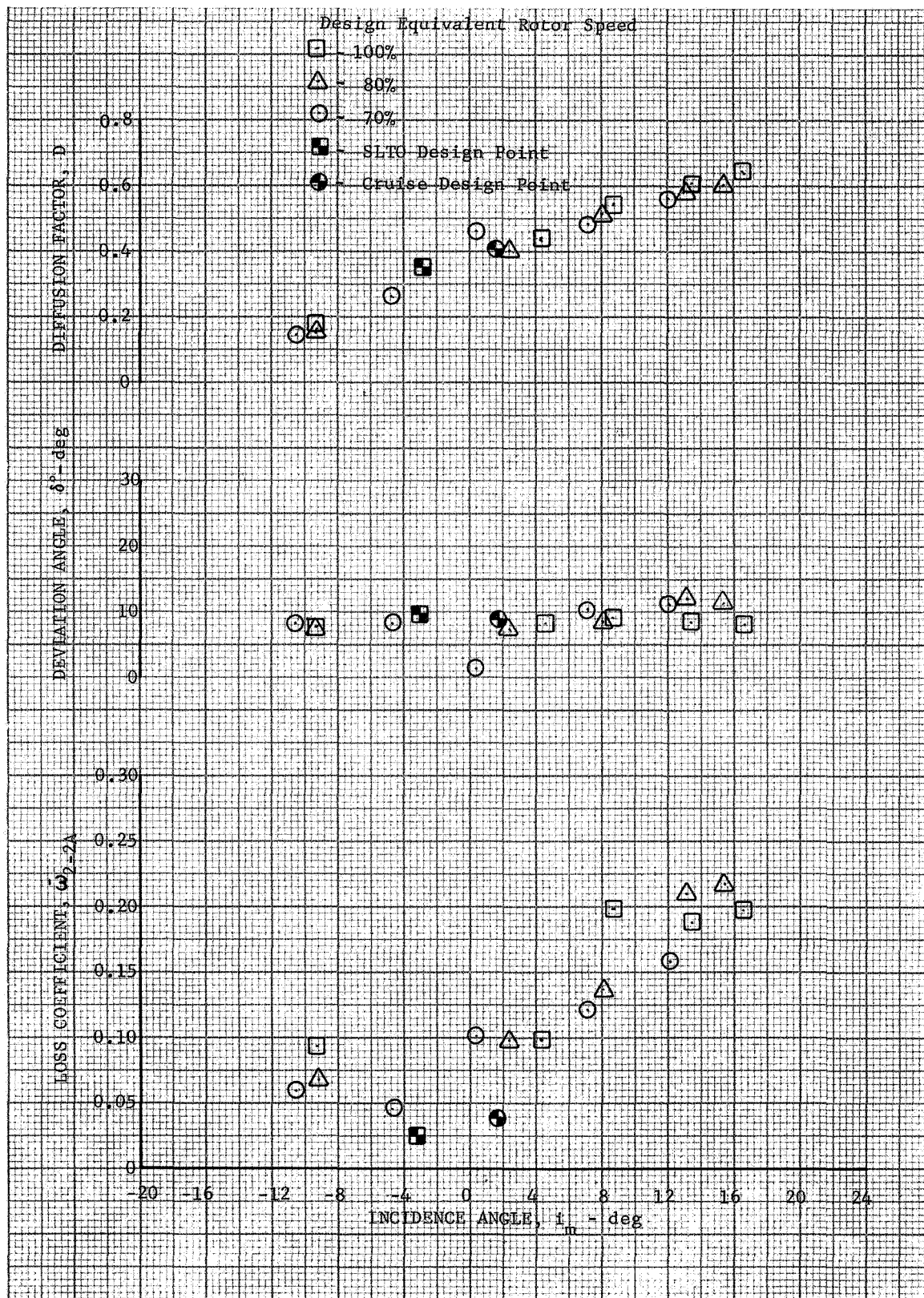
DF 68935

Figure 66. Stator Blade Element Performance: Intermediate Configuration, 30% Span From Tip



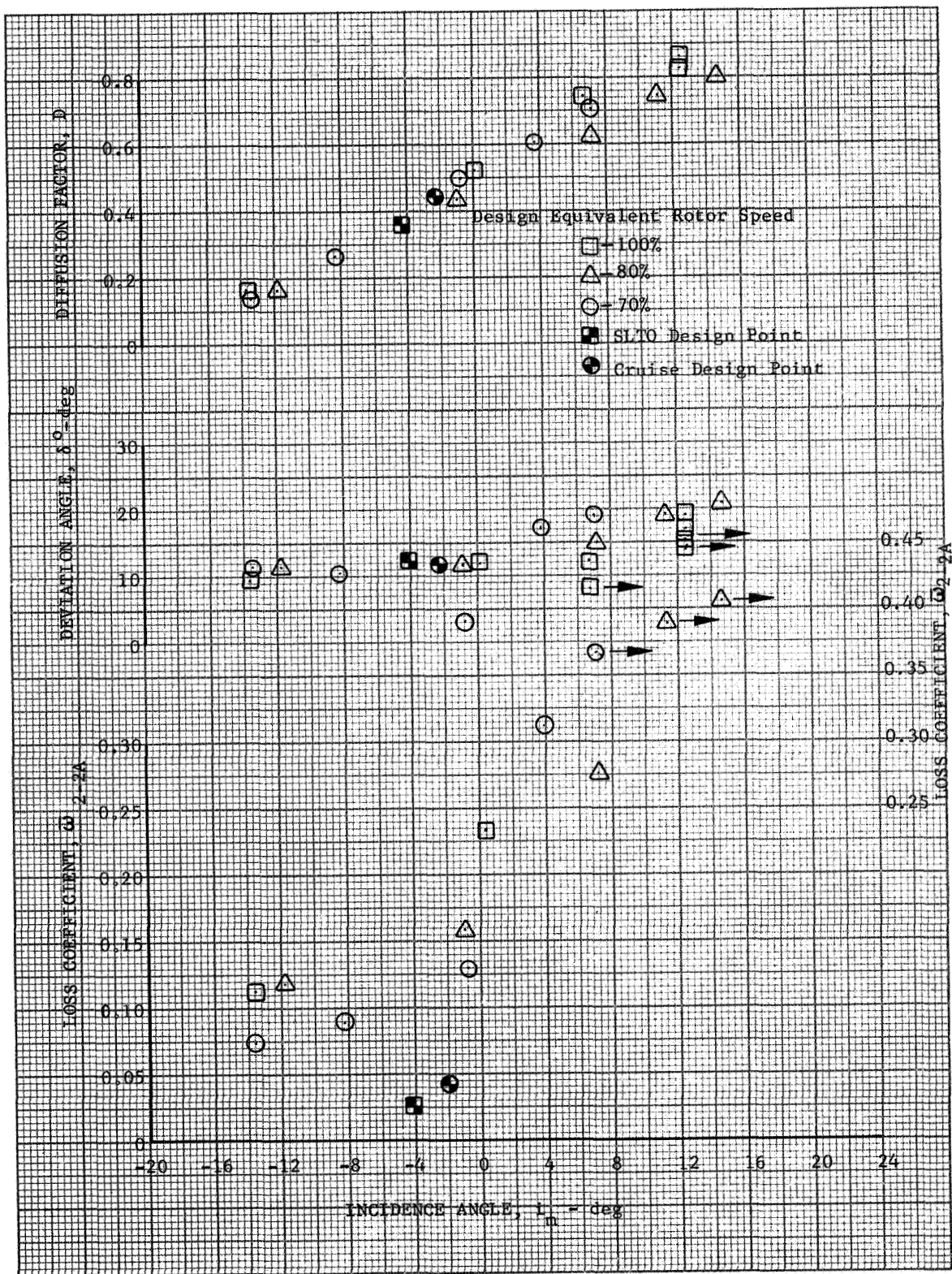
DF 68942

Figure 67. Stator Blade Element Performance: Intermediate Configuration, 50% Span From Tip



DF 68808

Figure 68. Stator Blade Element Performance: Intermediate Configuration, 70% Span From Tip



DF 68809

Figure 69. Stator Blade Element Performance: Intermediate Configuration, 90% Span From Tip

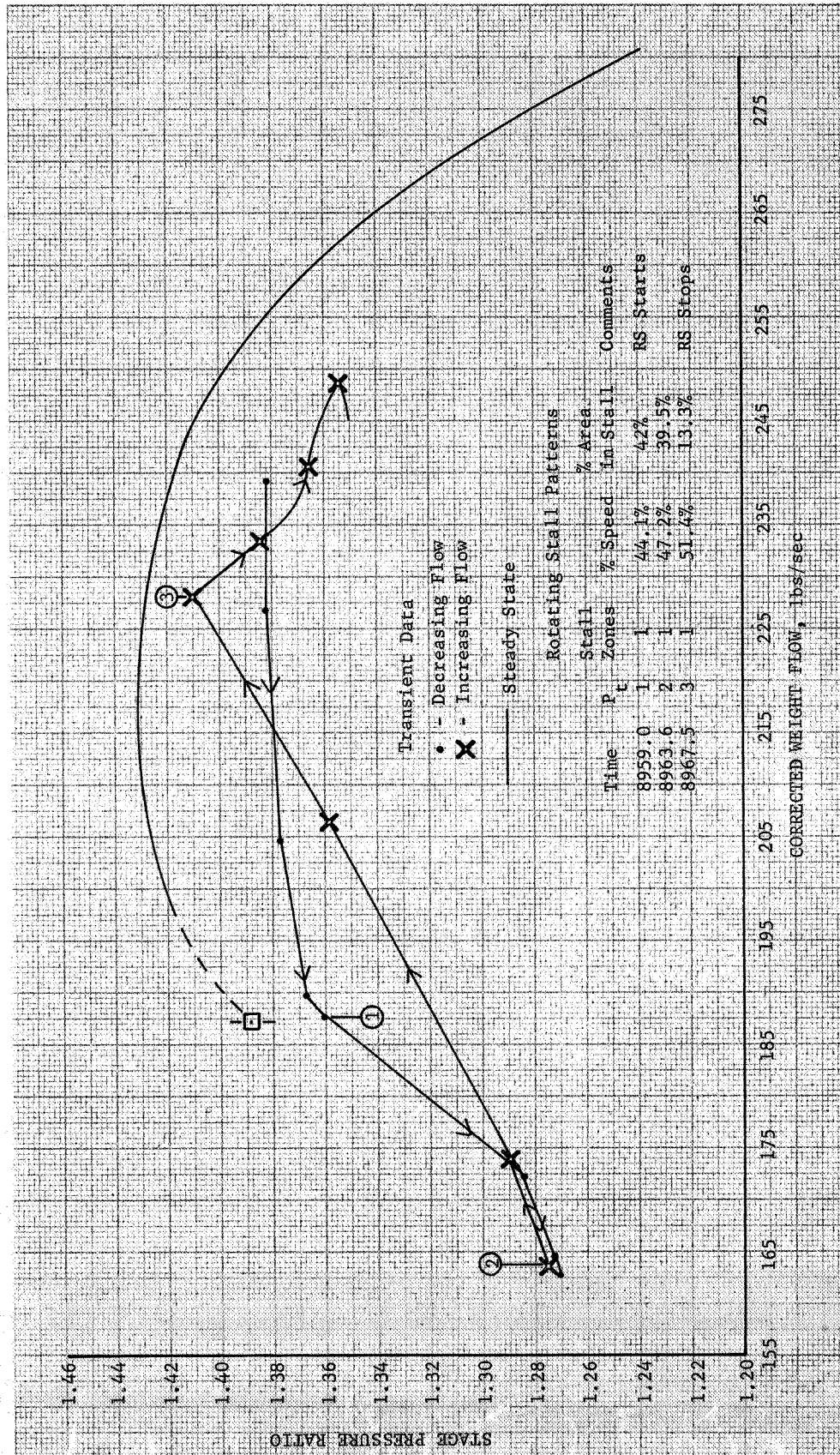


Figure 70. Transient Stall Data: SLTO Configuration, 100% Design Equivalent Rotor Speed

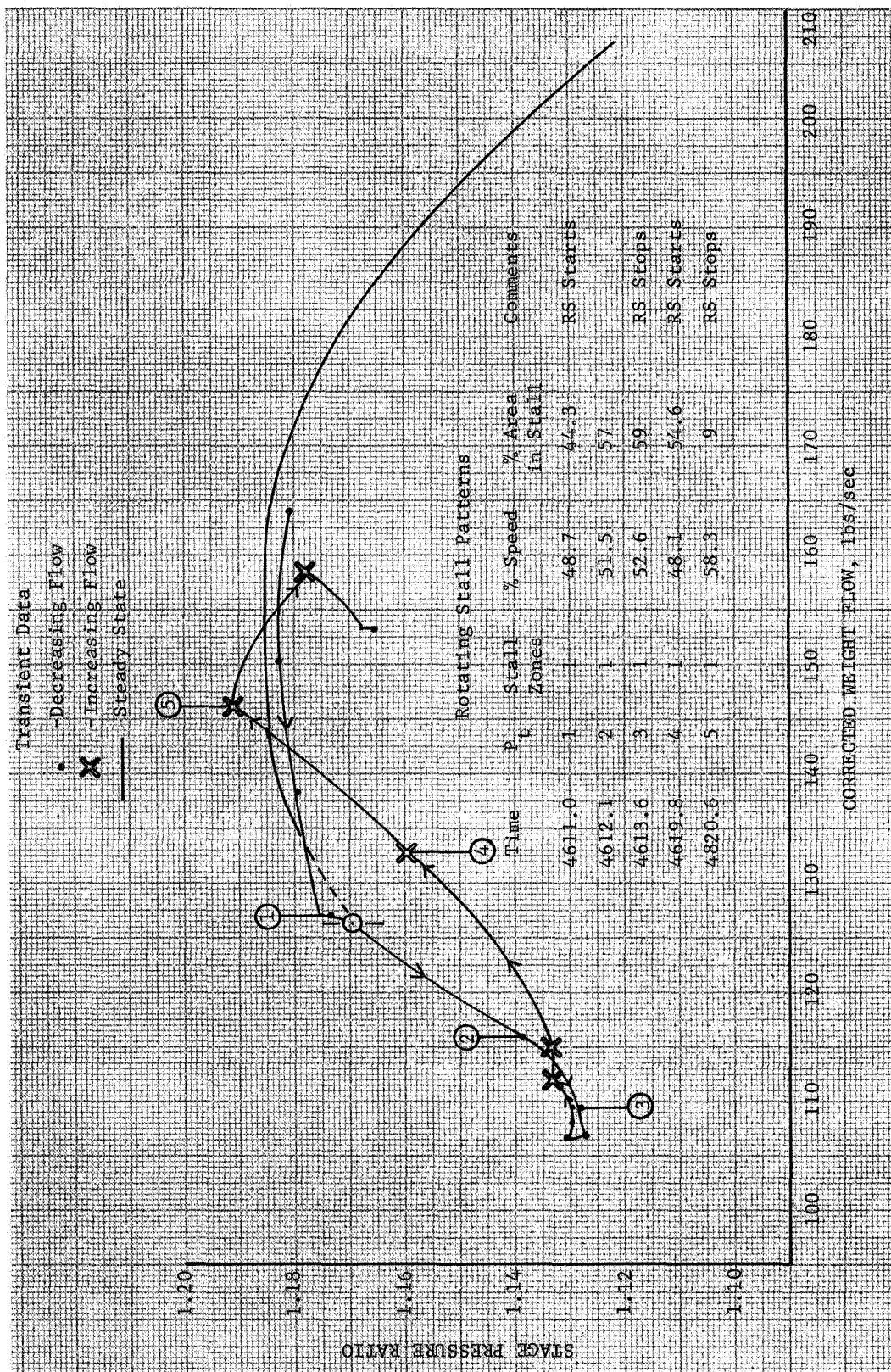


Figure 71. Transient Stall Data: SLTO Configuration, 70% Design Equivalent Rotor Speed

DF 68811

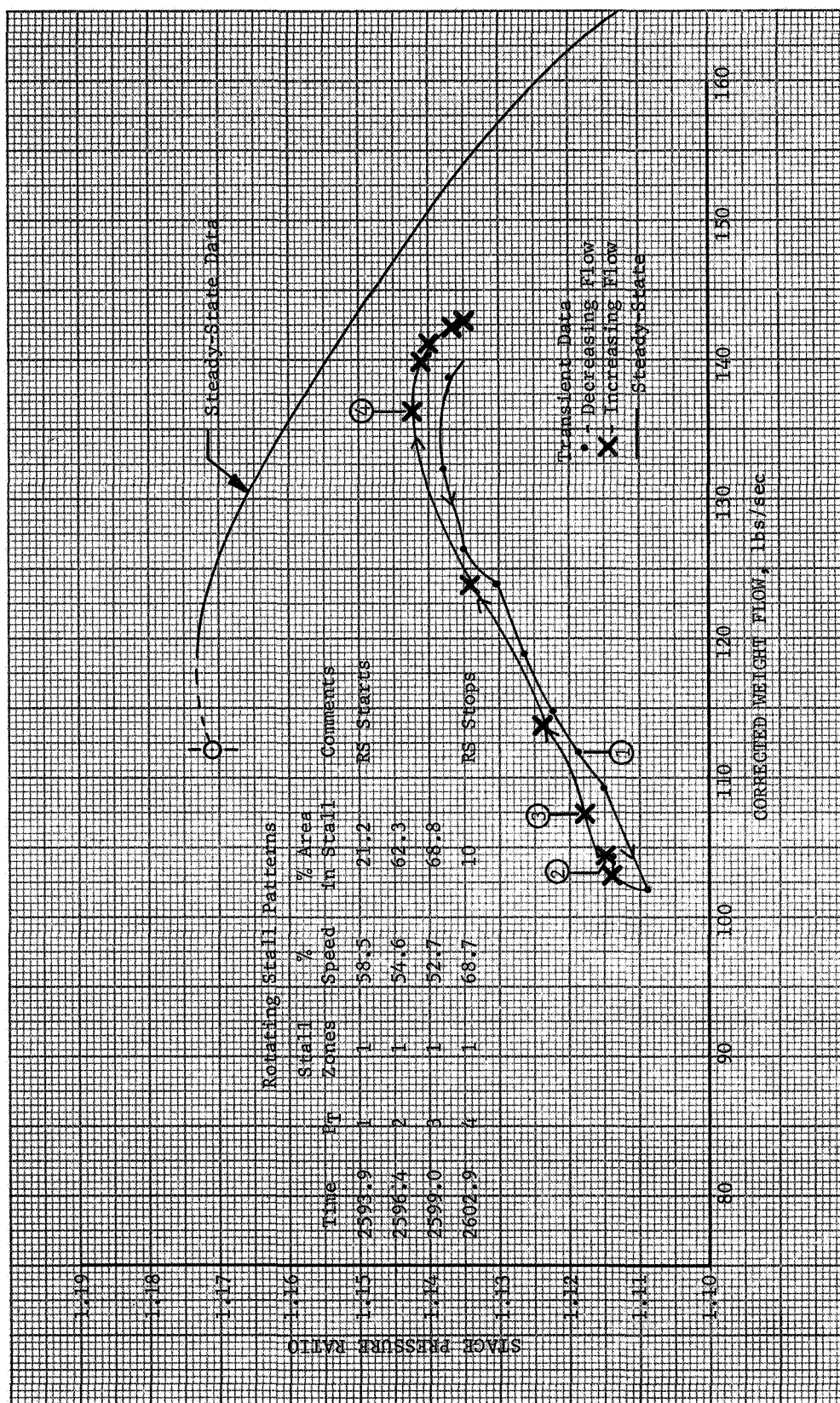
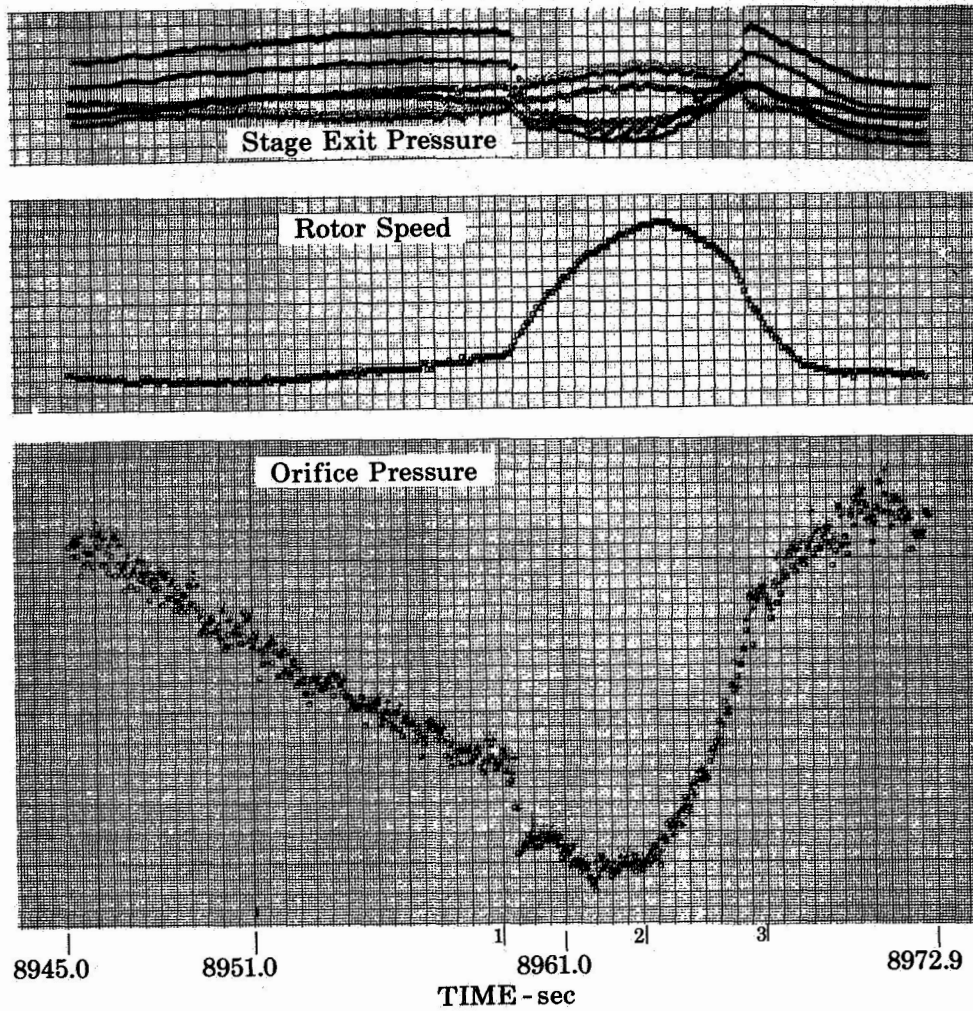


Figure 72. Transient Stall Data: Cruise Configuration, 70% Design Equivalent Rotor Speed

Transient Data



Rotating Stall Data

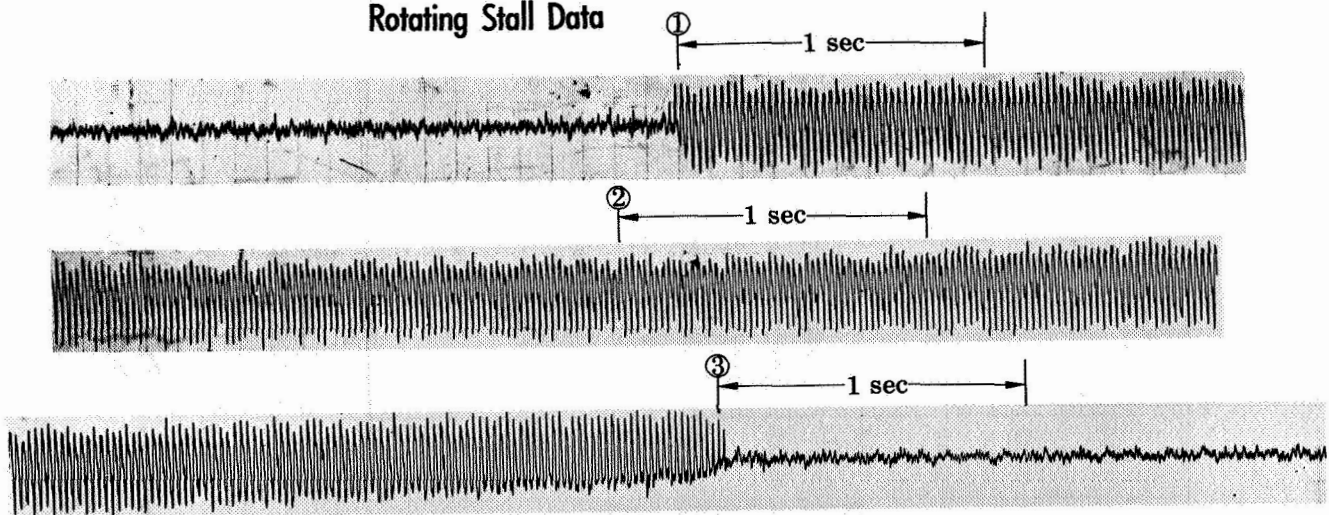


Figure 73. Transient and Rotating Stall FD 25734
(Kistler Trace) Data

APPENDIX A
DEFINITION OF SYMBOLS

A_A	Flowpath annular area, in. ²
a'_{0_1}	Inlet relative stagnation velocity of sound, ft/sec
c	Chord length, in.
C_P	Specific heat of air, 0.24 Btu/lb ^o F
d	Diameter
D	Diffusion factor
g	Gravitational constant, 32.2 ft/sec ²
i_m	Incidence angle, deg (based on equivalent circular arc for stators)
J	Mechanical equivalent of heat, 778 ft-lb/Btu
\bar{m}	Mass flow parameter, $\sqrt{\gamma g/R} \ M \left[1 + \frac{\gamma-1}{2} M^2 \right]^{1/2} p/p$
M	Absolute Mach number
P	Total pressure, psia
p	Static pressure, psia
q	Pressure equivalent of the velocity head, psia
R	Gas constant for air, 53.3 $\frac{\text{ft-lbs}}{\text{lb-}^{\circ}\text{R}}$
S	Blade spacing, in.
s	Blade span, in.
T	Total temperature, ^o R
T_s	Static temperature, ^o R
t	Blade maximum thickness, in.
U	Rotor speed, ft/sec
V	Absolute velocity, ft/sec
W	Actual flow rate, lb _m /sec
β	Absolute air angle, deg
γ	Ratio of specific heats

γ°	Blade-Chord angle, deg
δ	Ratio of total pressure to NASA standard sea level pressure of 2116 psf
δ°	Deviation angle, deg
η_{ad}	Adiabatic efficiency
η_p	Polytropic Efficiency
θ	Ratio of total temperature to NASA standard sea level temperature of 518.7°R
κ	Blade metal angle, deg (based on equivalent circular arc for stators)
ρ	Density, lb _m /ft ³
σ	Solidity, c/S
ϕ	Blade camber angle, deg (based on equivalent circular arc for stator B and on the 0.5% chord and 95% chord for the IGV)
ψ	Pressure rise coefficient
$\bar{\omega}$	Total pressure loss coefficient

Subscripts

0	Compressor inlet instrumentation station
1	Guide vane exit/rotor inlet instrumentation station
2	Rotor exit/stator inlet instrumentation station
2A	Stator exit instrumentation station
fs	Freestream value of guide vane or stator rake total pressure
id	Isentropic condition
le	Leading edge
te	Trailing edge
m	Midspan
s	Static condition
z	Axial component
θ	Tangential component

Superscripts

'	Relative rotor blade
-	Mass average value (used for overall and blade element performance)

Nomenclature used for blade element data tabulation (tables B-2, B-3, and B-4)

PCT SPAN	Percent Span
DIA	Diameter, inches
BETA	Absolute air angle, degrees
BETA (PR)	Relative air angle, degrees
V	Absolute velocity, ft/sec
VZ	Axial component of velocity, ft/sec
V-THETA	Tangential component of absolute velocity, ft/sec
V(PR)	Relative velocity, ft/sec
V-THETA PR	Tangential component of relative velocity, ft/sec
U	Wheel speed, ft/sec
M	Absolute Mach number
M(PR)	Relative Mach number
TURN	Air Turning, degrees
TURN (PR)	Relative air turning, degrees
$\bar{\omega}$	Loss coefficient
DFAC	Diffusion factor
EFFP	Polytropic efficiency
EFF	Adiabatic efficiency
INCIDM	Incidence, degrees
DEVM	Deviation, degrees

NOTE

The numbers following the nomenclature in Tables B-2, B-3, and B-4 refer to the leading and trailing edges of the respective blade rows, as indicated below:

3	IGV _{1e}
4	IGV _{te}
6	Rotor _{1e}
7	Rotor _{te}
9	Stator _{1e}
10	Stator _{te}

DEFINITION OF CALCULATED PERFORMANCE VARIABLES

Overall Performance Data

Inlet Mach Number:

$$\frac{W\sqrt{\theta}\sqrt{518.7}}{\delta P_o A_{A_o}} = g\sqrt{\gamma/g} M_o \left[1 + \frac{\gamma-1}{2} M_o^2 \right]^{1/2}$$

Pressure Ratio:

$$\text{Rotor: } \frac{\bar{P}_2}{\bar{P}_1} \quad \text{Rotor-Guide Vane: } \frac{\bar{P}_2}{14.696} \quad \text{Stage: } \frac{\bar{P}_{2A}}{14.696}$$

Corrected Flow

$$W\sqrt{\theta}/\delta$$

Corrected Specific Flow:

$$\frac{W\sqrt{\theta}}{\delta A_A}$$

Corrected Rotor Speed:

$$N/\sqrt{\theta}$$

Adiabatic Efficiency (overall):

$$\eta_{ad} = \frac{(\bar{P}_2/\bar{P}_1)^{\frac{\gamma-1}{\gamma}} - 1}{\bar{T}_{2A}/\bar{T}_1 - 1} \quad (\text{e.g., for rotor})$$

Pressure Rise Coefficient:

$$\psi = \frac{C_p J g T_o \left[(\bar{P}_{2A}/P_o)^{\frac{\gamma-1}{\gamma}} - 1 \right]}{U_{1e_m}^2|_{\text{rotor}}}$$

Flow Coefficient:

$$\phi = V_{z1e_m}/U_{1e_m}|_{\text{rotor}}$$

Blade Element Data

Incidence Angle:

$$i_m = \beta_{1e} - \kappa_{1e}$$

Diffusion Factor:

$$\text{Rotor: } D = 1 - \frac{v'_{te}}{v'_{1e}} + \left(\frac{d_{te} v_{\theta te} - d_{1e} v_{\theta 1e}}{(d_{1e} + d_{te}) \sigma v'_{1e}} \right)$$

$$\text{Stator: } D = 1 - \frac{v_{te}}{v_{1e}} + \left(\frac{d_{1e} v_{\theta 1e} - d_{te} v_{\theta te}}{(d_{1e} + d_{te}) \sigma v_{1e}} \right)$$

Deviation Angle:

$$\delta^o = \beta_{te} - \kappa_{te}$$

Loss Coefficient:

$$\text{IGV: } \bar{\omega} = \frac{P_{fs} - \bar{P}_1}{P_{fs} - P_o}$$

(P_o was found from linear interpolation of inner and outer wall static pressures at Station 0.)

$$\text{Rotor: } \bar{\omega} = \frac{\bar{P}'_{2id} - P'_2}{\bar{P}'_1 - P_1}$$

where:

$$P'_{2id} = P'_1 \left\{ 1 + \frac{\gamma-1}{2} \left(\frac{U_{te}}{a_{01}'} \right)^2 \left[1 - \left(\frac{d_{1e}}{d_{te}} \right)^2 \right] \right\}^{\frac{\gamma}{\gamma-1}}$$

$$P' \text{ is found from } p/P' = \left[1 + \frac{\gamma-1}{2} M'^2 \right]^{\frac{\gamma}{\gamma-1}}$$

and M' is calculated using trigonometric functions and the measurements of U , β , P , and p .

$$\text{Stator: } \bar{\omega} = \frac{P_{fs} - \bar{P}_{2a}}{P_{fs} - P_2}$$

Adiabatic Efficiency:

$$\text{Rotor: } \frac{\left[(P_{te}/P_{le})^{\frac{\gamma-1}{\gamma}} - 1 \right]}{T_{te}/T_o - 1}$$

Polytropic Efficiency:

$$\text{IGV: } \eta_p = \frac{\ln \left(T_{sle}/T_{ste} \right)}{\frac{\gamma-1}{\gamma} \ln P_{le}/P_{te}}$$

$$\text{Rotor: } \eta_p = \frac{\frac{\gamma-1}{\gamma} \ln (P_{te}/P_{le})}{\ln T_{te}/T_o}$$

$$\text{Stator: } \eta_p = \frac{\frac{\gamma-1}{\gamma} \ln (P_{te}/P_{le})}{\ln T_{ste}/T_{sle}}$$

APPENDIX B

TABULATED PERFORMANCE

The stage overall performance, rotor performance, and corrected weight flow for each test point are presented in table B-1.

Tables B-2 and B-3, respectively, present SLTO and cruise configuration blade element data for each test point. Table B-4 presents blade element data for the intermediate configuration.* Definition of the blade element velocity triangle and performance variables as tabulated in the computer printout is given in Appendix A.

*At 110% of design rotor speed for the SLTO configuration, three data points were omitted from the data presentation because of questionable plenum pressure data.

Table B-1. Overall Performance
SLTO Configuration

Point Number	Corrected Weight Flow $W/\theta/\delta$ lb/sec	Rotor			Guide Vane Rotor			Stage		
		\bar{P}_2/\bar{P}_1	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2/P_0	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2A/P_0	$\eta_{ad}\%$	$\eta_p\%$
50% Design Equivalent Rotor Speed										
147	92.57	1.0980	86.12	86.4	1.0978	85.95	86.2	1.0862	76.04	76.3
148	100.76	1.0943	89.47	89.7	1.0941	89.25	89.6	1.0876	83.32	83.5
149	106.95	1.0938	86.64	86.9	1.0937	86.53	86.7	1.0908	83.97	84.2
150	121.23	1.0877	106.52	-----	1.0874	106.09	-----	1.0858	104.20	-----
151	133.18	1.0781	96.45	96.6	1.0778	96.18	96.3	1.0768	95.00	95.1
146	149.44	1.0650	94.81	94.9	1.0643	93.76	93.8	1.0608	88.78	88.9
70% Design Equivalent Rotor Speed										
140	133.86	1.1959	85.74	86.1	1.1953	85.47	85.8	1.1783	78.44	78.9
141	140.30	1.1951	85.48	85.9	1.1945	85.23	85.6	1.1831	80.55	81.0
142	151.31	1.1918	90.40	90.7	1.1908	89.96	90.3	1.1844	87.11	87.4
143	171.33	1.1893	98.07	98.2	1.1877	97.29	97.4	1.1809	93.99	94.1
144	187.25	1.1657	96.66	96.8	1.1641	95.39	95.5	1.1607	93.53	93.7
139	207.02	1.1329	99.83	99.9	1.1311	98.55	98.6	1.1217	91.79	91.9

Table B-1. Overall Performance (Continued)
SLTO Configuration

Point Number	Corrected Weight Flow $W\sqrt{\theta}/\delta$ lb/sec	Rotor			Guide Vane Rotor			Stage		
		\bar{P}_2/\bar{P}_1	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2/P_0	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_{2A}/P_0	$\eta_{ad}\%$	$\eta_p\%$
		80% Design Equivalent Rotor Speed								
137	157.40	1.2581	82.32	82.9	1.2569	81.97	82.6	1.2379	76.34	77.1
133	164.03	1.2647	92.59	92.9	1.2652	92.75	93.0	1.2495	87.67	88.1
134	176.83	1.2534	89.95	90.3	1.2507	89.05	89.5	1.2524	89.63	90.0
135	198.66	1.2521	89.61	90.0	1.2516	89.46	89.9	1.2527	89.80	90.1
136	222.84	1.2107	96.58	96.7	1.2098	96.12	96.3	1.2073	95.05	95.2
132	238.36	1.1800	102.18	----	1.1748	99.35	99.4	1.1599	91.33	91.5
100% Design Equivalent Rotor Speed										
154	197.35	1.4423	82.90	83.8	1.4440	83.18	84.1	1.4194	79.09	80.1
155	209.34	1.4416	87.23	87.9	1.4437	87.60	88.2	1.4301	85.24	86.0
156	224.10	1.4252	89.34	89.9	1.4237	89.07	89.7	1.4312	90.45	90.9
157	242.44	1.4105	91.40	91.6	1.4078	90.85	91.3	1.4149	92.27	92.6
158	260.52	1.3686	92.92	93.2	1.3646	92.02	92.4	1.3629	91.63	92.0
153	280.52	1.2646	91.04	91.4	1.2601	89.61	90.0	1.2400	83.19	83.7

Table B-1. Overall Performance (Continued)
SLTO Configuration

Point Number	Corrected Weight Flow $W\sqrt{\theta}/\delta$ lb/sec	Rotor		Guide Vane Rotor			Stage			
		\bar{P}_2/\bar{P}_1	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2/P_0	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_{2A}/P_0	$\eta_{ad}\%$	$\eta_p\%$
		110% Design Equivalent Rotor Speed								
161	223.62	1.5422	77.85	79.2	1.5385	77.72	79.1	1.5237	75.88	77.4
165	279.67	1.4131	90.11	90.6	1.4106	89.74	90.2	1.3896	85.66	86.3
160	290.69	1.2956	87.62	88.1	1.2907	86.43	86.9	1.2737	81.77	82.4

Table B-1. Overall Performance (Continued)
Cruise Configuration

Point Number	Corrected Weight Flow $w\sqrt{\theta}/\delta$ lb/sec	Rotor			Guide Vane Rotor			Stage		
		\bar{P}_2/\bar{P}_1	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2/P_0	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_{2A}/P_0	$\eta_{ad}\%$	$\eta_p\%$
50% Design Equivalent Rotor Speed										
182	85.57	1.0912	83.59	84.0	1.0908	83.30	83.7	1.0875	80.32	80.6
183	91.01	1.0890	91.11	91.3	1.0884	90.52	90.7	1.0837	85.85	86.0
184	99.51	1.0835	94.06	94.1	1.0830	93.53	93.7	1.0786	88.71	88.8
185	106.21	1.0763	95.23	95.3	1.0754	94.12	94.2	1.0710	88.75	88.9
186	114.70	1.0647	95.88	96.0	1.0639	96.68	96.8	1.0598	88.83	88.9
181	121.02	1.0586	96.13	96.2	1.0573	94.06	94.1	1.0519	85.32	85.4
70% Design Equivalent Rotor Speed										
168	118.88	1.1800	85.03	85.4	1.1792	84.70	85.1	1.1739	82.33	82.7
169	125.53	1.1796	89.35	89.6	1.1781	88.64	88.9	1.1704	85.03	85.4
170	131.27	1.1744	91.93	92.2	1.1733	91.41	91.7	1.1648	87.14	87.4
171	140.07	1.1600	93.68	93.9	1.1604	93.93	94.0	1.1536	90.15	90.4
173	152.68	1.1441	101.47	----	1.1433	100.93	----	1.1365	96.35	96.4
167	166.46	1.1187	100.01	100.0	1.1168	98.49	98.6	1.1022	86.56	86.7

Table B-1. Overall Performance (Continued)
Cruise Configuration

Point Number	Corrected Weight Flow $w\sqrt{\theta}/\delta$ lb/sec	Rotor		Guide Vane Rotor			Stage			
		\bar{P}_2/\bar{P}_1	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2/P_0	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_{2A}/P_0		
		80% Design Equivalent Rotor Speed								
175	138.10	1.2421	86.29	86.7	1.2406	85.82	86.3	1.2341	83.65	84.1
177	147.54	1.2312	88.22	88.6	1.2298	87.72	88.1	1.2217	84.83	85.3
178	157.91	1.2170	88.66	89.0	1.2147	87.80	88.2	1.2140	87.53	87.9
179	166.19	1.2059	94.71	94.8	1.2022	93.10	93.3	1.1955	90.19	90.4
180	176.48	1.1815	95.82	96.0	1.1790	94.55	94.7	1.1701	90.09	90.3
174	189.09	1.1507	97.86	98.0	1.1480	96.15	96.3	1.1334	87.08	87.3
100% Design Equivalent Rotor Speed										
190	175.83	1.3922	85.12	85.8	1.3908	84.85	85.5	1.3732	81.43	82.3
191	184.90	1.3889	86.26	86.9	1.3865	85.79	86.4	1.3856	85.60	86.3
192	193.09	1.3822	89.67	90.2	1.3795	89.10	89.6	1.3758	88.32	88.8
193	205.90	1.3582	93.61	93.9	1.3551	92.87	93.2	1.3436	90.15	90.6
194	217.46	1.3126	93.31	93.6	1.3099	92.58	92.9	1.2991	89.62	90.0
189	232.54	1.2526	95.94	96.1	1.2460	93.60	93.8	1.2216	84.96	85.4

Table B-1 Overall Performance (Continued)
Intermediate Configuration

Point Number	Corrected Weight Flow $W\sqrt{\theta}/\delta$ lb/sec	Rotor			Guide Vane Rotor			Stage		
		\bar{P}_2/\bar{P}_1	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2/P_0	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_{2A}/P_0	$\eta_{ad}\%$	$\eta_p\%$
50% Design Equivalent Rotor Speed										
215	89.71	1.0937	86.28	86.5	1.0933	85.95	86.2	1.0872	80.44	80.7
216	97.29	1.0888	81.87	82.2	1.0886	81.66	82.0	1.0859	79.28	79.5
217	113.09	1.0810	90.35	90.6	1.0801	89.40	89.6	1.0798	89.06	89.2
218	121.61	1.0719	99.21	99.2	1.0710	97.99	98.1	1.0684	94.45	94.5
214	138.41	1.0572	92.93	93.1	1.0563	91.52	91.7	1.0545	88.61	88.7
70% Design Equivalent Rotor Speed										
209	129.82	1.1899	82.79	83.3	1.1892	82.49	83.0	1.1777	77.76	78.3
210	142.02	1.1820	88.24	88.6	1.1809	87.73	88.0	1.1764	85.65	86.0
211	163.55	1.1610	90.15	90.4	1.1509	89.05	89.3	1.1572	88.09	88.3
212	180.36	1.1367	92.59	92.8	1.1397	94.53	94.6	1.1384	93.73	93.8
208	193.55	1.1138	97.61	97.7	1.1128	96.78	96.9	1.1097	94.23	94.3

Table B-1. Overall Performance (Continued)
Intermediate Configuration

Point Number	Corrected Weight Flow $W\sqrt{\theta}/\delta$ lb/sec	Rotor			Guide Vane Rotor			Stage		
		\bar{P}_2/\bar{P}_1	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_2/P_0	$\eta_{ad}\%$	$\eta_p\%$	\bar{P}_{2A}/P_0	$\eta_{ad}\%$	$\eta_p\%$
		80% Design Equivalent Rotor Speed								
203	146.41	1.2555	85.27	85.8	1.2542	84.87	85.4	1.2364	79.36	80.0
204	150.21	1.2473	83.35	84.0	1.2462	83.01	83.6	1.2362	79.86	80.5
205	171.33	1.2394	88.92	89.3	1.2375	88.28	88.6	1.2328	86.65	87.0
206	188.93	1.2158	92.47	92.7	1.2132	91.45	91.8	1.2085	89.55	89.8
202	217.93	1.1558	97.29	97.4	1.1520	95.03	95.1	1.1410	88.50	88.7
100% Design Equivalent Rotor Speed										
197	190.95	1.4165	83.83	84.7	1.4165	83.83	84.7	1.4105	82.75	83.6
198	199.54	1.4242	83.59	84.4	1.4239	83.53	84.4	1.4158	82.12	83.0
199	216.15	1.4030	86.30	86.9	1.4010	85.93	86.6	1.4085	87.36	88.0
200	238.11	1.3645	91.75	92.1	1.3615	91.07	91.4	1.3565	89.92	90.3
196	262.57	1.2362	94.44	94.6	1.2325	93.06	93.3	1.2230	89.52	89.8

Table B-2. Blade Element Data SLTO Configuration

50 PERCENT SPEED PT. 147				CONFIGURATION		N/√e		ω√e/δ	
				SLTO	3010.	92.57			
INLET GUIDE VANE									
ROTOR									
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	0.240	0.874	-0.462
BETA 4	0.284	0.984	-0.426	-1.177	-3.459	BETA 7	48.502	54.707	51.618
V 3	138.93	146.54	151.75	155.09	155.85	BETA(PR) 6	63.368	65.938	68.607
V 4	139.12	154.69	160.03	165.16	163.45	BETA(PR) 7	30.876	47.209	54.368
VZ 3	138.93	146.32	151.74	155.08	155.84	V 6	153.73	161.86	165.85
VZ 4	139.11	153.83	160.02	165.12	163.16	V 7	291.19	266.26	260.66
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	153.69	161.67	165.81
V-THETA 4	0.69	2.66	-1.19	-3.39	-9.86	VZ 7	191.80	153.66	161.75
M 3	0.1246	0.1315	0.1362	0.1392	0.1399	V-THETA 6	0.64	2.47	-1.34
M 4	0.1248	0.1388	0.1436	0.1483	0.1467	V-THETA 7	218.10	217.32	204.33
TURN	-0.28	-0.98	0.43	1.18	3.46	V(PR) 6	342.9	396.6	454.6
(REFER TO 70% & 100% SPEED DATA)									
ω						V(PR) 7	224.9	226.5	277.8
DFAC	0.001	-0.046	-0.059	-0.076	-0.080	VTHETA PR6	-306.5	-362.1	-423.3
EFFP	0.2800	1.1960	0.8399	1.0855	0.7933	VTHETA PR7	-115.4	-166.2	-225.8
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	307.13	364.52	421.92
DEVM	5.116	2.716	2.326	1.477	1.859	U 7	333.46	383.50	430.13
STATOR B									
PCT SPAN	90	70	50	30	10	DFAC	0.1318	0.2543	0.3237
DIA 9	26.021	29.683	33.345	37.007	40.669	EFFP	.5765	.6423	.5792
BETA 9	47.597	53.512	50.039	44.301	43.054	EFF	1.0564	0.9054	0.8970
BETA 10	0.288	1.200	0.030	1.342	2.281	INCIDM	1.0570	0.9043	0.8957
V 9	288.30	265.71	262.73	285.37	285.89	DEVM	14.768	13.438	12.507
V 10	166.33	184.79	187.35	219.23	249.19		12.076	14.709	12.268
VZ 9	191.48	156.71	168.04	204.06	208.90				
VZ 10	164.62	183.02	186.51	218.93	249.00				
V-THETA 9	212.88	213.62	201.38	199.31	195.18				
V-THETA 10	0.84	3.87	0.10	5.14	9.92				
M 9	0.2570	0.2362	0.2333	0.2533	0.2205				
M 10	0.1477	0.1639	0.1660	0.1941	0.2205				
TURN	47.309	52.312	50.009	42.958	40.773				
ω	0.2531	0.1709	0.1983	0.2049	0.1413				
DFAC	0.7034	0.6286	0.6240	0.5503	0.4439				
EFFP	0.4012	0.5218	0.2770	0.1000	-0.0443				
INCIDM	6.697	16.412	15.339	10.301	7.054				
DEVM	8.988	8.700	7.530	9.342	11.161				

Table B-2. Blade Element Data SLTO Configuration (Continued)

CONFIGURATION				$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/g$	
50 PERCENT SPEED PT. 148				3000.		100.7	
INLET GUIDE VANE				ROTOR			
PCT SPAN				PCT SPAN			
DIA 3				DIA 6			
BETA 3				BETA 6			
BETA 4				BETA 7			
V 3				BETA(PR) 6			
V 4				BETA(PR) 7			
VZ 3				V 6			
VZ 4				V 7			
V-THETA 3				VZ 6			
V-THETA 4				VZ 7			
M 3				V-THETA 6			
M 4				V-THETA 7			
TURN				V(PR) 6			
ω				V(PR) 7			
DFAC				VTHETA PR6			
EFFP				VTHETA PR7			
INCIDM				U 6			
DEVM				U 7			
				M 6			
				M 7			
				M(PR) 6			
				M(PR) 7			
				TURN(PR)			
				ω			
				DFAC			
				EFFP			
				EFF			
				INCIDM			
				DEVM			

Table B-2. Blade Element Data SLTO Configuration (Continued)

INLET GUIDE VANE				ROTOR			
50 PERCENT SPEED		PT. 150		CONFIGURATION		N/√θ	
		SLTO				121.2	

Table B-2. Blade Element Data SLTO Configuration (Continued)

50 PERCENT SPEED PT. 146				CONFIGURATION		N/√σ		ω√σδ					
				SLTO		3040.4		149.4					
INLET GUIDE VANE													
ROTOR													
PCT SPAN	90	70	50	30	10		PCT SPAN	90	70	50	30	10	
DIA 3	21.150	26.050	30.950	35.850	40.750		DIA 6	23.385	27.755	32.125	36.495	40.865	
BETA 3	0.000	0.000	0.000	0.000	0.000		BETA 6	0.766	0.187	-0.842	-1.116	-3.870	
BETA 4	0.880	0.320	-0.864	-1.107	-3.890		BETA 7	30.180	26.316	23.190	20.672	17.097	
V 3	227.95	241.46	249.67	253.41	253.36		BETA(PR) 6	49.819	53.086	56.860	59.912	64.249	
V 4	236.38	255.73	263.69	269.01	264.48		BETA(PR) 7	26.199	37.425	44.726	50.253	54.903	
VZ 3	227.92	241.16	249.65	253.39	253.35		V 6	259.12	276.10	280.99	283.75	270.90	
VZ 4	236.35	254.37	263.64	268.95	263.87		V 7	362.89	342.99	333.08	327.64	324.29	
V-THETA 3	0.00	0.00	0.00	0.00	0.00		VZ 6	259.03	275.89	280.90	283.68	270.28	
V-THETA 4	3.63	1.43	-3.98	-5.20	-17.94		VZ 7	312.35	307.15	306.02	306.50	309.96	
M 3	0.2050	0.2173	0.2248	0.2282	0.2281		V-THETA 6	3.47	0.90	-4.13	-5.53	-18.28	
M 4	0.2127	0.2303	0.2375	0.2424	0.2382		V-THETA 7	182.43	152.05	131.16	115.66	95.34	
TURN	-0.88	-0.32	0.86	1.11	3.89		V(PR) 6	401.5	459.5	513.9	565.9	522.1	
(REFER TO 70% & 100% SPEED DATA)													
ω	-0.029	-0.056	-0.064	-0.072	-0.079		V(PR) 7	349.6	387.1	430.9	479.4	539.1	
DFAC	0.8320	1.0064	0.8527	1.0275	0.7459		VTHETA PR6	-306.7	-367.3	-430.3	-489.6	-560.3	
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200		VTHETA PR7	-154.4	-235.3	-303.2	-368.6	-441.1	
INCIDM							U 6	310.19	368.16	426.12	484.09	542.05	
DEVM	4.520	3.380	2.764	1.407	2.290		U 7	336.79	387.32	434.41	484.29	536.42	
STATOR B													
PCT SPAN	90	70	50	30	10		PCT SPAN	90	70	50	30	10	
DIA 9	26.021	25.683	33.345	37.007	40.669		DFAC	.2926	.2875	.2726	.2486	.2188	
BETA 9	28.330	24.832	22.059	19.654	17.069		EFFP	0.9984	0.9826	0.9791	0.9651	0.9606	
BETA 10	1.614	0.717	0.147	-0.655	1.044		EFF	0.9984	0.9825	0.9790	0.9647	0.9602	
V 9	370.11	353.28	341.80	337.11	326.57		INCIDM	1.219	0.586	0.760	0.312	1.049	
V 10	367.32	349.12	340.57	336.21	337.98		DEVM	7.399	4.925	2.626	1.653	2.003	
VZ 9	320.57	318.63	316.31	317.46	312.12								
VZ 10	365.17	346.90	340.09	336.18	337.83								
V-THETA 9	175.64	148.36	128.37	113.38	95.86								
V-THETA 10	10.34	4.37	0.87	-3.84	6.16								
M 9	0.3320	0.3167	0.3062	0.3019	0.2922								
M 10	0.3295	0.3129	0.3050	0.3011	0.3026								
TURN	26.717	24.115	21.911	20.309	16.025								
ω	0.0465	0.0301	0.0245	0.0264	0.0436								
DFAC	0.1773	0.1791	0.1677	0.1654	0.0988								
EFFP	-3.4494	-1.9349	-8.3529	-15.6625	1.4936								
INCIDM	-12.570	-12.268	-12.641	-14.306	-18.931								
DEVM	10.314	8.217	7.647	7.345	9.944								

Table B-2. Blade Element Data SLTO Configuration (Continued)

CONFIGURATION				$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/\delta$						
70 PERCENT SPEED	PT. 140	SLTO	4255.	133.8								
INLET GUIDE VANE												
ROTOR												
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10	
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865	
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	0.965	0.571	-0.480	-0.878	-3.052	
BETA 4	1.113	0.772	-0.424	-0.839	-2.870	BETA 7	47.772	50.060	50.163	44.975	47.970	
V 3	205.49	216.49	222.70	225.74	225.24	BETA(PR) 6	62.364	64.923	67.707	69.335	72.246	
V 4	202.69	224.35	234.07	241.83	239.48	BETA(PR) 7	33.673	46.543	54.452	55.874	58.853	
VZ 3	205.49	216.09	222.69	225.73	225.23	V 6	225.46	240.25	245.48	257.11	247.49	
VZ 4	202.61	223.41	233.97	241.77	239.18	V 7	396.71	375.37	365.32	387.20	396.76	
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	225.34	240.01	245.37	257.05	247.14	
V-THETA 4	3.94	3.02	-1.73	-3.54	-11.99	VZ 7	264.90	240.62	233.84	273.84	290.31	
M 3	0.1847	0.1947	0.2003	0.2030	0.2026	V-THETA 6	3.80	2.39	-2.06	-3.94	-13.18	
M 4	0.1822	0.2018	0.2106	0.2176	0.2155	V-THETA 7	293.76	287.80	280.52	273.67	270.44	
TURN	-1.11	-0.77	0.42	0.84	2.87	V(PR) 6	485.8	566.4	646.9	728.4	810.5	
W	0.0201	0.0257	0.0257	-0.079	0.0240	V(PR) 7	320.4	350.4	402.5	488.2	561.3	
DFAC	0.023	-0.029	-0.055	-0.079	-0.090	VTHETA PR6	-430.4	-512.9	-598.5	-681.5	-771.9	
EFFP	6.8871	1.0923	0.8388	1.1130	0.8089	VTHETA PR7	-177.6	-254.3	-327.5	-404.2	-480.4	
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	434.16	515.30	596.43	677.56	758.70	
DEVM	4.287	2.928	2.324	1.139	1.270	U 7	471.39	542.13	608.03	677.84	750.81	
STATOR B								M 6	0.2028	0.2162	0.2315	0.2228
PCT SPAN	90	70	50	30	10	M 7	0.3522	0.3318	0.3219	0.3407	0.3479	
DIA 9	26.021	29.683	33.345	37.007	40.669	M(PR) 6	0.4370	0.5097	0.5823	0.6559	0.7294	
BETA 9	46.618	48.963	48.717	43.577	42.177	M(PR) 7	0.2844	0.3097	0.3547	0.4296	0.4922	
BETA 10	2.473	1.747	-0.207	1.092	4.663	TURN(PR)	28.680	18.360	13.247	13.458	13.393	
V 9	394.58	376.19	369.48	394.81	401.99	W	0.1346	0.2803	0.3620	0.2696	0.3202	
V 10	238.07	264.76	260.57	300.34	346.98	DFAC	.5590	.5794	.5617	.5001	.4712	
VZ 9	267.33	245.69	243.35	286.00	297.82	EFFP	1.0937	0.9790	0.9121	0.9195	0.8634	
VZ 10	235.12	263.02	260.06	300.25	345.74	EFF	1.0959	0.9785	0.9100	0.9173	0.8595	
V-THETA 9	286.78	283.75	277.65	272.15	269.91	INCIDM	13.764	12.423	11.607	9.735	9.046	
V-THETA 10	10.27	8.07	-0.94	5.72	28.21	DEVM	14.873	14.043	12.352	7.274	5.953	
M 9	0.3502	0.3325	0.3255	0.3475	0.3525							
M 10	0.2097	0.2329	0.2285	0.2631	0.3035							
TURN	44.146	47.216	48.924	42.484	37.515							
W	0.2298	0.1710	0.1975	0.2128	0.1285							
DFAC	0.6633	0.5970	0.6266	0.5552	0.4297							
EFFP	0.4058	0.4741	0.3223	0.2466	0.2754							
INCIDM	5.718	11.863	14.017	9.577	6.177							
DEVM	11.173	9.247	7.293	9.092	13.563							

Table B-2. Blade Element Data SLTO Configuration (Continued)

70 PERCENT SPEED										PT. 141		PT. 140.		PT. 140.		PT. 140.		PT. 140.	
CONFIGURATION										SLTO		SLTO		SLTO		SLTO		SLTO	
INLET GUIDE VANE										SLTO		SLTO		SLTO		SLTO		SLTO	
INLET GUIDE VANE										SLTO		SLTO		SLTO		SLTO		SLTO	
INLET GUIDE VANE										SLTO		SLTO		SLTO		SLTO		SLTO	
INLET GUIDE VANE										SLTO		SLTO		SLTO		SLTO		SLTO	
INLET GUIDE VANE										SLTO		SLTO		SLTO		SLTO		SLTO	
INLET GUIDE VANE										SLTO		SLTO		SLTO		SLTO		SLTO	
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Table B-2. Blade Element Data SLTO Configuration (Continued)

CONFIGURATION			N/√σ		ω√σ/s	
70 PERCENT SPEED PT. 143			SLIO		171.3	
INLET GUIDE VANE			ROTOR			
PCT SPAN	90	70	50	30	10	10
DIA 3	21.150	26.050	30.950	35.850	40.750	36.495
BETA 3	0.000	0.000	0.000	0.000	0.000	-1.450
BETA 4	0.987	0.444	-0.424	-1.494	-4.780	34.272
V 3	260.21	276.65	287.95	294.28	295.33	63.946
V 4	269.90	293.28	304.54	314.98	309.70	52.731
VZ 3	260.19	276.31	287.93	294.26	295.32	335.54
VZ 4	269.86	291.01	304.53	314.87	308.63	411.03
V-THEIA 3	0.00	0.00	0.00	0.00	0.00	323.94
V-THEIA 4	4.65	2.27	-2.25	-8.21	-25.81	323.53
M 3	0.2344	0.2493	0.2597	0.2654	0.2664	339.62
M 4	0.2432	0.2645	0.2748	0.2844	0.2796	231.46
TURN	-0.99	-0.44	0.42	1.49	4.78	763.7
ω	0.0199		0.0216		0.032	560.9
DFAC	-0.028	-0.056	-0.062	-0.084	-0.092	-686.1
EFFP	0.7169	0.9698	0.8290	1.0426	0.7068	-446.4
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	677.56
DEVM	4.413	3.256	2.324	1.794	3.180	677.84
STATOR B.						
PCT SPAN	90	70	50	30	10	10
DIA 9	26.021	29.683	33.345	37.007	40.669	40.865
BETA 9	43.432	37.720	35.157	32.729	29.005	-4.753
BETA 10	4.192	1.087	0.871	-0.258	1.669	30.345
V 9	423.95	404.56	412.28	423.91	423.95	67.596
V 10	325.14	343.27	359.27	384.88	413.62	56.905
VZ 9	301.72	316.86	335.84	356.42	370.76	325.00
VZ 10	321.82	339.55	357.83	384.63	413.43	410.43
V-THEIA 9	291.46	247.51	237.39	229.19	205.57	323.85
V-THEIA 10	23.77	6.51	5.46	-1.73	12.05	428.12
M 9	0.3769	0.3593	0.3656	0.3753	0.3746	290.20
M 10	0.2874	0.3038	0.3176	0.3400	0.3652	311.99
TURN	39.240	36.633	34.285	32.986	27.336	399.82
ω	0.1676	0.0189	0.0567	0.0303	0.0253	311.63
DFAC	0.4728	0.3960	0.3760	0.3471	0.2466	253.74
EFFP	0.4739	0.5314	0.3883	0.2866	0.0131	600.7
INCIDM	2.532	.620	.457	-1.271	-6.995	422.7
DEVM	12.892	8.587	8.371	7.742	10.569	513.4

Table B-2. Blade Element Data SLTO Configuration (Continued)

CONFIGURATION				$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/\delta$					
70 PERCENT SPEED		PT. 144	SLTO	4255.	187.2						
INLET GUIDE VANE											
ROTOR											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	0.783	0.186	-0.786	-1.669	-4.373
BETA 4	0.891	0.275	-0.828	-1.693	-4.454	BETA 7	38.029	33.782	30.616	28.315	24.638
V 3	288.13	305.89	317.51	323.43	323.91	BETA(PR) 6	52.825	55.895	59.185	61.970	65.908
V 4	300.27	324.89	336.27	344.71	339.68	BETA(PR) 7	27.627	40.073	46.991	51.610	55.126
VZ 3	288.10	305.52	317.48	323.41	323.89	V 6	325.97	348.49	358.79	366.58	352.28
VZ 4	300.23	322.75	336.23	344.56	338.65	V 7	458.41	431.88	424.65	427.54	436.23
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	325.87	348.18	358.70	366.41	351.26
V-THETA 4	4.67	1.56	-4.86	-10.18	-26.38	VZ 7	359.44	358.63	365.28	376.33	396.51
M 3	0.2598	0.2761	0.2867	0.2922	0.2926	V-THETA 6	4.46	1.13	-4.92	-10.67	-26.86
M 4	0.2709	0.2935	0.3040	0.3118	0.3071	V-THETA 7	282.40	240.14	216.26	202.79	181.86
TURN	-0.89	-0.27	0.83	1.69	4.45	V(PR) 6	539.3	621.1	700.2	779.7	860.5
\vec{w}	0.0213		0.0191		0.027	V(PR) 7	407.6	469.1	535.8	606.1	693.5
DFAC	-0.034	-0.060	-0.067	-0.082	-0.089	VTHETA PR6	-429.7	-514.2	-501.4	-688.2	-785.6
EFFP	0.8244	0.9939	0.8608	1.0135	0.7361	VTHETA PR7	-189.0	-302.0	-391.8	-475.0	-569.0
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	434.16	515.30	596.43	677.56	758.70
DEVM	4.509	3.425	2.728	1.993	2.854	U 7	471.39	542.13	608.03	677.84	750.81
						M 6	0.2945	0.3152	0.3248	0.3320	0.3187
						M 7	0.4091	0.3847	0.3777	0.3798	0.3869
						M(PR) 6	0.4873	0.5619	0.6338	0.7001	0.7786
						M(PR) 7	0.3637	0.4178	0.4766	0.5383	0.6151
						TURN(PR)	25.192	15.799	12.190	10.359	10.782
						\vec{w}	0.0862	0.0772	0.0823	0.0912	0.1120
PCT SPAN	90	70	50	30	10	DFAC	.5022	.3965	.3678	.3452	.3078
DIA 9	26.021	29.683	33.345	37.007	40.669	EFFP	1.0078	1.0195	1.0188	0.9928	0.9941
BETA 9	36.247	32.121	29.374	27.047	24.385	EFF	1.0079	1.0199	1.0192	0.9926	0.9940
BETA 10	2.280	0.458	0.549	-0.576	1.032	INCIDM	4.225	3.395	3.085	2.370	2.708
V 9	460.43	440.56	435.90	440.42	443.00	DEVM	8.827	7.573	4.891	3.010	2.226
V 10	410.63	398.19	404.44	414.18	436.08						
VZ 9	364.60	370.09	378.90	392.18	403.43						
VZ 10	407.83	394.79	403.40	414.07	435.93						
V-THETA 9	272.24	234.24	213.81	200.27	182.90						
V-THETA 10	16.33	3.18	3.88	-4.16	7.85						
M 9	0.4110	0.3927	0.3880	0.3915	0.3931						
M 10	0.3653	0.3539	0.3592	0.3675	0.3868						
TURN	33.967	31.663	28.825	27.623	23.353						
\vec{w}	0.1514	0.0405	0.0455	0.205	0.0201						
DFAC	0.3194	0.3115	0.2840	0.2769	0.2080						
EFFP	0.6005	0.4975	0.4429	0.2585	-0.3022						
INCIDM	-4.653	-4.979	-5.326	-6.953	-11.615						
DEVM	10.980	7.958	8.049	7.424	9.932						

Table B-2. Blade Element Data SLTO Configuration (Continued)

70 PERCENT SPEED				PT.139		CONFIGURATION		N/√δ		ω√δ/δ		207.	
INLET GUIDE VANE												ROTOR	
POT SPAN		90	70	50	30	10	POT SPAN		90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865		
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	0.832	0.204	-0.845	-1.774	-3.971		
BETA 4	0.960	0.344	-0.866	-1.839	-3.970	BETA 7	29.596	25.612	22.628	19.678	16.128		
V 3	323.81	343.06	354.75	360.03	359.84	BETA(PR) 6	49.231	52.484	56.223	59.482	63.792		
V 4	336.82	363.85	376.09	384.09	379.10	BETA(PR) 7	27.520	38.413	45.491	51.072	54.942		
VZ 3	323.76	342.64	354.71	360.01	359.81	V 6	370.29	395.30	403.51	407.54	388.06		
VZ 4	336.76	361.93	376.02	383.88	378.19	V 7	498.40	473.07	459.88	451.67	456.46		
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	370.16	395.01	403.38	407.32	387.13		
V-THETA 4	5.64	2.19	-5.68	-12.32	-26.25	VZ 7	431.47	426.17	424.26	425.23	438.49		
M 3	0.2925	0.3102	0.3210	0.3259	0.3257	V-THETA 6	5.38	1.41	-5.95	-12.62	-26.87		
M 4	0.3045	0.3294	0.3408	0.3482	0.3436	V-THETA 7	246.15	204.50	176.94	152.09	126.80		
TURN	-0.96	-0.34	0.87	1.84	3.97	V(PR) 6	566.9	648.8	725.6	802.1	876.5		
ω	0.0219		0.0188		0.0250	V(PR) 7	488.7	544.4	605.5	676.9	763.4		
DFAC	-0.031	-0.057	-0.068	-0.084	-0.090	VTHETA PR6	-429.3	-514.5	-603.1	-691.0	-786.5		
EFFP	0.7928	0.9298	0.8270	0.9917	0.7970	VTHETA PR7	-225.8	-338.3	-431.8	-526.5	-624.9		
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	434.67	515.90	597.13	678.36	759.59		
DEVM	4.440	3.356	2.766	2.139	2.370	U 7	471.94	542.76	608.75	678.64	751.69		
STATOR B													
POT SPAN		90	70	50	30	10	POT SPAN		90	70	50	30	10
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC	0.0589	0.0530	0.0726	0.0876	0.1113		
BETA 9	27.739	24.050	21.461	18.634	15.594	EFFP	.2939	.2842	.2719	.2476	.2114		
BETA 10	1.597	0.556	0.212	-0.325	1.833	EFF	1.0542	1.0482	1.0329	1.0023	1.0068		
V 9	509.07	489.18	473.48	467.78	462.05	INCIDM	1.0552	1.0491	1.0335	1.0070	.592		
V 10	509.30	486.00	474.36	468.22	469.56	DEVM	.631	-0.016	.123	-1.118	2.042		
VZ 9	443.27	443.96	440.03	443.25	444.92								
VZ 10	506.28	482.93	473.72	468.21	469.15								
V-THETA 9	236.94	199.35	173.23	149.47	124.21								
V-THETA 10	14.19	4.71	1.76	-2.66	15.02								
M 9	0.4574	0.4389	0.4242	0.4188	0.4132								
M 10	0.4576	0.4359	0.4250	0.4193	0.4202								
TURN	26.142	23.494	21.248	18.960	13.761								
ω	0.0418	0.0279	0.0213	0.0178	0.0315								
DFAC	0.1659	0.1698	0.1575	0.1513	0.0989								
EFFP	70.4983	-6.0340	24.2595	36.6491	2.6005								
INCIDM	-13.161	-13.050	-13.239	-15.366	-20.406								
DEVM	10.297	8.056	7.712	7.675	10.733								

Table B-2. Blade Element Data SLTO Configuration (Continued)

INLET GUIDE VANE						ROTOR							
80 PERCENT SPEED PT. 137			CONFIGURATION SLTO			$N/\sqrt{\theta}$	$w\sqrt{\theta}/s$						
						4815.	157.3						
PCT SPAN	90	70	50	30	10			PCT SPAN	90	70	50	30	
DIA 3	21.150	26.050	30.950	35.850	40.750			DIA 6	23.385	27.755	32.125	36.495	17
BETA 3	0.000	0.000	0.000	0.000	0.000			BETA 6	2.104	1.549	0.424	-0.470	40.845
BETA 4	2.414	1.800	0.611	-0.446	-1.657			BETA 7	49.203	51.105	50.062	46.436	-1.867
V 3	243.97	256.99	264.24	267.32	266.65			BETA(PR) 6	60.956	63.663	66.453	68.431	43.310
V 4	241.01	265.16	277.42	286.32	285.70			BETA(PR) 7	28.727	42.596	51.196	54.187	71.282
VZ 3	243.96	256.51	264.23	267.31	266.64			V 6	267.67	285.22	293.32	304.15	49.430
VZ 4	240.74	263.94	277.25	286.23	285.57			V 7	478.35	452.55	439.66	456.66	294.15
V-THETA 3	0.00	0.00	0.00	0.00	0.00			VZ 6	267.37	284.84	293.18	304.09	294.15
V-THETA 4	8.33	8.33	2.96	-2.23	-8.26			VZ 7	310.47	283.71	281.98	314.61	341.32
M 3	0.2196	0.2314	0.2380	0.2314	0.2402			V-THETA 6	9.87	7.71	2.17	-2.49	-9.50
M 4	0.2169	0.2389	0.2500	0.2582	0.2576			V-THETA 7	362.12	352.21	337.09	330.90	321.87
TURN	-2.41	-1.80	-0.61	0.45	1.66			V(PR) 6	550.8	642.2	733.9	827.2	628.5
ω								V(PR) 7	356.4	386.0	450.4	537.8	628.5
(REFER TO 70% & 100% SPEED DATA)								VTHETA PR6	-481.5	-575.4	-672.8	-769.2	-868.1
DFAC	0.033	-0.015	-0.044	-0.075	-0.087			U 6	-171.3	-261.3	-351.0	-436.2	-527.7
EFFP	-3.5686	0.8482	0.7890	0.9801	0.8348			U 7	491.30	583.12	674.93	766.74	858.55
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200			M 6	533.43	613.47	688.06	767.05	849.62
DEVM	2.986	1.900	1.289	.746	.057			M 7	0.2412	0.2572	0.2646	0.2745	0.2655
								M(PR) 6	0.4237	0.3985	0.3856	0.3995	0.4091
								M(PR) 7	0.4962	0.5790	0.6620	0.7465	0.8268
								TURN(PR)	0.3157	0.3399	0.3951	0.4705	0.5480
								ω	32.219	21.045	15.248	14.240	14.176
									0.1258	0.2678	0.3319	0.2699	0.2557
PCT SPAN	90	70	50	30	10			DFAC	.5880	.6104	.5780	.5299	.4837
DIA 9	26.021	29.683	33.345	37.007	40.669			EFFP	1.0354	0.9370	0.8850	0.8720	0.8460
BETA 9	48.166	49.802	48.840	45.068	42.220			EFF	1.0365	0.9350	0.8813	0.8675	0.8463
BETA 10	2.534	2.107	0.543	1.529	4.834			INCIDM	12.356	11.163	10.353	8.831	8.082
V 9	474.23	454.64	443.18	465.52	477.51			DEVM	9.927	10.096	9.096	5.587	4.206
V 10	266.33	312.99	304.14	344.36	402.59								
VZ 9	311.86	291.97	291.30	328.78	353.50								
VZ 10	262.98	310.96	303.65	344.22	401.02								
V-THETA 9	353.34	347.26	333.66	379.56	320.88								
V-THETA 10	11.77	11.51	2.88	9.19	33.93								
M 9	0.4199	0.4004	0.3887	0.4073	0.4165								
M 10	0.2331	0.2736	0.2648	0.2993	0.3496								
TURN	45.633	47.695	48.298	43.538	37.386								
ω	0.2278	0.1577	0.1733	0.0967	0.1049								
DFAC	0.7125	0.6147	0.6421	0.5824	0.4496								
EFFP	0.4476	0.5617	0.4437	0.3897	0.4939								
INCIDM	7.266	12.702	14.140	11.068	6.220								
DEVM	11.234	9.607	8.043	9.529	13.734								

Table B-2. Blade Element Data SLTO Configuration (Continued)

80 PERCENT SPEED PT. 133					CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/\delta$		
					SLTO		4820.		164.0		
INLET GUIDE VANE											
ROTOR											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	2.080	1.927	0.425	0.056	-1.959
BETA 4	2.355	2.140	0.531	0.093	-2.023	BETA 7	49.464	49.353	47.906	44.554	39.847
V 3	250.80	265.14	274.71	280.43	282.01	BETA(PRI) 6	60.078	62.964	65.630	67.586	70.094
V 4	254.41	276.29	291.84	298.62	297.07	BETA(PRI) 7	33.106	45.433	54.169	56.329	58.269
VZ 3	250.79	264.74	274.70	280.42	282.00	V 6	277.52	293.33	305.10	316.46	315.30
VZ 4	254.18	274.45	291.81	298.62	296.88	V 7	451.08	432.45	412.33	433.51	451.83
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	277.26	292.85	305.02	316.44	315.11
V-THETA 4	10.45	10.31	2.71	0.49	-10.49	VZ 7	291.38	281.36	276.24	308.86	346.89
M 3	0.2258	0.2388	0.2476	0.2528	0.2542	V-THETA 6	10.07	9.86	2.27	0.31	-10.78
M 4	0.2291	0.2490	0.2632	0.2694	0.2680	V-THETA 7	342.82	328.12	305.97	304.14	289.51
TURN	-2.35	-2.14	-0.53	-0.09	2.02	V(PRI) 6	555.9	644.4	739.3	829.9	925.5
(REFER TO 70% & 100% SPEED DATA)						V(PRI) 7	350.0	401.4	472.2	557.2	659.6
$\bar{\omega}$						VTHETA PR6	-481.7	-573.9	-673.4	-767.2	-870.2
DFAC	0.007	-0.022	-0.057	-0.064	-0.072	VTHETA PR7	-191.2	-286.0	-382.8	-463.7	-561.0
EFFP	-2.2464	1.3886	1.1912	1.1254	0.9722	U 6	491.81	583.72	675.63	767.53	859.44
INCIDIM	-20.200	-20.200	-20.200	-20.200	-20.200	U 7	533.98	614.11	688.77	767.85	850.50
DEVM	3.045	1.560	1.369	.207	.423	M 6	0.2501	0.2646	0.2754	0.2858	0.2847
STATOR B						M 7	0.3994	0.3813	0.3623	0.3800	0.3951
PCT SPAN	90	70	50	30	10	M(PRI) 6	0.5010	0.5813	0.6672	0.7495	0.8357
DIA 9	26.021	29.683	33.345	37.007	40.669	M(PRI) 7	0.3099	0.3539	0.4148	0.4884	0.5768
BETA 9	48.594	47.954	46.492	42.869	37.905	TURN(PRI)	26.964	17.506	11.456	11.256	11.825
BETA 10	3.984	2.670	1.012	2.155	0.467	$\bar{\omega}$	0.1563	0.2292	0.2976	0.2138	0.1668
V 9	444.50	433.82	415.91	444.50	470.68	DFAC	.5900	.5717	.5338	.4924	.4393
V 10	261.26	307.46	324.39	354.25	424.76	EFFP	1.0734	1.0242	0.9644	0.9557	0.9540
VZ 9	268.86	288.16	285.25	325.49	371.36	EFF	1.0756	1.0249	0.9633	0.9579	0.9540
VZ 10	257.82	304.16	322.95	353.62	424.75	INCIDIM	11.478	10.464	9.530	7.986	6.894
V-THETA 9	333.35	322.16	301.65	302.40	289.16	DEVM	14.306	12.933	12.069	7.729	5.369
V-THETA 10	18.15	14.32	5.73	13.32	3.46						
M 9	0.3933	0.3825	0.3654	0.3897	0.4121						
M 10	0.2289	0.2693	0.2836	0.3090	0.3708						
TURN	44.610	45.284	45.481	40.715	37.438						
$\bar{\omega}$	0.2099	0.1264	0.1400	0.1473	0.0729						
DFAC	0.6819	0.5824	0.5331	0.5075	0.3930						
EFFP	0.4053	0.4323	0.4840	0.2303	0.1149						
INCIDIM	7.694	10.854	11.792	8.869	1.905						
DEVM	12.684	10.170	8.512	1.155	9.367						

Table B-2. Blade Element Data SLTO Configuration (Continued)

80 PERCENT SPEED PT. 134										CONFIGURATION		N/√σ	
										SLTO		4845.	
INLET GUIDE VANE													
ROTOR													
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10		
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865		
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	1.915	1.577	0.357	0.282	-1.584		
BETA 4	2.186	1.777	0.469	-0.281	-1.428	BETA 7	46.950	46.156	43.625	38.371	36.149		
V 3	273.06	288.54	298.24	303.37	303.12	BETA(PR) 6	58.180	60.950	63.811	65.702	68.957		
V 4	274.15	301.47	315.62	323.95	325.26	BETA(PR) 7	27.711	43.004	52.123	52.626	55.560		
VZ 3	273.05	288.05	298.23	303.35	303.10	V 6	300.80	321.44	333.13	349.15	376.07		
VZ 4	273.92	299.78	315.53	323.90	325.15	V 7	492.74	451.48	427.22	468.59	483.60		
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	300.52	320.99	333.00	349.10	375.94		
V-THETA 4	10.46	9.35	2.58	-1.59	-8.11	VZ 7	334.22	312.30	309.02	367.27	380.44		
M 3	0.2461	0.2602	0.2691	0.2738	0.2735	V-THETA 6	10.05	8.85	2.07	-1.72	-9.20		
M 4	0.2471	0.2720	0.2850	0.2927	0.2939	V-THETA 7	360.07	325.62	294.75	290.87	285.27		
TURN	-2.19	-1.78	-0.47	0.28	1.43	V(PR) 6	570.0	661.2	754.6	848.4	935.4		
(REFER TO 70% & 100% SPEED DATA)													
ω	0.016	-0.028	-0.054	-0.070	-0.086	V(PR) 7	379.9	427.6	503.7	603.2	-873.2		
DFAC	0.1121	0.7070	0.7142	0.8453	0.7804	VTHETA PR6	-484.3	-577.9	-397.6	-481.0	-569.6		
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PR7	-176.7	-291.7	-677.1	-773.2	-869.6		
INCIDM	3.214	1.923	1.431	.581	-.172	U 6	494.37	586.75	679.13	771.52	863.91		
DEVM						U 7	536.75	617.30	692.34	771.83	854.91		
STATOR B													
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10		
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC	.5596	.5118	.4951	.4409	.4097		
BETA 9	45.863	44.769	41.831	36.501	35.818	EFFP	1.0517	0.9817	0.9448	0.9667	0.9320		
BETA 10	7.367	1.955	0.797	1.469	2.613	EFF	1.0533	0.9812	0.9432	0.9656	0.9295		
V 9	487.45	453.81	436.14	487.67	492.37	INCIDM	9.580	8.450	7.711	6.102	5.757		
V 10	296.96	340.43	347.23	393.55	439.24	DEVM	8.911	10.504	10.023	4.026	2.669		
VZ 9	333.93	319.87	324.15	391.97	399.08								
VZ 10	291.27	337.46	346.26	393.36	438.63								
V-THETA 9	349.83	319.59	290.88	290.08	288.14								
V-THETA 10	38.08	11.62	4.83	10.09	20.02								
M 9	0.4320	0.4007	0.3841	0.4294	0.4320								
M 10	0.2602	0.2986	0.3042	0.3445	0.3842								
TURN	38.496	42.814	41.034	35.032	33.205								
ω	0.2182	0.0190	0.0960	0.0828	0.0384								
DFAC	0.6332	0.5284	0.4924	0.4618	0.3731								
EFFP	0.4286	0.6723	0.7977	0.6728	1.2023								
INCIDM	4.963	7.669	7.131	2.501	-.182								
DEVM	16.067	9.455	8.297	9.469	11.513								

Table B-2. Blade Element Data SLTO Configuration (Continued)

80 PERCENT SPEED PT. 135				CONFIGURATION		SLTO		$N/\sqrt{\theta}$		$\omega\sqrt{\theta}/\delta$		198.6			
INLET GUIDE VANE												ROTOR			
POT SPAN				70		50		30		10		30		10	
DIA 3	90	21.150	26.050	0.000	0.000	3.090	35.850	0.000	0.000	4.0750	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	2.002	1.610	0.250	-0.292	-2.459
BETA 4	2.248	1.755	0.320	0.320	0.320	0.320	0.320	0.320	0.320	0.320	47.510	42.020	38.799	35.864	32.860
V 3	306.61	325.57	338.66	345.53	346.11	346.11	346.11	346.11	346.11	346.11	54.913	51.859	60.494	62.888	66.537
V 4	315.17	339.75	359.17	368.26	368.55	368.55	368.55	368.55	368.55	368.55	23.339	39.144	46.182	49.508	53.202
VZ 3	306.59	325.07	338.65	345.51	346.09	346.09	346.09	346.09	346.09	346.09	335.61	363.18	383.89	396.49	382.87
VZ 4	314.92	337.26	359.14	368.23	368.22	368.22	368.22	368.22	368.22	368.22	522.25	485.01	481.70	514.09	514.09
V-THETA 3	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	339.30	362.62	383.78	396.45	382.52
V-THETA 4	12.36	10.40	2.01	-2.06	-15.64	-15.64	-15.64	-15.64	-15.64	-15.64	350.84	359.85	375.15	407.81	433.76
M 3	0.2767	0.2941	0.3062	0.3125	0.3130	0.3130	0.3130	0.3130	0.3130	0.3130	11.87	10.21	1.67	-2.02	-16.50
M 4	0.2846	0.3072	0.3251	0.3335	0.3338	0.3338	0.3338	0.3338	0.3338	0.3338	385.10	324.66	301.83	294.88	275.92
TURN	-2.25	-1.75	-0.32	0.32	2.43	2.43	2.43	2.43	2.43	2.43	590.3	681.8	779.3	869.9	960.7
(REFER TO 70% & 100% SPEED DATA)															
W	-0.007	-0.027	-0.058	-0.069	-0.087	-0.087	-0.087	-0.087	-0.087	-0.087	384.2	464.6	542.2	628.2	724.2
DFAC	0.6211	0.7399	0.9946	1.1072	0.9970	0.9970	0.9970	0.9970	0.9970	0.9970	-483.0	-577.1	-678.2	-774.3	-881.3
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	-20.200	-20.200	-20.200	-20.200	-20.200	494.88	587.35	679.83	772.31	864.79
INCIDM	3.152	1.945	1.580	-0.020	.832	.832	.832	.832	.832	.832	537.31	617.93	693.06	772.63	855.80
DEVM											0.3071	0.3288	0.3480	0.3597	0.3471
											0.4633	0.4293	0.4253	0.4436	0.4520
											0.5338	0.6173	0.7064	0.7893	0.8709
											0.3408	0.4113	0.4788	0.5537	0.6367
											31.566	18.686	14.305	13.378	13.335
											0.1069	0.1096	0.1159	0.0883	0.1036
STATOR B															
POT SPAN	90	70	50	30	10	10	30	50	70	90	5814	5008	4660	4303	3889
DIA 9	26.021	25.683	33.345	37.007	40.669	40.669	37.007	33.345	25.683	26.021	0.9332	0.9784	0.9590	0.9578	0.9363
BETA 9	46.161	40.347	37.202	34.312	31.783	31.783	34.312	37.202	40.347	46.161	0.9312	0.9778	0.9577	0.9564	0.9339
BETA 10	4.224	C.960	0.344	-0.561	2.341	2.341	-0.561	0.344	C.960	4.224	6.313	5.359	4.394	3.288	3.337
V 9	516.63	490.39	493.67	521.66	524.74	524.74	521.66	493.67	490.39	516.63					
V 10	358.69	388.17	409.09	438.84	475.53	475.53	438.84	409.09	388.17	358.69					
VZ 9	350.44	370.31	392.02	430.80	445.95	445.95	430.80	392.02	370.31	350.44					
VZ 10	354.53	384.32	407.79	438.70	475.02	475.02	438.70	407.79	384.32	354.53					
V-THETA 9	372.21	317.49	298.49	294.06	276.39	276.39	294.06	298.49	317.49	372.21					
V-THETA 10	26.42	6.50	2.45	-4.30	19.43	19.43	-4.30	2.45	6.50	26.42					
M 9	0.4575	0.4343	0.4361	0.4603	0.4616	0.4616	0.4603	0.4361	0.4343	0.4575					
M 10	0.3416	0.3414	0.3593	0.3849	0.4169	0.4169	0.3849	0.3593	0.3414	0.3416					
TURN	41.937	39.387	36.858	34.873	29.442	29.442	34.873	36.858	39.387	41.937					
W	0.01853	0.0482	0.0623	0.0292	0.0162	0.0162	0.0292	0.0623	0.0482	0.01853					
DFAC	0.5595	0.4688	0.4352	0.4265	0.3323	0.3323	0.4265	0.4352	0.4688	0.5595					
EFFP	0.5970	0.7643	0.6825	0.6825	1.0839	1.0839	0.6825	0.6825	0.7643	0.5970					
INCIDM	5.261	3.247	2.502	.312	-4.217	-4.217	.312	2.502	3.247	5.261					
DEVM	12.924	8.460	7.844	7.439	11.241	11.241	7.439	7.844	8.460	12.924					

Table B-2. Blade Element Data SLTO Configuration (Continued)

80 PERCENT SPEED PT. 132				CONFIGURATION		SLTO		$\frac{N\sqrt{\sigma}}{s}$		239.3	
INLET GUIDE VANE				ROTOR							
				PCT SPAN		90		70		50	
PCT SPAN	90	70	50	30	10	PCT SPAN		90		70	
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	2.045	1.235	0.158	0.717	-1.579
BETA 4	2.339	1.421	0.107	-0.540	-1.673	BETA 7	30.335	26.460	23.032	20.197	18.068
V 3	379.32	407.20	416.35	423.05	429.91	PFTA(PRI) 6	47.735	51.022	54.855	58.041	62.389
V 4	395.48	428.63	443.72	455.45	466.84	BETA(PRI) 7	26.235	37.342	44.853	50.528	53.776
VZ 3	379.26	401.70	416.31	423.01	429.88	V 6	436.00	467.55	477.76	484.92	490.11
VZ 4	395.14	426.22	443.19	455.41	466.65	V 7	577.49	547.53	530.46	520.33	532.23
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	435.61	467.11	477.66	484.86	488.94
V-THETA 4	16.14	10.63	7.82	-4.29	-13.04	VZ 7	496.21	489.68	488.32	488.25	488.25
M 3	0.3438	0.3650	0.3782	0.3845	0.3844	V-THETA 6	15.56	10.08	1.32	-6.07	-12.65
M 4	0.3588	0.3897	0.4034	0.4140	0.4268	V-THETA 7	201.67	243.06	207.28	179.65	165.07
TURN	-2.34	-1.42	-0.11	0.54	1.67	V(PRI) 6	647.8	742.8	829.8	917.1	990.0
(REFER TO 70% & 100% SPEED DATA)											
W	-0.021	-0.052	-0.064	-0.082	-0.072	V(PRI) 7	555.7	616.5	688.8	748.2	856.2
DFAC	0.6021	0.8046	0.7327	0.9542	0.6935	VTHETA PPA	-470.3	-577.3	-678.5	-778.4	-877.4
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PPT	-245.6	-374.0	-485.8	-593.0	-690.7
INCIDM	3.061	2.279	1.793	.840	.073	U 6	494.84	587.35	679.83	772.31	864.79
DEVM						U 7	537.31	617.93	693.06	772.63	855.80
STATOR B											
PCT SPAN	90	70	50	30	10	M 6	0.3065	0.4264	0.4360	0.4428	0.4184
DIA 9	26.021	29.683	33.345	37.007	40.669	M 7	0.5191	0.4909	0.4750	0.4655	0.4760
BETA 9	28.411	24.969	21.986	18.746	17.734	M(PRI) 6	0.5893	0.6773	0.7573	0.8374	0.9024
BETA 10	1.567	0.456	-0.007	-0.526	1.432	M(PRI) 7	0.4005	0.5528	0.6168	0.6872	0.7657
V 9	590.20	566.69	546.94	526.71	507.43	TURN(PRI)	21.493	13.660	9.937	7.552	9.613
V 10	592.15	564.57	547.81	527.74	507.33	3	0.0680	0.0654	0.0844	0.0957	0.0874
VZ 9	510.59	510.84	506.76	511.35	514.62	DFAC	.2989	.2942	.2744	.2524	.2192
VZ 10	588.62	560.89	547.00	542.65	545.99	EFFP	1.0600	1.0605	1.0645	1.0186	1.0551
V-THETA 9	280.82	238.32	203.88	173.54	164.62	EFF	1.0709	1.0620	1.0660	1.0100	1.0514
V-THETA 10	16.19	4.49	-0.06	-8.77	13.72	INCIDM	-865	-1.478	-1.245	-1.519	-811
M 9	0.5312	0.5090	0.4805	0.4338	0.4336	DEVM	7.435	4.842	2.753	1.928	.876
M 10	0.5330	0.5070	0.4813	0.4364	0.4302						
TURN	26.844	24.414	21.893	19.472	16.303						
W	0.0476	0.0444	0.0364	0.0262	0.0286						
DFAC	0.1672	0.1731	0.1625	0.1530	0.1195						
EFFP	14.267	-11.738	32.8036	10.2093	30.690						
INCIDM	-12.489	-12.231	-12.844	-15.254	-18.266						
DEVM	10.267	7.956	7.493	7.074	10.332						

Table B-2. Blade Element Data SLTO Configuration (Continued)

100 PERCENT SPEED	PT. 154	CONFIGURATION	N/√θ	√θ/s	197.3	ROTOR
INLET GUIDE VANE						
PCT SPAN	90	70	50	30	10	PCT SPAN
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6
BETA 4	1.299	1.056	-0.173	0.831	4.149	BETA 7
V 3	303.52	320.83	334.49	343.55	346.36	VECTA (PR) 6
V 4	307.81	333.39	357.90	360.06	375.85	BETA (PR) 7
VZ 3	303.51	320.28	334.48	343.54	346.35	V 6
VZ 4	307.71	331.05	357.89	360.01	374.87	V 7
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6
V-THETA 4	6.98	6.14	-1.08	5.22	27.19	VZ 7
M 3	0.2739	0.3023	0.3107	0.3133	0.3133	V-THETA 6
M 4	0.2778	0.3013	0.3239	0.3259	0.3406	V-THETA 7
TURN	-1.30	-1.06	0.17	-0.83	-4.15	V (PR) 6
ω	0.0170	0.0212			0.0230	V (PR) 7
DFAC	-0.002	-0.029	-0.072	-0.040	-0.046	VTHETA PR6
EFFP	0.4715	0.5512	1.0339	0.8095	1.4089	VTHETA PR7
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6
DEVM	4.101	2.644	2.073	-5.31	-5.749	U 7
						M 6
						M 7
						M (PR) 6
						M (PR) 7
						TURN (PR)
						ω
						DFAC
						EFFP
						EFF
						INCIDM
						DEVM
						ω
						DFAC
						EFFP
						INCIDM
						DEVM

Table B-2. Blade Element Data SLTO Configuration (Continued)

100 PERCENT SPEED PT. 155										$\omega\sqrt{\sigma}/\delta$ 209.3									
CONFIGURATION										209.3									
INLET GUIDE VANE										6050.									
SLTO										209.3									
INLET GUIDE VANE										6050.									
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Table B-2. Blade Element Data SLTO Configuration (Continued)

100 PERCENT SPEED PT. 157				CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/s$		242.	
				SLTO		6054.					
INLET GUIDE VANE											
ROTOR											
PCT SPAN											
DIA 3											
BETA 3											
BETA 4											
V 3											
V 4											
VZ 3											
VZ 4											
V-THETA 3											
V-THETA 4											
M 3											
M 4											
TURN											
ω											
DFAC											
EFFP											
INCIDM											
DEVM											
PCT SPAN											
DIA 9											
BETA 9											
BETA 10											
V 9											
V 10											
VZ 9											
VZ 10											
V-THETA 9											
V-THETA 10											
M 9											
M 10											
TURN											
ω											
DFAC											
EFFP											
INCIDM											
DEVM											

Table B-2. Blade Element Data SLTO Configuration (Continued)

100 PERCENT SPEED PT. 158				CONFIGURATION		N/√σ		ω√δ/δ	
				SLTO		6040.		260.5	
INLET GUIDE VANE									
ROTOR									
PCT SPAN	90	70	50	30	10				
DIA 3	21.150	26.050	30.950	35.850	40.750				
BETA 3	0.000	0.000	0.000	0.000	0.000				
BETA 4	1.223	0.450	-0.974	-1.433	-0.695				
V 3	416.80	443.11	461.87	471.37	473.54				
V 4	438.04	468.15	497.46	503.22	510.20				
VZ 3	416.77	442.52	461.84	471.35	473.51				
VZ 4	437.93	464.69	497.38	503.05	510.17				
V-THETA 3	0.000	0.000	0.000	0.000	0.000				
V-THETA 4	9.35	3.68	-8.46	-12.58	-6.19				
M 3	0.3787	0.4033	0.4210	0.4300	0.4320				
M 4	0.3986	0.4269	0.4547	0.4602	0.4669				
TURN	-1.22	-0.45	0.97	1.43	0.70				
ω	0.0184		0.0214		0.0240				
DFAC	-0.039	-0.052	-0.086	-0.081	-0.084				
EFFP	0.7110	0.6677	0.9482	0.7759	0.9205				
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200				
DEVM	4.177	3.250	2.874	1.733	-9.95				
STATOR B									
PCT SPAN	90	70	50	30	10				
DIA 9	26.021	29.683	33.345	37.007	40.669				
BETA 9	40.135	35.480	31.351	29.924	30.403				
BETA 10	2.901	0.051	0.050	-1.261	1.350				
V 9	609.69	588.24	600.14	609.60	615.01				
V 10	489.01	507.26	529.51	555.16	604.83				
VZ 9	456.62	474.31	510.87	528.16	530.40				
VZ 10	484.78	502.02	527.69	554.80	604.59				
V-THETA 9	393.00	341.43	312.24	304.09	311.24				
V-THETA 10	24.75	0.45	0.46	-12.21	14.25				
M 9	0.5996	0.5194	0.5291	0.5356	0.5372				
M 10	0.4284	0.4449	0.4640	0.4856	0.5280				
TURN	37.233	35.430	31.301	31.184	29.053				
ω	0.1448	0.0344	0.0445	0.0088	0.0143				
DFAC	0.4275	0.3757	0.3463	0.3322	0.2517				
EFFP	0.4870	0.5575	0.4318	0.5210	3.5717				
INCIDM	-0.765	-1.620	-3.349	-4.076	-5.597				
DEVM	11.601	7.551	7.550	6.739	10.250				

Table B-2. Blade Element Data SLTO Configuration (Continued)

CONFIGURATION						$\frac{W}{\sqrt{\theta}}$	$\frac{W}{\sqrt{\theta}}/\delta$	ROTOR				
INLET GUIDE VANE						PT SPAN	90	70	50	30	10	
100 PERCENT SPEED	PT.153	SLTO	6020.	$\frac{W}{\sqrt{\theta}}$	280.5							
POT SPAN	90	70	50	30	10	DIA 6	23.385	27.755	32.125	36.495	40.865	
DIA 3	21.150	26.050	30.950	35.850	40.750	BETA 6	1.382	0.717	-0.240	-1.658	-3.297	
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 7	31.426	26.761	24.522	19.349	63.078	
BETA 4	1.606	0.907	-0.383	-1.653	-3.249	BETA(PR) 6	48.081	51.129	54.996	58.867	63.078	
V 3	463.70	491.39	507.59	514.29	514.05	BETA(PR) 7	27.224	38.376	45.873	53.654	56.251	
V 4	483.04	523.00	544.97	557.14	545.55	V 6	540.09	582.15	592.80	589.62	562.45	
VZ 3	463.62	490.80	507.54	514.25	514.05	V 7	694.44	662.72	635.81	586.14	609.27	
VZ 4	482.83	520.49	544.90	556.89	544.68	VZ 6	539.81	581.78	592.69	589.34	561.52	
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 7	589.99	591.17	578.17	542.62	574.86	
V-THETA 4	13.54	8.28	-3.64	-16.07	-30.92	V-THETA 6	13.03	7.29	-2.49	-17.06	-32.35	
M 3	0.4227	0.4489	0.4644	0.4708	0.4706	V-THETA 7	362.07	298.40	263.88	221.46	201.86	
M 4	0.4410	0.4791	0.5002	0.5120	0.5008	V(PR) 6	808.1	927.2	1033.3	1139.9	1240.2	
TURN	-1.61	-0.91	0.38	1.65	3.25	V(PR) 7	666.4	754.8	830.8	915.7	1034.8	
ω	0.0281	0.0216	0.0216	0.0220	0.0220	VTHETA PR6	-601.2	-721.8	-846.3	-975.7	-1105.8	
DFAC	-0.027	-0.056	-0.077	-0.099	-0.091	VTHETA PR7	-304.8	-468.6	-596.4	-737.6	-860.4	
EFFP	0.8119	0.9487	0.9143	1.0087	0.7728	U 6	614.26	729.05	843.83	958.62	1073.41	
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 7	666.92	767.00	860.25	959.02	1062.25	
DEVM	3.794	2.793	1.517	1.953	1.649	M 6	0.4955	0.5362	0.5466	0.5435	0.5171	
						M 7	0.6234	0.5934	0.5672	0.5203	0.5402	
						M(PR) 6	0.7414	0.8541	0.9528	1.0508	1.1402	
						M(PR) 7	0.5982	0.6759	0.7411	0.8129	0.9174	
						TURN(PR)	20.851	12.738	9.118	5.211	6.826	
						ω	0.0423	0.0550	0.1296	0.1682	0.2093	
POT SPAN	90	70	50	30	10	DFAC	.3341	.3096	.3044	.2901	.2543	
DIA 9	26.021	29.683	33.345	37.007	40.669	EFFP	1.0467	1.0583	0.9908	0.8358	0.872R	
BETA 9	29.165	24.917	23.710	20.645	18.538	EFF	1.0485	1.0604	0.9905	0.8309	0.8685	
BETA 10	1.843	C.592	0.193	-1.393	0.652	INCIDM	-.519	-1.3/1	-1.104	-.733	-.122	
V 9	715.34	652.76	648.96	614.17	616.27	DEVM	8.424	5.876	3.773	5.054	3.351	
V 10	725.69	650.43	654.82	614.44	652.85							
VZ 9	614.96	624.98	593.57	574.71	584.18							
VZ 10	721.41	686.70	654.12	614.25	652.64							
V-THETA 9	348.60	291.86	260.95	216.54	195.93							
V-THETA 10	23.34	7.13	2.21	-15.00	7.43							
M 9	0.6437	0.6223	0.5796	0.5466	0.5465							
M 10	0.6538	C.6201	0.5852	0.5468	0.5813							
TURN	27.321	24.325	23.517	22.043	17.885							
ω	0.1357	0.1339	0.1386	0.1160	0.0905							
DFAC	0.1583	0.1721	0.1664	0.1760	0.0896							
EFFP	4.2002	-14.7944	6.7688	117.9792	1.1808							
INCIDM	-11.735	-12.183	-10.990	-13.355	-17.462							
DEVM	10.513	8.092	7.693	6.602	9.552							

Table B-2. Blade Element Data SITO Configuration (Continued)

110 PERCENT SPEED Pt. 161									
INLET GUIDE VANE									
CONFIGURATION									
SITO									
N/√S $\omega \sqrt{S}$									
6654.0 223.62									
PCT SPAN									
DIA 3	90	70	50	30	10	90	70	50	10
BETA 3	21.150	26.050	30.950	35.850	40.750	23.385	27.755	32.125	36.495
BETA 4	0.000	0.000	0.000	0.000	0.000	4.220	4.946	5.899	7.218
V 3	4.859	5.395	5.124	7.590	7.757	50.533	52.310	54.280	51.122
V 4	356.13	375.33	386.90	392.72	393.36	58.983	61.872	64.144	65.874
VZ 3	352.51	388.39	416.07	426.56	429.34	29.404	44.611	52.656	57.965
VZ 4	356.13	374.61	386.89	392.71	393.35	392.10	416.96	435.75	454.13
VZ 4	351.16	384.99	414.26	422.77	425.41	652.23	607.96	602.93	648.34
V-THETA 3	0.00	0.00	0.00	0.00	0.00	390.88	415.02	433.98	450.47
V-THETA 4	29.86	36.52	37.16	56.34	57.95	411.50	371.12	351.73	406.83
M 3	0.3223	0.3401	0.3508	0.3552	0.3568	28.86	35.95	37.21	57.06
M 4	0.3190	0.3522	0.3780	0.3878	0.3904	503.52	481.10	489.51	504.72
TURN	-4.86	-5.39	-5.12	-7.59	-7.76	758.6	874.8	995.2	1099.1
ω	0.053	0.015	-0.027	-0.014	-0.018	475.9	522.1	580.3	688.4
DFAC	-0.8387	0.6735	0.9356	0.8945	0.8693	-650.1	-765.9	-895.5	-1002.5
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	-233.6	-366.7	-461.3	-555.3
INCIDM	0.541	-1.695	-3.224	-7.290	-9.357	678.95	805.83	932.70	1059.58
DEVN						737.16	847.78	950.85	1060.01
						0.3556	0.3788	0.3964	0.4137
						0.5721	0.5287	0.5198	0.5573
						0.6881	0.7948	0.9053	1.0012
						0.4174	0.4540	0.5003	0.5918
						29.569	17.039	11.479	12.036
						.0688	.2295	.3851	.1804
						.6032	.6042	.6084	.5555
						1.0152	0.9529	0.8733	0.9036
						1.0161	0.9503	0.8661	0.8973
						10.383	9.172	8.044	6.204
						10.604	12.111	10.556	5.165
									5.066
STATOR B									
PCT SPAN	90	70	50	30	10	90	70	50	10
DIA 9	26.021	29.683	33.345	37.007	40.569	23.385	27.755	32.125	36.495
BETA 9	49.936	51.295	53.127	49.652	51.367	4.220	4.946	5.899	7.218
BETA 10	-7.170	3.191	-1.317	1.257	4.493	50.533	52.310	54.280	51.122
V 9	641.29	607.34	606.90	665.42	691.10	58.983	61.872	64.144	65.874
V 10	319.48	380.16	388.56	454.35	574.98	29.404	44.611	52.656	57.965
VZ 9	406.78	377.53	363.30	430.67	431.46	392.10	416.96	435.75	454.13
VZ 10	312.55	376.76	387.30	453.99	573.19	652.23	607.96	602.93	648.34
V-THETA 9	490.80	473.96	485.50	507.13	539.86	390.88	415.02	433.98	450.47
V-THETA 10	-39.88	21.16	-8.93	9.97	45.04	411.50	371.12	351.73	406.83
M 9	0.5618	0.5280	0.5231	0.5725	0.5880	28.86	35.95	37.21	57.06
M 10	0.2736	0.3256	0.3303	0.3849	0.4848	503.52	481.10	489.51	504.72
TURN	57.106	48.105	54.444	48.395	46.874	758.6	874.8	995.2	1099.1
ω	.0927	.1791	.1849	.1980	.1217	475.9	522.1	580.3	688.4
DFAC	0.8184	0.6800	0.7183	0.6669	0.5167	-650.1	-765.9	-895.5	-1002.5
EFFP	0.5514	0.6942	0.6367	0.4551	0.7280	-233.6	-366.7	-461.3	-555.3
INCIDM	9.036	14.195	18.427	15.652	15.367	678.95	805.83	932.70	1059.58
DEVN	1.530	10.691	6.183	9.257	13.393	737.16	847.78	950.85	1060.01

Table B-2. Blade Element Data SLTO Configuration (Continued)

110 PERCENT SPEED Pt. 165				N/√σ		ω√σ/δ	
				6652.0		279.67	
INLET GUIDE VANE				ROTOR			
				POT SPAN			
				90	70	50	30
				10			

Table B-3. Blade Element Data Cruise Configuration

CONFIGURATION					$N/\sqrt{\theta}$	$\omega\sqrt{\theta}/\delta$
CRUISE					3078.	85.5
INLET GUIDE VANE						
ROTOR						
PCT SPAN	90	70	50	30	10	
DIA 6	21.150	26.050	30.950	35.850	40.750	DIA 6
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6
BETA 4	30.821	27.804	26.025	23.444	20.421	BETA 7
V 3	130.94	137.07	140.43	142.15	142.04	BETA(PR) 6
V 4	153.75	158.86	162.70	165.09	160.03	BETA(PR) 7
VZ 3	130.94	136.88	140.43	142.14	142.04	V 6
VZ 4	132.02	139.92	146.15	151.43	149.97	V 7
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6
V-THETA 4	78.78	74.10	71.39	65.68	55.84	VZ 7
M 3	0.1175	0.1230	0.1260	0.1275	0.1274	V-THETA 6
M 4	0.1380	0.1426	0.1460	0.1482	0.1436	V-THETA 7
TURN	-30.82	-27.80	-26.03	-23.44	-20.42	V(PR) 6
(REFER TO 70% & 100% SPEED DATA)						
DFAC	0.134	0.115	0.097	0.071	0.070	V(PR) 7
EFFP	0.9635	0.9575	0.9458	1.0498	0.8521	VTHETA PR6
INCID/M	-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PR7
DEVM	5.179	6.496	6.475	7.456	8.579	U 6
STATOR B						
PCT SPAN	90	70	50	30	10	
DIA 9	26.021	29.683	33.345	37.007	40.669	U 6
BETA 9	57.942	55.735	56.814	53.012	47.628	M 6
BETA 10	30.949	26.816	25.991	25.801	25.328	M 7
V 9	306.98	302.18	302.80	315.56	322.35	M(PR) 6
V 10	126.13	204.73	209.41	238.75	257.81	M(PR) 7
VZ 9	160.15	169.21	165.45	189.83	217.19	TURN(PR)
VZ 10	106.30	181.40	187.79	214.90	232.97	ω
V-THETA 9	260.17	249.73	253.41	252.06	238.15	DFAC
V-THETA 10	64.86	92.36	91.77	103.91	110.29	EFFP
M 9	0.2741	0.2696	0.2696	0.2807	0.2865	EFF
M 10	0.1119	0.1820	0.1859	0.2117	0.2285	INCIDM
TURN	26.993	28.919	30.823	27.212	22.301	DEVM
ω	0.4963	0.2149	0.2296	0.2039	0.2438	
DFAC	0.8264	0.5335	0.5413	0.4618	0.3900	
EFFP	0.4558	0.6940	0.6494	0.5562	0.3993	
INCID/M	8.042	9.635	13.114	10.012	2.628	
DEVM	13.649	8.316	7.491	7.801	8.228	

Table B-3. Blade Element Data Cruise Configuration (Continued)

50PERCENT SPEED		PT.183		CONFIGURATION		N/√δ		ω√δ/δ		91.0	
INLET GUIDE VANE											
ROTOR											
PCT SPAN											
DIA 6											
BETA 6											
BETA 7											
BETA(PR) 6											
BETA(PR) 7											
V 6											
V 7											
VZ 6											
VZ 7											
V-THETA 6											
V-THETA 7											
V(PR) 6											
V(PR) 7											
VTHETA PR6											
VTHETA PR7											
U 6											
U 7											
M 6											
M 7											
M(PR) 6											
M(PR) 7											
TURN(PR)											
ω											
DFAC											
EFFP											
EFF											
INCIDM											
DEVM											
STATOR B											
PCT SPAN											
DIA 9											
BETA 9											
BETA 10											
V 9											
V 10											
VZ 9											
VZ 10											
V-THETA 9											
V-THETA 10											
M 9											
M 10											
TURN											
ω											
DFAC											
EFFP											
INCIDM											
DEVM											

Table B-3. Blade Element Data Cruise Configuration (Continued)

[illegible]

Table B-3. Blade Element Data Cruise Configuration (Continued)

50PERCENT SPEED PT.185					CONFIGURATION		N/√θ		ω√θ/δ		
					CRUISE		3062.		106.2		
INLET GUIDE VANE										ROTOR	
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	27.817	25.760	24.332	22.182	18.692
BETA 4	30.713	27.731	25.678	23.433	19.829	BETA 7	49.454	49.474	50.438	46.655	42.411
V 3	160.31	169.17	174.77	178.06	178.96	BETA(PR) 6	49.863	55.647	60.604	64.055	67.877
V 4	194.81	202.34	204.18	204.69	197.58	BETA(PR) 7	21.446	35.844	45.294	50.697	55.665
VZ 3	160.30	169.00	174.76	178.05	178.95	V 6	206.18	211.76	211.52	213.79	205.98
VZ 4	167.48	178.05	184.01	187.81	185.86	V 7	334.13	317.30	309.35	311.54	307.80
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	182.31	190.56	192.71	197.96	195.11
V-THETA 4	99.50	94.15	88.47	81.40	67.02	VZ 7	216.17	206.02	196.96	213.82	227.25
M 3	0.1439	0.1519	0.1569	0.1599	0.1607	V-THETA 6	96.21	92.03	87.15	80.72	66.01
M 4	0.1750	0.1818	0.1835	0.1840	0.1775	V-THETA 7	253.90	241.18	238.49	226.56	207.60
TURN	-30.71	-27.73	-25.68	-23.43	-19.83	V(PR) 6	282.8	337.8	392.6	452.5	518.1
(REFER TO 70% & 100% SPEED DATA)						V(PR) 7	233.4	254.3	280.1	337.6	402.9
ω						VTHETA PR6	-216.2	-278.8	-342.1	-406.9	-480.0
DFAC	0.103	0.086	0.086	0.080	0.083	VTHETA PR7	-85.3	-148.0	-199.1	-261.2	-332.7
EFFP	0.8489	0.9378	0.9098	0.9789	0.7922	U 6	312.44	370.82	429.21	487.59	545.98
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 7	339.22	390.13	437.56	487.79	540.30
DEVM	5.287	6.569	6.822	7.467	9.171	M 6	0.1853	0.1904	0.1902	0.1922	0.1851
STATOR B						M 7	0.2893	0.2838	0.2763	0.2781	0.2745
PCT SPAN	90	70	50	30	10	M(PR) 6	0.2542	0.3037	0.3530	0.4068	0.4657
DIA 9	26.021	29.683	33.345	37.007	40.669	M(PR) 7	0.2090	0.2275	0.2501	0.3013	0.3593
BETA 9	47.982	48.115	49.104	45.006	40.523	TURN(PR)	28.411	19.781	15.306	13.357	12.213
BETA 10	27.511	27.300	27.240	26.255	26.028	ω	0.0143	0.0321	0.0629	0.0462	0.0745
V 9	332.03	318.88	312.09	317.04	315.61	DFAC	.3915	.4259	.4505	.3978	.3500
V 10	255.82	254.57	251.94	277.19	292.81	EFFP	0.9813	0.9821	0.9334	0.9823	0.9956
VZ 9	218.47	211.34	203.73	224.04	239.91	EFF	0.9812	0.9820	0.9327	0.9821	0.9956
VZ 10	225.33	224.32	223.28	248.44	263.11	INCIDM	1.263	3.147	4.504	4.455	4.677
V-THETA 9	246.68	237.40	235.91	224.21	205.07	DEVM	2.646	3.344	3.194	2.097	2.765
V-THETA 10	118.17	116.76	115.32	122.62	128.49						
M 9	0.2974	0.2852	0.2787	0.2830	0.2815						
M 10	0.2283	0.2270	0.2244	0.2270	0.2609						
TURN	20.471	20.815	21.864	18.751	14.495						
ω	0.1431	0.1410	0.1727	0.1203	0.1068						
DFAC	0.3699	0.3538	0.3606	0.2748	0.1890						
EFFP	0.7028	0.6642	0.5218	0.4702	0.2947						
INCIDM	-1.918	2.015	5.404	2.006	-4.477						
DEVM	10.211	8.800	8.740	8.255	8.928						

Table B-3. Blade Element Data Cruise Configuration (Continued)

50PERCENT SPEED		PT.186		CONFIGURATION		$\frac{W}{\sqrt{\theta}}$		$\frac{W\sqrt{\theta}}{\delta}$			
				CRUISE		3056.		114.7			
INLET GUIDE VANE											
ROTOR											
POT SPAN	90	70	50	30	10	POT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	27.652	25.675	24.405	22.428	21.945
BETA 4	30.554	27.842	26.059	23.698	22.190	BETA 7	44.651	44.424	43.792	41.488	43.177
V 3	178.12	187.18	192.02	193.01	185.99	BETA(PR) 6	46.790	53.470	58.733	62.361	64.718
V 4	209.38	213.44	213.86	216.12	227.54	BETA(PR) 7	22.668	35.948	44.008	50.165	55.965
VZ 3	178.11	186.84	192.01	193.00	185.98	V 6	221.70	224.48	224.11	226.77	233.11
VZ 4	180.28	187.90	191.81	197.40	210.42	V 7	338.59	319.71	314.32	311.99	305.69
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6	196.27	202.11	203.89	209.52	216.22
V-THETA 4	106.44	99.68	93.95	86.86	85.94	VZ 7	239.44	227.93	226.61	233.57	222.81
M 3	0.1600	0.1681	0.1725	0.1734	0.1671	V-THETA 6	102.89	97.26	92.60	86.52	87.12
M 4	0.1882	0.1919	0.1923	0.1943	0.2047	V-THETA 7	237.96	223.79	217.53	206.68	209.14
TURN	-30.55	-27.84	-26.06	-23.70	-22.19	VIPR) 6	286.7	339.7	392.9	451.7	506.3
ω	(REFER TO 70% & 100% SPEED DATA)					VIPR) 7	261.0	282.0	315.5	364.8	398.3
DFAC	0.131	0.130	0.132	0.106	0.007	VTHETA PR6	-208.9	-272.8	-335.8	-400.1	-457.8
EFFP	0.9193	0.8754	0.8556	1.0065	0.9144	VTHETA PR7	-100.6	-165.6	-219.2	-280.2	-330.1
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	311.82	370.09	428.36	486.64	544.91
DEVM	5.446	6.458	6.441	7.202	6.810	U 7	338.56	389.36	436.70	486.84	539.24
STATOR B											
POT SPAN	90	70	50	30	10	POT SPAN	90	70	50	30	10
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC	.2728	.3217	.3326	.3111	.3256
BETA 9	42.946	42.802	42.312	40.123	41.385	EFFP	1.0160	1.0083	0.9967	1.0137	1.0156
BETA 10	27.160	27.257	27.608	27.113	26.266	EFF	1.0161	1.0084	0.9967	1.0138	1.0158
V 9	339.21	324.50	320.22	319.43	314.22	INCIDM	-1.810	0.970	2.633	2.761	1.518
V 10	288.79	287.23	281.83	297.75	308.66	DEVM	3.868	3.448	1.908	1.565	3.065
VZ 9	244.23	236.57	236.49	244.17	231.36						
VZ 10	255.05	253.63	249.37	265.03	276.37						
V-THETA 9	231.10	220.49	215.56	205.85	207.73						
V-THETA 10	131.82	131.55	130.61	135.70	136.59						
M 9	0.3043	0.2908	0.2866	0.2858	0.2808						
M 10	0.2584	0.2569	0.2518	0.2661	0.2757						
TURN	15.786	15.545	14.704	13.009	15.119						
ω	0.1832	0.1193	0.1713	0.1020	0.1208						
DFAC	0.2527	0.2238	0.2344	0.1701	0.1268						
EFFP	0.4222	0.5146	0.3179	0.2278	0.3793						
INCIDM	-6.954	-3.298	-1.388	-2.877	-3.615						
DEVM	9.860	8.757	9.108	9.112	9.166						

Table B-3. Blade Element Data Cruise Configuration (Continued)

70 PERCENT SPEED PT. 168				CONFIGURATION		N/√θ		ω√θ/δ							
				CRUISE		4240.		118.8							
INLET GUIDE VANE															
ROTOR															
PCT SPAN															
	90	70	50	30	10										
DIA 3	21.150	26.050	30.950	35.850	40.750										
BETA 3	0.000	0.000	0.000	0.000	0.000										
BETA 4	27.752	27.942	26.346	24.417	34.815										
V 3	184.67	192.96	197.38	198.82	196.41										
V 4	212.14	222.14	228.94	235.57	255.22										
VZ 3	184.67	192.69	197.38	198.81	196.41										
VZ 4	187.71	195.43	205.02	214.42	209.53										
V-THETA 3	0.00	0.00	0.00	0.00	0.00										
V-THETA 4	98.78	104.09	101.60	97.38	145.71										
M 3	0.1659	0.1734	0.1774	0.1787	0.1765										
M 4	0.1907	0.1998	0.2059	0.2120	0.2298										
TURN	-27.75	-27.94	-26.35	-24.42	-34.81										
ω	0.0481		0.0476		0.0420										
DFAC	0.125	0.122	0.099	0.061	0.071										
EFFP	0.9195	0.8930	0.9092	1.0001	0.9537										
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200										
DEVM	8.248	6.358	6.154	6.483	-5.815										
STATOR B															
	90	70	50	30	10										
DIA 9	26.021	29.683	33.345	37.007	40.669										
BETA 9	57.059	56.334	55.120	53.905	61.280										
BETA 10	37.892	26.491	25.119	25.604	23.217										
V 9	405.26	404.65	414.88	431.61	475.93										
V 10	138.20	253.06	295.86	332.72	355.65										
VZ 9	216.29	223.08	236.91	254.27	228.56										
VZ 10	107.54	224.73	267.26	299.99	326.79										
V-THETA 9	340.11	336.79	340.35	348.76	417.38										
V-THETA 10	85.50	112.88	125.59	143.79	140.20										
M 9	0.3607	0.3597	0.3679	0.3818	0.4214										
M 10	0.1225	0.2234	0.2610	0.2928	0.3125										
TURN	19.166	29.843	30.001	28.301	38.063										
ω	0.5982	0.3502	0.2426	0.2095	0.1604										
DFAC	0.8908	0.5992	0.5126	0.4501	0.5307										
EFFP	0.3561	0.4970	0.6198	0.5706	0.6810										
INCIDM	7.159	10.234	11.420	10.905	16.280										
DEVM	20.592	7.991	6.619	7.604	6.117										
PCT SPAN															
	90	70	50	30	10										
DIA 6	23.385	27.755	32.125	36.495	40.865										
BETA 6	25.011	26.050	24.810	23.131	37.590										
BETA 7	56.382	57.580	56.766	55.241	61.586										
BETA(PI) 6	58.711	63.018	66.221	68.591	70.114										
BETA(PI) 7	27.473	42.261	49.454	53.176	60.080										
V 6	226.11	233.13	239.76	246.59	269.95										
V 7	419.16	405.78	410.19	426.69	438.45										
VZ 6	204.83	209.30	217.54	226.74	213.90										
VZ 7	229.56	217.27	224.64	243.21	208.62										
V-THETA 6	95.60	102.38	100.60	96.87	164.67										
V-THETA 7	349.06	342.53	343.10	350.55	385.63										
V(PI) 6	394.4	461.4	539.6	621.2	628.8										
V(PI) 7	261.6	293.9	345.8	405.9	418.3										
VTHETA PR6	-337.0	-411.1	-493.7	-578.3	-591.4										
VTHETA PR7	-120.7	-197.7	-262.8	-324.9	-362.5										
U 6	432.63	513.48	594.33	675.18	756.02										
U 7	469.73	540.21	605.89	675.45	748.16										
M 6	0.2034	0.2097	0.2158	0.2220	0.2432										
M 7	0.3734	0.3607	0.3637	0.3774	0.3872										
M(PI) 6	0.3548	0.4151	0.4856	0.5592	0.5666										
M(PI) 7	0.2330	0.2613	0.3067	0.3590	0.3694										
TURN(PI)	31.229	20.741	16.759	15.412	10.033										
ω	0.1144	0.1253	0.2031	0.1658	-0.0184										
DFAC	.5794	.5720	.5497	.5289	.4984										
EFFP	0.9734	1.0011	0.9703	0.9423	0.9110										
EFF	0.9729	1.0011	0.9696	0.9409	0.9086										
INCIDM	10.111	10.518	10.121	8.991	6.914										
DEVM	8.673	9.761	7.354	4.576	7.180										

Table B-3. Blade Element Data Cruise Configuration (Continued)

70 PERCENT SPEED				PT. 169	CONFIGURATION		CRUISE	$N/\sqrt{\sigma}$	425.5.	$\omega\sqrt{\sigma}/\delta$	125.
INLET GUIDE VANE											
ROTOR											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	25.479	26.108	24.920	23.289	21.775
BETA 4	28.104	28.214	26.129	24.339	22.887	BETA 7	57.453	52.476	57.514	56.204	51.807
V 3	193.01	202.45	207.55	210.51	211.44	BETA(PR) 6	56.659	61.041	65.017	67.742	70.630
V 4	227.28	236.38	243.91	247.75	239.88	BETA(PR) 7	26.653	40.527	49.592	54.049	58.066
VZ 3	193.00	202.20	207.55	210.50	211.43	V 6	240.94	249.95	251.95	256.71	251.85
VZ 4	200.47	207.36	218.95	225.73	221.00	V 7	423.54	412.64	412.39	424.18	422.28
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	217.45	224.28	228.44	235.77	233.88
V-THETA 4	107.07	111.75	107.42	102.11	93.29	VZ 7	226.08	251.08	221.37	235.91	261.10
M 3	0.1734	0.1819	0.1866	0.1892	0.1901	V-THETA 6	103.65	110.00	106.16	101.49	93.43
M 4	0.2044	0.2127	0.2195	0.2230	0.2159	V-THETA 7	357.02	327.26	347.86	352.50	331.98
TURN	-28.10	-28.21	-26.13	-24.34	-22.89	V(PR) 6	395.7	463.3	540.9	622.5	705.2
ω	0.0559	0.0297	0.085	0.067	0.0470	V(PR) 7	254.9	330.7	341.7	401.9	493.6
DFAC	0.107	0.112	0.085	0.067	0.086	VTHETA PR6	-330.5	-405.3	-490.3	-576.1	-665.3
EFFP	0.8755	0.8710	0.8889	0.9645	0.8260	VTHETA PR7	-114.4	-214.9	-260.2	-325.3	-418.9
INCIDIM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	434.16	515.30	596.43	677.56	758.70
DEVM	7.896	6.086	6.371	6.561	6.113	U 7	471.39	542.13	608.03	677.84	750.81
STATOR B						M 6	0.2168	0.2250	0.2268	0.2312	0.2268
						M 7	0.3773	0.3676	0.3657	0.3752	0.3729
						M(PR) 6	0.3561	0.4171	0.4870	0.5605	0.6349
						M(PR) 7	0.2271	0.2946	0.3030	0.3554	0.4359
						TURN(PR)	29.999	20.497	15.420	13.691	12.564
						ω	0.1112	0.0938	0.1518	0.1260	0.1661
PCT SPAN	90	70	50	30	10	DFAC	.5983	.4936	.5578	.5344	.4581
DIA 9	26.021	25.683	33.345	37.007	40.669	EFFP	0.9734	1.1008	0.9597	0.9376	0.9653
BETA 9	57.432	51.155	56.208	54.559	49.411	EFF	0.9729	1.1029	0.9588	0.9360	0.9644
BETA 10	29.828	25.604	25.779	25.819	23.346	INCIDIM	8.059	8.541	8.917	8.142	7.430
V 9	412.33	413.43	415.40	430.00	436.45	DEVM	7.853	8.027	7.492	5.449	5.166
V 10	185.29	290.13	292.46	337.50	363.94						
VZ 9	178.86	257.90	262.54	249.19	283.97						
VZ 10	158.02	259.67	262.54	303.56	334.15						
V-THETA 9	347.49	322.00	345.22	350.32	331.44						
V-THETA 10	92.16	125.38	127.19	146.99	144.22						
M 9	0.3671	0.3684	0.3684	0.3804	0.3858						
M 10	0.1632	0.2575	0.2579	0.2970	0.3203						
TURN	27.604	25.550	30.429	28.739	26.065						
ω	0.1871	0.1942	0.2387	0.1844	0.1740						
DFAC	0.7813	0.4907	0.5249	0.4351	0.3712						
EFFP	0.4153	1.0757	0.6335	0.5565	0.4033						
INCIDIM	7.532	5.055	12.508	11.559	4.411						
DEVM	12.528	7.104	7.279	7.819	6.246						

Table B-3. Blade Element Data Cruise Configuration (Continued)

70 PERCENT SPEED PT. 170				CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/\delta$	
				CRUISE		4260.		131.	
INLET GUIDE VANE									
ROTOR									
PCT SPAN									
DIA 6									
BETA 6									
BETA (PR) 6									
BETA (PR) 7									
V 6									
V 7									
VZ 6									
VZ 7									
V-THETA 6									
V-THETA 7									
V (PR) 6									
V (PR) 7									
VTHETA PR6									
VTHETA PR7									
U 6									
U 7									
M 6									
M 7									
M (PR) 6									
M (PR) 7									
TURN (PR)									
ω									
DFAC									
EFFP									
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INCIDM									
DEVM									

Table B-3. Blade Element Data Cruise Configuration (Continued)

70 PERCENT SPEED PT.171				CONFIGURATION		N/√σ		ω√σ/δ	
				CRUISE		4245.		140.06	
INLET GUIDE VANE									
ROTOR									
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	25.436	26.298	25.162
BETA 4	28.166	28.252	26.588	24.418	23.912	BETA 7	51.330	51.357	54.049
V 3	214.18	225.45	232.77	236.88	237.32	BETA(PR) 6	52.067	57.383	61.277
V 4	256.86	265.72	276.96	278.11	270.99	BETA(PR) 7	22.410	37.118	45.866
VZ 3	214.17	225.20	232.76	236.89	237.31	V 6	272.78	278.96	286.55
VZ 4	226.44	232.79	247.66	253.23	247.73	V 7	452.88	431.44	434.21
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	246.29	249.89	259.31
V-THETA 4	121.25	125.78	123.96	114.97	109.84	VZ 7	281.44	269.16	251.64
M 3	0.1926	0.2028	0.2094	0.2132	0.2135	V-THETA 6	117.16	123.59	121.84
M 4	0.2313	0.2394	0.2496	0.2507	0.2442	V-THETA 7	353.59	336.97	347.13
TURN	-28.17	-28.25	-26.59	-24.42	-23.91	V(PR) 6	400.7	463.7	539.6
ω	0.0460		0.0307		0.0320	V(PR) 7	306.1	337.9	361.5
DFAC	0.091	0.104	0.078	0.070	0.089	VTHETA PR6	-316.0	-390.5	-473.2
EFFP	0.8887	0.8867	1.0825	1.2081	1.0722	VTHETA PR7	-116.7	-203.9	-259.5
INCID/M	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	433.14	514.09	595.03
DEVM	7.834	6.048	5.912	6.482	5.068	U 7	470.28	540.85	606.61
STATOR B									
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC	0.0686	0.0593	0.0706
BETA 9	50.157	50.079	52.668	48.193	46.102	EFFP	0.9564	0.9573	0.9575
BETA 10	27.622	26.812	26.757	25.624	26.087	EFF	0.9557	1.0084	0.9174
V 9	447.37	432.74	431.63	442.14	442.40	INCID/M	3.467	4.883	5.177
V 10	303.72	317.13	318.63	364.24	385.04	DEVM		4.618	3.766
VZ 9	281.61	275.80	261.04	294.67	306.73				
VZ 10	266.74	280.73	283.63	328.24	345.79				
V-THETA 9	343.49	331.88	343.21	329.52	318.78				
V-THETA 10	140.81	143.05	143.45	157.52	169.32				
M 9	0.3997	0.3861	0.3838	0.3929	0.3926				
M 10	0.2691	0.2811	0.2815	0.3221	0.3404				
TURN	22.536	23.267	25.911	22.560	20.015				
ω	0.1768	0.1283	0.1837	0.1146	0.1065				
DFAC	0.4872	0.4434	0.4633	0.3572	0.2916				
EFFP	0.7280	0.7807	0.6571	0.6727	0.6165				
INCID/M	0.257	3.979	8.968	5.183	1.102				
DEVM	10.322	8.312	8.257	7.624	6.987				

Table B-3. Blade Element Data Cruise Configuration (Continued)

CONFIGURATION									
80PERCENT SPEED					PT.175				
INLET GUIDE VANE									
CRUISE									
W/√θ/δ 138.0									
ROTOR									
PCT SPAN									
DIA 3	90	70	50	30	10	W/√θ/δ 481.0			
BETA 3	21.150	26.050	30.950	35.850	40.750				
BETA 4	0.000	0.000	0.000	0.000	0.000				
BETA 10	30.178	27.880	26.088	23.381	20.275				
V 3	212.16	222.91	229.13	232.97	234.08				
V 4	245.99	258.51	269.00	273.09	266.24				
VZ 3	212.15	222.54	229.12	232.96	234.07				
VZ 4	212.62	227.36	241.52	250.65	249.75				
V-THETA 3	0.000	0.000	0.000	0.000	0.000				
V-THETA 4	123.66	120.88	118.29	108.37	92.26				
M 3	0.1907	0.2005	0.2061	0.2096	0.2106				
M 4	0.2214	0.2328	0.2424	0.2461	0.2399				
TURN	-30.18	-27.88	-26.09	-23.38	-20.28				
(REFER TO 70% & 100% SPEED DATA)									
DFAC	0.139	0.115	0.086	0.061	0.059				
EFFP	0.8606	0.8844	0.9345	0.9926	0.8707				
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200				
DEVM	5.822	6.420	6.412	7.519	8.725				
STATOR B									
PCT SPAN									
DIA 9	90	70	50	30	10	W/√θ/δ 481.0			
BETA 9	26.021	29.683	33.345	37.007	40.669				
BETA 10	60.417	55.432	57.408	53.796	47.454				
V 9	37.741	26.327	25.407	25.791	21.343				
V 10	461.79	462.78	467.38	486.77	500.75				
VZ 9	143.68	275.69	346.45	383.11	412.74				
VZ 10	223.88	260.78	250.99	287.32	338.60				
V-THETA 9	111.28	244.73	311.68	344.62	383.96				
V-THETA 10	401.59	381.08	393.78	392.78	368.92				
M 9	87.94	122.27	148.64	166.69	150.03				
M 10	0.4101	0.4107	0.4129	0.4289	0.4407				
TURN	0.1257	0.2422	0.3044	0.3355	0.3606				
TURN	22.676	29.104	32.001	28.005	26.110				
DFAC	0.6653	0.4558	0.2493	0.2496	0.2452				
EFFP	0.9428	0.6315	0.4874	0.4291	0.3856				
INCIDM	0.2665	0.3138	0.5901	0.3639	0.1756				
DEVM	10.517	9.332	13.708	10.796	2.454				
DEVM	20.441	7.827	6.907	7.791	4.243				

Table B-3. Blade Element Data Cruise Configuration (Continued)

INLET GUIDE VANE										ROTOR									
80PERCENT SPEED					CONFIGURATION					N/√S					ω√S/δ				
PT.177					CRUISE					4820.					147.5				
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30			
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865	DIA 6	23.385	27.755	32.125	36.495			
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	27.053	25.837	24.631	21.802	20.781	BETA 6	27.053	25.837	24.631	21.802			
BETA 4	30.299	27.744	26.072	23.137	21.434	BETA 7	56.570	55.141	56.926	54.527	51.003	BETA 7	56.570	55.141	56.926	54.527			
V 3	228.14	240.02	246.95	250.46	247.07	BETA(PR) 6	55.491	60.419	64.290	66.962	68.513	BETA(PR) 6	55.491	60.419	64.290	66.962			
V 4	266.98	276.09	282.36	286.16	297.29	BETA(PR) 7	23.869	39.393	48.040	53.035	56.822	BETA(PR) 7	23.869	39.393	48.040	53.035			
VZ 3	228.14	239.59	246.94	250.45	247.06	V 6	280.73	289.00	293.31	300.52	314.88	V 6	280.73	289.00	293.31	300.52			
VZ 4	230.48	243.09	253.43	262.91	276.63	V 7	495.19	476.08	476.69	484.30	489.15	V 7	495.19	476.08	476.69	484.30			
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6	248.55	259.85	266.45	278.94	285.34	VZ 6	248.55	259.85	266.45	278.94			
V-THETA 4	134.70	128.52	124.09	112.44	108.64	V7 7	270.51	271.65	259.85	280.91	307.13	V7 7	270.51	271.65	259.85	280.91			
M 3	0.2052	0.2160	0.2223	0.2255	0.2224	V-THETA 6	130.29	125.95	122.24	111.61	109.14	V-THETA 6	130.29	125.95	122.24	111.61			
M 4	0.2405	0.2488	0.2546	0.2580	0.2682	V-THETA 7	413.27	390.66	399.45	394.41	380.64	V-THETA 7	413.27	390.66	399.45	394.41			
TURN	-30.30	-27.74	-26.07	-23.14	-21.43	V(PR) 6	438.8	526.5	614.3	712.8	806.3	V(PR) 6	438.8	526.5	614.3	712.8			
ω	0.132	0.121	0.109	0.083	0.016	V(PR) 7	298.3	352.1	389.1	467.4	561.4	V(PR) 7	298.3	352.1	389.1	467.4			
DFAC	0.9113	0.8911	0.9012	0.9676	0.9257	VTHETA PR6	-361.5	-457.8	-553.4	-655.9	-750.3	VTHETA PR6	-361.5	-457.8	-553.4	-655.9			
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PR7	-120.7	-223.5	-289.3	-373.4	-469.9	VTHETA PR7	-120.7	-223.5	-289.3	-373.4			
INCIDM	5.701	6.556	6.428	7.763	7.566	U 6	491.81	583.72	675.63	767.53	859.44	U 6	491.81	583.72	675.63	767.53			
DEVM	5.701	6.556	6.428	7.763	7.566	U 7	533.98	614.11	688.77	767.85	859.50	U 7	533.98	614.11	688.77	767.85			
						M 6	0.2531	0.2606	0.2646	0.2712	0.2843	M 6	0.2531	0.2606	0.2646	0.2712			
						M 7	0.4412	0.4730	0.4217	0.4270	0.4305	M 7	0.4412	0.4730	0.4217	0.4270			
						M(PR) 6	0.3956	0.4748	0.5541	0.6432	0.7281	M(PR) 6	0.3956	0.4748	0.5541	0.6432			
						M(PR) 7	0.2658	0.3128	0.3448	0.4121	0.4941	M(PR) 7	0.2658	0.3128	0.3448	0.4121			
						TURN(PR)	31.610	21.001	16.234	13.920	11.691	TURN(PR)	31.610	21.001	16.234	13.920			
						ω	-0.0069	0.0694	0.1172	0.1070	0.1606	ω	-0.0069	0.0694	0.1172	0.1070			
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30			
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC	.5652	.5336	.5580	.5214	.5336	DFAC	.5652	.5336	.5580	.5214			
BETA 9	56.443	53.726	55.523	52.642	49.095	EFFP	0.9859	1.0125	0.9859	0.9422	0.9394	EFFP	0.9859	1.0125	0.9859	0.9394			
BETA 10	30.201	25.654	26.018	25.775	24.436	EFF	0.9855	1.0128	0.9855	0.9406	0.9375	EFF	0.9855	1.0128	0.9855	0.9375			
V 9	482.17	476.26	480.42	491.76	504.79	INCIDM	6.891	7.919	8.190	7.362	5.313	INCIDM	6.891	7.919	8.190	7.362			
V 10	214.20	328.88	339.41	393.41	425.00	DEVM	5.069	6.893	5.940	4.435	3.922	DEVM	5.069	6.893	5.940	4.435			
VZ 9	261.56	279.60	271.16	298.36	328.67														
VZ 10	182.34	293.62	303.51	354.04	386.80														
V-THETA 9	401.81	383.96	396.04	390.88	381.51														
V-THETA 10	107.75	142.38	148.71	171.07	175.81														
M 9	0.4291	0.4231	0.4250	0.4337	0.4448														
M 10	0.1879	0.2897	0.2975	0.2975	0.3723														
TURN	26.243	28.072	29.505	26.867	24.658														
ω	0.1783	0.1884	0.2250	0.1543	0.1957														
DFAC	0.7828	0.5150	0.5188	0.4080	0.3530														
EFFP	0.4739	0.7058	0.6581	0.5952	0.3437														
INCIDM	6.543	7.626	11.823	9.642	4.095														
DEVM	12.901	7.154	7.518	7.275	7.336														

Table B-3. Blade Element Data Cruise Configuration (Continued)

CONFIGURATION						$N/\sqrt{\sigma}$	$w\sqrt{\sigma}/\delta$
INLET GUIDE VANE						4804.	157.9
CRUISE							
80 PERCENT SPEED	PT. 178						
PCT SPAN	90	70	50	30	10	PCT SPAN	90
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	28.114
BETA 4	30.897	27.828	26.109	23.333	18.815	BETA 7	54.156
V 3	243.56	256.10	263.74	268.37	269.70	BETA(PR) 6	51.128
V 4	296.61	301.03	307.24	309.55	304.39	BETA(PR) 7	19.655
VZ 3	243.55	255.80	263.74	268.36	269.69	V 6	313.19
VZ 4	254.51	264.78	275.85	284.21	288.12	V 7	521.90
V-T-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	276.16
V-T-THETA 4	152.31	140.52	135.21	122.60	98.17	VZ 7	303.70
M 3	0.2192	0.2306	0.2376	0.2418	0.2430	V-T-THETA 6	147.58
M 4	0.2676	0.2716	0.2773	0.2794	0.2747	V-T-THETA 7	423.06
TURN	-30.90	-27.83	-26.11	-23.33	-18.82	V(PR) 6	440.1
(REFER TO 70% & 100% SPEED DATA)						V(PR) 7	324.5
DFAC	0.103	0.103	0.093	0.076	0.053	V-THETA PR6	-342.6
EFPF	0.9134	0.8416	0.8447	0.9286	0.8634	V-THETA PR7	-109.2
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	490.18
DEVN	5.103	6.472	6.391	7.567	10.185	U 7	532.21
						M 6	0.2828
						M 7	0.4662
						M(PR) 6	0.3974
						M(PR) 7	0.2899
						TURN(PR)	31.466
						W	0.0068
PCT SPAN	90	70	50	30	10	DFAC	.5019
DIA 9	26.021	29.683	33.345	37.007	40.669	EFPF	0.9496
BETA 9	53.383	50.694	53.417	49.003	41.684	EFF	0.9484
BETA 10	27.931	26.422	26.224	25.541	26.223	INCIDM	2.528
V 9	511.98	495.12	498.25	513.31	509.15	DEVN	0.855
V 10	294.95	349.24	360.76	409.96	435.48		
VZ 9	299.77	311.48	296.20	336.65	380.18		
VZ 10	257.79	312.15	322.70	369.77	390.59		
V-T-THETA 9	410.93	383.11	400.09	387.42	338.60		
V-T-THETA 10	138.15	155.40	159.41	176.76	192.43		
M 9	0.4570	0.4410	0.4417	0.4543	0.4495		
M 10	0.2597	0.3082	0.3171	0.3603	0.3824		
TURN	25.452	24.272	27.193	23.462	15.461		
W	0.2718	0.1197	0.1465	0.1027	0.1292		
DFAC	0.6210	0.4806	0.4864	0.3923	0.2825		
EFPF	0.6733	0.8138	0.7808	0.7558	0.7778		
INCIDM	3.483	4.594	9.717	6.003	-3.316		
DEVN	10.631	7.922	7.724	7.541	9.123		

Table B-3. Blade Element Data Cruise Configuration (Continued)

80 PERCENT SPEED PT.179				CONFIGURATION		N/√θ		ω√θ/δ			
				CRUISE		4842.		166.1			
INLET GUIDE VANE										ROTOR	
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	36.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	27.767	25.601	24.407	22.037	18.435
BETA 4	30.527	27.592	25.825	23.415	19.640	BETA 7	51.010	49.730	52.719	48.934	42.843
V 3	255.08	269.45	278.54	284.09	285.32	BETA(PT) 6	49.290	55.283	60.124	63.421	67.401
V 4	313.56	322.38	327.52	329.95	317.04	BETA(PT) 7	21.507	36.358	45.415	50.728	54.835
VZ 3	255.07	269.16	278.53	284.08	285.31	V 6	330.69	338.44	339.68	346.09	332.65
VZ 4	270.09	283.99	294.81	302.78	298.60	V 7	523.24	497.98	490.64	495.30	496.52
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	292.55	304.96	309.27	320.79	315.58
V-THETA 4	159.27	149.32	142.67	131.12	106.56	VZ 7	327.56	321.61	297.05	325.33	364.05
M 3	0.2297	0.2428	0.2511	0.2561	0.2573	V-THETA 6	154.06	146.24	140.36	129.86	105.20
M 4	0.2831	0.2912	0.2959	0.2982	0.2863	V-THETA 7	406.69	379.97	390.39	373.43	337.63
TURN	-30.53	-27.59	-25.82	-23.41	-19.64	V(PT) 6	448.6	535.6	620.9	717.0	821.2
(REFER TO 70% & 100% SPEED DATA)											
ω	0.091	0.084	0.082	0.070	0.075	V(PT) 7	353.9	399.7	423.4	514.0	632.1
DFAC	0.8932	0.8930	0.8907	0.9843	0.7083	VTHETA PR6	-340.0	-440.1	-538.4	-641.2	-758.2
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PR7	-129.7	-236.9	-301.5	-397.9	-516.8
INCID/M						U 6	494.06	586.39	678.71	771.04	863.36
DEVM	5.473	6.708	6.675	7.485	9.360	U 7	536.42	616.91	691.92	771.35	854.38
						M 6	0.2989	0.3060	0.3071	0.3130	0.3007
						M 7	0.4683	0.4444	0.4357	0.4391	0.4395
						M(PT) 6	0.4054	0.4842	0.5614	0.6485	0.7422
						M(PT) 7	0.3167	0.3567	0.3760	0.4557	0.5595
						TURN(PT)	27.777	18.903	14.705	12.692	12.566
STATOR B											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 9	26.021	29.683	33.345	37.007	40.869	DFAC	4.276	4.305	4.893	4.348	3.625
BETA 9	49.553	48.400	51.389	46.976	40.500	EFFP	0.9878	1.0215	0.9250	0.9690	1.0229
BETA 10	27.500	27.129	27.202	25.853	26.501	EFF	0.9876	1.0220	0.9231	0.9681	1.0237
V 9	518.19	500.40	494.43	504.97	506.47	INCIDM	0.690	2.783	4.024	3.821	4.201
V 10	373.41	382.24	374.88	423.37	453.97	DEVM	2.707	3.858	3.315	2.128	1.935
VZ 9	330.13	329.68	307.56	344.34	387.66						
VZ 10	328.74	337.16	332.26	380.74	406.25						
V-THETA 9	394.35	374.20	386.34	369.16	325.89						
V-THETA 10	172.42	174.30	171.37	184.62	202.57						
M 9	0.4636	0.4466	0.4391	0.4479	0.4487						
M 10	0.3307	0.3384	0.3303	0.3734	0.4006						
TURN	22.053	21.271	24.187	21.123	13.549						
ω	0.1543	0.1209	0.1866	0.1299	0.1163						
DFAC	0.4358	0.3971	0.4310	0.3316	0.2208						
EFFP	0.7342	0.7638	0.6009	0.5815	0.5549						
INCIDM	-0.347	2.300	7.689	3.976	-4.950						
DEVM	10.200	8.629	8.702	7.853	9.401						

Table B-3. Blade Element Data Cruise Configuration (Continued)

CONFIGURATION						$N/\sqrt{\theta}$	$w\sqrt{\theta}/g$
80PERCENT SPEED PT.180						4829.	176.4
INLET GUIDE VANE							
PCT SPAN	90	70	50	30	10		
DIA 3	21.150	26.050	30.950	35.850	40.750		
BETA 3	0.000	0.000	0.000	0.000	0.000		
BETA 4	31.087	27.876	26.439	23.559	18.269		
V 3	273.48	288.16	297.40	302.65	303.80	BETA(PR) 6	
V 4	338.39	344.86	349.41	349.06	339.23	BETA(PR) 7	
VZ 3	273.46	287.87	297.39	302.64	303.79	V 6	
VZ 4	289.79	303.12	312.85	319.96	322.13	V 7	
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6	
V-THETA 4	174.72	161.24	155.58	139.52	106.34	VZ 7	
M 3	0.2465	0.2599	0.2683	0.2731	0.2742	V-THETA 6	
M 4	0.3059	0.3119	0.3161	0.3158	0.3067	V-THETA 7	
TURN	-31.09	-27.88	-26.44	-23.56	-18.27	V(PR) 6	
	(REFER TO 70% & 100% SPEED DATA)					V(PR) 7	
DFAC	0.090	0.087	0.088	0.078	0.058	VTHETA PR6	
EFFP	0.9295	0.8987	0.9070	0.9738	0.8301	VTHETA PR7	
INCIDIM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	
DEVM	4.913	6.424	6.061	7.341	10.731	U 7	
						M 6	
						M(PR) 6	
						M(PR) 7	
						TURN(PR)	
						\bar{w}	
						DFAC	
						EFFP	
						EFF	
						INCIDIM	
						DEVM	
						\bar{w}	
						DFAC	
						EFFP	
						EFF	
						INCIDIM	
						DEVM	

Table B-3. Blade Element Data Cruise Configuration (Continued)

80PERCENT SPEED PT.174					CONFIGURATION		N/√θ		ω√θ/δ		
					CRUISE		4850.		189.0		
INLET GUIDE VANE										ROTOR	
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	28.035	25.712	24.478	22.189	18.693
BETA 4	31.109	27.957	26.242	23.706	19.921	BETA 7	42.882	42.004	41.703	39.425	34.799
V 3	295.40	310.87	320.48	325.73	326.80	BETA(PR) 6	41.826	49.459	55.156	59.448	64.188
V 4	369.11	373.79	375.61	376.99	368.23	BETA(PR) 7	21.703	35.757	43.628	49.526	53.900
VZ 3	295.38	310.58	320.46	325.72	326.78	V 6	392.83	395.12	394.92	396.82	379.48
VZ 4	316.03	328.41	336.87	345.16	346.20	V 7	552.70	513.12	503.33	501.60	504.26
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	346.67	355.76	359.36	367.41	359.45
V-THETA 4	190.71	175.24	166.08	151.56	125.46	VZ 7	403.07	380.99	375.60	387.39	414.06
M 3	0.2665	0.2806	0.2895	0.2943	0.2953	V-THETA 6	184.64	171.42	163.63	149.87	121.62
M 4	0.3343	0.3386	0.3403	0.3416	0.3335	V-THETA 7	376.11	343.37	334.85	318.55	287.78
TURN	-31.11	-27.96	-26.24	-23.71	-19.92	V(PR) 6	465.3	547.5	629.0	722.8	825.5
(REFER TO 70% & 100% SPEED DATA)											
W	0.081	0.083	0.089	0.076	0.065	V(PR) 7	435.9	469.9	519.2	596.9	702.0
DFAC	0.9360	0.8949	0.8631	0.9615	0.8679	VTHETA PR6	-310.2	-415.9	-516.2	-622.4	-743.2
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PR7	-161.2	-274.6	-358.2	-454.1	-568.0
INCLD/M	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	494.88	587.35	679.83	772.31	864.79
DEVM	4.891	6.343	6.258	7.194	9.079	U 7	537.31	617.93	693.06	772.63	855.80
STATOR B											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC	.2256	.2711	.2912	.2783	.2423
BETA 9	41.238	40.419	40.238	37.721	32.394	EFFP	1.0361	1.0653	1.0465	1.0370	1.0327
BETA 10	28.342	28.414	27.665	26.431	26.566	EFF	1.0367	1.0665	1.0474	1.0378	1.0327
V 9	582.29	521.05	513.60	514.38	502.56	INCLD/M	-6.774	-3.041	-0.944	-0.152	0.988
V 10	501.71	490.39	480.11	496.66	504.42	DEVM	2.903	3.257	1.528	0.926	0.109
VZ 9	408.32	394.11	391.34	406.83	424.26						
VZ 10	438.91	428.24	424.42	444.70	451.04						
V-THETA 9	364.06	337.83	331.77	314.71	269.24						
V-THETA 10	238.18	233.34	222.91	221.07	225.59						
M 9	0.4981	0.4686	0.4608	0.4608	0.4488						
M 10	0.4505	0.4400	0.4296	0.4443	0.4506						
TURN	12.896	12.005	12.574	11.290	5.829						
W	0.2215	0.1491	0.1817	0.1224	0.1493						
DFAC	0.1704	0.1373	0.1561	0.1192	0.0391						
EFFP	-0.0368	-0.1641	-0.2726	-0.7171	5.4352						
INCLD/M	-8.662	-5.681	-3.462	-5.279	-12.606						
DEVM	11.042	9.914	9.165	8.431	9.466						

Table B-3. Blade Element Data Cruise Configuration (Continued)

100 PERCENT SPEED		PI.190		CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/\delta$									
				CRUISE		6045.		175.8									
INLET GUIDE VANE																	
MOTOR																	
PCT SPAN																	
DIA 3																	
BETA 3																	
BETA 4																	
V 3																	
V 4																	
VZ 3																	
VZ 4																	
V-THETA 3																	
V-THETA 4																	
M 3																	
M 4																	
TURN																	
$\bar{\omega}$																	
DFAC																	
EFFP																	
INCIDM																	
DEVN																	
STATOR B																	
PCT SPAN																	
DIA 9																	
BETA 9																	
BETA 10																	
V 9																	
V 10																	
VZ 9																	
VZ 10																	
V-THETA 9																	
V-THETA 10																	
M 9																	
M 10																	
TURN																	
$\bar{\omega}$																	
DFAC																	
EFFP																	
INCIDM																	
DEVN																	

Table B-3. Blade Element Data Cruise Configuration (Continued)

[illegible]

Table B-3. Blade Element Data Cruise Configuration (Continued)

CONFIGURATION						$N/\sqrt{\theta}$	$w\sqrt{\theta}/s$
100 PERCENT SPEED PT. 192			CRUISE			6045.	193.0
INLET GUIDE VANE							
				ROTOR			
STATOR B							
POT SPAN	90	70	50	30	10		
DIA 3	21.150	26.050	30.950	35.850	40.750		
BETA 3	0.000	0.000	0.000	0.000	0.000		
BETA 4	30.800	27.432	25.636	23.143	22.016		
V 3	301.61	316.93	326.99	333.52	335.78		
V 4	361.17	376.67	385.73	390.91	384.40		
VZ 3	301.60	316.54	326.98	333.51	335.77		
VZ 4	310.22	330.66	347.73	359.44	356.36		
V-THETA 3	0.000	0.000	0.000	0.000	0.000		
V-THETA 4	184.93	172.61	166.88	153.64	144.10		
M 3	0.2722	0.2862	0.2954	0.3015	0.3035		
M 4	0.3270	0.3395	0.3497	0.3545	0.3485		
TURN	-30.80	-27.43	-25.64	-23.14	-22.02		
ω	0.0602		0.0289		0.0260		
DFAC	0.117	0.094	0.077	0.059	0.070		
EFFP	0.8848	0.8967	0.9113	0.9870	0.9197		
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200		
DEVN	5.200	6.868	6.864	7.757	6.984		
ROTOR							
POT SPAN	90	70	50	30	10		
DIA 9	26.021	29.683	33.345	37.007	40.669		
BETA 9	55.995	54.101	55.307	51.142	47.305		
BETA 10	30.942	25.716	25.876	26.280	27.306		
V 9	609.09	597.23	606.74	633.43	642.05		
V 10	246.37	413.29	435.31	504.79	548.98		
VZ 9	334.04	347.38	344.05	397.06	435.37		
VZ 10	208.09	368.50	389.58	493.26	452.10		
V-THETA 9	504.93	483.79	498.87	493.26	471.89		
V-THETA 10	126.67	179.33	189.98	223.50	251.84		
M 9	0.5411	0.5287	0.5338	0.5556	0.5616		
M 10	0.2138	0.3611	0.3785	0.4381	0.4762		
TURN	25.054	28.385	29.431	24.862	19.999		
ω	0.5785	0.2120	0.2295	0.1756	0.1547		
DFAC	0.8271	0.5146	0.5045	0.4012	0.3092		
EFFP	0.4312	0.7034	0.6845	0.5870	0.4669		
INCIDM	6.095	8.001	11.607	8.142	2.305		
DEVN	13.642	7.216	7.376	8.280	10.206		

Table B-3. Blade Element Data Cruise Configuration (Continued)

100 PERCENT SPEED		PT. 193		CONFIGURATION		CRUISE		N/√σ		ω√σ/δ		205.	
INLET GUIDE VANE												ROTOR	
PCT SPAN		90	70	50	30	10	PCT SPAN		90	70	50	30	10
DIA 3	21.150	26.050	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.405	40.685	
BETA 3	0.000	0.000	0.000	0.000	0.000	0.000	BETA 6	27.880	25.541	24.392	22.016	20.167	
BETA 4	30.690	27.632	27.632	25.995	23.571	21.180	BETA 7	53.064	50.978	53.452	50.319	45.664	
V 3	323.93	341.57	341.57	352.24	357.94	359.90	RETA(PR) 6	48.544	54.474	59.775	62.746	66.500	
V 4	307.61	409.33	409.33	418.45	418.38	412.32	RETA(PR) 7	10.578	35.539	43.970	42.838	53.443	
VZ 3	323.92	341.19	341.19	352.23	357.93	358.89	V 6	420.16	432.17	438.37	442.68	429.77	
VZ 4	341.92	360.61	360.61	376.09	383.45	384.47	V 7	661.08	627.88	626.95	630.97	643.43	
V-THETA 3	0.000	0.000	0.000	0.000	0.000	0.000	VZ 6	371.31	389.65	399.17	410.37	402.49	
V-THETA 4	202.94	189.84	189.84	167.30	148.97	148.97	VZ 7	394.81	394.97	373.16	412.81	449.66	
M 3	0.2926	0.3089	0.3187	0.3249	0.3249	0.3249	V-THETA 6	196.47	186.33	181.03	165.05	147.82	
M 4	0.3604	0.3717	0.3802	0.3801	0.3745	0.3745	V-THETA 7	528.41	487.80	503.67	495.60	460.22	
TURN	-30.69	-27.63	-26.00	-23.57	-21.19	-21.19	V(PR) 6	560.9	670.7	776.8	846.1	1013.4	
ω	0.0634	0.0299	0.074	0.066	0.0390	0.0390	V(PR) 7	421.5	495.8	518.7	524.7	755.7	
DFAC	0.084	0.083	0.074	0.066	0.058	0.058	VTHETA PR6	-420.3	-545.7	-666.3	-796.7	-937.7	
EFFP	0.9302	0.9180	0.9372	0.9356	0.9064	0.9064	VTHETA PR7	-141.3	-292.4	-360.2	-477.4	-676.4	
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	616.81	732.07	847.34	962.60	1077.87	
DEVM	5.310	6.668	6.505	7.329	7.820	7.820	U 7	669.69	770.19	863.82	963.07	1066.66	
STATOR B													
PCT SPAN		90	70	50	30	10	PCT SPAN		90	70	50	30	10
DIA 9	26.021	29.683	29.683	33.345	37.007	40.669	DIA 6	23.385	27.755	32.125	36.405	40.685	
BETA 9	52.180	49.590	49.590	52.020	48.220	43.884	BETA 6	27.880	25.541	24.392	22.016	20.167	
BETA 10	27.948	27.129	27.129	27.075	25.991	28.260	BETA 7	53.064	50.978	53.452	50.319	45.664	
V 9	648.83	629.42	629.42	633.24	645.58	654.99	V 6	420.16	432.17	438.37	442.68	429.77	
V 10	390.48	446.54	446.54	452.50	516.61	555.35	V 7	661.08	627.88	626.95	630.97	643.43	
VZ 9	390.36	405.09	405.09	388.73	430.03	472.02	VZ 6	371.31	389.65	399.17	410.37	402.49	
VZ 10	341.57	393.96	393.96	401.91	464.21	489.08	VZ 7	394.81	394.97	373.16	412.81	449.66	
V-THETA 9	512.53	479.25	479.25	499.14	481.41	454.04	V-THETA 6	196.47	186.33	181.03	165.05	147.82	
V-THETA 10	183.01	203.42	203.42	205.61	226.39	262.94	V-THETA 7	528.41	487.80	503.67	495.60	460.22	
M 9	0.5791	0.5603	0.5603	0.5597	0.5691	0.5758	M 6	0.3818	0.3931	0.3989	0.4029	0.3890	
M 10	0.3413	0.3916	0.3916	0.3943	0.4504	0.4838	M 7	0.5908	0.5588	0.5541	0.5556	0.5651	
TURN	24.232	22.461	22.461	24.995	22.229	15.624	M(PR) 6	0.5097	0.6100	1.7068	0.8156	0.9214	
ω	0.2915	0.1615	0.2196	0.1214	0.1306	0.1306	M(PR) 7	0.3768	0.4324	0.4585	0.5501	0.6631	
DFAC	0.5856	0.4674	0.4873	0.3836	0.2922	0.2922	TURN(PR) 1	28.960	18.915	15.100	12.907	13.156	
EFFP	0.6554	0.7445	0.6506	0.7139	0.6848	0.6848	ω	-0.0613	-0.0000	0.0368	0.0523	0.0584	
INCIDM	2.280	3.490	8.320	5.220	-1.116	-1.116	DFAC	0.7755	0.5088	0.5088	0.4621	0.3989	
DEVM	10.648	8.629	8.525	7.991	11.160	11.160	EFFP	0.9775	1.0446	0.9627	0.9796	1.0081	
							FFF	0.9767	1.0463	0.9767	0.9767	1.0085	
							INCIDM	-0.056	1.974	2.975	3.146	3.399	
							DEVM	0.778	3.039	1.870	1.238	0.543	

Table B-3. Blade Element Data Cruise Configuration (Continued)

100 PERCENT SPEED PT. 194				CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/\delta$	
				CRUISE		6045.		217.4	
INLET GUIDE VANE									
ROTOR									
POT SPAN									
DIA 6									
BETA 6									
BETA 7									
BETA(PR) 6									
BETA(PR) 7									
V 6									
V 7									
VZ 6									
VZ 7									
V-THETA 6									
V-THETA 7									
M 6									
M 7									
TURN									
ω									
DFAC									
EFFP									
INCIDM									
DEVM									
STATOR B									
POT SPAN									
DIA 9									
BETA 9									
BETA 10									
V 9									
V 10									
VZ 9									
VZ 10									
V-THETA 9									
V-THETA 10									
M 9									
M 10									
TURN									
ω									
DFAC									
EFFP									
INCIDM									
DEVM									

Table B-3. Blade Element Data Cruise Configuration (Continued)

100PERCENT SPEED					PT.189		CONFIGURATION		CRUISE		$N/\sqrt{\sigma}$		6055.		$\omega\sqrt{\sigma}/g$		232.5	
INLET GUIDE VANE																		
ROTOR																		
PCT SPAN		90	70	50	30	10	PCT SPAN		90	70	50	30	10					
DIA 3		21.150	26.050	30.950	35.850	40.750	DIA 6		23.385	27.755	32.125	36.495	40.865					
BETA 3		0.000	0.000	0.000	0.000	0.000	BETA 6		27.653	25.450	24.218	21.962	18.406					
BETA 4		30.687	27.709	26.007	23.428	19.704	BETA 7		43.384	43.011	43.180	41.269	37.310					
V 3		370.78	391.33	404.25	411.12	412.31	BETA(PR) 6		41.504	48.547	54.091	58.647	63.178					
V 4		465.50	476.56	482.51	482.42	473.06	BETA(PR) 7		21.753	35.331	43.086	49.834	52.986					
VZ 3		370.75	390.92	404.23	411.10	412.29	V 6		495.16	505.26	508.40	508.52	492.46					
VZ 4		400.31	419.59	433.63	442.63	445.36	V 7		686.68	642.64	633.26	622.28	643.20					
V-THETA 3		0.00	0.00	0.00	0.00	0.00	VZ 6		438.50	455.91	463.57	471.58	467.27					
V-THETA 4		237.57	221.59	211.57	191.81	159.50	VZ 7		496.59	469.49	461.54	467.63	511.56					
M 3		0.3359	0.3549	0.3670	0.3734	0.3745	V-THETA 6		229.81	217.12	208.55	190.18	155.49					
M 4		0.4244	0.4349	0.4405	0.4404	0.4316	V-THETA 7		471.67	438.37	433.33	410.45	389.86					
TURN		-30.69	-27.71	-26.01	-23.43	-19.70	V(PR) 6		585.6	688.9	790.5	906.4	1035.6					
ω		0.0519	0.0230	0.0230	0.0320	0.0320	V(PR) 7		537.3	576.0	632.3	725.1	849.8					
DFAC		0.073	0.069	0.070	0.061	0.046	VTHETA PR6		-388.0	-516.2	-640.2	-774.0	-924.2					
EFFP		0.9021	0.8948	0.9049	0.9616	0.8298	VTHETA PR7		-199.1	-333.1	-431.9	-554.1	-678.6					
INCIDM		-20.200	-20.200	-20.200	-20.200	-20.200	U 6		617.83	733.28	848.74	964.19	1079.65					
DEVM		5.313	6.591	6.493	7.472	9.296	U 7		670.80	771.46	865.25	964.59	1068.42					
							M 6		0.4525	0.4621	0.4652	0.4653	0.4500					
							M 7		0.6212	0.5784	0.5675	0.5557	0.5722					
							M(PR) 6		0.5352	0.6301	0.7232	0.8293	0.9462					
							M(PR) 7		0.4861	0.5184	0.5666	0.6476	0.7560					
							TURN(PR)		19.746	13.197	10.999	8.812	10.192					
							ω		0.0232	0.0838	0.0679	0.0575	0.0578					
							DFAC		.2453	.2960	.3219	.3085	.2849					
							EFFP		1.0492	1.0578	1.0238	1.0111	0.9643					
							EFF		1.0506	1.0595	1.0246	1.0115	0.9630					
							INCIDM		-7.096	-3.953	-2.009	-0.953	-0.022					
							DEVM		2.953	2.831	0.986	1.234	0.086					
STATOR B																		
PCT SPAN		90	70	50	30	10	PCT SPAN		90	70	50	30	10					
DIA 9		26.021	29.683	33.345	37.007	40.669	DIA 6		23.385	27.755	32.125	36.495	40.865					
BETA 9		41.783	41.337	41.751	39.515	31.108	BETA 6		27.653	25.450	24.218	21.962	18.406					
BETA 10		28.417	28.667	27.851	25.906	28.591	BETA 7		43.384	43.011	43.180	41.269	37.310					
V 9		685.92	652.94	645.07	639.83	645.02	BETA(PR) 6		41.504	48.547	54.091	58.647	63.178					
V 10		613.98	604.97	593.75	612.97	636.40	BETA(PR) 7		21.753	35.331	43.086	49.834	52.986					
VZ 9		502.64	486.86	480.25	493.53	552.12	V 6		495.16	505.26	508.40	508.52	492.46					
VZ 10		536.57	526.78	523.81	551.28	558.68	V 7		686.68	642.64	633.26	622.28	643.20					
V-THETA 9		457.03	431.26	425.55	407.11	333.25	VZ 6		438.50	455.91	463.57	471.58	467.27					
V-THETA 10		292.18	290.21	277.39	267.80	304.55	VZ 7		496.59	469.49	461.54	467.63	511.56					
M 9		0.6205	0.5882	0.5786	0.5722	0.5746	V-THETA 6		229.81	217.12	208.55	190.18	155.49					
M 10		0.5512	0.5424	0.5299	0.5468	0.5659	V-THETA 7		471.67	438.37	433.33	410.45	389.86					
TURN		13.366	12.670	13.900	13.609	2.517	V(PR) 6		585.6	688.9	790.5	906.4	1035.6					
ω		0.2122	0.1591	0.1957	0.1351	0.1260	V(PR) 7		537.3	576.0	632.3	725.1	849.8					
DFAC		0.1885	0.1583	0.1810	0.1433	0.0361	VTHETA PR6		-388.0	-516.2	-640.2	-774.0	-924.2					
EFFP		0.0376	0.0325	-0.1113	-0.4927	-0.8383	VTHETA PR7		-199.1	-333.1	-431.9	-554.1	-678.6					
INCIDM		-8.117	-4.763	-1.949	-3.485	-13.892	U 6		617.83	733.28	848.74	964.19	1079.65					
DEVM		11.117	10.167	9.351	7.906	11.491	U 7		670.80	771.46	865.25	964.59	1068.42					

Table B-4. Blade Element Data Intermediate Configuration (Continued)

50 PERCENT SPEED P1. 215				CONFIGURATION		N/δ		ω√σ / δ			
				INTERMEDIATE		3020.		89.71			
INLET GUIDE VANE											
ROTOR											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	13.502	13.764	13.216	9.699	6.830
BETA 4	15.052	14.771	13.883	10.292	7.446	BETA 7	52.735	54.316	55.432	51.362	48.257
V 3	135.77	142.77	146.86	149.55	150.34	BETA(PR) 6	61.691	64.466	67.552	69.558	72.049
V 4	140.12	152.54	158.51	163.00	160.34	BETA(PR) 7	25.727	43.810	52.743	55.167	59.448
VZ 3	135.77	142.53	146.86	149.55	150.33	V 6	151.20	161.19	163.81	170.97	169.14
VZ 4	135.25	146.84	153.85	160.43	158.95	V 7	307.62	280.48	274.98	286.65	284.35
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6	146.97	156.40	159.42	168.51	167.94
V-THETA 4	56.39	38.89	38.03	29.13	20.76	VZ 7	184.97	163.39	155.90	178.94	189.31
M 3	0.1218	0.1261	0.1318	0.1342	0.1345	V-THETA 6	35.30	38.35	37.45	28.80	20.11
M 4	0.1257	0.1365	0.1423	0.1484	0.1435	V-THETA 7	244.82	227.82	226.43	223.90	212.16
TURN	-15.05	-14.77	-13.86	-10.25	-7.45	V(PR) 6	309.9	362.9	417.5	482.5	544.9
(REFER TO 70% & 100% SPEED DATA)						V(PR) 7	206.8	226.7	257.7	313.3	372.4
ω	0.106	0.070	0.051	0.038	0.005	VTHETA PR6	-272.8	-327.4	-385.9	-452.1	-518.4
DFAC	0.8572	1.0482	0.8412	0.9682	0.7647	VTHETA PR7	-89.8	-157.0	-205.1	-257.2	-320.7
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	308.15	365.73	423.32	480.90	538.49
INCIDM	5.648	4.229	3.317	5.308	7.854	U 7	334.57	394.78	431.55	481.10	532.89
DEVM						M 6	0.1357	0.1447	0.1471	0.1535	0.1519
STATOR B.						M 7	0.2745	0.2499	0.2446	0.2546	0.2522
PCT SPAN	90	70	50	30	10	M(PR) 6	0.2781	0.3257	0.3748	0.4332	0.4892
DIA 9	26.021	25.683	33.345	37.007	40.565	M(PR) 7	0.1845	0.2020	0.2292	0.2783	0.3304
BETA 9	52.366	53.407	54.154	49.489	45.857	TURN(PR)	35.955	20.633	14.802	14.389	12.602
BETA 10	21.572	18.535	15.440	14.703	11.737	ω	0.1229	0.2067	0.2780	0.2199	0.2665
V 9	259.70	275.36	276.74	492.72	293.52	DFAC	-5839	.5828	.5739	.5312	.4818
V 10	151.84	185.60	183.52	235.36	250.54	EFFP	1.0038	0.9792	0.9283	0.9031	0.8774
VZ 9	160.36	165.51	161.56	189.99	204.42	EFF	1.0038	0.9792	0.9274	0.9018	0.8757
V-THETA 9	237.34	224.29	224.34	222.55	210.63	INCIDM	13.019	11.966	11.422	9.958	8.849
V-THETA 10	55.83	59.02	48.86	59.74	50.57	DEVM	6.927	11.310	10.643	6.567	6.588
M 9	0.2674	0.2489	0.2462	0.2600	0.2605						
M 10	0.1546	0.1645	0.1625	0.2037	0.2219						
TURN	30.794	34.872	36.716	34.736	34.120						
ω	0.3605	0.1866	0.2603	0.1306	0.1152						
DFAC	0.7204	0.5740	0.6159	0.4552	0.4120						
EFFP	0.4501	0.2350	0.2155	0.6027	0.4558						
INCIDM	6.966	11.807	14.959	10.989	5.357						
DEVM	17.227	13.023	9.940	9.703	7.637						

Table B-4. Blade Element Data Intermediate Configuration (Continued)

SUPERCENT SPEED				PT. 216	INLET GUIDE VANE		CONFIGURATION	$N/\sqrt{\sigma}$	3027.	$w\sqrt{\sigma}/s$	37.2	ROTOR			
				70	50	30	10					70	50	30	10
PCT SPAN															
DIA 3				21.150	26.050	35.850	40.750					27.755	32.175	36.495	40.905
BETA 3				0.000	0.000	0.000	0.000					13.502	12.719	9.091	6.359
BETA 4				15.387	14.925	9.762	6.804					52.785	55.697	47.303	45.780
V 3				147.69	154.82	162.35	162.70					65.651	65.651	65.651	70.085
V 4				163.71	164.55	174.47	173.55					40.661	40.152	32.412	57.011
VZ 3				147.68	154.68	162.34	162.69					173.52	170.44	134.81	190.024
VZ 4				157.84	150.11	171.93	172.33					203.00	207.14	238.42	308.022
V-THETA 3				0.00	0.00	0.00	0.00					149.27	174.02	142.43	170.13
V-THETA 4				43.44	42.36	29.58	20.56					177.11	191.02	202.33	207.95
M 3				0.1325	0.1390	0.1457	0.1460					42.14	38.90	29.27	10.94
M 4				0.1470	0.1477	0.1567	0.1558					233.41	222.14	219.33	213.75
TURN				-15.39	-14.93	-9.76	-6.80					366.0	422.5	485.2	560.8
ω				0.042	0.076	0.017	-0.004					233.7	278.1	331.4	301.0
DFAC				4.3323	0.9059	0.8476	0.7223					-324.9	-305.0	-452.3	-320.8
EFFP				-20.200	-20.200	-20.200	-20.200					-152.3	-210.4	-267.3	-320.4
INCIDM				5.313	4.075	5.838	8.496					308.86	424.30	482.02	530.76
DEVM												335.35	432.55	482.22	534.12
												0.1547	0.1502	0.1660	0.1610
												0.2612	0.2557	0.2653	0.2644
												0.3286	0.3705	0.4395	0.4037
												0.1931	0.2475	0.2950	0.3383
												21.034	14.522	15.638	13.072
ω												0.1252	0.1897	0.1587	0.1919
PCT SPAN												0.5698	0.5206	0.4942	0.4706
DIA 9				26.021	29.683	37.007	40.669					0.0190	0.0183	0.0183	0.0183
BETA 9				51.209	51.492	45.708	44.537					0.0100	0.0100	0.0100	0.0100
BETA 10				22.584	17.216	14.820	9.562					10.115	9.576	8.451	7.785
V 9				312.79	292.95	305.20	304.75								
V 10				182.73	207.39	238.12	266.30								
VZ 9				192.34	181.05	213.07	217.20								
VZ 10				167.11	196.34	230.11	262.57								
V-THETA 9				243.80	229.24	218.46	213.74								
V-THETA 10				70.18	61.38	60.91	44.24								
M 9				0.2792	0.2611	0.2714	0.2705								
M 10				0.1623	0.1842	0.2112	0.2360								
TURN				28.624	34.276	30.887	34.976								
ω				0.2858	0.1323	0.1623	0.0715					8.161	7.052	3.812	4.113
DFAC				0.6229	0.5270	0.4604	0.3976								
EFFP				0.6030	0.7519	0.6276	0.7923								
INCIDM				5.809	9.892	7.208	4.037								
DEVM				18.248	11.761	9.820	5.462								

Table B-4. Blade Element Data Intermediate Configuration (Continued)

		CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/s$				
SUPERCRANT SPEED		PT.217	INTERMEDIATE	3037.		113.0				
INLET GUIDE VANE										
PCT SPAN	50	70	50	30	10	PCT SPAN	70	50	30	10
DIA 3	21.150	26.050	36.950	35.850	43.750	DIA 6	23.385	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	14.333	12.586	9.030	6.623
BETA 4	15.814	14.632	13.608	9.668	0.965	BETA 7	46.468	44.236	39.809	38.140
V 3	170.63	150.48	186.54	189.77	190.13	BETA(PR) 6	54.260	61.451	64.580	68.189
V 4	182.40	156.28	200.90	205.25	201.09	PETA(PR) 7	22.894	46.615	51.777	56.118
VZ 3	176.62	180.25	186.53	189.76	190.12	V 6	194.46	208.41	216.39	208.48
VZ 4	175.45	188.84	195.31	202.33	195.61	V 7	331.21	298.11	299.46	299.57
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	188.36	202.44	213.70	207.09
V-THETA 4	45.71	45.56	47.20	34.47	24.39	VZ 7	226.96	217.68	230.00	235.61
M 3	0.1532	0.1621	0.1676	0.1705	0.1708	V-THETA 6	48.14	46.12	33.96	24.05
M 4	0.1638	0.1764	0.1806	0.1845	0.1807	V-THETA 7	249.12	212.17	191.72	185.01
TURN	-15.81	-14.63	-13.61	-9.67	-6.96	V(PR) 6	322.5	378.0	497.9	557.4
(REFER TO 70% & 100% SPEED DATA)								432.1	497.9	557.4
DFAC	0.081	0.052	0.050	0.010	0.006	V(PR) 7	247.6	317.5	371.8	422.6
EFFP	0.7571	1.0127	0.8173	0.9329	0.6683	VTHETA PR6	-261.7	-379.5	-449.6	-517.5
INCIDIM	-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PR7	-96.3	-230.7	-292.1	-350.9
DEVN	4.886	4.368	3.592	5.912	8.335	U 6	309.88	425.70	483.61	541.52
STATOR B								367.79	483.61	541.52
PCT SPAN	50	70	50	30	10	U 7	336.45	433.04	483.81	535.89
DIA 9	26.021	25.683	35.345	37.007	43.609	M 6	0.1747	0.1873	0.1946	0.1874
BETA 9	45.218	42.880	41.437	38.393	37.153	M 7	0.2962	0.2717	0.2669	0.2566
BETA 10	17.642	15.060	14.080	13.932	11.570	M(PR) 6	0.2893	0.3397	0.4476	0.5010
V 9	325.61	306.08	303.04	306.72	336.80	M(PR) 7	0.2214	0.2495	0.3314	0.3761
V 10	243.22	244.42	249.86	270.14	235.72	TURN(PR)	31.359	14.831	12.801	12.071
VZ 9	225.51	222.65	226.65	240.34	244.52	DFAC	0.0714	0.0838	0.0999	0.1051
VZ 10	225.98	234.14	241.75	262.06	279.48	EFFP	4547	4191	3947	3820
V-THETA 9	251.12	266.28	260.55	190.49	185.37	EFF	0.9587	0.9885	0.9492	0.8990
V-THETA 10	73.71	63.51	60.79	65.27	59.26	EFF	0.9582	0.9885	0.9487	0.8978
M 9	0.2311	0.2734	0.2704	0.2734	0.2731	INCIDIM	5.660	5.351	4.980	4.989
M 10	0.2160	0.2178	0.2225	0.2434	0.2541	DEVN	4.094	4.515	3.177	3.288
TURN	27.576	27.820	27.356	24.412	25.193					
DFAC	0.1375	0.0814	0.0843	0.2490	-0.0076					
EFFP	0.4229	0.3551	0.3761	0.3096	0.2596					
INCIDIM	0.7115	0.7642	0.7444	0.9252	1.2011					
DEVN	-0.182	1.280	2.237	-0.107	-3.337					
	13.342	9.560	8.580	8.982	7.870					

Table B-4. Blade Element Data Intermediate Configuration (Continued)

SUPERCENT SPEED		CONFIGURATION		INLET GUIDE VANE		ROTOR		N/√σ	
PT.218		INTERMEDIATE						121.6	
INLET GUIDE VANE									
(REFER TO 70% & 10% SPEED DATA)									
STATOR B									
PCT SPAN									
90	70	50	30	10	PCT SPAN	90	70	50	30
21.150	26.050	36.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495
1.100	1.000	0.900	0.800	0.700	BETA 6	14.365	13.479	12.237	8.806
15.822	15.574	15.271	9.392	5.623	BETA 7	50.211	39.381	37.241	33.882
183.62	154.31	201.02	204.59	204.59	BETA(PR) 6	50.777	54.528	58.769	66.357
197.99	212.30	216.77	220.40	215.55	BETA(PR) 7	24.003	38.118	45.656	51.096
183.60	194.08	201.01	204.50	204.50	V 6	211.52	225.95	228.65	232.27
190.43	204.28	210.98	217.44	214.12	V 7	333.13	305.35	299.35	298.68
0.000	0.000	0.000	0.000	0.000	VZ 5	204.86	219.17	223.39	229.47
54.15	55.42	49.76	35.97	24.66	VZ 7	251.74	235.80	238.19	247.92
0.1645	0.1746	0.1839	0.1839	0.1842	V-THETA 6	52.48	52.58	48.46	35.55
0.1779	0.1909	0.1944	0.1902	0.1930	V-THETA 7	215.84	193.73	181.15	146.51
-15.00	-16.57	-13.27	-9.54	-6.62	V(PR) 6	324.0	377.8	430.0	494.5
0.075	0.047	0.040	0.011	0.005	V(PR) 7	270.4	300.0	340.0	447.1
0.250	1.0062	0.8423	0.9535	0.6592	VTHETA PR6	-251.0	-307.6	-368.4	-438.0
-20.200	-20.200	-20.200	-20.200	-20.200	VTHETA PR7	-112.6	-185.2	-243.8	-307.3
4.817	4.426	3.929	6.208	8.677	U 6	303.46	360.16	416.87	473.58
					U 7	329.47	378.91	424.98	473.77
					M 6	0.1072	0.2029	0.2057	0.2089
					M 7	0.2984	0.2732	0.2676	0.2668
					M(PR) 6	0.2413	0.3398	0.3876	0.4449
					M(PR) 7	0.2480	0.2684	0.3048	0.3527
					TURN(PR)	26.763	16.386	13.108	11.254
					Σ	0.0360	0.0354	0.0346	0.0659
					DFAC	0.3364	0.3556	0.3393	0.3038
					EFFP	1.0491	1.0650	1.0782	0.9938
					EFF	1.0496	1.0656	1.0790	0.9938
					INCIDM	2.177	2.208	2.669	3.157
					DEVN	5.208	5.618	3.556	2.496
					Σ				2.551
					DFAC				
					EFFP				
					INCIDM				
					DEVN				
					Σ				
					DFAC				
					EFFP				
					INCIDM				
					DEVN				
					Σ				

Table B-4. Blade Element Data Intermediate Configuration (Continued)

CONFIGURATION					$M/\sqrt{\sigma}$	ROTOR					
INTERMEDIATE					3017.	138.4					
INLET GUIDE VANE											
					10						
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	13.820	13.159	12.001	9.093	5.844
BETA 4	15.396	14.370	13.033	9.668	6.298	BETA 7	34.954	33.404	32.142	28.365	23.002
V 3	211.01	222.89	229.99	233.65	234.02	BETA(PR) 6	46.307	50.430	55.144	59.285	63.780
V 4	227.86	242.60	248.65	251.36	246.56	BETA(PR) 7	21.643	24.641	43.287	49.152	55.707
VZ 3	210.98	222.63	229.97	235.63	234.01	V 6	245.29	260.04	262.18	263.99	263.55
VZ 4	219.68	233.86	242.23	247.78	245.07	V 7	372.15	340.98	324.26	321.97	322.51
V-THETA 3	0.00	0.00	0.00	0.00	0.00	V7 6	238.13	253.03	256.32	260.66	262.23
V-THETA 4	60.50	60.21	56.07	42.21	27.05	V7 7	303.71	284.30	274.43	283.27	264.44
M 3	0.1897	0.2005	0.2069	0.2102	0.2106	V-THETA 6	58.59	59.19	54.87	41.70	25.92
M 4	0.2050	0.2183	0.2238	0.2263	0.2219	V-THETA 7	213.21	187.72	172.51	152.97	131.14
TURN	-15.40	-14.37	-13.03	-9.67	-6.30	V(PR) 6	344.9	397.3	448.5	510.3	570.0
(REFER TO 70% & 100% SPEED DATA)						V(PR) 7	328.2	346.0	377.2	433.2	497.0
DFAC	0.067	0.049	0.042	0.015	0.004	VTHETA PR6	-249.2	-306.2	-348.0	-438.7	-512.1
EFFP	0.8879	0.9432	0.8516	0.9386	0.7475	VTHETA PR7	-121.0	-196.7	-259.6	-327.7	-401.2
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	307.84	345.37	422.00	490.43	537.05
DEVM	5.304	4.630	4.167	5.932	9.002	U 7	334.24	384.39	431.12	490.62	532.36
STATOR B						M 6	0.2208	0.2342	0.2361	0.2378	0.2383
PCT SPAN	90	70	50	30	10	M 7	0.3343	0.3058	0.2905	0.2884	0.2887
DIA 9	26.021	29.683	33.345	37.007	40.669	M(PR) 6	0.3103	0.3578	0.4040	0.4597	0.5140
BETA 9	33.385	31.906	30.921	27.211	23.375	M(PR) 7	0.2947	0.3103	0.3379	0.3679	0.4456
BETA 10	15.756	14.730	14.113	13.465	13.356	TURN(PR)	24.657	15.748	11.552	10.132	10.072
V 9	371.28	348.29	330.41	330.48	327.75	W	0.0326	0.0388	0.0508	0.0611	0.0910
V 10	351.45	333.98	320.69	321.52	326.62	DFAC	.2175	.2592	.2703	.2484	.2144
VZ 9	305.36	293.96	283.00	293.88	300.81	EFFP	0.9642	0.8800	0.8301	0.9415	0.9714
VZ 10	336.36	321.03	310.53	312.66	317.71	EFF	0.9639	0.9450	0.9347	0.9410	0.9711
V-THETA 9	204.30	184.08	169.78	151.12	130.04	INCIDM	-2.293	-2.070	-0.956	-0.315	1.580
V-THETA 10	95.43	84.92	78.20	74.87	75.45	DEVM	2/843	2.141	1.187	0.552	0.807
M 9	0.3335	0.3125	0.2961	0.2961	0.2935						
M 10	0.3153	0.2994	0.2873	0.2879	0.2924						
TURN	17.628	17.176	16.807	13.746	10.020						
W	0.0656	0.0445	0.0277	0.0667	0.0915						
DFAC	0.1604	0.1567	0.1506	0.1347	0.0854						
EFFP	0.5142	0.5365	0.5276	-0.1006	-8.9810						
INCIDM	-12.015	-9.694	-8.279	-11.289	-17.125						
DEVM	11.456	9.230	8.613	8.465	9.256						

Table B-4. Blade Element Data Intermediate Configuration (Continued)

70PERCENT SPEED		PT.209	CONFIGURATION		$N/\sqrt{\sigma}$	$\omega\sqrt{\sigma}/d$
			INTERMEDIATE		4223.	129.8
INLET GUIDE VANE						
ROTOR						
PCT SPAN	90	70	50	30	10	PCT SPAN
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6
BETA 4	15.450	14.805	13.770	9.872	6.553	BETA 7
V 3	198.07	208.47	214.94	218.65	219.27	BETA(PR) 6
V 4	205.42	223.38	232.27	238.25	235.88	BETA(PR) 7
VZ 3	198.06	208.12	214.94	218.65	219.26	V 6
VZ 4	197.97	214.93	225.54	234.70	234.34	V 7
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6
V-THETA 4	54.72	57.08	55.28	40.84	26.92	VZ 7
M 3	0.1780	0.1874	0.1933	0.1966	0.1972	V-THETA 6
M 4	0.1846	0.2009	0.2090	0.2144	0.2122	V-THETA 7
TURN	-15.45	-14.80	-13.77	-9.87	-6.55	V(PR) 6
$\bar{\omega}$	0.0250	0.0240	0.049	0.004	0.0200	V(PR) 7
DFAC	0.105	0.067	0.049	0.004	-0.014	VTHETA PR6
EFFP	0.8255	1.0221	0.9021	0.9789	0.8144	VTHETA PR7
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6
DEVM	5.250	4.195	3.430	5.728	8.747	U 7
STATOR B						
PCT SPAN	90	70	50	30	10	$\bar{\omega}$
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC
BETA 9	52.409	53.685	52.804	49.016	46.602	EFFP
BETA 10	23.853	16.868	14.378	13.971	10.724	EFF
V 9	419.38	388.73	394.27	416.16	427.50	INCIDM
V 10	211.81	264.72	264.12	310.97	360.01	DEVM
ROTOR						
PCT SPAN	90	70	50	30	10	$\bar{\omega}$
DIA 6	23.385	27.755	32.125	36.495	40.865	DIA 9
BETA 6	13.898	13.794	12.978	9.258	5.984	BETA 9
BETA 7	52.725	54.769	54.129	50.530	48.622	BETA 10
BETA(PR) 6	60.428	63.345	66.316	68.571	71.327	V 9
BETA(PR) 7	25.565	44.342	51.359	54.033	57.730	V 10
V 6	220.94	235.55	247.07	251.36	247.11	VZ 9
V 7	431.01	389.72	391.02	408.19	414.62	VZ 10
VZ 6	214.40	228.52	235.91	248.05	245.70	V-THETA 9
VZ 7	259.22	224.51	228.95	259.41	274.07	V-THETA 10
V-THETA 6	53.07	56.16	54.36	40.44	25.77	M 9
V-THETA 7	342.97	316.33	316.86	315.11	311.12	M 10
V(PR) 6	434.5	509.5	587.1	679.0	767.6	TURN
V(PR) 7	289.4	314.4	366.9	441.8	513.3	$\bar{\omega}$
VTHETA PR6	-377.8	-455.3	-537.6	-632.0	-727.2	DFAC
VTHETA PR7	-124.9	-219.7	-286.6	-357.6	-434.7	EFFP
U 6	430.90	511.42	591.95	672.47	752.99	INCIDM
U 7	467.84	538.05	603.46	672.74	745.14	DEVM
M 6	0.1987	0.2119	0.2179	0.2263	0.2225	$\bar{\omega}$
M 7	0.3834	0.3455	0.3457	0.3599	0.3641	DFAC
M(PR) 6	0.3907	0.4584	0.5293	0.6113	0.6917	EFFP
M(PR) 7	0.2574	0.2787	0.3244	0.3895	0.4500	INCIDM
TURN(PR)	34.853	18.980	14.949	14.531	13.596	DEVM
$\bar{\omega}$	0.0863	0.1686	0.2466	0.2125	0.2588	$\bar{\omega}$
DFAC	.5821	.5875	.5639	.5299	.5055	DFAC
EFFP	1.0037	0.9638	0.8353	0.9048	0.8524	EFFP
EFF	1.0038	0.9631	0.9338	0.9023	0.8492	INCIDM
INCIDM	11.828	10.845	10.216	8.991	8.126	DEVM
DEVM	6.765	11.812	9.259	5.438	4.830	$\bar{\omega}$

$$w\sqrt{\theta}/\delta$$

142.0

σ/\sqrt{N}
4246.

CONFIRMATION
INTERMEDIATE

70 PERCENT SPEED PT. 210

ROTOR

INLET GUIDE VANE

PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	6.000	6.000	0.000	0.000	0.000	BETA 6	14.548	13.565	12.556	9.306	5.898
BETA 4	16.104	14.628	13.501	9.995	6.416	BETA 7	49.874	50.049	48.843	45.435	41.430
V 3	217.55	229.42	236.60	240.10	239.43	BETA(PR) 6	57.508	60.766	63.935	66.478	69.921
V 4	228.53	246.09	254.85	261.25	259.69	BETA(PR) 7	25.997	42.107	49.269	52.435	55.172
VZ 3	217.55	229.06	236.59	240.09	239.42	V 6	244.71	261.04	269.04	278.40	268.09
VZ 4	219.54	236.92	247.74	257.25	258.06	V 7	435.99	401.63	399.91	416.30	430.74
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6	236.78	253.52	262.52	274.70	266.67
V-THETA 4	63.39	62.15	59.50	45.35	29.02	VZ 7	279.15	257.57	263.00	292.06	322.95
M 3	0.1956	0.2064	0.2129	0.2161	0.2155	V-THETA 6	61.47	61.23	58.49	45.02	27.55
M 4	0.2056	0.2215	0.2295	0.2353	0.2339	V-THETA 7	333.37	307.89	301.09	296.59	285.02
TURN	-16.10	-14.63	-13.50	-10.00	-6.42	V(PR) 6	440.8	519.2	597.5	688.3	776.8
W	0.0220		0.0226		0.0220	V(PR) 7	312.6	347.6	403.4	479.2	565.5
DFAC	0.099	0.065	0.050	0.007	-0.024	VTHETA PR6	-371.8	-453.0	-536.7	-631.1	-729.5
EFFP	0.8201	0.9621	0.8723	0.9802	0.8045	VTHETA PR7	-137.0	-233.1	-305.7	-379.8	-464.2
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	433.25	514.21	595.17	676.13	757.09
DEVM	4.596	4.372	3.699	5.605	8.884	U 7	470.39	540.98	606.75	676.41	749.22
						M 6	0.2203	0.2351	0.2424	0.2509	0.2415
						M 7	0.3884	0.3567	0.3544	0.3680	0.3799
						M(PR) 6	0.3968	0.4677	0.5363	0.6204	0.6998
						M(PR) 7	0.2785	0.3088	0.3575	0.4236	0.4988
						TURN(PR)	31.501	18.636	14.657	14.041	14.749

[illegible]

Table B-4. Blade Element Data Intermediate Configuration (Continued)

70PERCENT SPEED PT.211				CONFIGURATION		$N/\sqrt{\theta}$		$\omega\sqrt{\theta}/\delta$	
				INTERMEDIATE		4227.		153.5	
INLET GUIDE VANE									
ROTOR									
PCT SPAN									
DIA 6									
BETA 6									
BETA 7									
BETA(2) 6									
BETA(2) 7									
V 6									
VZ 6									
VZ 7									
V-THETA 6									
V-THETA 7									
M 6									
TURN(2) 6									
W									
DFAC									
EFFP									
INCIDM									
DEVM									
STATOR B									
PCT SPAN									
DIA 9									
BETA 9									
BETA 10									
V 9									
V 10									
VZ 9									
VZ 10									
V-THETA 9									
V-THETA 10									
M 9									
M 10									
TURN									
W									
DFAC									
EFFP									
INCIDM									
DEVM									

Table B-4. Blade Element Data Intermediate Configuration (Continued)

70PERCENT SPEED PT.212				CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/\delta$		180.3	
				INTERMEDIATE		4230.					
INLET GUIDE VANE											
ROTOR											
POT SPAN	90	70	50	30	10	POT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	C.000	C.000	C.000	C.000	C.000	BETA 6	13.868	13.530	12.092	9.086	5.124
BETA 4	15.445	14.792	13.199	9.715	5.467	BETA 7	38.397	38.408	36.119	32.973	28.685
V 3	279.56	295.34	304.90	309.63	309.72	BETA(PR) 6	48.505	52.332	56.778	60.611	65.186
V 4	300.10	320.65	327.97	332.32	327.61	BETA(PR) 7	22.269	35.884	43.771	49.490	53.522
VZ 3	279.53	294.98	304.88	309.61	309.71	V 6	322.84	343.26	348.34	352.47	336.19
VZ 4	289.25	308.44	319.28	327.54	326.12	V 7	497.45	453.59	443.37	441.54	447.88
V-THETA 3	C.000	C.000	C.000	C.000	C.000	VZ 6	313.35	333.48	340.53	348.02	334.85
V-THETA 4	79.92	81.86	74.85	56.08	31.21	VZ 7	388.03	355.09	357.96	370.35	392.90
M 3	0.2520	C.2664	0.2752	0.2795	0.2796	V-THETA 6	77.38	80.31	72.97	55.66	30.03
M 4	0.2708	C.2896	0.2964	0.3003	0.2960	V-THETA 7	308.97	281.80	261.35	240.30	214.98
TURN	-15.45	-14.79	-13.20	-9.71	-5.47	V(PR) 6	473.0	545.9	621.6	709.2	797.9
ω	0.0180		0.0228		0.0230	V(PR) 7	421.3	438.7	496.0	570.2	660.9
DFAC	C.073	0.055	0.048	C.018	-0.007	VTHETA PR6	-354.2	-432.0	-520.0	-617.9	-724.2
EFFP	1.9216	1.7942	1.4591	1.7423	1.4428	VTHETA PR7	-159.6	-257.1	-343.1	-433.6	-531.4
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	431.61	512.27	592.93	673.58	754.24
DEVM	5.255	4.208	4.001	5.885	9.833	U 7	468.62	538.94	604.46	673.86	746.39
						M 6	0.2916	0.3104	0.3151	0.3189	0.3039
						M 7	0.4466	0.4059	0.3961	0.3938	0.3988
						M(PR) 6	0.4273	0.4937	0.5623	0.6417	0.7213
						M(PR) 7	0.3782	0.3925	0.4431	0.5086	0.5885
						TURN(PR)	26.228	16.426	13.001	11.120	11.664
						ω	0.0278	0.0187	0.0342	0.0745	0.1075
						DFAC	.2936	.3113	.3105	.3122	.2803
						EFFP	0.9990	0.9840	0.9967	0.9629	0.9370
						EFF	0.9990	0.9837	0.9966	0.9622	0.9357
						INCIDM	-0.095	-0.168	0.678	1.011	1.986
						DEVM	3.469	3.384	1.671	0.890	0.622
						ω					
POT SPAN	90	70	50	30	10	POT SPAN	90	70	50	30	10
DIA 9	26.021	29.683	33.345	37.007	40.669	DFAC	.2936	.3113	.3105	.3122	.2803
BETA 9	37.055	36.847	34.636	31.700	28.368	EFFP	0.9990	0.9840	0.9967	0.9629	0.9370
BETA 10	15.212	13.869	13.445	13.162	12.935	EFF	0.9990	0.9837	0.9966	0.9622	0.9357
V 9	452.91	486.99	452.70	455.48	455.47	INCIDM	-0.095	-0.168	0.678	1.011	1.986
V 10	423.88	409.06	404.94	414.35	427.59	DEVM	3.469	3.384	1.671	0.890	0.622
VZ 9	387.19	366.59	371.86	387.51	400.66						
VZ 10	406.47	354.50	393.20	403.45	416.60						
V-THETA 9	297.02	276.45	257.30	239.34	216.41						
V-THETA 10	111.22	98.06	94.16	94.35	95.71						
M 9	C.4424	0.4127	0.4047	0.4065	0.4057						
M 10	C.3785	0.3649	0.3688	0.3688	0.3802						
TURN	21.843	22.978	21.191	18.538	15.433						
ω	0.0880	0.0444	0.0553	0.0576	0.0445						
DFAC	0.2794	C.2707	0.2636	0.2387	0.1910						
EFFP	C.7183	C.8066	C.7234	0.7319	1.0414						
INCIDM	-8.345	-4.753	-4.564	-6.800	-12.132						
DEVM	10.912	8.369	7.945	8.162	8.835						

Table B-4. Blade Element Data Intermediate Configuration

CONFIGURATION				$\omega\sqrt{\theta}/\delta$	
INTERMEDIATE				102.5	
70PERCENT SPEED PT.208				4245.	
INLET GUIDE VANE				ROTOR	
POT SPAN				POT SPAN	
DIA 3				DIA 6	
BETA 3				BETA 6	
V 3				V 6	
VZ 3				VZ 6	
V-THETA 3				V-THETA 6	
M 3				M 6	
M 4				M 7	
TURN				TURN	
DFAC				DFAC	
EFFP				EFFP	
INCIDM				INCIDM	
DEVN				DEVN	
STATOR B				STATOR B	
POT SPAN				POT SPAN	
DIA 9				DIA 9	
BETA 9				BETA 9	
V 9				V 9	
VZ 9				VZ 9	
V-THETA 9				V-THETA 9	
M 9				M 9	
M 10				M 10	
TURN				TURN	
DFAC				DFAC	
EFFP				EFFP	
INCIDM				INCIDM	
DEVN				DEVN	

Table B-4. Blade Element Data Intermediate Configuration (Continued)

80PERCENT SPEED PT.2C3				CONFIGURATION		$N/\sqrt{\sigma}$		$\omega\sqrt{\sigma}/s$	
				INTERMEDIATE		4942.		146.4	
INLET GUIDE VANE									
ROTOR									
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.395	27.755	32.125
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	14.730	17.939	19.044
BETA 4	15.372	14.756	14.680	10.471	7.576	BETA 7	6.1100	58.485	52.478
V 3	226.27	233.85	243.83	249.58	256.56	BETA(PP) 6	61.150	63.927	66.241
V 4	236.48	253.11	262.21	271.34	269.66	BETA(PP) 7	22.239	43.683	51.626
VZ 3	226.26	233.54	243.82	249.57	255.55	V 6	246.42	263.96	278.36
VZ 4	228.02	242.56	254.33	266.82	267.30	V 7	500.90	456.39	442.89
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6	238.70	255.81	271.12
V-THETA 4	62.69	64.47	63.79	49.31	35.55	VZ 7	249.22	238.30	269.57
M 3	0.1981	0.2104	0.2195	0.2247	0.2256	V-THETA 6	60.61	63.59	62.82
M 4	0.2128	0.2279	0.2362	0.2445	0.2430	V-THETA 7	43.4.31	383.07	351.26
TURN	-15.37	-14.76	-14.68	-10.47	-7.58	V(PR) 6	494.9	592.2	672.9
W	0.0806	0.1975	0.2936	0.2517	0.3063	V(PR) 7	269.9	329.9	434.5
DFAC	0.072	0.057	0.056	0.112	-0.005	VTHETA PRA	-433.5	-522.8	-615.9
EFEP	0.8219	0.9152	0.8703	0.9864	0.7904	VTHETA PR7	-102.1	-227.8	-340.7
INCIDM	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	494.66	586.39	678.71
DEVM	5.328	4.244	3.120	5.129	7.724	U 7	536.42	616.91	691.92
STATOR B									
PCT SPAN	90	70	50	30	10	M 6	0.2219	0.2378	0.2509
DIA 9	26.021	29.683	33.345	37.307	40.669	M 7	0.4634	0.4024	0.3902
BETA 9	60.076	57.187	50.682	47.030	47.114	M(PR) 6	0.4455	0.5244	0.6066
BETA 10	24.851	17.066	13.945	13.185	11.562	M(PR) 7	0.2389	0.2910	0.3830
V 9	484.80	453.55	445.85	462.05	473.83	TURN(PR)	38.005	20.211	14.610
V 10	217.82	237.02	290.13	340.44	407.44	W	0.0806	0.1975	0.2936
VZ 9	236.45	242.75	281.15	314.73	327.41	DFAC	0.7353	0.6554	0.5353
VZ 10	195.50	280.01	290.02	331.13	399.57	EFEP	0.8967	0.9259	0.9967
V-THETA 9	420.17	381.18	344.93	338.08	344.99	CFR	0.8038	0.9238	0.9966
V-THETA 10	91.54	87.17	69.02	77.65	81.75	INCIDM	12.550	11.027	10.141
M 9	0.4285	0.4000	0.3931	0.4361	0.4116	DEVM	3.438	11.183	6.612
M 10	0.1498	0.2596	0.2535	0.2973	0.3552				
TURN	35.225	40.122	36.738	33.945	35.552				
W	0.4073	0.2175	0.3109	0.2671	0.1257				
DFAC	0.8051	0.6113	0.6208	0.5260	0.4067				
EFEP	0.5021	0.5819	0.3573	0.3117	0.4150				
INCIDM	14.576	15.587	11.482	8.530	6.614				
DEVM	20.551	11.566	8.445	8.185	7.462				

Table B-4. Blade Element Data Intermediate Configuration (Continued)

80 PERCENT SPEED		PT. 204	CONFIGURATION		$N/\sqrt{\sigma}$	$\omega\sqrt{\sigma}/\delta$
			IN. IMMEDIATE		4.924.	150.2
INLET GUIDE VANE						
POT SPAN						
	90	70	50	30	10	
DIA 3	21.150	26.950	30.950	35.950	40.750	1'
BETA 3	0.000	0.000	0.000	0.000	0.000	36.495
BETA 4	15.372	14.873	13.964	10.459	8.267	3.648
V 3	220.33	242.54	251.12	255.24	254.02	47.701
V 4	241.82	259.96	269.82	277.65	276.59	67.812
VZ 3	229.33	242.17	251.12	255.23	254.91	53.434
VZ 4	233.16	249.67	251.90	272.42	273.71	207.22
V-THETA 3	0.00	0.00	0.00	0.00	0.00	295.31
V-THETA 4	64.10	66.73	64.65	50.29	39.77	481.30
M 3	0.2063	0.2183	0.2261	0.2298	0.2295	292.59
M 4	0.2176	0.2341	0.2431	0.2497	0.2493	327.57
TURN	-15.37	-14.87	-13.86	-10.46	-8.27	313.93
(REFER TO 70% & 100% SPEED DATA)						
W	0.089	0.068	0.055	0.014	0.007	441.6
DFAC	0.9277	0.9604	0.9182	0.9828	0.9327	527.1
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	-612.6
INCIDM	5.328	4.127	3.336	5.141	7.033	-718.4
DEVM						-423.4
STATOR B						
	90	70	50	30	10	
DIA 9	26.021	29.683	33.345	37.007	40.669	36.495
BETA 9	56.647	54.609	49.473	46.182	46.414	3.648
BETA 10	23.820	16.810	13.579	13.555	14.212	47.701
V 9	487.29	461.97	455.25	478.00	489.59	67.812
V 10	235.25	305.77	302.58	351.02	410.27	53.434
VZ 9	262.81	264.64	295.76	330.93	337.41	207.22
VZ 10	212.68	289.79	293.22	341.13	397.59	295.31
V-THETA 9	407.03	377.02	346.04	344.80	356.63	481.30
V-THETA 10	95.01	98.43	71.04	82.27	100.73	527.1
M 9	0.4314	0.4080	0.4017	0.4204	0.4284	506.5
M 10	0.2054	0.2676	0.2647	0.3066	0.3574	-612.6
TURN	32.827	37.889	35.803	32.627	32.202	-718.4
W	0.3910	0.2101	0.2766	0.2437	0.1399	-423.4
DFAC	0.7573	0.5946	0.6013	0.5218	0.4153	-423.4
EFFP	0.5261	0.6226	0.4692	0.4971	0.6499	-423.4
INCIDM	11.247	13.099	10.273	7.682	5.914	36.495
DEVM	19.520	11.310	8.079	8.555	10.112	47.701

ROTOR						
	90	70	50	30	1'	
DIA 6	23.385	27.755	32.125	36.495	40.865	1'
BETA 6	14.064	13.981	12.800	9.648	7.037	36.495
BETA 7	56.778	56.001	51.043	47.701	47.117	47.701
ETA (PP) 6	60.065	62.856	65.063	67.812	71.000	71.000
ETA (PP) 7	72.479	74.451	76.424	78.397	80.370	80.370
V 6	255.43	274.12	287.18	299.22	311.26	311.26
V 7	502.63	463.39	451.50	466.61	481.30	481.30
VZ 6	247.69	265.81	270.95	292.08	297.59	297.59
VZ 7	273.52	258.77	283.65	313.03	327.57	327.57
V-THETA 6	62.07	65.76	63.63	49.81	30.36	30.36
V-THETA 7	420.43	394.15	351.09	345.12	362.74	362.74
V (PP) 6	406.4	582.8	673.5	775.8	868.1	868.1
V (PP) 7	208.0	346.8	441.6	577.1	696.5	696.5
VTHETA DR6	-430.2	-518.4	-612.6	-718.4	-820.8	-820.8
VTHETA DR7	-113.2	-230.5	-338.3	-423.4	-498.5	-498.5
U 6	402.22	584.21	676.19	768.17	860.15	860.15
U 7	534.43	614.62	680.34	748.49	811.21	811.21
M 6	0.2300	0.2470	0.2590	0.2681	0.2753	0.2753
M 7	0.4455	0.4093	0.3983	0.4103	0.4212	0.4212
M (DR) 6	0.4470	0.5252	0.6074	0.6099	0.7827	0.7827
M (DR) 7	0.2642	0.3063	0.3896	0.4635	0.5210	0.5210
TURN (PP) 6	37.570	21.178	15.436	14.375	14.315	14.315
W	0.0730	0.1719	0.2574	0.2097	0.2578	0.2578
DFAC	.6680	.6221	.5247	.4905	.4820	.4820
EFFP	0.9140	0.9313	0.9776	0.9478	0.9478	0.9478
EFF	0.9125	0.9204	0.9769	0.9460	0.9460	0.9460
INCIDM	11.466	10.356	9.339	8.212	7.802	7.802
DEVM	3.678	9.151	7.895	4.834	3.788	3.788

ROTOR

Table B-4. Blade Element Data Intermediate Configuration (Continued)

80PERCENT SPEED				CONFIGURATION		$\omega/\sqrt{\theta}$	
PT.205				INTERMEDIATE		4854.	
INLET GUIDE VANE							
ROTOR							
$\omega/\sqrt{\theta}/\delta$							
171.3							
PCT SPAN							
DIA 6							
HETA 6							
RFTA 7							
BETA(PP) 6							
HETA(PP) 7							
V 6							
V 7							
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Table B-4. Blade Element Data Intermediate Configuration (Continued)

PERCENT SPEED				CONFIGURATION		N/√σ		ω√σ/δ	
PERCENT SPEED				INTERMEDIATE		4834.		122.0	
INLET GUIDE VANE									
POT SPAN									
DIA 3									
BETA 3									
BETA 4									
V 3									
V 4									
VZ 3									
VZ 4									
V-THETA 3									
V-THETA 4									
M 3									
M 4									
TURN									
(REFER TO 70% & 100% SPEED DATA)									
ω									
DFAC									
EFPF									
INCIDM									
DEVM									
STATOR B									
POT SPAN									
DIA 9									
BETA 9									
BETA 10									
V 9									
V 10									
VZ 9									
VZ 10									
V-THETA 9									
V-THETA 10									
M 9									
M 10									
TURN									
ω									
DFAC									
EFPF									
INCIDM									
DEVM									

Table B-4. Blade Element Data Intermediate Configuration (Continued)

80PERCENT SPEED		PT.202		CONFIGURATION		$N/\sqrt{\sigma}$		$w\sqrt{\sigma}/\delta$		217.9	
				INTERMEDIATE		4832.					
INLET GUIDE VANE											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	14.016	13.210	11.949	9.309	5.172
BETA 4	15.650	14.543	13.138	9.951	5.830	BETA 7	35.014	33.789	32.045	29.543	26.640
V 3	343.02	362.78	375.11	381.70	382.19	BETA (PR) 6	45.328	49.439	54.172	58.242	62.844
V 4	373.55	398.92	409.72	415.77	407.85	BETA (PR) 7	22.057	35.750	44.199	49.844	55.190
VZ 3	342.98	362.36	375.08	381.68	382.16	V 6	403.01	428.64	433.64	438.21	424.25
VZ 4	359.70	384.14	398.98	409.52	405.76	V 7	501.10	533.29	509.64	514.80	494.35
V-THETA 3	0.00	0.00	0.00	0.00	0.00	VZ 6	390.93	417.02	424.17	432.42	422.57
V-THETA 4	100.77	100.17	93.13	71.85	41.22	VZ 7	481.97	442.81	431.78	430.11	449.37
M 3	0.3102	0.3284	0.3399	0.3460	0.3464	V-THETA 6	97.61	97.95	89.78	70.89	37.72
M 4	0.3384	0.3620	0.3720	0.3777	0.3703	V-THETA 7	339.16	296.58	270.41	248.90	206.18
TURN	-15.65	-14.54	-13.14	-9.95	-5.90	V (PR) 6	556.1	641.5	724.7	821.6	925.0
(REFER TO 70% & 100% SPEED DATA)											
W	0.062	0.040	0.033	0.025	-0.013	VTHETA 0R6	522.4	546.1	602.6	681.3	787.3
DFAC	0.8546	0.9318	0.8483	0.9356	0.7104	VTHETA PR7	-196.2	-487.2	-587.5	-598.6	-823.9
EFFP	-20.200	-20.200	-20.200	-20.200	-20.200	U 6	493.04	585.17	677.31	769.44	861.59
INCIDM	5.050	4.457	4.062	5.619	9.500	U 7	535.31	615.64	690.49	769.76	852.62
DEVM						M 6	0.3658	0.3897	0.3944	0.3987	0.3956
STATOR B											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 9	26.021	29.683	33.345	37.007	40.669	DIA 6	23.385	27.755	32.125	36.495	40.865
BETA 9	33.498	32.236	30.495	28.229	22.249	BETA 6	14.016	13.210	11.949	9.309	5.172
BETA 10	15.273	13.348	13.081	12.570	12.433	BETA 7	35.014	33.789	32.045	29.543	26.640
V 9	588.01	545.10	520.68	519.75	509.52	BETA (PR) 6	45.328	49.439	54.172	58.242	62.844
V 10	552.26	526.71	508.02	509.07	526.18	BETA (PR) 7	22.057	35.750	44.199	49.844	55.190
VZ 9	482.69	458.09	447.74	457.85	471.52	V 6	403.01	428.64	433.64	438.21	424.25
VZ 10	529.82	508.94	493.84	497.58	513.77	V 7	501.10	533.29	509.64	514.80	494.35
V-THETA 9	324.52	290.76	264.23	245.84	192.92	VZ 6	390.93	417.02	424.17	432.42	422.57
V-THETA 10	145.47	121.59	114.98	110.97	113.28	VZ 7	481.97	442.81	431.78	430.11	449.37
M 9	0.5303	0.4902	0.4671	0.4654	0.4555	V-THETA 6	97.61	97.95	89.78	70.89	37.72
M 10	0.4964	0.4729	0.4552	0.4562	0.4710	V-THETA 7	339.16	296.58	270.41	248.90	206.18
TURN	18.225	18.888	17.414	15.659	9.815	V (PR) 6	556.1	641.5	724.7	821.6	925.0
W	0.1171	0.0687	0.0922	0.1205	0.0761	V (PR) 7	522.4	546.1	602.6	681.3	787.3
DFAC	0.1722	0.1601	0.1498	0.1400	0.0442	VTHETA 0R6	-196.2	-487.2	-587.5	-598.6	-823.9
EFFP	0.1713	0.0341	-0.5561	-1.9369	1.7692	VTHETA PR7	493.04	585.17	677.31	769.44	861.59
INCIDM	-11.902	-9.364	-8.705	-10.271	-18.251	U 6	535.31	615.64	690.49	769.76	852.62
DEVM	10.973	7.808	7.581	7.570	8.333	M 6	0.3658	0.3897	0.3944	0.3987	0.3956
						M 7	0.5332	0.4701	0.4568	0.4515	0.4414
						M (PR) 6	0.5048	0.5833	0.4592	0.7475	0.8416
						M (PR) 7	0.4712	0.4906	0.5401	0.6093	0.7023
						TURN (PR)	23.265	13.671	9.968	3.377	7.647
						W	0.0070	0.0265	0.0561	0.0690	0.1026
						DFAC	0.2250	0.2734	0.2743	0.2675	0.2348
						EFFP	1.0151	1.0262	1.0086	0.9687	0.9874
						INCIDM	1.0154	1.0267	1.0088	0.9680	0.9822
						DEVM	-3.272	-3.061	-1.928	-1.358	-0.354
							3.257	3.250	2.099	1.264	2.299

Table B-4. Blade Element Data Intermediate Configuration (Continued)

CONFIGURATION				$\omega \sqrt{\sigma} / \delta$	
				$N/\sqrt{\sigma}$	100°
100 PERCENT SPEED	PT.197	INTERMEDIATE		6000.	
INLET GUIDE VANE					
PCT SPAN	ac	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750
BETA 3	0.000	0.000	0.000	0.000	0.000
BETA 4	15.774	15.136	16.048	12.582	13.982
V 3	296.38	311.60	324.18	330.67	337.94
V 4	314.60	337.80	348.80	361.60	367.10
VZ 3	294.38	311.25	324.17	330.66	330.92
VZ 4	302.75	323.61	338.48	352.88	356.23
V-THETA 3	C.OO	C.OO	C.OO	C.OO	C.OO
V-THETA 4	85.52	88.20	84.70	78.77	89.70
M 3	C.2655	C.2814	C.2029	C.2088	C.2991
M 4	C.2841	C.3054	C.3157	C.3324	C.3324
TURN	-15.77	-15.14	-14.05	-12.53	-13.98
$\bar{\omega}$	0.0180	0.0260	0.0260	0.0260	0.0170
DFAC	0.080	0.060	0.055	0.025	0.025
EFP	0.0492	0.0900	0.0178	1.0386	1.0181
IACIOM	-20.200	-20.200	-20.200	-20.200	-20.200
DEVM	-4.926	-3.864	-3.152	-3.018	-1.318
STATOR B					
PCT SPAN	ac	70	50	30	10
DIA 9	26.021	29.683	33.345	37.007	40.669
BETA 9	57.915	58.129	50.346	47.702	49.397
BETA 10	21.730	13.812	8.691	10.651	16.202
V 9	613.52	575.68	567.82	613.82	613.96
V 10	245.00	364.51	357.73	415.03	529.26
VZ 9	318.98	300.52	300.94	412.99	399.48
VZ 10	224.61	340.45	351.99	407.73	508.15
V-THETA 9	519.82	488.89	437.17	454.02	466.14
V-THETA 10	90.71	97.02	54.06	76.72	147.67
M 9	0.5393	0.5034	0.4966	0.5347	0.5307
M 10	0.2104	0.3141	0.3084	0.3569	0.4548
TURN	36.186	44.318	41.655	37.051	33.195
$\bar{\omega}$	0.4518	0.1987	0.3017	0.3220	0.1288
DFAC	0.8640	0.6540	0.6674	0.6103	0.3915
EFP	0.5275	0.6718	0.4608	0.4080	0.7288
IACIOM	12.515	16.529	11.146	9.202	8.897
DEVM	17.430	8.312	3.191	5.651	12.102
ROTOR					
PCT SPAN	ac	70	50	30	10
DIA 6	23.385	27.755	32.125	36.495	40.845
BETA 6	14.490	14.209	12.910	11.574	12.945
BETA 7	57.662	59.319	52.339	49.109	50.602
BETA(PP) 6	59.170	62.204	64.630	64.581	60.602
BETA(PR) 7	21.588	42.774	50.713	52.647	57.051
V 6	330.78	352.33	372.83	391.92	391.90
V 7	330.41	352.33	372.83	391.92	391.90
VZ 6	330.41	352.33	372.83	391.92	391.90
VZ 7	330.41	352.33	372.83	391.92	391.90
V-THETA 6	87.77	87.02	93.30	78.63	85.84
V-THETA 7	536.86	498.48	445.60	453.04	472.28
V(PP) 6	624.0	731.4	848.0	965.9	1062.0
V(PR) 7	624.0	731.4	848.0	965.9	1062.0
VTHETA PP6	624.0	731.4	848.0	965.9	1062.0
VTHETA PR7	624.0	731.4	848.0	965.9	1062.0
U 6	671.34	772.10	865.97	965.39	1069.31
U 7	671.34	772.10	865.97	965.39	1069.31
M 6	0.2989	0.3188	0.3377	0.3555	0.3662
M 7	0.5599	0.5070	0.4921	0.5210	0.5280
M(PP) 6	0.5599	0.5070	0.4921	0.5210	0.5280
M(PR) 7	0.3220	0.3524	0.4748	0.5611	0.6150
TURN(PR)	27.533	10.401	13.911	13.932	12.438
$\bar{\omega}$	0.0597	0.1840	0.2781	0.2405	0.2149
DFAC	0.6819	0.6728	0.5101	0.5058	0.5008
EFP	0.9140	0.9000	0.9827	0.9878	0.9156
EFE	0.9102	0.8956	0.9819	0.9660	0.9107
IACIOM	10.529	9.704	8.530	6.981	6.289
DEVM	2.788	10.274	8.613	4.047	4.151

Table B-4. Blade Element Data Intermediate Configuration (Continued)

PCT SPAN	INLET GUIDE VANE			CONFIGURATION			N/√θ	ω√θ/δ	214.	ROTOR
	90	70	50	30	INTERMEDIATE	60°30.				
DIA 3	21.150	26.050	30.950	35.850			40.750			
BETA 3	0.000	0.000	0.000	0.000			0.000			
BETA 4	16.025	14.911	13.422	11.091			11.235			
V 3	341.44	361.00	373.22	378.39			375.55			
V 4	362.47	398.70	423.69	416.61			422.02			
VZ 3	341.43	360.46	373.20	378.36			375.53			
VZ 4	348.36	373.48	392.55	408.74			413.92			
V-THETA 3	0.000	0.00	0.00	0.00			0.00			
V-THETA 4	100.06	93.69	80.07	82.23			82.23			
M 3	0.308	0.338	0.349	0.340			0.340			
M 4	0.328	0.352	0.366	0.375			0.383			
TURN	-16.02	-14.91	-13.42	-11.09			-11.24			
Σ	0.0200	0.0256	0.030	0.0210			0.0210			
DFAC	0.089	0.064	0.045	0.005			-0.014			
EFFP	0.0425	0.0404	0.0027	0.0036			0.0294			
INCIDIM	-20.200	-20.200	-20.200	-20.200			-20.200			
DEVH	-4.675	-4.089	-3.780	-4.519			-4.065			
STATOR B										
PCT SPAN	90	70	50	30			10			
DIA 9	26.021	29.683	33.345	37.007			40.669			
BETA 9	52.108	50.343	45.569	42.203			47.420			
BETA 10	16.492	14.620	12.925	13.339			15.018			
V 9	632.48	608.89	601.19	642.71			669.38			
V 10	316.38	419.03	423.80	475.81			560.68			
VZ 9	381.10	385.46	420.02	476.11			451.42			
VZ 10	290.69	401.74	412.21	462.96			541.22			
V-THETA 9	80.76	463.80	429.31	431.75			493.44			
V-THETA 10	0.5591	105.93	94.07	109.77			145.20			
M 9	0.2734	0.3637	0.3677	0.3636			0.5432			
M 10	35.626	35.719	32.744	28.864			0.4846			
TURN	0.4153	0.1983	0.2207	0.2070			0.0773			
Σ	0.7434	0.5566	0.5405	0.4932			0.4165			
DFAC	0.5395	0.6834	0.6144	0.6722			0.9591			
INCIDIM	6.708	8.748	6.369	3.703			6.990			
DEVH	12.182	9.129	7.325	8.339			10.918			

Table B-4. Blade Element Data Intermediate Configuration (Continued)

[illegible]

Table B-4. Blade Element Data Intermediate Configuration (Continued)

100 PERCENT SPEED				PT. 196		CONFIGURATION		$\omega\sqrt{\sigma}/\delta$		$\omega\sqrt{\sigma}/\delta$	
				INTERMEDIATE				605.9.		262.	
INLET GUIDE VANE											
ROTOR											
PCT SPAN	90	70	50	30	10	PCT SPAN	90	70	50	30	10
DIA 3	21.150	26.050	30.950	35.850	40.750	DIA 6	23.185	27.755	32.125	36.495	40.865
BETA 3	0.000	0.000	0.000	0.000	0.000	BETA 6	13.043	13.200	11.905	9.560	6.657
BETA 4	15.753	14.682	13.093	10.359	6.933	BETA 7	33.167	33.808	31.793	28.509	25.368
V 3	428.60	452.65	466.99	473.76	473.69	BETA (PR) 6	47.639	46.151	53.732	59.037	62.866
V 4	466.03	496.52	514.37	519.44	510.79	BETA (PR) 7	24.270	36.304	44.923	51.547	54.176
VZ 3	428.55	452.09	466.96	473.73	473.66	V 6	508.00	542.07	551.63	552.49	525.94
VZ 4	448.50	478.18	500.93	510.95	507.05	V 7	726.06	661.58	629.91	609.40	636.33
V-THETA 3	0.000	0.000	0.000	0.000	0.000	VZ 6	493.79	527.43	539.65	544.79	522.30
V-THETA 4	126.52	125.85	116.52	93.40	61.65	VZ 7	604.99	549.16	535.11	535.41	574.06
M 3	0.3897	0.4123	0.4258	0.4322	0.4372	V-THETA 6	122.62	123.79	113.79	91.75	60.97
M 4	0.4249	0.4538	0.4708	0.4757	0.4674	V-THETA 7	397.21	368.11	331.87	290.87	272.62
TURN	-15.75	-14.68	-13.09	-10.36	-6.93	V (PR) 6	699.7	806.6	912.3	1029.1	1145.5
$\bar{\omega}$	0.0200	0.0231	0.0241	0.003	0.0190	V (PR) 7	666.7	682.1	756.2	961.1	982.4
DFAC	0.064	0.044	0.024	0.003	-0.013	VTHETA PR6	-495.6	-610.0	-735.5	-873.1	-1119.4
EFFP	0.9195	0.9238	0.9435	0.9794	0.8262	VTHETA DR7	-274.0	-403.9	-534.0	-674.4	-796.5
INCIDIM	-20.200	-20.200	-20.200	-20.200	0.8262	U 6	618.24	773.77	849.30	944.83	1080.36
DEVM	-4.947	-4.318	-4.107	-5.241	-8.367	U 7	671.24	771.97	865.82	965.23	1069.13
		STATOR 1				M 6	0.4656	0.4974	0.5066	0.5075	0.4819
						M 7	0.6585	0.5953	0.5650	0.5453	0.5681
						M (PR) 6	0.6402	0.7402	0.8379	0.9453	1.0406
						M (PR) 7	0.6046	0.6138	0.6782	0.7705	0.8770
		STATOR B				TURN (PR)	20.829	12.830	8.804	6.488	4.691
						$\bar{\omega}$	0.0200	0.0460	0.0922	0.1054	0.1142
PCT SPAN	90	70	50	30	10	DFAC	1.1961	0.2761	0.2727	0.2196	0.2288
DIA 9	26.021	29.683	33.345	37.007	40.669	EFFP	1.0806	1.0789	0.9883	0.9572	0.9559
BETA 9	31.691	32.216	30.252	26.822	25.493	EFF	1.0932	1.0298	0.9880	0.9560	0.9544
BETA 10	14.236	13.249	12.369	10.727	12.562	INCIDIM	-3.094	-3.349	-2.368	-1.563	-0.333
V 9	725.70	678.77	634.24	645.51	651.66	DEVM	5.470	3.804	2.823	2.947	1.276
V 10	679.72	642.88	612.45	596.71	636.76						
VZ 9	608.43	571.22	556.90	565.99	588.06						
VZ 10	654.96	622.26	597.48	586.28	621.33						
V-THETA 9	381.24	361.86	325.21	286.18	280.47						
V-THETA 10	167.15	147.33	131.19	111.07	138.49						
M 9	0.6581	0.6119	0.5799	0.5688	0.5824						
M 10	0.6132	0.5774	0.5483	0.5332	0.5685						
TURN	17.455	18.967	17.883	16.095	12.931						
$\bar{\omega}$	0.1101	0.0915	0.0938	0.1300	0.0719						
DFAC	0.1715	0.1816	0.1829	0.1879	0.1297						
EFFP	0.3036	0.3258	0.3258	0.2427	0.0740						
INCIDIM	-13.709	-9.384	-8.948	-11.678	-15.007						
DEVM	9.936	7.749	6.869	5.727	8.462						

APPENDIX C

Method of Deducing Rotating Stall Characteristics

The number, rotational speed, and size of rotating stall zones were determined from the results obtained with three Kistler high response pressure transducers located at Station 2, 10% of span from the outer wall, and spaced 30 and 60 degrees apart, respectively. A set of representative transducer signal traces is presented in figure C-1 to illustrate the method used. The method is predicated on a maximum number of stall zones equal to twelve, and equal rotational speed for all zones.

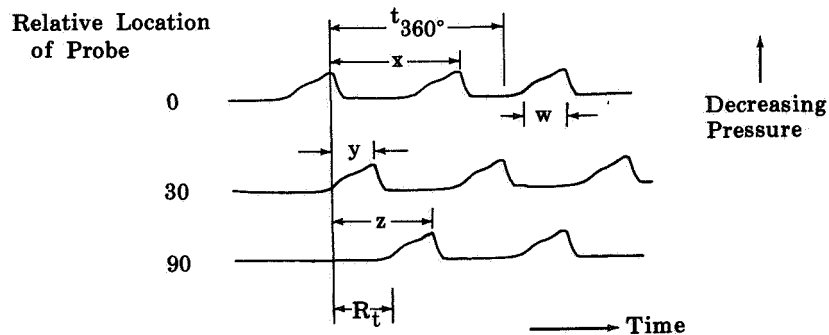


Figure C-1. Transducer Signal Diagram

FD 25199

In figure C-1:

x = time between any two adjacent stall zones (or the cycle time of a single stall zone)

y = time for one of twelve or less stall zones to move from the 0-degree to the 30-degree probe.

z = time between a simultaneous stall zone intercept at the 0-degree probe and a stall zone intercept at the 90-degree probe.

w = the circumferential extent of a stall zone

R_t = time for one rotor revolution

t_{360° = time equivalent of 360 degrees on the transducer signal trace

The minimum angular spacing of stall zones is obtained from

$$\frac{30}{y/x}$$

Thus, if $y = x$, the minimum angular spacing would be 30 degrees. With the angular spacing known for any two adjacent zones, it is possible to identify the time-equivalent of 360 degrees on the transducer signal trace. The number of stall zones within this time period is the number of stall zones present. The rotational speed of the stall zone in terms of rotor speed is obtained from the ratio

$$\frac{R_t}{t_{360^\circ}}$$

The extent of the stall zone is

$$\frac{w}{t_{360^\circ}}$$

The third probe can be used for a check on the number of stall zones when the minimum spacing is equal to or greater than 60 or 90 degrees. The third probe also defines the direction of rotation of the stall zones, although this direction is usually that of the rotor.

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